

AIRBUS A320 Family

LANDING GEAR

NLG PN: D23757500-7

MLG RH PN: 201582002-040

MLG LH PN: 201582001-040

- *SPECIFICATION*
- *FAA 8130 - OVHL*
- *FOLIO 12*
- *LLP*
- *TCI*
- *NIS (LAN CHILE)*
- *NIS (AVIANCA)*
- *NIS (AIRBUS)*
- *NON OPERATION STATEMENT*
- *AVIANCA HOURS AND CYCLES REPORT*



NOSE LANDING GEAR

POSITION:	NLG
PART NUMBER:	D23757500-7
SERIAL NUMBER:	D3137
MFR DATE	OCT2008
TOTAL HOURS	21.765
TOTAL CYCLES:	16.758



MAIN LANDING GEAR

POSITION:	MLG LH	MLG RH
PART NUMBER:	201582001-040	201582002-040
SERIAL NUMBER:	MDL3642	MDL3642
MFR DATE:	OCT2008	OCT2008
TOTAL HOURS	21.765	21.765
TOTAL CYCLES	16.758	16.758

Email: salesteam@airparts.aero

MEMO

AIRBUS
Airbus Delivery Centre Hamburg
Conformity Department



FROM
BDQS

REFERENCE
777.0666/2008

DATE
05-NOV-08

TO

COPIES

Owner / operator
LAN

Subject : A318 - 121 **MSN 3642**
List of identified parts for Landing Gears and Wing Box Pintle Pin
Assemblies

Please find attached for your information the lists of identified parts fitted at time of delivery to MSN 3642 Main and Nose Landing gears (lists provided by Messier-Dowty) plus Wing box Pintle Pin Assemblies (lists provided by AIRBUS UK).

Best regards,

A handwritten signature in black ink, appearing to read 'R. Jahn', is located to the left of the circular stamp.



Ronald Jahn
Aircraft Conformity Department
Airbus Delivery Centre Hamburg

COMPANY: Messier Dowty SA FACTORY : MD	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 1 / 2 PROGRAM : A318 VERSION/RANK : LAN04 STANDARD : ST4 MSN : 3642
CA CODE : DB282 DESIGNATION: NOSE LANDING GEAR	SERIAL NUMBER: B3137 PART NUMBER: NA20284-12	

IDENTIFIABLE PARTS	Original date of issue: 12/07/2008
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DESCRIPTION OF PART	PART NUMBER	SERIAL NUMBER	FORMAT
NOSE LANDING GEAR	D23757500-7	B3137	0
BARREL	D67583	08B0300X8772	2
RACK	D66677	07CZ79057X07	2
PIN	D59954	07CZ82036X22	2
TURNING TUBE	D65520	07MGP70395X2469	2
TORQUE LINKS	D65234	08FGA239526X472	2
TORQUE LINK PIN	D65566-1	08JL779X85	2
CENTER PIN	D64433-1	07JL249X84	2
HINGE PIN (BARREL/STRUCTURE)	D65588-1	07JL113X78	2
HINGE PIN (BARREL/STRUCTURE)	D65588-1	07JL113X64	2
PIN	D64125	07COU45617X0021	2
PIN (RETRACTION ACTUATOR/BARREL)	D65589-1	07JL120X05	2
PIN	D65599-1	07JL121X17	2
SHOCK ABSORBER	D23592030	B443	0
CYLINDER	D65519	07TOF3816P9X003	2
SLIDING TUBE	D66679	08B0713X3712	2
WHEEL AXLE	D66680	08MPS13036X6	2
TORQUE LINKS	D65234	08FGA234441X385	2
TORQUE LINK PIN	D65566-1	08JL779X39	2
STRUT COMPLETE	D23596000-3	WIA283	0

INSPECTION APPROVAL F0189XU	ISSUE	A
	DATE	29/07/2008

COMPANY: Messier Dowty SA FACTORY: MD	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 2 / 2 PROGRAM : A318 VERSION/RANK : LAN04 STANDARD : ST4 MSN : 3642
CA CODE : DB282 DESIGNATION: NOSE LANDING GEAR	SERIAL NUMBER: B3137 PART NUMBER: NA20284-12	

IDENTIFIABLE PARTS	Original date of issue: 12/07/2008
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DESCRIPTION OF PART	PART NUMBER	SERIAL NUMBER	FORMAT
FORESTAY UPPER PANNEL	D65237	WIA033-291	2
FORESTAY LOWER LINK	D67802	08B0060X11	2
UNIVERSAL JOINT	D59638	WIA025-244	2
PIN (LOWER LINK/UNIVERSAL JOINT)	D65773	WIA024-289	2
PIN (UPPER PANNEL/UNIVERSAL JOINT)	D65887	WIA023-325	2
PIN	D67928	WIA022-381	2
PIN	D67932	WIA011-168	2
PIN (UPPER PANNEL/STRUCTURE)	D65616	WIA029-465	2
PIN (UPPER PANNEL/STRUCTURE)	D65616	WIA029-464	2
LOCK LINK UPPER ARM	D59636-1	WIA036-338	2
LOCK LINK LOWER ARM	D59635	WIA034-318	2
LOCK LINK CENTRAL PIN	D60054	WIA017-234	2
PIN (LOCK ACTUATOR/UPPER LINK)	D60739	WIA021-304	2
UNLOCKING SPRING	D23611000	WIA031-565	0
DOWNLOCK SPRING	D59800-0004	05810	2
UNLOCKING SPRING	D23611000	WIA031-566	0
DOWNLOCK SPRING	D59800-0004	05566	2

INSPECTION APPROVAL F0189XU	ISSUE	A
	DATE	29/07/2008

COMPANY: Messier Dowty Ltd FACTORY : MDL	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 1 / 5 PROGRAM : A318 VERSION/RANK : LAN04 STANDARD : ST4 MSN : 3642
CA CODE : DQ364 DESIGNATION: M.L.G. LEFT	SERIAL NUMBER: MDL3642 PART NUMBER: 201581001	

IDENTIFIABLE PARTS	Original date of issue: 17/07/2008
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DESCRIPTION OF PART	PART NUMBER	SERIAL NUMBER	FORMAT
M.L.G. LEG & DRESSINGS	201582001-040	MDG4845	0
MAIN FITTING	201540300	08MDM2051	2
UPPER TORQUE LINK	201540303	08BEL0060	2
LOWER TORQUE LINK	201540302	08BEL0113	2
AFT PINTLE BEARING	201042106	08NM132600X1177	2
PIN (FORWARD PINTLE)	201173600	08MDG6974	2
CROSS BOLT (PINTLE PIN)	201540616	PA79539X437	2
CROSS BOLT (PINTLE PIN)	201540616	PA79539X443	2
PIN (RETAINING LOWER BEARING)	201383608	08BEV81759009	2
PIN (RETAINING LOWER BEARING)	201383608	08BEV81759010	2
PIN (RETAINING LOWER BEARING)	201383608	08BEV81759011	2
HOUSING	201056669	08BEV81828072	2
PIN (UPLOCK)	201383648	08BEV82156012	2
PIN (TORQUE LINK APEX)	201383606	08MDG5303	2
NUT (TORQUE LINK APEX PIN)	201587612	08MSL5902386	2
PIN (TORQUE LINK/MAIN FITTING)	201160603	08MDG8381	2
PIN (TORQUE LINKSLIDING TUBE)	201160602	08MDG6745	2
PIN	201056885	08BEV81971X214	2
PIN	201056909	08BEV81754019	2
COMPLETE CARDAN	201058306	07COU45295X5723	2

INSPECTION APPROVAL M-DG187	ISSUE	A
	DATE	28/07/2008

COMPANY: Messier Dowty Ltd FACTORY : MDL	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 2 / 5 PROGRAM : A318 VERSION/RANK : LAN04 STANDARD : ST4 MSN : 3642
CA CODE : DQ364 DESIGNATION: M.L.G. LEFT	SERIAL NUMBER: MDL3642 PART NUMBER: 201581001	

IDENTIFIABLE PARTS	Original date of issue: 17/07/2008
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DESCRIPTION OF PART	PART NUMBER	SERIAL NUMBER	FORMAT
SHOCK ABSORBER	201371281	08B5151X6200	0
SLIDING ROD	201371304	08B5151X6200	2
UPPER DIAPHRAGM TUBE	201371615	08B0314X1	2
PIN	201160317	08SOP87792X006	2
NUT	D52041-1	08CZ85600X148	2
NUT	D52041-1	08CZ85600X34	2
DAMPER TORQUE LINK	201419001-020	MG2205	0
PISTON	201160693	08MG10734	2
SIDESTAY ASSY	201166013-025	AP1269	0
MAIN BRACE STAY ASSY (Including)	201163004-015	WITHOUT	0
UPPER SIDE STAY	201163300	AP042206	2
LOWER SIDE STAY	201163301	07AP0228	2
UPPER CARDAN	201163620	08B0549X5	2
LOWER CARDAN PIN	201163307	08B0893X1768	2
PIN	201163609	08AP0110	2
PIN	201163618	07AP0042	2
PIN	201163619	07AP0267	2
PIN	201057627	07AP0219	2
NUT	201163944	2186/07	2
NUT	201163945	2762/08	2

INSPECTION APPROVAL M-DG187	ISSUE	A
	DATE	28/07/2008

COMPANY: Messier Dowty Ltd FACTORY : MDL	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 3 / 5 PROGRAM : A318 VERSION/RANK : LAN04 STANDARD : ST4 MSN : 3642
CA CODE : DQ364 DESIGNATION: M.L.G. LEFT	SERIAL NUMBER: MDL3642 PART NUMBER: 201581001	

IDENTIFIABLE PARTS	Original date of issue: 17/07/2008
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DESCRIPTION OF PART	PART NUMBER	SERIAL NUMBER	FORMAT
CUFF	201661613	07AP0226	2
PIN	201661605	08AP0030	2
PIN	201661606	08AP0123	2
PIN	201661607	08AP0148	2
LINK	201661307	08AP0202	2
PIN	201661609	08AP0130	2
PIN	201661612	08AP0169	2
PIN	201661612	08AP0163	2
LINK BANANA	201661306	08AP0132	2
LINK BANANA	201661306	08AP0133	2
LOCKING SPRING ASSY	201059001	AP2569	0
SPRING	201059602	017450	2
LOCKING SPRING ASSY	201059001	AP2570	0
SPRING	201059602	017494	2
LOCK LINK ASSY	201058003	AP1313	0
UPPER ARM	201058310	08AP0005	2
LOWER ARM	201058305	08AP0041	2
PIN	201058608	07AP0223	2
PIN	201058611	07AP0071	2
LOCK STAY ACTUATOR	114095004-010	AP1285	0

INSPECTION APPROVAL M-DG187		ISSUE	A
		DATE	28/07/2008

COMPANY: Messier Dowty Ltd FACTORY : MDL	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 4 / 5 PROGRAM : A318 VERSION/RANK : LAN04 STANDARD : ST4 MSN : 3642
CA CODE : DQ364 DESIGNATION: M.L.G. LEFT	SERIAL NUMBER: MDL3642 PART NUMBER: 201581001	

IDENTIFIABLE PARTS	Original date of issue: 17/07/2008
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DESCRIPTION OF PART	PART NUMBER	SERIAL NUMBER	FORMAT
CYLINDER	114095305	07AP0407	2
PISTON ROD	114095667	07AP0122	2
EYE END	114095602	07AP0247	2
RETRACTION ACTUATOR	201590002-020	B5241	0
PIN	201590901	07BA58521X46	2
PIN	201590900	08BA59459X44	2
CYLINDER	201590906	08MDC00078	2
PISTON HEAD	201590909	08JL985X150	2
NUT	201590910	08JL1090X127	2
GLAND	201590913	08JL986X50	2
PISTON ROD	201590908	07MDC00557	2
BRAKE MANIFOLD	C24837102-2	H8270	0
PRESS XDCR-NORM BRK	CZ9284E	HEK44379	0
PRESS XDCR-NORM BRK	CZ9284E	HEK44417	0
BRAKE (WHEEL 1)	C20225510	11834	0
BRAKE (WHEEL 2)	C20225510	11823	0
MONITORING UNIT-BRK TEMP, L	35-1H5-1002	12582	0
TACHOMETER OUTBOARD (WHEEL 1)	C20105000-2	57893	0
TACHOMETER INBOARD (WHEEL 2)	C20105000-2	57878	0
MOTOR ASSEMBLY-BRAKE FAN (WHEEL 1)	AE1502U02	09895	0

INSPECTION APPROVAL M-DG187	ISSUE	A
	DATE	28/07/2008

COMPANY: Messier Dowty Ltd FACTORY: MDL	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 5 / 5 PROGRAM : A318 VERSION/RANK : LAN04 STANDARD : ST4 MSN : 3642
CA CODE : DQ364 DESIGNATION: M.L.G. LEFT	SERIAL NUMBER: MDL3642 PART NUMBER: 201581001	

IDENTIFIABLE PARTS	Original date of issue: 17/07/2008
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DESCRIPTION OF PART	PART NUMBER	SERIAL NUMBER	FORMAT
MOTOR ASSEMBLY-BRAKE FAN (WHEEL 2)	AE1502U02	09898	0

INSPECTION APPROVAL M-DG187	ISSUE	A
	DATE	28/07/2008

COMPANY: Messier Dowty Ltd FACTORY: MDL	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 1 / 5 PROGRAM : A318 VERSION/RANK : LAN04 STANDARD : ST4 MSN : 3642
CA CODE : DR364 DESIGNATION: M.L.G. RIGHT	SERIAL NUMBER: MDL3642 PART NUMBER: 201581002	

IDENTIFIABLE PARTS	Original date of issue: 17/07/2008
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DESCRIPTION OF PART	PART NUMBER	SERIAL NUMBER	FORMAT
M.L.G. LEG & DRESSINGS	201582002-040	MDG4846	0
MAIN FITTING	201540301	08MDM2048	2
UPPER TORQUE LINK	201540303	08BEL0061	2
LOWER TORQUE LINK	201540302	08BEL0129	2
AFT PINTLE BEARING	201042106	08NM132600X1179	2
PIN (FORWARD PINTLE)	201173600	08MDG6980	2
CROSS BOLT (PINTLE PIN)	201540616	08PA79539X444	2
CROSS BOLT (PINTLE PIN)	201540616	08PA79539X438	2
PIN (RETAINING LOWER BEARING)	201383608	08BEV81759X012	2
PIN (RETAINING LOWER BEARING)	201383608	08BEV81759X013	2
PIN (RETAINING LOWER BEARING)	201383608	08BEV81759X014	2
HOUSING	201056669	08BEV81828077	2
PIN (UPLOCK)	201383648	08BEV82156017	2
PIN (TORQUE LINK APEX)	201383606	08MDG5306	2
NUT (TORQUE LINK APEX PIN)	201587612	08MSL5902352	2
PIN (TORQUE LINK/MAIN FITTING)	201160603	08MDG8380	2
PIN (TORQUE LINKSLIDING TUBE)	201160602	08MDG6742	2
PIN	201056885	08BEV81971X219	2
PIN	201056909	08BEV81754025	2
COMPLETE CARDAN	201058306	08COU46164X5769	2

INSPECTION APPROVAL M-DG43	ISSUE	A
	DATE	24/07/2008

COMPANY: Messier Dowty Ltd FACTORY : MDL	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 2 / 5 PROGRAM : A318 VERSION/RANK : LAN04 STANDARD : ST4 MSN : 3642
CA CODE : DR364 DESIGNATION: M.L.G. RIGHT	SERIAL NUMBER: MDL3642 PART NUMBER: 201581002	

IDENTIFIABLE PARTS	Original date of issue: 17/07/2008
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DESCRIPTION OF PART	PART NUMBER	SERIAL NUMBER	FORMAT
SHOCK ABSORBER	201371281	08B5178X9131	0
SLIDING ROD	201371304	08B5178X9131	2
UPPER DIAPHRAGM TUBE	201371615	08B0314X2	2
PIN	201160317	08SOP87792X018	2
NUT	D52041-1	08CZ85600X172	2
NUT	D52041-1	08CZ85600X151	2
DAMPER TORQUE LINK	201419001-020	MG2204	0
PISTON	201160693	08MG10733	2
SIDESTAY ASSY	201166014-025	AP1270	0
MAIN BRACE STAY ASSY (Including)	201163004-015	WITHOUT	0
UPPER SIDE STAY	201163300	AP042106	2
LOWER SIDE STAY	201163301	07AP0223	2
UPPER CARDAN	201163620	08B0563X3	2
LOWER CARDAN PIN	201163307	08B0893X1767	2
PIN	201163609	08AP0124	2
PIN	201163618	07AP0048	2
PIN	201163619	07AP0268	2
PIN	201057627	07AP0220	2
NUT	201163944	2185/07	2
NUT	201163945	2763/08	2

INSPECTION APPROVAL M-DG43	ISSUE	A
	DATE	24/07/2008

COMPANY: Messier Dowty Ltd FACTORY: MDL	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 3 / 5 PROGRAM : A318 VERSION/RANK : LAN04 STANDARD : ST4 MSN : 3642
CA CODE : DR364 DESIGNATION: M.L.G. RIGHT	SERIAL NUMBER: MDL3642 PART NUMBER: 201581002	

IDENTIFIABLE PARTS	Original date of issue: 17/07/2008
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DESCRIPTION OF PART	PART NUMBER	SERIAL NUMBER	FORMAT
CUFF	201661613	07AP0218	2
PIN	201661605	08AP0054	2
PIN	201661606	08AP0118	2
PIN	201661607	08AP0140	2
LINK	201661307	08AP0195	2
PIN	201661609	08AP0129	2
PIN	201661612	08AP0170	2
PIN	201661612	08AP0172	2
LINK BANANA	201661306	08AP0114	2
LINK BANANA	201661306	08AP0121	2
LOCKING SPRING ASSY	201059001	AP2571	0
SPRING	201059602	017503	2
LOCKING SPRING ASSY	201059001	AP2572	0
SPRING	201059602	017511	2
LOCK LINK ASSY	201058003	AP1316	0
UPPER ARM	201058310	08AP0006	2
LOWER ARM	201058305	08AP0051	2
PIN	201058608	07AP0224	2
PIN	201058611	07AP0073	2
LOCK STAY ACTUATOR	114095004-010	AP1286	0

INSPECTION APPROVAL M-DG43	ISSUE	A
	DATE	24/07/2008

COMPANY: Messier Dowty Ltd FACTORY : MDL	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 4 / 5 PROGRAM : A318 VERSION/RANK : LAN04 STANDARD : ST4 MSN : 3642
CA CODE : DR364 DESIGNATION: M.L.G. RIGHT	SERIAL NUMBER: MDL3642 PART NUMBER: 201581002	

IDENTIFIABLE PARTS	Original date of issue: 17/07/2008
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DESCRIPTION OF PART	PART NUMBER	SERIAL NUMBER	FORMAT
CYLINDER	114095305	07AP0414	2
PISTON ROD	114095667	07AP0125	2
EYE END	114095602	07AP0479	2
RETRACTION ACTUATOR	201590002-020	B5242	0
PIN	201590901	07BA58521X25	2
PIN	201590900	08BA59459X02	2
CYLINDER	201590906	08MDC00080	2
PISTON HEAD	201590909	08JL985X155	2
NUT	201590910	08JL1090X140	2
GLAND	201590913	08JL986X54	2
PISTON ROD	201590908	08MDC00160	2
BRAKE MANIFOLD	C24837102-2	H8275	0
PRESS XDCR-NORM BRK	CZ9284E	HEK44656	0
PRESS XDCR-NORM BRK	CZ9284E	HEK44430	0
BRAKE (WHEEL 3)	C20225510	11824	0
BRAKE (WHEEL 4)	C20225510	11829	0
MONITORING UNIT-BRK TEMP, R	35-1H5-1002	12325	0
TACHOMETER INBOARD (WHEEL 3)	C20105000-2	57862	0
TACHOMETER OUTBOARD (WHEEL 4)	C20105000-2	57892	0
MOTOR ASSEMBLY-BRAKE FAN (WHEEL 3)	AE1502U02	09893	0

INSPECTION APPROVAL M-DG43	ISSUE	A
	DATE	24/07/2008

COMPANY: Messier Dowty Ltd FACTORY : MDL	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 5 / 5 PROGRAM : A318 VERSION/RANK : LAN04 STANDARD : ST4 MSN : 3642
CA CODE : DR364 DESIGNATION: M.L.G. RIGHT	SERIAL NUMBER: MDL3642 PART NUMBER: 201581002	

IDENTIFIABLE PARTS	Original date of issue: 17/07/2008
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DESCRIPTION OF PART	PART NUMBER	SERIAL NUMBER	FORMAT
MOTOR ASSEMBLY-BRAKE FAN (WHEEL 4)	AE1502U02	09899	0

INSPECTION APPROVAL M-DG43	ISSUE	A
	DATE	24/07/2008

COMPAGNY : AIRBUS UK FACTORY : CHESTER	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 001/001 PROGRAM : A318 VERS-RK : LAN04-0015 STANDARD : ST4 MSN : 3642
CA CODE : DQ000 S/N : DHB13642 DESIGNATION : WING EQUIPPED LH	PART NUMBER : D57048000000	

PVR12L LIST OF IDENTIFIED PARTS FITTED TO THE ASSEMBLY

DESCRIPTION =====	PART NUMBER =====	SERIAL NUMBER =====
SIDE STAY FITTING	D57252570002	EEPF293011
FORWARD PINTLE FITTING	D57252560004A	EEP266671
GEAR SUPPORT RIB	D57252101012	3449
RETRACTION JACK FITTING	D57259162000	EEPF270500

FOLIO INSPECTION APPROVAL CQ0240	FOLIO ISSUE : B FOLIO DATE : 15-JUL-2008
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COMPAGNY : AIRBUS UK FACTORY : CHESTER	CONSTITUENT ASSEMBLY INSPECTION REPORT	FOLIO : 12 PAGE : 001/001 PROGRAM : A318 VERS-RK : LAN04-0015 STANDARD : ST4 MSN : 3642
CA CODE : DR000 S/N : DHB13642 DESIGNATION : WING EQUIPPED RH	PART NUMBER : D57048000001	

PVR12L LIST OF IDENTIFIED PARTS FITTED TO THE ASSEMBLY

DESCRIPTION =====	PART NUMBER =====	SERIAL NUMBER =====
SIDE STAY FITTING	D57252570003	EEPF294034
FORWARD PINTLE FITTING	D57252560005A	EEP265651
GEAR SUPPORT RIB	D57252181013	679
RETRACTION JACK FITTING	D57259162001	EEPF271512

FOLIO INSPECTION APPROVAL CQ0240	FOLIO ISSUE : B FOLIO DATE : 15-JUL-2008
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INSPECTION REPORT

DATE: 16.10.2008 TIME: 08:35:02

BSLAQ6

TH61IA

PAGE: 3

FIRM I INSPECTION REPORT IAIRCRAFT: 3642 I CHAPTER: 12
 AIRBUS DEUTSCHLAND GMBH I WORKINGLIST (ALL) I TYPE: A318 -121/00 I PAGE: 1

DIVISION I SECTION I ASSY.-TITLE: AIRCRAFT ASSEMBLY I STANDARD: ST040066 I FIRST DATE 16.10.2008
 HAMBURG FAL I FAL I ASSY.-PART NO.: D0007508000000 I VERSION: LAN040015 I

CODE SERIAL NUMBER I IDENTIFIABLE PARTS I LINEA AEREA NACIONAL DE I ISSUE DOC: 01
 DA999 TF 4642 I I CHILE I DATE DOC: 02.07.2008

CODE	SERIAL-NUMBER	TITLE	PART-NUMBER	SERIAL-NUMBER	PREPARED
POINT	INDEX	VENDOR-CODE	PART-INDEX	NUMBER OF DESCRIPTION	DATE
		VENDOR	FOR INFORMATION	REFERENCE	

DA999		AIRCRAFT ASSEMBLY	D3214022000200	M337550-12801	DAH1604
TF 4642		NUT SPECIAL STA-LOCK	IND: L	BDE-NO:	19.08.2008
00001	006	MLG/RH	SLN41193	CHARGE:	COMPLETED
				REL.DATE:	

DA999		AIRCRAFT ASSEMBLY	D3214022000200	M337550-12804	DAH1604
TF 4642		NUT SPECIAL STA-LOCK	IND: L	BDE-NO:	19.08.2008
00002	004	MLG/LH	SLN41193	CHARGE:	COMPLETED
				REL.DATE:	

DA999		AIRCRAFT ASSEMBLY	D3215303000800	F07647-35	DAH1604
TF 4642		PIN MACHINED	IND: R	BDE-NO:	19.08.2008
00003	002	MLG/RH		CHARGE:	COMPLETED
				REL.DATE:	

DA999		AIRCRAFT ASSEMBLY	D3215303000800	F07647-45	DAH1604
TF 4642		PIN MACHINED	IND: R	BDE-NO:	19.08.2008
00004	002	MLG/LH		CHARGE:	COMPLETED
				REL.DATE:	

**MSN 3642 CC-CZS LEFT HAND MAIN LANDING GEAR STATUS**

AIRCRAFT TYPE : A318-121
 M S N : 3642
 AIRCRAFT REG. : CC-CZS
 MLG ASSY P/N : 201581001
 MLG ASSY S/N : MDL3642
 LAST SHOP VISIT : NEW BY MANUFACTURE

MANUFACTURE : MESSIER - DOWTY Ltd

DATE MFG : July-08

DATE INSTALLED : 28-Oct-08
 A/C TSN : 0
 A/C CSN : 0
 CURRENT STATUS : 21-Nov-13
 A/C TSN : 11,952.1
 A/C CSN : 9,526

No	PART No	DESCRIPTION	SERIAL No	LIFE LIMIT PARTS			SPEC LIMIT OVERHAUL				CURRENT STATUS :				LAST OVERHAUL ACCOMPLISHMENT		REMAINING TO NEXT OVERHAUL		REMAINING TO LIFE		OVH NEXT DUE (DATE)
				CYCLES	CYCLES	YR	TSN	CSN	TSO	CSO	CYCLES	DATE	CYCLES	DAYS	CYCLES	DATE					
1	SLN41193	REAR PINTLE PIN NUT	M337550-12804	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
2	D3215303000800	PIN PINTLE REAR	F07647-45	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
3	201582001-040	M.L.G. LEG & DRESSINGS	MDG4845	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18					
4	201371281	SHOCK ABSORB/SLIDING TUBE	08B5151X6200	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18					
5	201587612	TORQUE LINK APEX PIN NUT	08MSL5902386	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
6	201160602	LOWER TORQUE LINK PIN	08MDG6745	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
7	201383606	TORQUE LINK APEX PIN	08MDG5303	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
8	201540302	LOWER TORQUE LINK	08BEL0113	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
9	201371615	UPPER DIAPHRAGM TUBE	08B0314X1	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
10	201371304	SLIDING TUBE	08B5151X6200	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
11	201059009	LOCK STAY CARDAN PIN	08BEV81754019	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
12	201160603	UPPER TORQUE LINK PIN	08MDG8381	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
13	201540616	FORWARD PINTLE PIN STUB BOLT	PA79539X437	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
14	201540616	FORWARD PINTLE PIN STUB BOLT	PA79539X443	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
15	201058306	LOCK STAY CARDAN	07COU45295X5/23	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
16	201540303	UPPER TORQUE LINK	08BEL0060	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
17	201042108	AFT PINTLE SPHERICAL BEARING	08NM132800X1177	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
18	201540300	MAIN FITTING	08MDM2051	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
19	201160013-025	SIDE STAY COMPLETE	AP1269	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18					
20	201058003	LOCK LINK STAY ASSEMBLY	AP1313	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18					
21	201058608	LOCK STAY PIN-CENTER	07AP0223	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
22	201058611	LOCK STAY PIN-UPPER	07AP0071	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
23	201058310	UPPER LOCK LINK	08AP0005	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
24	201058305	LOWER LOCK LINK	08AP0041	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
25	201163004-015	BASIC SIDE STAY	WITHOUT	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18					
26	201661605	PIN (LOCK/LINK BANANA LINK)	08AP0030	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
27	201057627	LOCK STAY PIN-LOWER	07AP0219	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
28	201661606	LOWER PIN (SPRINGS)	08AP0123	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
29	201163609	SIDE STAY PIVOT PIN	08AP0110	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
30	201661609	TRIANGULAR LINK PIN	08AP0130	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
31	201661612	CUFF PIN	08AP0169	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
32	201661612	CUFF PIN	08AP0163	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
33	201661607	UPPER PIN (SPRINGS)	08AP0148	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
34	201163944	UPPER CARDAN NUT	2186/07	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
35	201163945	LOWER CARDAN NUT	2762/08	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
36	201163619	SIDE STAY CARDAN PIN-LOWER	07AP0267	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
37	201163618	SIDE STAY CARDAN PIN-UPPER	07AP0042	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
38	201661613	CUFF	07AP0226	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
39	201661306	BANANA LINK	08AP0132	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
40	201661306	BANANA LINK	08AP0133	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
41	201661307	TRIANGULATION LINK	08AP0202	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
42	201163620	UPPER CARDAN	08B0549X5	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
43	201163300	UPPER SIDE STAY	AP042206	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
44	201163307	LOWER CARDAN	08B0893X1768	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
45	201163301	LOWER SIDE STAY	07AP0228	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
46	201059001	AFT LOCK SPRING ASSY	AP2569	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18					
47	201059001	FWD LOCK SPRING ASSY	AP2570	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18					
48	201590002-020	RETRACTION ACTUATOR	B5241	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18					
49	201590001	RETRACTION ACT PIN-REAR SPAR SIDE	07BA58521X46	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
50	201590000	RETRACTION ACT PIN-MAIN FITTING S/S	08BA59459X44	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
51	201590913	ACTUATOR PISTON ROD GLAND	08LJ986X50	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
52	201590910	ACTUATOR PISTON HEAD NUT	08LJ1090X127	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
53	201590909	ACTUATOR PISTON HEAD	08LJ985X150	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
54	201590908	RETRACTION ACTUATOR PISTON ROD	07MDC00557	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
55	201590906	RETRACTION ACTUATOR CYLINDER	08MDC00078	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					
56	201173600	PIN PINTLE-FORWARD	08MDG6974	60,000	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	50,474	28-Oct-18					

END OF STATUS

JOSE GARAY V.
 QUALITY CONTROL INSPECTOR
 LAN AIRLINES S.A.





MSN 3642 CC-CZS RIGHT HAND MAIN LANDING GEAR STATUS

AIRCRAFT TYPE : A318-121
M S N : 3642
AIRCRAFT REG. : CC-CZS
MLG ASSY P/N : 201581002
MLG ASSY S/N : MDL3642
LAST SHOP VISIT : NEW by MANUFACTURE

MANUFACTURE : MESSIER - DOWTY Ltd

DATE MFG July-08

DATE INSTALLED : 28-Oct-08
A/C TSN : 0
A/C CSN : 0
CURRENT STATUS : 21-Nov-13
A/C TSN : 11,952.1
A/C CSN : 9,526

Table with columns: No, PART No, DESCRIPTION, SERIAL No, LIFE LIMIT PARTS, SPEC LIMIT OVERHAUL, CURRENT STATUS (CYCLES, YR, TSN, CSN, TSO, CSO), LAST OVERHAUL ACCOMPLISHMENT (CYCLES, DATE), REMAINING TO OVERHAUL (CYCLES, DAYS), REMAINING TO LIFE (CYCLES), OVH NEXT DUE (DATE). Rows 1-56.

END OF STATUS

JOSE GARAY V.
QUALITY CONTROL INSPECTOR
LAN AIRLINES S.A.





MSN 3642 CC-CZS NOSE LANDING GEAR STATUS

AIRCRAFT TYPE : A318-121
M S N : 3642
AIRCRAFT REG. : CC-CZS
NLG ASSY P/N : NA20284-12
NLG ASSY S/N : B3137
LAST SHOP VISIT : NEW by MANUFACTURE

MANUFACTURE : MESSIER - DOWTY Ltd

DATE MFG : July-08

DATE INSTALLED : 28-Oct-08
A/C TSN : 0
A/C CSN : 0
CURRENT STATUS : 21-Nov-13
A/C TSN : 11,952.1
A/C CSN : 9,526

No	PART No	DESCRIPTION	SERIAL No	LIFE LIMIT PARTS			SPEC LIMIT OVERHAUL				CURRENT STATUS :				LAST OVERHAUL ACCOMPLISHMENT		REMAINING TO NEXT OVERHAUL		REMAINING TO LIFE		OVH NEXT DUE (DATE)	
				CYCLES	CYCLES	YR	TSN	CSN	TSO	CSO	CYCLES	DATE	CYCLES	DAYS	CYCLES	DATE						
1	D23596000-3	FORESTAY (STRUT) ASSY	WIA283	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18						
2	D65616	PIN (UPPER PANEL/STRUCTURE)	WIA029-465	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
3	D65616	PIN (UPPER PANEL/STRUCTURE)	WIA029-464	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
4	D67928	LOCK LINK LOWER PIN	WIA022-381	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
5	D65237	FORESTAY UPPER PANEL	WIA033-291	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
6	D59638	UNIVERSAL JOINT	WIA025-244	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
7	D65887	PIN (UPPER PANEL/UNIVERSAL JOINT)	WIA023-325	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
8	D67802	FORESTAY LOWER LINK	08B0060X11	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
9	D65773	PIN (LOWER LINK/UNIVERSAL JOINT)	WIA024-289	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
10	D67932	CROSS PIN	WIA011-168	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
11	D23611000	LOCKING SPRING ASSY	WIA031-565	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18						
12	D23611000	LOCKING SPRING ASSY	WIA031-566	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18						
13	D60739	UNLOCKING ACT/UPP LINK PIN	WIA021-304	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18						
14	D59636-1	LOCK LINK UPPER LINK	WIA036-338	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
15	D59635	LOCK LINK LOWER LINK	WIA034-318	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
16	D60054	LOCK LINK CENTRE PIN	WIA017-234	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
17	D23757500-7	NOSE LANDING GEAR LEG ASSY	B3137	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18						
18	D23592030	SHOCK ABSORBER	B443	N/A	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	N/A	28-Oct-18						
19	D66679	SLIDING TUBE	08B0713X3712	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
20	D65519	CYLINDER	07TOF3816P9X003	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
21	D66680	WHEEL AXLE	08MPS13036X6	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
22	D65588-1	HINGE PIN	07JL113X78	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
23	D65588-1	HINGE PIN	07JL113X64	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
24	D65599-1	PIN (FORESTAY/BARREL)	07JL121X17	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
25	D65589-1	PIN (RETRACTION ACT/BARREL)	07JL120X05	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						
26	D67583	BARREL	08B0300X8772	31,400	20,000	10	11,952	9,526	11,952	9,526	0	28-Oct-08	10,474	1,802	21,874	28-Oct-18						

END OF STATUS

QC
6.59

JOSE GARAY V.
 QUALITY CONTROL INSPECTOR
 LAN AIRLINES S.A.

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21/07/2019
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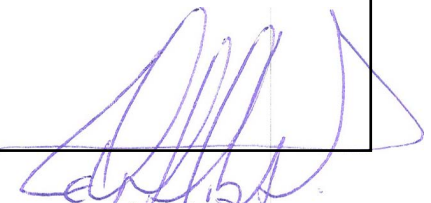
Remaining potentials	PR-ONR	Use up to now	
Effectivity	*	Simulation according to A/C	N
From position	32	The whole kardex or just the COSL	*
To position	3299999	Only expired potentials	N
Protocol type	LLT	All equipment	*
Maintenance level	All		

Simulations to be added to the actual ageing

- Hours**
- Cycles**
- Months**

Grey Line = Expired Potential

Grey header = Sub assembly life limit expired



Leandro Ortibas
 Engenheiro de Produção
 CREA 5060931772

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

From Position 32 To 3299999 Protocol Type LLT Part level * Maintenance level All
 Simulation 0 Months 0,00 FH 0 CY
 Ageing on 21/07/2019 21 764,97 FH 16 758 CY

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)				PN Description		Specific Reference										
	WO number	WP	Scheduled check														
	Inspection				Inspection 2				Overhaul				Life Limit				
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32LMA00 28/10/2008	MAIN LAINDING GEAR LH PN : 201581001				SN : MDL3642	MAIN LANDING GEAR LH				TCI							
LLT													OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		
32LMG00 28/10/2008	MAIN FITTING PN : 201540300				SN : 08MDM2051	MAIN FITTING											
LLT													OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		
32LMG01 28/10/2008	LOWER CARDAN PIN PN : 201163307				SN : 08B0893X1768	LOWER CARDAN PIN											
LLT													OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		


Leandro Ortibas
 Engenheiro de Produção
 CREA 5060921772

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)				PN Description		Specific Reference										
	WO number	WP	Scheduled check														
	Inspection				Inspection 2				Overhaul				Life Limit				
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32LMG02 28/10/2008	SIDE STAY LOWER CARDAN PIN PN : 201163619				SN : 07AP0267		SIDE STAY LWR CARDAN PIN										
LLT													OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		
32LMG03 28/10/2008	REAR PINTLE PIN PN : D3215303000800				SN : F07647-45		REAR PINTLE PIN										
LLT													OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		
32LMG04 28/10/2008	REAR PINTLE PIN NUT PN : SLN41193				SN : M337550-12804		REAR PINTLE PIN NUT										
LLT													OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		
32LMG05 28/10/2008	AFT PINTLE SPHERICAL BEARING PN : 201042106				SN : 08NM132600X1177		AFT PINTLE BEARING										
LLT													OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance						
	Installed Equipment (PN/SN)				PN Description		Specific Reference									
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32LMG06 28/10/2008	FORWARD PINTLE PIN PN : 201173600				SN : 08MDG6974		PIN									
LLT CY		16758,00				16758,00		20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918		3650	3918	-268		26/10/2018		3918		
32LMG07 28/10/2008	FORWARD PINTLE PIN STUB BOLTS PN : 201540616				SN : PA79539X437		CROSS BOLT									
LLT CY		16758,00				16758,00		20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918		3650	3918	-268		26/10/2018		3918		
32LMG08 28/10/2008	FORWARD PINTLE PIN STUB BOLTS AT LH MLG PN : 201540616				SN : PA79539X443		CROSS BOLT									
LLT CY		16758,00				16758,00		20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918		3650	3918	-268		26/10/2018		3918		
32LMG09 28/10/2008	UPPER TORQUE LINK PIN PN : 201160603				SN : 08MDG8381		PIN									
LLT CY		16758,00				16758,00		20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918		3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance						
	Installed Equipment (PN/SN)				PN Description		Specific Reference									
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32LMG10 28/10/2008	UPPER TORQUE LINK PN : 201540303				SN : 08BEL0060		UPPER TORQUE LINK									
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG11 28/10/2008	TORQUE LINK APEX PIN PN : 201383606				SN : 08MDG5303		MLG TORQUE LINK APEX PIN									
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG12 28/10/2008	TORQUE LINK APEX PIN NUT PN : 201587612				SN : 08MSL5902386		NUT									
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG13 28/10/2008	LOWER TORQUE LINK PN : 201540302				SN : 08BEL0113		LOWER TORQUE LINK									
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32LMG14 28/10/2008	LOWER TORQUE LINK PIN PN : 201160602		SN : 08MDG6745	PIN												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG15 28/10/2008	SLIDING TUBE AT LH MLG PN : 201371304		SN : 08B5151X6200	SLIDING ROD												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG16 28/10/2008	UPPER DIAPHRAGM TUBE PN : 201371615		SN : 08B0314X1	UPPER DIAPHRAGM TUBE												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG17 28/10/2008	LOWER CARDAN NUT PN : 201163945		SN : 2762/08	LOWER CARDAN NUT												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description										AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance
	Installed Equipment (PN/SN)										PN Description		Specific Reference			
	WO number	WP	Scheduled check		Inspection		Inspection 2		Overhaul		Life Limit					
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32LMG18 28/10/2008	LOWER SIDE STAY AT LH MLG PN : 201163301										SN : 07AP0228		LOWER SIDE STAY			
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG19 28/10/2008	SIDE STAY PIVOT PIN PN : 201163609										SN : 08AP0110		SIDE STAY PIVOT PIN			
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG20 28/10/2008	UPPER SIDE STAY PN : 201163300										SN : AP042206		UPPER SIDE STAY			
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG21 28/10/2008	SIDE STAY UPPER CARDAN PIN PN : 201163618										SN : 07AP0042		SIDE STAY UPR CARDAN PIN			
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
		Inspection		Inspection 2		Overhaul				Life Limit						
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32LMG22 28/10/2008	UPPER CARDAN PN : 201163620		SN : 08B0549X5	UPPER CARDAN												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG23 28/10/2008	UPPER CARDAN NUT PN : 201163944		SN : 2186/07	UPPER CARDAN NUT												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG24 28/10/2008	UPPER PIN PN : 201661607		SN : 08AP0148	UPPER PIN (SPRINGS)												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG25 28/10/2008	LOWER PIN PN : 201661606		SN : 08AP0123	LOWER PIN (SPRINGS)												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32LMG26 28/10/2008	TRIANGULATION LINK PIN PN : 201661609		SN : 08AP0130	PIN												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG27 28/10/2008	TRIANGULATION LINK PN : 201661307		SN : 08AP0202	LINK												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG28 28/10/2008	BANANA LINK PN : 201661306		SN : 08AP0132	LINK BANANA												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG29 28/10/2008	BANANA LINK PN : 201661306		SN : 08AP0133	LINK BANANA												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description															
	Installed Equipment (PN/SN)															
	WO number WP Scheduled check															
AMM																
F.I.N																
Maint. level																
Zone																
Phasing																
Tolerance																
PN Description																
Specific Reference																
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32LMG30 28/10/2008	CUFF PIN PN : 201661612 SN : 08AP0169 PIN															
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00	
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			
32LMG31 28/10/2008	CUFF PIN PN : 201661612 SN : 08AP0163 PIN															
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00	
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			
32LMG32 28/10/2008	CUFF PN : 201661613 SN : 07AP0226 CUFF															
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00	
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			
32LMG33 28/10/2008	PIN PN : 201661605 SN : 08AP0030 PIN (LOCK/BANANA LINK)															
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00	
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance									
	Installed Equipment (PN/SN)		PN Description		Specific Reference												
	WO number	WP	Scheduled check														
		Inspection				Inspection 2				Overhaul				Life Limit			
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32LMG34 28/10/2008	LOCK LINK LOWER PIN PN : 201057627				SN : 07AP0219	LOCK LINK LOWER PIN											
LLT																	
CY		16758,00				16758,00				20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918				3650	3918	-268	26/10/2018		3918		
32LMG35 28/10/2008	LOWER LOCK LINK PN : 201058305				SN : 08AP0041	LOWER ARM											
LLT																	
CY		16758,00				16758,00				20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918				3650	3918	-268	26/10/2018		3918		
32LMG36 28/10/2008	LOCK LINK CENTER PIN PN : 201058608				SN : 07AP0223	LOCK LINK CENTER PIN											
LLT																	
CY		16758,00				16758,00				20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918				3650	3918	-268	26/10/2018		3918		
32LMG37 28/10/2008	UPPER LOCK LINK PN : 201058310				SN : 08AP0005	UPPER ARM											
LLT																	
CY		16758,00				16758,00				20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918				3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32LMG38 28/10/2008	LOCK LINK UPPER PIN PN : 201058611		SN : 07AP0071	LOCK LINK UPPER PIN												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG39 28/10/2008	LOCK STAY CARDAN PN : 201058306		SN : 07COU45295X5723	COMPLETE CARDAN												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG40 28/10/2008	LOCK STAY CARDAN PIN PN : 201056909		SN : 08BEV81754019	LOCK STAY CARDAN PIN												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG46 28/10/2008	SIDESTAY ASSY PN : 201166013-025		SN : AP1269	STAY ASSY-SIDE,MLG											TCI	
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)		PN Description		Specific Reference										
	WO number	WP	Scheduled check												
Inspection				Inspection 2				Overhaul				Life Limit			
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32LMG47 28/10/2008	DAMPER TORQUE LINK														O/C
	PN : 201419001-020		SN : MG2205		DAMPER, TORQUE LINK - MLG										
	LLT														
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	21764,97	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	16758,00	O/C	O/C
Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C	O/C	O/C	3918	O/C	O/C	3918	O/C	
32LMG49 28/10/2008	RETRACTION ACTUATOR														TCI
	PN : 201590002-020		SN : B5241		ACTUATOR-RETRACTION				2503GM				731		
	LLT											OVH			
	CY		16758,00				16758,00		20000	16758,00	3242,00	20000,00		16758,00	
Cal Total		3918				3918		3650	3918	-268	26/10/2018		3918		
32LMG50 28/10/2008	MONITORING UNIT BRK TEMP L														O/C
	PN : 35-1H5-1002		SN : 11944		BTMU-BRAKE TEMPERATURE MO										
	LLT														
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	21764,97	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	16758,00	O/C	O/C
Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C	O/C	O/C	3918	O/C	O/C	3918	O/C	
32LMG51 28/10/2008	HOUSING														TCI
	PN : 201056669		SN : 08BEV81828072		HOUSING										
	LLT											OVH			
	CY		16758,00				16758,00		20000	16758,00	3242,00	20000,00		16758,00	
Cal Total		3918				3918		3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description										AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance		
	Installed Equipment (PN/SN)										PN Description		Specific Reference					
	WO number	WP	Scheduled check		Inspection		Inspection 2		Overhaul		Life Limit							
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline		
32LMG52 28/10/2008	PIN (UPLOCK) PN : 201383648										SN : 08BEV82156012		PIN				TCI	
LLT																		
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918				
32LMG53 28/10/2008	PIN PN : 201056885										SN : 08BEV81971X214		PIN				TCI	
LLT																		
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918				
32LMG54 28/10/2008	SHOCK ABSORBER PN : 201371281										SN : 08B5151X6200		SHOCK ABSORBER				TCI	
LLT																		
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918				
32LMG55 28/10/2008	PIN PN : 201160317										SN : 08SOP87792X006		PIN				TCI	
LLT																		
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918				

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance		
	Installed Equipment (PN/SN)				PN Description				Specific Reference												
	WO number	WP	Scheduled check																		
		Inspection				Inspection 2				Overhaul				Life Limit							
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline				
32LMG56 28/10/2008	NUT PN : D52041-1				SN : 08CZ85600X148				NUT				TCI								
	LLT												OVH								
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00						
Cal Total			3918				3918			3650	3918	-268	26/10/2018		3918						
32LMG57 28/10/2008	NUT PN : D52041-1				SN : 08CZ85600X34				NUT				TCI								
	LLT												OVH								
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00						
Cal Total			3918				3918			3650	3918	-268	26/10/2018		3918						
32LMG58 28/10/2008	PISTON PN : 201160693				SN : 08MG10734				PISTON				TCI								
	LLT												OVH								
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00						
Cal Total			3918				3918			3650	3918	-268	26/10/2018		3918						
32LMG59 28/10/2008	LOCKING SPRING ASSY PN : 201059001				SN : AP2569				LOCK SPRING				O/C								
	LLT																				
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	O/C	
	CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	O/C	
	Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C					
Cal Total			3918				3918			3650	3918	-268	26/10/2018		3918						

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance									
	Installed Equipment (PN/SN)		PN Description		Specific Reference												
	WO number	WP	Scheduled check														
	Inspection				Inspection 2				Overhaul				Life Limit				
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32LMG60 28/10/2008	SPRING PN : 201059602				SN : 017450	SPRING										TCI	
LLT																	
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		
32LMG61 28/10/2008	LOCKING SPRING ASSY PN : 201059001				SN : AP2570	LOCK SPRING										O/C	
LLT																	
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C		O/C	O/C	21764,97	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C		O/C	O/C	16758,00	O/C	O/C
Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C		O/C		3918	O/C	
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		
32LMG62 28/10/2008	SPRING PN : 201059602				SN : 017494	SPRING										TCI	
LLT																	
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		
32LMG63 28/10/2008	LOCK LINK ASSY PN : 201058003				SN : AP1313	STAY-LOCK										TCI	
LLT																	
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance									
	Installed Equipment (PN/SN)		PN Description		Specific Reference												
	WO number	WP	Scheduled check														
		Inspection				Inspection 2				Overhaul				Life Limit			
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32LMG65 28/10/2008	CYLINDER PN : 114095305				SN : 07AP0407	CYLINDER											TCI
	LLT																
	CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
	Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG66 28/10/2008	PISTON ROD PN : 114095667				SN : 07AP0122	PISTON ROD											
	LLT																
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00		
	Cal A/C Days		3918				3918			3653	3918	-265	29/10/2018		3918		
	Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG67 28/10/2008	EYE END PN : 114095602				SN : 07AP0247	EYE END											TCI
	LLT																
	CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
	Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG68 28/10/2008	CYLINDER PN : 201590906				SN : 08MDC00078	RTCN ACT CYLINDER											
	LLT																
	CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
	Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32LMG69 28/10/2008	PISTON HEAD PN : 201590909		SN : 08JL985X150	PISTON HEAD												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG70 28/10/2008	NUT PN : 201590910		SN : 08JL1090X127	RTCN ACT PISTON HEAD NUT												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG71 28/10/2008	GLAND PN : 201590913		SN : 08JL986X50	RETRACTION ACT GLAND												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG72 28/10/2008	PISTON ROD PN : 201590908		SN : 07MDC00557	PISTON ROD												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance	
	Installed Equipment (PN/SN)				PN Description				Specific Reference											
	WO number	WP	Scheduled check																	
	Inspection				Inspection 2				Overhaul				Life Limit							
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline				
32LMG73 28/10/2008	PIN PN : 201590901				SN : 07BA58521X46				RTCN ACTUATOR PIN (REAR)											
LLT													OVH							
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00			60000,00	
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					
32LMG74 28/10/2008	PIN PN : 201590900				SN : 08BA59459X44				RTCN ACTUATOR PIN (MAIN)											
LLT													OVH							
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00			60000,00	
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					
32LMG77 28/10/2008	PIN (RETAINING LOWER BEARING) PN : 201383608				SN : 08BEV81759009				RETAINIG PIN				TCI							
LLT													OVH							
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00		16758,00					
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					
32LMG78 28/10/2008	PIN (RETAINING LOWER BEARING) PN : 201383608				SN : 08BEV81759010				RETAINIG PIN				TCI							
LLT													OVH							
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00		16758,00					
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)		PN Description		Specific Reference										
	WO number	WP	Scheduled check												
		Inspection		Inspection 2		Overhaul		Life Limit							
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32LMG79 28/10/2008	PIN (RETAINING LOWER BEARING) PN : 201383608		SN : 08BEV81759011	RETAINIG PIN										TCI	
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32LMG83 28/10/2008	BASIC SIDE STAY PN : 201163004-015		SN : AP1269	STAY-SIDE										TCI	
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32LMG84 28/10/2008	LEG AND DRESSINGS-MLG PN : 201582001-040		SN : MDG4845	LEG AND DRESSINGS-MLG										TCI	
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG00 28/10/2008	HINGE PIN PN : D65588-1		SN : 07JL113X78	PIN, HINGE											
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH	31400	16758,00	14642,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	31400,00

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM		F.I.N		Maint. level		Zone		Phasing		Tolerance					
	Installed Equipment (PN/SN)		PN Description				Specific Reference											
	WO number	WP	Scheduled check															
		Inspection				Inspection 2				Overhaul				Life Limit				
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32NG01 28/10/2008	HINGE PIN PN : D65588-1		SN : 07JL113X64		PIN, HINGE													
LLT																		
CY		16758,00				16758,00				20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918				3918				3650	3918	-268		26/10/2018		3918		
32NG02 28/10/2008	BARREL PN : D67583		SN : 08B0300X8772		BARREL													
LLT																		
CY		16758,00				16758,00				20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918				3918				3650	3918	-268		26/10/2018		3918		
32NG03 28/10/2008	PIN PN : D65589-1		SN : 07JL120X05		PIN													
LLT																		
CY		16758,00				16758,00				20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918				3918				3650	3918	-268		26/10/2018		3918		
32NG04 28/10/2008	PIN PN : D65599-1		SN : 07JL121X17		PIN													
LLT																		
CY		16758,00				16758,00				20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918				3918				3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description								AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance		
	Installed Equipment (PN/SN)								PN Description		Specific Reference					
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32NG05 28/10/2008	NGL SHOCK ABSORBER CYLINDER PN : D65519 SN : 07TOF3816P9X003 CYLINDER															
LLT CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00	
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			
32NG06 28/10/2008	NGL SLIDING TUBE PN : D66679 SN : 08B0713X3712 SLIDING TUBE															
LLT CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00	
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			
32NG07 28/10/2008	NGL WHEEL AXLE PN : D66680 SN : 08MPS13036X6 WHEEL AXLE															
LLT CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00	
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			
32NG08 28/10/2008	FORESTAY LOWER LINK PN : D67802 SN : 08B0060X11 FORESTAY LOWER LINK															
LLT CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00	
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM		F.I.N		Maint. level		Zone		Phasing		Tolerance					
	Installed Equipment (PN/SN)		PN Description		Specific Reference													
	WO number	WP	Scheduled check															
		Inspection				Inspection 2				Overhaul				Life Limit				
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32NG09 28/10/2008	PIN PN : D65773		SN : WIA024-289		PIN													
LLT																		
CY		16758,00				16758,00				20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918				3918				3650	3918	-268		26/10/2018		3918		
32NG10 28/10/2008	LOCK LINK LOWER PIN PN : D67928		SN : WIA022-381		PIN, LOCK LINK LOWER													
LLT																		
CY		16758,00				16758,00				20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918				3918				3650	3918	-268		26/10/2018		3918		
32NG11 28/10/2008	CROSS PIN PN : D67932		SN : WIA011-168		PIN CROSS													
LLT																		
CY		16758,00				16758,00				20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918				3918				3650	3918	-268		26/10/2018		3918		
32NG12 28/10/2008	UNIVERSAL JOINT PN : D59638		SN : WIA025-244		UNIVERSAL JOINT													
LLT																		
CY		16758,00				16758,00				20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918				3918				3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32NG13 28/10/2008	PIN PN : D65887				SN : WIA023-325				PIN							
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	31400	16758,00	14642,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32NG14 28/10/2008	FORESTAY UPPER PANEL PN : D65237				SN : WIA033-291				FORESTAY UPPER PANEL							
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	31400	16758,00	14642,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32NG15 28/10/2008	PIN PN : D65616				SN : WIA029-465				PIN							
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	31400	16758,00	14642,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32NG16 28/10/2008	PIN PN : D65616				SN : WIA029-464				PIN							
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	31400	16758,00	14642,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
		Inspection		Inspection 2		Overhaul				Life Limit						
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32NG17 28/10/2008	LOCK LINK LOWER LINK PN : D59635		SN : WIA034-318	LOCK LINK LOWER LINK												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32NG18 28/10/2008	LOCK LINK CENTER PIN PN : D60054		SN : WIA017-234	PIN, LOCK LINK CENTER												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32NG19 28/10/2008	LOCK LINK UPPER LINK PN : D59636-1		SN : WIA036-338	LOCK LINK UPPER LINK												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32NG21 28/10/2008	STEERING PIN PN : D59954		SN : 07CZ82036X22	PIN STEERING												TCI
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32NG22 28/10/2008	TORNING TUBE PN : D65520		SN : 07MGP70395X2469	TORNING TUBE						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32NG23 28/10/2008	TORQUE LINK PN : D65234		SN : 08FGA239526X472	TORQUE LINK						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32NG24 28/10/2008	TORQUE LINK PN : D65234		SN : 08FGA234441X385	TORQUE LINK						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32NG25 28/10/2008	TORQUE LINK PIN PN : D65566-1		SN : 08JL779X85	TORQUE LINK PIN						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance	
	Installed Equipment (PN/SN)				PN Description				Specific Reference											
	WO number	WP	Scheduled check																	
	Inspection				Inspection 2				Overhaul				Life Limit							
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline				
32NG26 28/10/2008	TORQUE LINK PIN PN : D65566-1				SN : 08JL779X39				TORQUE LINK PIN				TCI							
LLT																				
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00					
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					
32NG27 28/10/2008	TORQUE LINK APEX PIN ASSY PN : D64433-1				SN : 07JL249X84				NGL AFT TORQUE LINKS CNTR				TCI							
LLT																				
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00					
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					
32NG28 28/10/2008	PLUG AXE ASSY PN : D64125				SN : 07COU45617X0021				PLUG AXE ASSY				TCI							
LLT																				
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00					
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					
32NG30 28/10/2008	PIN AT NLG PN : D60739				SN : WIA021-304				PIN				TCI							
LLT																				
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00					
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)		PN Description		Specific Reference										
	WO number	WP	Scheduled check												
Inspection		Inspection 2				Overhaul				Life Limit					
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32NG31 28/10/2008	UNLOCKING SPRING PN : D23611000		SN : WIA031-565	LOCK-SPRING											TCI
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG32 28/10/2008	DOWNLOCK SPRING PN : D59800-0004		SN : 05810	DOWNLOCK SPRING											TCI
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG33 28/10/2008	UNLOCKING SPRING PN : D23611000		SN : WIA031-566	LOCK-SPRING											TCI
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG34 28/10/2008	DOWNLOCK SPRING PN : D59800-0004		SN : 05566	DOWNLOCK SPRING											TCI
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)		PN Description		Specific Reference										
	WO number	WP	Scheduled check												
Inspection				Inspection 2				Overhaul				Life Limit			
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32NG35 28/10/2008	NGL SHOCK ABSORBER PN : D23592030		SN : B443	ABSORBER-SHOCK											TCI
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG36 28/10/2008	STRUT COMPLETE PN : D23596000-3		SN : WIA283	FORESTAY ASSY											TCI
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG37 28/10/2008	RACK PN : D66677		SN : 07CZ79057X07	RACK											TCI
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32RMA00 28/10/2008	MAIN LAINDING GEAR RH PN : 201581002		SN : MDL3642	MAIN LANDING GEAR RH											TCI
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG00 28/10/2008	MAIN FITTING PN : 201540301		SN : 08MDM2048	MAIN FITTING												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32RMG01 28/10/2008	LOWER CARDAN PIN PN : 201163307		SN : 08B0893X1767	LOWER CARDAN PIN												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32RMG02 28/10/2008	SIDE STAY LOWER CARDAN PIN PN : 201163619		SN : 07AP0268	SIDE STAY LWR CARDAN PIN												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32RMG03 28/10/2008	REAR PINTLE PIN PN : D3215303000800		SN : F07647-35	REAR PINTLE PIN												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance						
	Installed Equipment (PN/SN)				PN Description		Specific Reference									
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG04 28/10/2008	REAR PINTLE PIN NUT PN : SLN41193				SN : M337550-12801		REAR PINTLE PIN NUT									
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32RMG05 28/10/2008	AFT PINTLE SPHERICAL BEARING PN : 201042106				SN : 08NM132600X1179		AFT PINTLE BEARING									
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32RMG06 28/10/2008	FORWARD PINTLE PIN PN : 201173600				SN : 08MDG6980		PIN									
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32RMG07 28/10/2008	FORWARD PINTLE PIN STUB BOLTS PN : 201540616				SN : 08PA79539X444		CROSS BOLT									
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG08 28/10/2008	FORWARD PINTLE PIN STUB BOLTS															
	PN : 201540616		SN : 08PA79539X438		CROSS BOLT											
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32RMG09 28/10/2008	UPPER TORQUE LINK PIN															
	PN : 201160603		SN : 08MDG8380		PIN											
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32RMG10 28/10/2008	UPPER TORQUE LINK															
	PN : 201540303		SN : 08BEL0061		UPPER TORQUE LINK											
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32RMG11 28/10/2008	TORQUE LINK APEX PIN															
	PN : 201383606		SN : 08MDG5306		MLG TORQUE LINK APEX PIN											
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance	
	Installed Equipment (PN/SN)				PN Description				Specific Reference											
	WO number	WP	Scheduled check																	
Inspection				Inspection 2				Overhaul				Life Limit								
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline					
32RMG12 28/10/2008	TORQUE LINK APEX PIN NUT PN : 201587612				SN : 08MSL5902352				NUT											
LLT												OVH								
CY		16758,00				16758,00		20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00				
Cal Total		3918				3918		3650	3918	-268		26/10/2018		3918						
32RMG13 28/10/2008	LOWER TORQUE LINK AT RH MLG PN : 201540302				SN : 08BEL0129				LOWER TORQUE LINK											
LLT												OVH								
CY		16758,00				16758,00		20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00				
Cal Total		3918				3918		3650	3918	-268		26/10/2018		3918						
32RMG14 28/10/2008	LOWER TORQUE LINK PIN PN : 201160602				SN : 08MDG6742				PIN											
LLT												OVH								
CY		16758,00				16758,00		20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00				
Cal Total		3918				3918		3650	3918	-268		26/10/2018		3918						
32RMG15 28/10/2008	SLIDING TUBE PN : 201371304				SN : 08B5178X9131				SLIDING ROD											
LLT												OVH								
CY		16758,00				16758,00		20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00				
Cal Total		3918				3918		3650	3918	-268		26/10/2018		3918						

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
		Inspection		Inspection 2		Overhaul		Life Limit								
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32RMG16 28/10/2008	UPPER DIAPHRAGM TUBE PN : 201371615		SN : 08B0314X2	UPPER DIAPHRAGM TUBE												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG17 28/10/2008	LOWER CARDAN NUT PN : 201163945		SN : 2763/08	LOWER CARDAN NUT												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG18 28/10/2008	LOWER SIDE STAY PN : 201163301		SN : 07AP0223	LOWER SIDE STAY												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG19 28/10/2008	SIDE STAY PIVOT PIN PN : 201163609		SN : 08AP0124	SIDE STAY PIVOT PIN												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
		Inspection		Inspection 2		Overhaul		Life Limit								
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32RMG20 28/10/2008	UPPER SIDE STAY PN : 201163300		SN : AP042106	UPPER SIDE STAY												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG21 28/10/2008	SIDE STAY UPPER CARDAN PIN PN : 201163618		SN : 07AP0048	SIDE STAY UPR CARDAN PIN												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG22 28/10/2008	UPPER CARDAN PN : 201163620		SN : 08B0563X3	UPPER CARDAN												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG23 28/10/2008	UPPER CARDAN NUT PN : 201163944		SN : 2185/07	UPPER CARDAN NUT												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32RMG24 28/10/2008	UPPER PIN PN : 201661607		SN : 08AP0140	UPPER PIN (SPRINGS)												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG25 28/10/2008	LOWER PIN PN : 201661606		SN : 08AP0118	LOWER PIN (SPRINGS)												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG26 28/10/2008	TRIANGULATION LINK PIN PN : 201661609		SN : 08AP0129	PIN												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG27 28/10/2008	TRIANGULATION LINK PN : 201661307		SN : 08AP0195	LINK												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance									
	Installed Equipment (PN/SN)		PN Description		Specific Reference												
	WO number	WP	Scheduled check														
		Inspection				Inspection 2				Overhaul				Life Limit			
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG28 28/10/2008	BANANA LINK PN : 201661306				SN : 08AP0114			LINK BANANA									
LLT													OVH				
CY		16758,00				16758,00				20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918				3650	3918	-268	26/10/2018		3918		
32RMG29 28/10/2008	BANANA LINK PN : 201661306				SN : 08AP0121			LINK BANANA									
LLT													OVH				
CY		16758,00				16758,00				20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918				3650	3918	-268	26/10/2018		3918		
32RMG30 28/10/2008	CUFF PIN PN : 201661612				SN : 08AP0170			PIN									
LLT													OVH				
CY		16758,00				16758,00				20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918				3650	3918	-268	26/10/2018		3918		
32RMG31 28/10/2008	CUFF PIN PN : 201661612				SN : 08AP0172			PIN									
LLT													OVH				
CY		16758,00				16758,00				20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918				3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance			
	Installed Equipment (PN/SN)				PN Description				Specific Reference													
	WO number	WP	Scheduled check																			
		Inspection				Inspection 2				Overhaul				Life Limit								
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline					
32RMG32 28/10/2008	CUFF PN : 201661613				SN : 07AP0218				CUFF													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG33 28/10/2008	PIN PN : 201661605				SN : 08AP0054				PIN (LOCK/BANANA LINK)													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG34 28/10/2008	LOCK LINK LOWER PIN PN : 201057627				SN : 07AP0220				LOCK LINK LOWER PIN													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG35 28/10/2008	LOWER LOCK LINK PN : 201058305				SN : 08AP0051				LOWER ARM													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG36 28/10/2008	LOCK LINK CENTER PIN PN : 201058608		SN : 07AP0224	LOCK LINK CENTER PIN												
LLT																
CY		16758,00				16758,00				16758,00			60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG37 28/10/2008	UPPER LOCK LINK PN : 201058310		SN : 08AP0006	UPPER ARM												
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG38 28/10/2008	LOCK LINK UPPER PIN PN : 201058611		SN : 07AP0073	LOCK LINK UPPER PIN												
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG39 28/10/2008	LOCK STAY CARDAN PN : 201058306		SN : 08COU46164X5769	COMPLETE CARDAN												
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance									
	Installed Equipment (PN/SN)		PN Description		Specific Reference												
	WO number	WP	Scheduled check														
Inspection		Inspection 2				Overhaul				Life Limit							
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline		
32RMG40 28/10/2008	LOCK STAY CARDAN PIN																
	PN : 201056909		SN : 08BEV81754025				LOCK STAY CARDAN PIN										
	LLT											OVH					
CY		16758,00			16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00		
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918				
32RMG46 28/10/2008	SIDESTAY ASSY															TCI	
	PN : 201166014-025		SN : AP1270				STAY ASSY-SIDE,MLG										
	LLT											OVH					
CY		16758,00			16758,00			20000	16758,00	3242,00	20000,00		16758,00				
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918				
32RMG47 28/10/2008	DAMPER TORQUE LINK															O/C	
	PN : 201419001-020		SN : MG2204				DAMPER, TORQUE LINK - MLG										
	LLT																
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	
Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C		
32RMG49 28/10/2008	RETRACTION ACTUATOR															TCI	
	PN : 201590002-020		SN : B5242				ACTUATOR-RETRACTION										
	LLT											OVH					
CY		16758,00			16758,00			20000	16758,00	3242,00	20000,00		16758,00				
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918				

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance						
	Installed Equipment (PN/SN)				PN Description		Specific Reference									
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG50 28/10/2008	MONITORING UNIT BRK TEMP L PN : 35-1H5-1002				SN : 12325		BTMU-BRAKE TEMPERATURE MO				O/C					
LLT																
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C
Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C	
32RMG51 28/10/2008	HOUSING PN : 201056669				SN : 08BEV81828077		HOUSING				TCI					
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG52 28/10/2008	PIN (UPLOCK) PN : 201383648				SN : 08BEV82156017		PIN				TCI					
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG53 28/10/2008	PIN PN : 201056885				SN : 08BEV81971X219		PIN				TCI					
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG54 28/10/2008	SHOCK ABSORBER PN : 201371281		SN : 08B5178X9131	SHOCK ABSORBER						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG55 28/10/2008	PIN PN : 201160317		SN : 08SOP87792X018	PIN						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG56 28/10/2008	NUT PN : D52041-1		SN : 08CZ85600X172	NUT						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG57 28/10/2008	NUT PN : D52041-1		SN : 08CZ85600X151	NUT						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance			
	Installed Equipment (PN/SN)				PN Description				Specific Reference													
	WO number	WP	Scheduled check																			
		Inspection				Inspection 2				Overhaul				Life Limit								
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline					
32RMG58 28/10/2008	PISTON				PISTON																TCI	
	PN : 201160693				SN : 08MG10733																	
	LLT													OVH								
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00							
	Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918							
32RMG59 28/10/2008	LOCKING SPRING ASSY				LOCK SPRING																O/C	
	PN : 201059001				SN : AP2571																	
	LLT																					
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C				O/C	
	CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C				O/C	
	Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C						
	Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918							
32RMG60 28/10/2008	SPRING				SPRING																TCI	
	PN : 201059602				SN : 017503																	
	LLT													OVH								
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00							
	Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918							

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG61 28/10/2008	LOCKING SPRING ASSY PN : 201059001		SN : AP2572	LOCK SPRING												O/C
LLT																
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C
Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG62 28/10/2008	SPRING PN : 201059602		SN : 017511	SPRING												TCI
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG63 28/10/2008	LOCK LINK ASSY PN : 201058003		SN : AP1316	STAY-LOCK												TCI
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG65 28/10/2008	CYLINDER PN : 114095305		SN : 07AP0414	CYLINDER												TCI
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG66 28/10/2008	PISTON ROD PN : 114095667		SN : 07AP0125	PISTON ROD						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG67 28/10/2008	EYE END PN : 114095602		SN : 07AP0479	EYE END						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG68 28/10/2008	CYLINDER PN : 201590906		SN : 08MDC00080	RTCN ACT CYLINDER												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG69 28/10/2008	PISTON HEAD PN : 201590909		SN : 08JL985X155	PISTON HEAD												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance					
	Installed Equipment (PN/SN)				PN Description		Specific Reference								
	WO number	WP	Scheduled check												
Inspection				Inspection 2				Overhaul				Life Limit			
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG70 28/10/2008	NUT PN : 201590910				SN : 08JL1090X140		RTCN ACT PISTON HEAD NUT								
LLT CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918		
32RMG71 28/10/2008	GLAND PN : 201590913				SN : 08JL986X54		RETRACTION ACT GLAND								
LLT CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918		
32RMG72 28/10/2008	PISTON ROD PN : 201590908				SN : 08MDC00160		PISTON ROD								
LLT CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918		
32RMG73 28/10/2008	PIN PN : 201590901				SN : 07BA58521X25		RTCN ACTUATOR PIN (REAR)								
LLT CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description										AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance			
	Installed Equipment (PN/SN)										PN Description		Specific Reference						
	WO number	WP	Scheduled check		Inspection		Inspection 2		Overhaul		Life Limit								
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline			
32RMG74 28/10/2008	PIN PN : 201590900 SN : 08BA59459X02										RTCN ACTUATOR PIN (MAIN)								
	LLT																		
	CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918					
32RMG77 28/10/2008	PIN (RETAINING LOWER BEARING) PN : 201383608 SN : 08BEV81759X012										RETAINIG PIN								
	LLT																		
	CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918					
32RMG78 28/10/2008	PIN (RETAINING LOWER BEARING) PN : 201383608 SN : 08BEV81759X013										RETAINIG PIN								
	LLT																		
	CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918					
32RMG79 28/10/2008	PIN (RETAINING LOWER BEARING) PN : 201383608 SN : 08BEV81759X014										RETAINIG PIN								
	LLT																		
	CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918					

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG83 28/10/2008	BASIC SIDE STAY PN : 201163004-015		SN : AP1270	STAY-SIDE						TCI						
LLT CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG84 28/10/2008	LEG AND DRESSINGS-MLG PN : 201582002-040		SN : MDG4846	LEG AND DRESSINGS-MLG						TCI						
LLT CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
320000 28/10/2008	NOSE LANDING GEAR PN : D23757500-7		SN : B3137	NOSE LANDING GEAR						TCI						
		PCP 00339384														
LLT CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		


Leandro Ortibas
 Engenheiro de Produção
 CREA 5060931772

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)				PN Description		Specific Reference										
	WO number	WP	Scheduled check														
Inspection				Inspection 2				Overhaul				Life Limit					
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline		
32LMG40 28/10/2008	LOCK STAY CARDAN PIN				LOCK STAY CARDAN PIN												
	PN : 201056909				SN : 08BEV81754019												
	LLT											OVH					
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00	
Cal Total		3867			3867			3650	3867	-217		26/10/2018		3867			
32LMG41 28/10/2008	TACHOMETER OUTBOARD (WHEEL 1)				TACHOMETER		19GG						731			O/C	
	PN : C20105000-2				SN : 58867												
	O/C																
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	
Cal A/C Days	O/C	3867	O/C		O/C	3867	O/C		O/C	3867	O/C		O/C	3867	O/C		
32LMG42 28/10/2008	TACHOMETER INBOARD (WHEEL 2)				TACHOMETER		21GG						731			O/C	
	PN : C20105000-2				SN : 57878												
	O/C																
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	
Cal A/C Days	O/C	3867	O/C		O/C	3867	O/C		O/C	3867	O/C		O/C	3867	O/C		
32LMG43 28/10/2008	BRAKE MANIFOLD				MANIFOLD-NORMAL BRAKE ASS												O/C
	PN : C24837102-2				SN : H8270												
	O/C																
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	
Cal A/C Days	O/C	3867	O/C		O/C	3867	O/C		O/C	3867	O/C		O/C	3867	O/C		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32RMG40 28/10/2008	LOCK STAY CARDAN PIN PN : 201056909		SN : 08BEV81754025	LOCK STAY CARDAN PIN												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3867			3867			3650	3867	-217		26/10/2018		3867		
32RMG41 06/07/2016	TACHOMETER OUTBOARD (WHEEL 1) PN : C20105000-2		SN : 57892	TACHOMETER		22GG						741			O/C	
O/C																
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C
Cal A/C Days	O/C	3867	O/C		O/C	3867	O/C		O/C	3867	O/C		O/C	3867	O/C	
32RMG42 06/07/2016	TACHOMETER INBOARD (WHEEL 2) PN : C20105000-2		SN : 57862	TACHOMETER		20GG						741			O/C	
O/C																
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C
Cal A/C Days	O/C	3867	O/C		O/C	3867	O/C		O/C	3867	O/C		O/C	3867	O/C	
32RMG43 28/10/2008	BRAKE MANIFOLD PN : C24837102-2		SN : H8275	MANIFOLD-NORMAL BRAKE ASS											O/C	
O/C																
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C
Cal A/C Days	O/C	3867	O/C		O/C	3867	O/C		O/C	3867	O/C		O/C	3867	O/C	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)				PN Description		Specific Reference										
	WO number	WP	Scheduled check														
Inspection				Inspection 2				Overhaul				Life Limit					
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline		
32111 05/11/2008	MOTOR ASSY BREAK FAN POSS LH 1				7GS		731										
	PN : AE1502U02				SN : 09895		FAN - MOTOR ASSY-BRAKE										
	O/C																
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	33717,07	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	26284,00	O/C	O/C	
Cal A/C Days	O/C	3859	O/C		O/C	3859	O/C	O/C	O/C	3859	O/C	O/C		UNK	O/C		
32112 05/11/2008	MOTOR ASSY BREAK FAN POSS LH 2				9GS		731										
	PN : AE1502U02				SN : 09898		FAN - MOTOR ASSY-BRAKE										
	O/C																
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	33717,07	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	26284,00	O/C	O/C	
Cal A/C Days	O/C	3859	O/C		O/C	3859	O/C	O/C	O/C	3859	O/C	O/C		UNK	O/C		
32113 05/11/2008	MOTOR ASSY BREAK FAN POSS RH 1				8GS		741										
	PN : AE1502U02				SN : 09893		FAN - MOTOR ASSY-BRAKE										
	O/C																
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	33717,07	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	26284,00	O/C	O/C	
Cal A/C Days	O/C	3859	O/C		O/C	3859	O/C	O/C	O/C	3859	O/C	O/C		UNK	O/C		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance						
	Installed Equipment (PN/SN)				PN Description		Specific Reference									
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32114 05/11/2008	MOTOR ASSY BREAK FAN POSS RH 2 PN : AE1502U02				SN : 09899		10GS FAN - MOTOR ASSY-BRAKE				741					
O/C																
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	33717,07	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	26284,00	O/C	O/C
Cal A/C Days	O/C	3859	O/C		O/C	3859	O/C	O/C	O/C	3859	O/C	O/C		UNK	O/C	
3231000 05/11/2008	SELECTOR-DOOR,ELECTRO-HYD PN : 114079018				SN : SA0804291		SELECTOR-DOOR,ELECTRO-HYD									
O/C																
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	33717,07	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	26284,00	O/C	O/C
Cal A/C Days	O/C	3859	O/C		O/C	3859	O/C	O/C	O/C	3859	O/C	O/C		UNK	O/C	
3231001 05/11/2008	SELECTOR-DOOR,ELECTRO-HYD PN : 114079019				SN : SA0804320		SELECTOR-GEAR,ELECTRO-HYD									
O/C																
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	33717,07	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	26284,00	O/C	O/C
Cal A/C Days	O/C	3859	O/C		O/C	3859	O/C	O/C	O/C	3859	O/C	O/C		UNK	O/C	



STATEMENT

Subject: Aircraft Non Incident/Accident

Date: November 21st, 2013

To whom it may concern,

This is to certify that during the operation period of time from November 06th, 2008 until delivery date, the below aircraft and its components with Part Numbers and Serial Numbers described below, have not been involved in any Incident or Accident.

This statement includes the landing gears, engines, and APU currently installed on A318-121, MSN 3642 (CC-CZS), as follows:

POSITION	PART NUMBER	SERIAL NUMBER	INSTALLATION DATE
NLG	NA20284-12	B3137	28-OCT-2008 (ORIGINAL)
MLG LH	201581001	MDL3642	28-OCT-2008 (ORIGINAL)
MLG RH	201581002	MDL3642	28-OCT-2008 (ORIGINAL)
ENGINE 1	PW6122A	P318122	09-NOV-2013
ENGINE 2	PW6122A	P318125	01-NOV-2013
APU	4500001B	2955	10-SEP-2013

JOSE GARAY V.
QUALITY CONTROL INSPECTOR
LAN AIRLINES S.A.

Statement nº 1049 /2019

Date: São Paulo, June 07th, 2019

Incident/Accident Clearance Statement – Airframe

Aircraft: Airbus A318-100 | MSN 3642 (PR-ONR) | TSN 21764,97 | CSN 16758
Engine: PW6000 | ESN | TSN | CSN
Engine: PW6000 | ESN P318123 | TSN 27722,97 | CSN 20883
APU: 4500001B | SN 2924 | TSN 14569 | CSN 14920
NLG: NA28008-015 | SN B3137 | TSN 21764,97 | CSN 16758
MLG LH: 201581001 | SN MDL3642 | TSN 21764,97 | CSN 16758
MLG RH: 201581002 | SN MDL3642 | TSN 21764,97 | CSN 16758

To whom it may concern:

This letter is to certify that the referred aircraft has been operated by Oceanair Linhas Aéreas S.A. during the period stated below and to the best of my knowledge:

Operation period: from December 3, 2013 thru June 7, 2019.

1. Neither the aircraft, nor any part installed have been, unless its airworthiness status was re-established by an approved maintenance organization in accordance with the instructions of the type certificate holder and/or OEM of the part, and supported by an authorized release certificate:

a. Damaged during a reportable incident or accident as defined by Brazilian Civil Aviation Authority regulation RBAC 121.703 – Service Difficulty Reports, equivalent to FAA Title 14 - Chapter I - Subchapter G - Part 121 - Subpart V - §121.703, or

b. Subjected to severe stress or heat (such as in a major engine failure, accident or fire) or has been submersed in salt water.

2. No part has been installed on the aircraft which was obtained from military source or was previously fitted to a state aircraft.



Alberto Ottavio Spelta
Chief Inspector

Rua Tamoios 579 Jardim Aeroporto – São Paulo – SP – CEP 04630-001
Telefone: (11) 3475 8200

AIRBUS

A318 MSN 3642 Incident/Accident Clearance Statement

Statement issue date: 03-Nov-2020

To whom it may concern

The A318 MSN 3642 (PR-ONR), details of which are specified below, has been under the Airbus Financial Services Limited (AFS) responsibility during the period from November 21st, 2019 (start of the period) to November 3rd, 2020 (end of the period).

Configuration details as of the end of the period:

Description	Type / Part No.	Serial No.	TSN	CSN
Aircraft	A318-122	3642	21,764.97 FH	16,758 FC
Engine 1 (off-wing)	PW6124A	P318140	15,510 FH	11,007 FC
Engine 2	PW6124A	P318123	27,722.97 FH	20,883 FC
APU	APS 3200	2924	14,569 Hrs	14,920 Cyc
Nose Landing Gear	D23757500-7	B3137	21,764.97 FH	16,758 FC
LH Main Landing Gear	201581001	MDL3642	21,764.97 FH	16,758 FC
RH Main Landing Gear	201581002	MDL3642	21,764.97 FH	16,758 FC

I hereby certify that, to the best of my knowledge, during the period stated above, neither the aircraft, nor any part installed have been damaged during, or identified as the root cause of, a reportable incident or accident as defined by ICAO Annex 13, or subjected to severe stress or heat beyond limits (such as in a major engine failure, accident, or fire) or has been submersed in salt water or exposed to corrosive agents outside normal operation. No part was obtained from a military or government entity.

I further confirm that, during the period stated above, no PMA part was installed on the aircraft and no non-TCH (Type Certificate Holder) or DER repair was performed on the aircraft.

Authorized AFS Representative:

Signature: 

Name: James COTTLE

Position: Managing Director

AN AIRBUS GROUP COMPANY

AIRBUS FINANCIAL SERVICES LIMITED
REGISTERED IN IRELAND NO. 144820
DIRECTORS:
A. HERMANN (GERMANY), K. GUENAN
(FRANCE), A. KITCHER (U.K.), (J. B. PONS
(FRANCE), B. SAKRAUSKI (GERMANY)
J. COTTLE, F. DOWLING, S. HAUGHEY, T.
LYDON, M. WALSH

5TH FLOOR
6 GEORGE'S DOCK
IFSC
DUBLIN 1
IRELAND
PHONE +353 (0)1 790 55 00

AIRBUS

A318 MSN 3642 Non Operation Statement

Statement issue date: 03-Nov-2020

To whom it may concern

We confirm that the A318 MSN 3642 (PR-ONR) was flown by the operator ONE to Brasilia for its last flight on 27-Jun-2018.

A318 MSN 3642 Total Flight Hours / Total Flight Cycles upon arrival at Brasilia:

Time Since New: 21,764.97 FH

Cycles Since New: 16,758 FC

We confirm that no further flight occurred since arrival at Brasilia on 27-Jun-2018.

Authorized AFS Representative:

Signature:



Name:

James COTTLE

Position: Managing Director

AN AIRBUS GROUP COMPANY

AIRBUS FINANCIAL SERVICES LIMITED
REGISTERED IN IRELAND NO. 144820
DIRECTORS:
A. HERMANN (GERMANY), K. GUENAN
(FRANCE), A. KITCHER (U.K.), J. B. PONS
(FRANCE), B. SAKRAUSKI (GERMANY)
J. COTTLE, F. DOWLING, S. HAUGHEY, T.
LYDON, M. WALSH

5TH FLOOR
6 GEORGE'S DOCK
IFSC
DUBLIN 1
IRELAND
PHONE +353 (0)1 780 55 00

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*****
* MARM17E          *                HOURS AND CYCLES OF PR-ONR          *      25/06/19  *
*                  *                *                *                *                *
* 382124          *   -> 01/01/2001          to 25/06/2019          *                *
*                  *                *                *                *                *
* MFONSECA        *   IN HOURS AND HUNDREDTH          *                *      15:25:37  *
*****
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! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!20/11/2013!	3,03	3,03	1,00 !	11952,10	11952,10	9526,00!	3,03	3,03	1,00!		!
!09/12/2013!	10,37	11,17	2,00 !	11962,47	11963,27	9528,00!	13,40	14,20	3,00!		!
!10/12/2013!	3,18	3,48	1,00 !	11965,65	11966,75	9529,00!	16,58	17,68	4,00!		!
!01/01/2014!	7,03	8,97	5,00 !	11972,68	11975,72	9534,00!	23,61	26,65	9,00!		!
!02/01/2014!	8,96	13,48	8,00 !	11981,65	11989,20	9542,00!	32,57	40,13	17,00!		!
!03/01/2014!	10,73	14,48	8,00 !	11992,38	12003,69	9550,00!	43,30	54,61	25,00!		!
!04/01/2014!	11,10	14,65	9,00 !	12003,48	12018,36	9559,00!	54,40	69,26	34,00!		!
!05/01/2014!	9,05	11,61	7,00 !	12012,54	12029,97	9566,00!	63,45	80,87	41,00!		!
!06/01/2014!	9,27	10,81	5,00 !	12021,80	12040,79	9571,00!	72,72	91,68	46,00!		!
!07/01/2014!	8,22	9,72	4,00 !	12030,02	12050,50	9575,00!	80,94	101,40	50,00!		!
!08/01/2014!	10,36	12,25	5,00 !	12040,37	12062,75	9580,00!	91,30	113,65	55,00!		!
!09/01/2014!	9,57	11,18	5,00 !	12049,94	12073,93	9585,00!	100,87	124,83	60,00!		!
!10/01/2014!	9,30	11,89	5,00 !	12059,24	12085,82	9590,00!	110,17	136,72	65,00!		!
!11/01/2014!	14,59	17,11	8,00 !	12073,83	12102,92	9598,00!	124,76	153,83	73,00!		!
!12/01/2014!	14,59	17,05	8,00 !	12088,43	12119,98	9606,00!	139,35	170,88	81,00!		!
!13/01/2014!	9,63	11,47	5,00 !	12098,06	12131,45	9611,00!	148,98	182,35	86,00!		!
!14/01/2014!	10,25	12,75	6,00 !	12108,31	12144,20	9617,00!	159,23	195,10	92,00!		!
!15/01/2014!	12,32	14,55	7,00 !	12120,63	12158,77	9624,00!	171,55	209,65	99,00!		!
!16/01/2014!	0,25	1,13	1,00 !	12120,88	12159,90	9625,00!	171,80	210,78	100,00!		!
!17/01/2014!	8,13	9,89	5,00 !	12129,01	12169,79	9630,00!	179,93	220,67	105,00!		!
!18/01/2014!	7,39	9,19	5,00 !	12136,39	12178,99	9635,00!	187,32	229,86	110,00!		!
!19/01/2014!	7,39	8,88	5,00 !	12143,78	12187,87	9640,00!	194,71	238,74	115,00!		!
!20/01/2014!	8,40	10,54	5,00 !	12152,18	12198,41	9645,00!	203,11	249,28	120,00!		!
!21/01/2014!	8,79	11,62	8,00 !	12160,98	12210,04	9653,00!	211,90	260,90	128,00!		!
!22/01/2014!	10,98	14,06	8,00 !	12171,96	12224,10	9661,00!	222,88	274,96	136,00!		!
!23/01/2014!	14,31	17,19	8,00 !	12186,28	12241,30	9669,00!	237,19	292,15	144,00!		!
!24/01/2014!	14,49	16,92	8,00 !	12200,77	12258,22	9677,00!	251,68	309,07	152,00!		!
!25/01/2014!	6,24	7,82	4,00 !	12207,02	12266,04	9681,00!	257,92	316,89	156,00!		!
!26/01/2014!	10,33	13,24	8,00 !	12217,33	12279,30	9689,00!	268,25	330,13	164,00!		!
!27/01/2014!	10,54	12,97	8,00 !	12227,88	12292,26	9697,00!	278,79	343,10	172,00!		!
!28/01/2014!	10,53	13,80	8,00 !	12238,40	12306,06	9705,00!	289,32	356,90	180,00!		!
!29/01/2014!	10,12	13,08	6,00 !	12248,52	12319,14	9711,00!	299,44	369,98	186,00!		!
!30/01/2014!	12,77	15,06	7,00 !	12261,30	12334,20	9718,00!	312,21	385,04	193,00!		!
!31/01/2014!	9,50	11,13	5,00 !	12270,80	12345,33	9723,00!	321,71	396,17	198,00!		!
!01/02/2014!	8,47	9,92	4,00 !	12279,27	12355,24	9727,00!	330,18	406,09	202,00!		!
!02/02/2014!	10,95	14,14	8,00 !	12290,22	12369,39	9735,00!	341,13	420,23	210,00!		!
!03/02/2014!	13,99	16,49	8,00 !	12304,23	12385,88	9743,00!	355,12	436,72	218,00!		!
!04/02/2014!	12,21	14,48	7,00 !	12316,43	12400,36	9750,00!	367,33	451,20	225,00!		!
!05/02/2014!	14,35	16,95	8,00 !	12330,79	12417,30	9758,00!	381,68	468,15	233,00!		!
!06/02/2014!	14,15	16,95	8,00 !	12344,94	12434,25	9766,00!	395,83	485,10	241,00!		!
!07/02/2014!	14,19	17,30	8,00 !	12359,14	12451,54	9774,00!	410,02	502,40	249,00!		!
!08/02/2014!	14,35	17,08	8,00 !	12373,50	12468,62	9782,00!	424,37	519,48	257,00!		!
!09/02/2014!	14,63	17,30	8,00 !	12388,12	12485,90	9790,00!	439,00	536,78	265,00!		!
!10/02/2014!	14,43	16,70	8,00 !	12402,55	12502,60	9798,00!	453,43	553,48	273,00!		!
!11/02/2014!	10,83	13,36	7,00 !	12413,37	12515,97	9805,00!	464,26	566,84	280,00!		!
!12/02/2014!	9,31	11,55	5,00 !	12422,68	12527,52	9810,00!	473,57	578,39	285,00!		!
!15/02/2014!	2,03	2,78	3,00 !	12424,71	12530,30	9813,00!	475,60	581,17	288,00!		!
!16/02/2014!	7,29	8,91	4,00 !	12432,00	12539,21	9817,00!	482,89	590,08	292,00!		!
!17/02/2014!	9,23	11,10	5,00 !	12441,23	12550,31	9822,00!	492,12	601,18	297,00!		!
!18/02/2014!	11,04	14,44	8,00 !	12452,27	12564,74	9830,00!	503,16	615,62	305,00!		!
!19/02/2014!	12,60	14,91	7,00 !	12464,87	12579,66	9837,00!	515,76	630,53	312,00!		!
!20/02/2014!	9,33	11,22	5,00 !	12474,20	12590,88	9842,00!	525,09	641,75	317,00!		!
!21/02/2014!	10,13	12,79	6,00 !	12484,34	12603,67	9848,00!	535,22	654,54	323,00!		!
!22/02/2014!	9,57	13,08	8,00 !	12493,91	12616,75	9856,00!	544,79	667,62	331,00!		!
!23/02/2014!	10,98	13,72	8,00 !	12504,89	12630,47	9864,00!	555,77	681,34	339,00!		!
!24/02/2014!	7,99	10,05	5,00 !	12512,89	12640,52	9869,00!	563,76	691,39	344,00!		!
!25/02/2014!	7,35	9,20	5,00 !	12520,24	12649,72	9874,00!	571,11	700,59	349,00!		!
!26/02/2014!	7,37	9,83	6,00 !	12527,61	12659,56	9880,00!	578,48	710,42	355,00!		!
!27/02/2014!	8,93	11,09	6,00 !	12536,54	12670,64	9886,00!	587,41	721,51	361,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!28/02/2014!	10,27	13,09	6,00 !	12546,80	12683,72	9892,00!	597,68	734,60	367,00!		!
!01/03/2014!	8,97	11,01	5,00 !	12555,77	12694,73	9897,00!	606,65	745,61	372,00!		!
!02/03/2014!	5,87	6,75	3,00 !	12561,64	12701,48	9900,00!	612,52	752,36	375,00!		!
!06/03/2014!	1,98	2,59	2,00 !	12563,62	12704,06	9902,00!	614,50	754,95	377,00!		!
!07/03/2014!	7,31	9,00	5,00 !	12570,93	12713,06	9907,00!	621,81	763,95	382,00!		!
!08/03/2014!	7,07	9,01	5,00 !	12577,99	12722,08	9912,00!	628,88	772,96	387,00!		!
!09/03/2014!	8,04	10,06	5,00 !	12586,04	12732,14	9917,00!	636,92	783,02	392,00!		!
!10/03/2014!	7,48	9,05	4,00 !	12593,52	12741,19	9921,00!	644,40	792,07	396,00!		!
!11/03/2014!	8,04	9,78	5,00 !	12601,56	12750,97	9926,00!	652,44	801,85	401,00!		!
!12/03/2014!	8,37	10,50	6,00 !	12609,92	12761,48	9932,00!	660,81	812,35	407,00!		!
!13/03/2014!	8,21	10,39	6,00 !	12618,13	12771,87	9938,00!	669,02	822,74	413,00!		!
!14/03/2014!	8,45	10,60	5,00 !	12626,58	12782,47	9943,00!	677,47	833,34	418,00!		!
!15/03/2014!	6,56	8,05	4,00 !	12633,15	12790,52	9947,00!	684,03	841,39	422,00!		!
!16/03/2014!	8,10	10,26	5,00 !	12641,25	12800,78	9952,00!	692,13	851,65	427,00!		!
!17/03/2014!	8,77	10,89	6,00 !	12650,02	12811,67	9958,00!	700,90	862,54	433,00!		!
!18/03/2014!	7,63	9,16	4,00 !	12657,65	12820,83	9962,00!	708,53	871,70	437,00!		!
!19/03/2014!	7,72	9,28	4,00 !	12665,37	12830,11	9966,00!	716,25	880,98	441,00!		!
!20/03/2014!	9,37	12,03	4,00 !	12674,74	12842,14	9970,00!	725,62	893,01	445,00!		!
!21/03/2014!	7,82	9,55	4,00 !	12682,56	12851,69	9974,00!	733,44	902,56	449,00!		!
!22/03/2014!	8,57	10,89	6,00 !	12691,13	12862,57	9980,00!	742,01	913,45	455,00!		!
!23/03/2014!	7,23	8,63	5,00 !	12698,35	12871,20	9985,00!	749,24	922,08	460,00!		!
!24/03/2014!	5,20	6,52	3,00 !	12703,55	12877,72	9988,00!	754,44	928,60	463,00!		!
!25/03/2014!	9,21	12,36	6,00 !	12712,75	12890,08	9994,00!	763,65	940,96	469,00!		!
!26/03/2014!	7,97	10,87	7,00 !	12720,72	12900,94	10001,00!	771,62	951,83	476,00!		!
!27/03/2014!	8,08	11,03	5,00 !	12728,81	12911,97	10006,00!	779,70	962,86	481,00!		!
!28/03/2014!	7,15	8,86	4,00 !	12735,96	12920,83	10010,00!	786,85	971,72	485,00!		!
!29/03/2014!	7,60	9,01	4,00 !	12743,56	12929,84	10014,00!	794,45	980,73	489,00!		!
!30/03/2014!	8,46	10,62	6,00 !	12752,02	12940,46	10020,00!	802,91	991,35	495,00!		!
!31/03/2014!	9,79	11,85	7,00 !	12761,80	12952,31	10027,00!	812,70	1003,20	502,00!		!
!01/04/2014!	11,48	13,75	5,00 !	12773,28	12966,06	10032,00!	824,18	1016,95	507,00!		!
!02/04/2014!	14,05	16,51	8,00 !	12787,32	12982,57	10040,00!	838,23	1033,46	515,00!		!
!03/04/2014!	9,07	10,67	5,00 !	12796,39	12993,26	10045,00!	847,30	1044,13	520,00!		!
!04/04/2014!	8,31	9,95	4,00 !	12804,69	13003,21	10049,00!	855,61	1054,08	524,00!		!
!05/04/2014!	9,03	10,80	4,00 !	12813,72	13014,01	10053,00!	864,64	1064,88	528,00!		!
!06/04/2014!	10,17	14,56	9,00 !	12823,89	13028,56	10062,00!	874,81	1079,44	537,00!		!
!07/04/2014!	10,30	13,56	8,00 !	12834,18	13042,12	10070,00!	885,11	1093,00	545,00!		!
!08/04/2014!	10,45	14,11	8,00 !	12844,64	13056,23	10078,00!	895,56	1107,11	553,00!		!
!09/04/2014!	12,59	14,87	7,00 !	12857,23	13071,09	10085,00!	908,15	1121,98	560,00!		!
!10/04/2014!	14,03	16,66	8,00 !	12871,26	13087,74	10093,00!	922,18	1138,64	568,00!		!
!11/04/2014!	13,86	16,30	8,00 !	12885,12	13104,04	10101,00!	936,04	1154,94	576,00!		!
!12/04/2014!	9,57	11,21	5,00 !	12894,69	13115,26	10106,00!	945,61	1166,15	581,00!		!
!13/04/2014!	9,30	11,61	5,00 !	12903,99	13126,86	10111,00!	954,91	1177,76	586,00!		!
!14/04/2014!	9,20	11,23	5,00 !	12913,19	13138,10	10116,00!	964,11	1188,99	591,00!		!
!15/04/2014!	10,37	14,07	8,00 !	12923,55	13152,18	10124,00!	974,48	1203,06	599,00!		!
!16/04/2014!	10,53	12,09	6,00 !	12934,08	13164,28	10130,00!	985,01	1215,15	605,00!		!
!17/04/2014!	14,06	16,59	8,00 !	12948,15	13180,87	10138,00!	999,07	1231,74	613,00!		!
!18/04/2014!	12,86	14,96	7,00 !	12961,01	13195,82	10145,00!	1011,93	1246,70	620,00!		!
!19/04/2014!	9,56	11,25	5,00 !	12970,58	13207,07	10150,00!	1021,49	1257,95	625,00!		!
!20/04/2014!	9,97	13,49	8,00 !	12980,57	13220,55	10158,00!	1031,46	1271,44	633,00!		!
!21/04/2014!	10,35	13,47	8,00 !	12990,92	13234,02	10166,00!	1041,81	1284,91	641,00!		!
!22/04/2014!	12,47	14,81	7,00 !	13003,38	13248,83	10173,00!	1054,28	1299,72	648,00!		!
!23/04/2014!	14,00	16,15	8,00 !	13017,38	13264,97	10181,00!	1068,28	1315,87	656,00!		!
!24/04/2014!	10,50	12,57	7,00 !	13027,87	13277,55	10188,00!	1078,78	1328,44	663,00!		!
!25/04/2014!	7,76	9,73	5,00 !	13035,63	13287,28	10193,00!	1086,54	1338,17	668,00!		!
!26/04/2014!	13,17	16,44	9,00 !	13048,80	13303,73	10202,00!	1099,71	1354,61	677,00!		!
!27/04/2014!	10,85	13,13	8,00 !	13059,65	13316,86	10210,00!	1110,56	1367,74	685,00!		!
!28/04/2014!	12,37	15,03	8,00 !	13072,03	13331,89	10218,00!	1122,93	1382,77	693,00!		!
!29/04/2014!	10,09	12,18	6,00 !	13082,11	13344,07	10224,00!	1133,02	1394,95	699,00!		!
!30/04/2014!	9,00	12,10	6,00 !	13091,11	13356,17	10230,00!	1142,02	1407,05	705,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!01/05/2014!	8,31	10,58	5,00 !	13099,41	13366,75	10235,00!	1150,33	1417,63	710,00!		!
!02/05/2014!	7,18	8,76	4,00 !	13106,59	13375,51	10239,00!	1157,51	1426,39	714,00!		!
!03/05/2014!	7,92	9,61	5,00 !	13114,51	13385,12	10244,00!	1165,43	1436,00	719,00!		!
!04/05/2014!	8,74	10,80	6,00 !	13123,24	13395,91	10250,00!	1174,17	1446,80	725,00!		!
!05/05/2014!	9,17	11,49	6,00 !	13132,41	13407,41	10256,00!	1183,34	1458,29	731,00!		!
!06/05/2014!	7,96	9,80	5,00 !	13140,37	13417,21	10261,00!	1191,30	1468,09	736,00!		!
!07/05/2014!	7,96	10,17	5,00 !	13148,33	13427,38	10266,00!	1199,26	1478,26	741,00!		!
!08/05/2014!	7,80	9,87	5,00 !	13156,13	13437,25	10271,00!	1207,06	1488,13	746,00!		!
!09/05/2014!	6,86	8,40	4,00 !	13162,99	13445,65	10275,00!	1213,92	1496,53	750,00!		!
!10/05/2014!	9,58	12,24	8,00 !	13172,57	13457,90	10283,00!	1223,50	1508,77	758,00!		!
!11/05/2014!	10,71	13,26	8,00 !	13183,27	13471,17	10291,00!	1234,21	1522,03	766,00!		!
!12/05/2014!	11,01	12,73	5,00 !	13194,26	13483,90	10296,00!	1245,22	1534,76	771,00!		!
!13/05/2014!	9,88	13,05	7,00 !	13204,14	13496,95	10303,00!	1255,10	1547,81	778,00!		!
!14/05/2014!	11,05	13,79	8,00 !	13215,19	13510,74	10311,00!	1266,15	1561,60	786,00!		!
!15/05/2014!	8,84	11,49	7,00 !	13224,02	13522,23	10318,00!	1274,99	1573,09	793,00!		!
!16/05/2014!	7,18	9,95	5,00 !	13231,20	13532,18	10323,00!	1282,17	1583,04	798,00!		!
!17/05/2014!	1,43	2,11	2,00 !	13232,63	13534,29	10325,00!	1283,60	1585,15	800,00!		!
!19/05/2014!	7,53	9,43	5,00 !	13240,16	13543,73	10330,00!	1291,13	1594,58	805,00!		!
!20/05/2014!	14,27	16,52	8,00 !	13254,42	13560,25	10338,00!	1305,40	1611,10	813,00!		!
!21/05/2014!	9,23	10,99	5,00 !	13263,65	13571,23	10343,00!	1314,63	1622,09	818,00!		!
!22/05/2014!	10,39	12,82	7,00 !	13274,04	13584,05	10350,00!	1325,02	1634,91	825,00!		!
!23/05/2014!	10,19	12,32	5,00 !	13284,23	13596,37	10355,00!	1335,21	1647,23	830,00!		!
!24/05/2014!	8,53	11,19	6,00 !	13292,76	13607,56	10361,00!	1343,74	1658,42	836,00!		!
!25/05/2014!	7,68	9,27	4,00 !	13300,44	13616,83	10365,00!	1351,42	1667,69	840,00!		!
!26/05/2014!	9,39	11,59	5,00 !	13309,83	13628,42	10370,00!	1360,81	1679,28	845,00!		!
!27/05/2014!	9,31	11,14	5,00 !	13319,15	13639,56	10375,00!	1370,12	1690,42	850,00!		!
!28/05/2014!	11,83	13,23	4,00 !	13330,98	13652,79	10379,00!	1381,95	1703,65	854,00!		!
!29/05/2014!	11,79	13,38	4,00 !	13342,78	13666,17	10383,00!	1393,74	1717,03	858,00!		!
!30/05/2014!	8,26	10,22	4,00 !	13351,04	13676,38	10387,00!	1402,00	1727,25	862,00!		!
!31/05/2014!	11,73	13,05	4,00 !	13362,77	13689,43	10391,00!	1413,73	1740,30	866,00!		!
!01/06/2014!	11,42	13,17	5,00 !	13374,18	13702,59	10396,00!	1425,15	1753,47	871,00!		!
!02/06/2014!	9,48	12,09	7,00 !	13383,67	13714,68	10403,00!	1434,63	1765,56	878,00!		!
!03/06/2014!	8,32	10,01	5,00 !	13391,99	13724,68	10408,00!	1442,95	1775,57	883,00!		!
!04/06/2014!	9,27	11,87	6,00 !	13401,26	13736,55	10414,00!	1452,22	1787,44	889,00!		!
!05/06/2014!	8,86	11,37	6,00 !	13410,11	13747,92	10420,00!	1461,08	1798,81	895,00!		!
!06/06/2014!	9,02	11,03	6,00 !	13419,13	13758,95	10426,00!	1470,10	1809,84	901,00!		!
!07/06/2014!	7,82	9,56	5,00 !	13426,95	13768,50	10431,00!	1477,92	1819,40	906,00!		!
!09/06/2014!	7,65	9,10	4,00 !	13434,60	13777,60	10435,00!	1485,57	1828,50	910,00!		!
!10/06/2014!	8,20	9,95	5,00 !	13442,79	13787,55	10440,00!	1493,77	1838,45	915,00!		!
!11/06/2014!	9,11	11,30	6,00 !	13451,90	13798,85	10446,00!	1502,88	1849,75	921,00!		!
!12/06/2014!	5,72	6,84	4,00 !	13457,62	13805,70	10450,00!	1508,60	1856,59	925,00!		!
!13/06/2014!	7,14	9,38	5,00 !	13464,75	13815,08	10455,00!	1515,74	1865,97	930,00!		!
!14/06/2014!	7,93	9,92	5,00 !	13472,68	13825,00	10460,00!	1523,67	1875,89	935,00!		!
!15/06/2014!	10,11	12,18	5,00 !	13482,79	13837,18	10465,00!	1533,78	1888,07	940,00!		!
!16/06/2014!	13,43	16,89	8,00 !	13496,22	13854,06	10473,00!	1547,21	1904,96	948,00!		!
!17/06/2014!	2,88	3,50	2,00 !	13499,10	13857,56	10475,00!	1550,09	1908,46	950,00!		!
!18/06/2014!	11,53	14,60	7,00 !	13510,63	13872,16	10482,00!	1561,62	1923,06	957,00!		!
!19/06/2014!	8,73	11,86	7,00 !	13519,36	13884,01	10489,00!	1570,35	1934,92	964,00!		!
!20/06/2014!	11,89	14,81	8,00 !	13531,25	13898,83	10497,00!	1582,24	1949,73	972,00!		!
!21/06/2014!	9,73	11,41	5,00 !	13540,98	13910,24	10502,00!	1591,97	1961,14	977,00!		!
!22/06/2014!	7,68	9,41	4,00 !	13548,66	13919,65	10506,00!	1599,65	1970,55	981,00!		!
!23/06/2014!	9,51	12,14	7,00 !	13558,17	13931,79	10513,00!	1609,16	1982,69	988,00!		!
!24/06/2014!	7,74	10,86	8,00 !	13565,90	13942,64	10521,00!	1616,90	1993,55	996,00!		!
!25/06/2014!	9,17	12,43	8,00 !	13575,07	13955,07	10529,00!	1626,07	2005,98	1004,00!		!
!26/06/2014!	6,75	8,73	4,00 !	13581,82	13963,80	10533,00!	1632,82	2014,71	1008,00!		!
!27/06/2014!	6,10	8,22	5,00 !	13587,92	13972,02	10538,00!	1638,92	2022,93	1013,00!		!
!28/06/2014!	7,00	8,63	4,00 !	13594,92	13980,65	10542,00!	1645,92	2031,56	1017,00!		!
!29/06/2014!	1,48	1,88	1,00 !	13596,40	13982,53	10543,00!	1647,40	2033,44	1018,00!		!
!30/06/2014!	6,79	9,39	5,00 !	13603,18	13991,92	10548,00!	1654,19	2042,83	1023,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!01/07/2014!	7,31	9,28	5,00 !	13610,50	14001,20	10553,00!	1661,50	2052,11	1028,00!		!
!02/07/2014!	11,92	14,50	6,00 !	13622,42	14015,70	10559,00!	1673,42	2066,61	1034,00!		!
!03/07/2014!	9,36	11,22	5,00 !	13631,77	14026,92	10564,00!	1682,78	2077,83	1039,00!		!
!04/07/2014!	10,52	12,65	6,00 !	13642,30	14039,57	10570,00!	1693,30	2090,48	1045,00!		!
!05/07/2014!	12,06	14,07	6,00 !	13654,36	14053,63	10576,00!	1705,36	2104,55	1051,00!		!
!06/07/2014!	13,88	16,15	8,00 !	13668,25	14069,78	10584,00!	1719,24	2120,70	1059,00!		!
!07/07/2014!	11,15	13,53	7,00 !	13679,40	14083,31	10591,00!	1730,39	2134,23	1066,00!		!
!08/07/2014!	10,59	14,19	8,00 !	13690,00	14097,51	10599,00!	1740,98	2148,42	1074,00!		!
!09/07/2014!	10,65	14,27	8,00 !	13700,65	14111,77	10607,00!	1751,63	2162,69	1082,00!		!
!10/07/2014!	10,90	13,53	8,00 !	13711,55	14125,30	10615,00!	1762,53	2176,22	1090,00!		!
!11/07/2014!	14,38	16,87	8,00 !	13725,93	14142,17	10623,00!	1776,91	2193,09	1098,00!		!
!12/07/2014!	14,00	16,45	8,00 !	13739,93	14158,61	10631,00!	1790,91	2209,54	1106,00!		!
!13/07/2014!	14,04	16,15	8,00 !	13753,98	14174,75	10639,00!	1804,95	2225,69	1114,00!		!
!14/07/2014!	7,18	9,62	7,00 !	13761,16	14184,38	10646,00!	1812,13	2235,31	1121,00!		!
!15/07/2014!	10,09	12,76	8,00 !	13771,25	14197,13	10654,00!	1822,22	2248,07	1129,00!		!
!16/07/2014!	7,82	9,82	5,00 !	13779,08	14206,95	10659,00!	1830,04	2257,89	1134,00!		!
!18/07/2014!	4,97	6,52	4,00 !	13784,05	14213,47	10663,00!	1835,01	2264,41	1138,00!		!
!19/07/2014!	7,50	9,53	6,00 !	13791,55	14223,00	10669,00!	1842,51	2273,94	1144,00!		!
!20/07/2014!	11,08	13,45	7,00 !	13802,63	14236,45	10676,00!	1853,59	2287,39	1151,00!		!
!21/07/2014!	13,80	16,32	8,00 !	13816,43	14252,77	10684,00!	1867,39	2303,71	1159,00!		!
!22/07/2014!	14,10	16,53	8,00 !	13830,53	14269,29	10692,00!	1881,49	2320,24	1167,00!		!
!23/07/2014!	14,09	16,35	8,00 !	13844,62	14285,64	10700,00!	1895,58	2336,59	1175,00!		!
!24/07/2014!	7,24	8,91	4,00 !	13851,85	14294,54	10704,00!	1902,82	2345,50	1179,00!		!
!25/07/2014!	8,24	9,80	4,00 !	13860,09	14304,34	10708,00!	1911,06	2355,30	1183,00!		!
!26/07/2014!	10,13	12,50	6,00 !	13870,22	14316,84	10714,00!	1921,19	2367,80	1189,00!		!
!27/07/2014!	9,14	11,67	8,00 !	13879,36	14328,51	10722,00!	1930,33	2379,47	1197,00!		!
!28/07/2014!	5,41	6,59	3,00 !	13884,77	14335,10	10725,00!	1935,74	2386,06	1200,00!		!
!29/07/2014!	8,22	9,72	4,00 !	13892,99	14344,82	10729,00!	1943,96	2395,78	1204,00!		!
!30/07/2014!	3,27	4,16	2,00 !	13896,26	14348,98	10731,00!	1947,23	2399,94	1206,00!		!
!31/07/2014!	6,51	9,08	9,00 !	13902,75	14358,07	10740,00!	1953,74	2409,02	1215,00!		!
!01/08/2014!	6,80	8,25	4,00 !	13909,55	14366,32	10744,00!	1960,54	2417,27	1219,00!		!
!02/08/2014!	8,40	10,88	7,00 !	13917,95	14377,20	10751,00!	1968,94	2428,15	1226,00!		!
!03/08/2014!	10,42	12,57	6,00 !	13928,36	14389,77	10757,00!	1979,36	2440,72	1232,00!		!
!04/08/2014!	10,13	12,16	0,00 !	13938,49	14401,94	10763,00!	1989,49	2452,88	1232,00!		!
!05/08/2014!	12,80	15,33	7,00 !	13951,29	14417,26	10770,00!	2002,29	2468,21	1239,00!		!
!06/08/2014!	14,08	16,66	8,00 !	13965,37	14433,93	10778,00!	2016,37	2484,87	1247,00!		!
!07/08/2014!	14,26	16,65	8,00 !	13979,64	14450,57	10786,00!	2030,63	2501,52	1255,00!		!
!08/08/2014!	14,13	16,75	8,00 !	13993,77	14467,32	10794,00!	2044,76	2518,27	1263,00!		!
!09/08/2014!	13,97	16,29	8,00 !	14007,73	14483,63	10802,00!	2058,73	2534,56	1271,00!		!
!10/08/2014!	8,13	10,23	7,00 !	14015,87	14493,86	10809,00!	2066,86	2544,79	1278,00!		!
!11/08/2014!	7,96	9,98	6,00 !	14023,83	14503,84	10815,00!	2074,82	2554,77	1284,00!		!
!12/08/2014!	6,32	8,14	5,00 !	14030,15	14511,98	10820,00!	2081,14	2562,91	1289,00!		!
!13/08/2014!	11,19	13,22	7,00 !	14041,34	14525,19	10827,00!	2092,33	2576,13	1296,00!		!
!14/08/2014!	8,28	11,33	7,00 !	14049,62	14536,53	10834,00!	2100,61	2587,46	1303,00!		!
!15/08/2014!	8,40	10,79	6,00 !	14058,01	14547,32	10840,00!	2109,01	2598,25	1309,00!		!
!16/08/2014!	8,42	10,61	6,00 !	14066,42	14557,92	10846,00!	2117,43	2608,86	1315,00!		!
!17/08/2014!	7,80	9,87	5,00 !	14074,22	14567,79	10851,00!	2125,23	2618,73	1320,00!		!
!18/08/2014!	8,28	10,31	6,00 !	14082,50	14578,10	10857,00!	2133,51	2629,04	1326,00!		!
!19/08/2014!	8,12	10,48	6,00 !	14090,61	14588,59	10863,00!	2141,63	2639,52	1332,00!		!
!20/08/2014!	7,74	9,24	4,00 !	14098,35	14597,83	10867,00!	2149,37	2648,76	1336,00!		!
!21/08/2014!	6,12	7,98	5,00 !	14104,47	14605,81	10872,00!	2155,49	2656,74	1341,00!		!
!22/08/2014!	8,47	10,88	6,00 !	14112,94	14616,69	10878,00!	2163,96	2667,62	1347,00!		!
!23/08/2014!	7,31	9,36	6,00 !	14120,26	14626,05	10884,00!	2171,27	2676,98	1353,00!		!
!24/08/2014!	1,59	2,20	2,00 !	14121,84	14628,25	10886,00!	2172,86	2679,18	1355,00!		!
!25/08/2014!	11,16	13,47	5,00 !	14133,00	14641,72	10891,00!	2184,02	2692,65	1360,00!		!
!26/08/2014!	8,34	9,87	4,00 !	14141,34	14651,59	10895,00!	2192,36	2702,52	1364,00!		!
!27/08/2014!	9,31	11,37	5,00 !	14150,64	14662,96	10900,00!	2201,67	2713,89	1369,00!		!
!28/08/2014!	1,50	2,32	2,00 !	14152,14	14665,28	10902,00!	2203,17	2716,21	1371,00!		!
!29/08/2014!	7,90	9,94	5,00 !	14160,04	14675,23	10907,00!	2211,07	2726,15	1376,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!30/08/2014!	8,39	10,70	6,00 !	14168,42	14685,93	10913,00!	2219,46	2736,85	1382,00!		!
!31/08/2014!	8,83	11,24	7,00 !	14177,26	14697,17	10920,00!	2228,29	2748,09	1389,00!		!
!01/09/2014!	6,67	8,90	6,00 !	14183,93	14706,07	10926,00!	2234,96	2756,99	1395,00!		!
!02/09/2014!	6,12	7,30	4,00 !	14190,05	14713,37	10930,00!	2241,08	2764,29	1399,00!		!
!03/09/2014!	8,63	11,21	6,00 !	14198,68	14724,59	10936,00!	2249,71	2775,50	1405,00!		!
!04/09/2014!	8,76	10,70	6,00 !	14207,44	14735,29	10942,00!	2258,47	2786,20	1411,00!		!
!05/09/2014!	7,41	10,12	7,00 !	14214,84	14745,40	10949,00!	2265,88	2796,32	1418,00!		!
!06/09/2014!	7,01	9,75	7,00 !	14221,86	14755,13	10956,00!	2272,89	2806,07	1425,00!		!
!07/09/2014!	8,60	11,18	6,00 !	14230,46	14766,32	10962,00!	2281,49	2817,25	1431,00!		!
!08/09/2014!	9,15	12,25	7,00 !	14239,62	14778,57	10969,00!	2290,64	2829,50	1438,00!		!
!09/09/2014!	11,49	13,95	8,00 !	14251,10	14792,51	10977,00!	2302,13	2843,45	1446,00!		!
!10/09/2014!	7,73	10,29	6,00 !	14258,83	14802,78	10983,00!	2309,86	2853,74	1452,00!		!
!11/09/2014!	6,65	8,75	6,00 !	14265,47	14811,52	10989,00!	2316,51	2862,49	1458,00!		!
!12/09/2014!	9,27	11,00	5,00 !	14274,74	14822,52	10994,00!	2325,78	2873,49	1463,00!		!
!13/09/2014!	9,33	10,89	5,00 !	14284,07	14833,42	10999,00!	2335,11	2884,38	1468,00!		!
!14/09/2014!	14,06	16,16	8,00 !	14298,14	14849,59	11007,00!	2349,17	2900,54	1476,00!		!
!15/09/2014!	10,29	12,41	6,00 !	14308,42	14861,99	11013,00!	2359,46	2912,95	1482,00!		!
!16/09/2014!	14,03	16,47	8,00 !	14322,45	14878,46	11021,00!	2373,49	2929,42	1490,00!		!
!17/09/2014!	7,99	10,02	6,00 !	14330,44	14888,48	11027,00!	2381,48	2939,44	1496,00!		!
!18/09/2014!	9,26	10,73	4,00 !	14339,70	14899,21	11031,00!	2390,74	2950,17	1500,00!		!
!19/09/2014!	9,22	11,06	4,00 !	14348,92	14910,27	11035,00!	2399,96	2961,23	1504,00!		!
!20/09/2014!	8,38	9,72	4,00 !	14357,30	14919,99	11039,00!	2408,34	2970,95	1508,00!		!
!21/09/2014!	7,78	9,40	5,00 !	14365,09	14929,39	11044,00!	2416,12	2980,35	1513,00!		!
!22/09/2014!	7,30	8,83	5,00 !	14372,39	14938,23	11049,00!	2423,42	2989,18	1518,00!		!
!23/09/2014!	7,79	9,26	4,00 !	14380,18	14947,49	11053,00!	2431,21	2998,44	1522,00!		!
!24/09/2014!	6,17	7,83	5,00 !	14386,35	14955,33	11058,00!	2437,38	3006,27	1527,00!		!
!25/09/2014!	12,34	14,60	7,00 !	14398,70	14969,93	11065,00!	2449,72	3020,87	1534,00!		!
!26/09/2014!	13,94	16,62	8,00 !	14412,64	14986,55	11073,00!	2463,66	3037,49	1542,00!		!
!27/09/2014!	14,13	16,60	8,00 !	14426,77	15003,17	11081,00!	2477,79	3054,09	1550,00!		!
!28/09/2014!	6,53	7,98	5,00 !	14433,28	15011,15	11086,00!	2484,32	3062,07	1555,00!		!
!29/09/2014!	7,77	9,79	5,00 !	14441,05	15020,94	11091,00!	2492,09	3071,86	1560,00!		!
!30/09/2014!	8,37	10,01	5,00 !	14449,43	15030,96	11096,00!	2500,46	3081,87	1565,00!		!
!01/10/2014!	7,61	9,46	5,00 !	14457,03	15040,41	11101,00!	2508,07	3091,33	1570,00!		!
!02/10/2014!	12,90	15,65	7,00 !	14469,92	15056,06	11108,00!	2520,97	3106,98	1577,00!		!
!03/10/2014!	9,89	12,25	6,00 !	14479,81	15068,31	11114,00!	2530,86	3119,23	1583,00!		!
!04/10/2014!	10,50	13,26	8,00 !	14490,31	15081,56	11122,00!	2541,36	3132,49	1591,00!		!
!05/10/2014!	9,79	11,62	5,00 !	14500,11	15093,18	11127,00!	2551,15	3144,11	1596,00!		!
!06/10/2014!	10,11	13,56	9,00 !	14510,23	15106,75	11136,00!	2561,26	3157,67	1605,00!		!
!07/10/2014!	10,37	12,95	6,00 !	14520,59	15119,69	11142,00!	2571,63	3170,62	1611,00!		!
!08/10/2014!	9,39	11,44	5,00 !	14529,98	15131,13	11147,00!	2581,02	3182,06	1616,00!		!
!09/10/2014!	10,53	13,59	8,00 !	14540,51	15144,72	11155,00!	2591,55	3195,65	1624,00!		!
!10/10/2014!	8,30	10,69	6,00 !	14548,81	15155,40	11161,00!	2599,85	3206,34	1630,00!		!
!11/10/2014!	8,05	9,76	5,00 !	14556,86	15165,16	11166,00!	2607,90	3216,10	1635,00!		!
!12/10/2014!	5,32	6,69	4,00 !	14562,18	15171,85	11170,00!	2613,22	3222,79	1639,00!		!
!13/10/2014!	6,18	7,89	5,00 !	14568,37	15179,74	11175,00!	2619,40	3230,68	1644,00!		!
!14/10/2014!	6,22	7,71	4,00 !	14574,59	15187,45	11179,00!	2625,62	3238,39	1648,00!		!
!15/10/2014!	8,23	10,34	5,00 !	14582,82	15197,78	11184,00!	2633,85	3248,73	1653,00!		!
!16/10/2014!	7,46	9,45	5,00 !	14590,28	15207,23	11189,00!	2641,31	3258,18	1658,00!		!
!17/10/2014!	9,53	12,29	7,00 !	14599,81	15219,51	11196,00!	2650,84	3270,47	1665,00!		!
!18/10/2014!	14,15	16,60	8,00 !	14613,95	15236,11	11204,00!	2664,99	3287,07	1673,00!		!
!19/10/2014!	14,12	16,52	8,00 !	14628,06	15252,63	11212,00!	2679,11	3303,59	1681,00!		!
!20/10/2014!	8,03	10,23	6,00 !	14636,09	15262,86	11218,00!	2687,14	3313,82	1687,00!		!
!21/10/2014!	11,87	13,89	6,00 !	14647,95	15276,74	11224,00!	2699,01	3327,71	1693,00!		!
!22/10/2014!	8,37	9,83	4,00 !	14656,32	15286,57	11228,00!	2707,38	3337,54	1697,00!		!
!23/10/2014!	8,17	9,70	4,00 !	14664,49	15296,27	11232,00!	2715,55	3347,24	1701,00!		!
!24/10/2014!	9,72	11,75	5,00 !	14674,21	15308,02	11237,00!	2725,27	3358,99	1706,00!		!
!25/10/2014!	14,07	16,50	8,00 !	14688,27	15324,52	11245,00!	2739,34	3375,49	1714,00!		!
!26/10/2014!	6,53	7,62	3,00 !	14694,80	15332,14	11248,00!	2745,87	3383,11	1717,00!		!
!27/10/2014!	10,55	12,79	7,00 !	14705,35	15344,93	11255,00!	2756,42	3395,90	1724,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!28/10/2014!	7,08	8,77	5,00 !	14712,43	15353,70	11260,00!	2763,50	3404,67	1729,00!		!
!29/10/2014!	6,44	7,77	4,00 !	14718,87	15361,47	11264,00!	2769,94	3412,44	1733,00!		!
!30/10/2014!	7,44	9,30	5,00 !	14726,31	15370,77	11269,00!	2777,38	3421,74	1738,00!		!
!31/10/2014!	6,30	8,01	4,00 !	14732,61	15378,78	11273,00!	2783,68	3429,75	1742,00!		!
!01/11/2014!	8,58	11,16	7,00 !	14741,19	15389,94	11280,00!	2792,26	3440,91	1749,00!		!
!02/11/2014!	10,80	13,29	7,00 !	14752,00	15403,25	11287,00!	2803,06	3454,20	1756,00!		!
!03/11/2014!	12,21	14,87	7,00 !	14764,20	15418,12	11294,00!	2815,27	3469,07	1763,00!		!
!04/11/2014!	14,02	16,23	8,00 !	14778,23	15434,35	11302,00!	2829,29	3485,30	1771,00!		!
!05/11/2014!	13,91	16,54	8,00 !	14792,15	15450,88	11310,00!	2843,20	3501,84	1779,00!		!
!06/11/2014!	10,40	12,85	6,00 !	14802,55	15463,74	11316,00!	2853,60	3514,69	1785,00!		!
!07/11/2014!	6,52	8,39	5,00 !	14809,07	15472,12	11321,00!	2860,12	3523,08	1790,00!		!
!08/11/2014!	2,40	3,27	3,00 !	14811,47	15475,39	11324,00!	2862,52	3526,35	1793,00!		!
!09/11/2014!	9,20	11,59	7,00 !	14820,67	15486,97	11331,00!	2871,72	3537,94	1800,00!		!
!10/11/2014!	8,77	10,88	7,00 !	14829,43	15497,85	11338,00!	2880,49	3548,82	1807,00!		!
!11/11/2014!	8,33	10,78	5,00 !	14837,76	15508,63	11343,00!	2888,82	3559,60	1812,00!		!
!12/11/2014!	8,84	11,70	6,00 !	14846,59	15520,33	11349,00!	2897,66	3571,30	1818,00!		!
!13/11/2014!	9,29	12,21	7,00 !	14855,86	15532,54	11356,00!	2906,95	3583,51	1825,00!		!
!14/11/2014!	7,25	9,44	6,00 !	14863,11	15541,98	11362,00!	2914,20	3592,95	1831,00!		!
!16/11/2014!	8,07	10,80	6,00 !	14871,18	15552,79	11368,00!	2922,27	3603,75	1837,00!		!
!17/11/2014!	10,03	13,71	8,00 !	14881,20	15566,50	11376,00!	2932,30	3617,46	1845,00!		!
!18/11/2014!	13,88	16,37	8,00 !	14895,10	15582,87	11384,00!	2946,18	3633,83	1853,00!		!
!19/11/2014!	13,85	16,39	8,00 !	14908,95	15599,26	11392,00!	2960,03	3650,22	1861,00!		!
!20/11/2014!	10,33	12,97	6,00 !	14919,29	15612,25	11398,00!	2970,36	3663,19	1867,00!		!
!21/11/2014!	8,38	10,82	6,00 !	14927,67	15623,07	11404,00!	2978,74	3674,01	1873,00!		!
!22/11/2014!	7,54	9,78	5,00 !	14935,22	15632,84	11409,00!	2986,28	3683,79	1878,00!		!
!23/11/2014!	7,36	11,42	10,00 !	14942,57	15644,25	11419,00!	2993,64	3695,21	1888,00!		!
!24/11/2014!	6,40	7,75	4,00 !	14948,97	15652,00	11423,00!	3000,04	3702,96	1892,00!		!
!25/11/2014!	11,16	14,99	8,00 !	14960,13	15667,00	11431,00!	3011,20	3717,95	1900,00!		!
!26/11/2014!	8,70	11,38	6,00 !	14968,83	15678,38	11437,00!	3019,90	3729,33	1906,00!		!
!27/11/2014!	6,43	9,91	9,00 !	14975,27	15688,29	11446,00!	3026,33	3739,24	1915,00!		!
!28/11/2014!	4,73	7,14	5,00 !	14980,00	15695,42	11451,00!	3031,06	3746,38	1920,00!		!
!29/11/2014!	8,05	10,20	6,00 !	14988,05	15705,62	11457,00!	3039,11	3756,58	1926,00!		!
!30/11/2014!	6,87	9,08	7,00 !	14994,92	15714,68	11464,00!	3045,98	3765,66	1933,00!		!
!01/12/2014!	10,83	13,26	7,00 !	15005,76	15727,94	11471,00!	3056,81	3778,92	1940,00!		!
!02/12/2014!	11,36	13,64	7,00 !	15017,13	15741,59	11478,00!	3068,17	3792,56	1947,00!		!
!03/12/2014!	8,05	10,37	6,00 !	15025,18	15751,98	11484,00!	3076,22	3802,93	1953,00!		!
!04/12/2014!	9,31	11,81	7,00 !	15034,50	15763,77	11491,00!	3085,53	3814,74	1960,00!		!
!05/12/2014!	8,38	10,38	5,00 !	15042,88	15774,15	11496,00!	3093,91	3825,12	1965,00!		!
!06/12/2014!	7,71	9,89	6,00 !	15050,60	15784,05	11502,00!	3101,62	3835,01	1971,00!		!
!07/12/2014!	10,40	13,20	8,00 !	15061,00	15797,25	11510,00!	3112,02	3848,21	1979,00!		!
!08/12/2014!	10,40	12,63	6,00 !	15071,40	15809,87	11516,00!	3122,42	3860,84	1985,00!		!
!09/12/2014!	11,22	13,70	7,00 !	15082,63	15823,57	11523,00!	3133,64	3874,54	1992,00!		!
!10/12/2014!	9,64	12,14	7,00 !	15092,26	15835,72	11530,00!	3143,28	3886,68	1999,00!		!
!11/12/2014!	6,87	9,35	7,00 !	15099,13	15845,07	11537,00!	3150,15	3896,03	2006,00!		!
!12/12/2014!	2,32	2,52	1,00 !	15101,45	15847,59	11538,00!	3152,47	3898,55	2007,00!		!
!13/12/2014!	8,50	10,92	8,00 !	15109,95	15858,52	11546,00!	3160,97	3909,47	2015,00!		!
!14/12/2014!	10,44	12,68	6,00 !	15120,39	15871,20	11552,00!	3171,41	3922,15	2021,00!		!
!15/12/2014!	14,01	16,21	8,00 !	15134,39	15887,41	11560,00!	3185,42	3938,36	2029,00!		!
!16/12/2014!	9,78	11,93	6,00 !	15144,17	15899,34	11566,00!	3195,20	3950,29	2035,00!		!
!17/12/2014!	9,22	11,34	5,00 !	15153,39	15910,68	11571,00!	3204,42	3961,63	2040,00!		!
!18/12/2014!	9,95	13,07	7,00 !	15163,35	15923,75	11578,00!	3214,37	3974,70	2047,00!		!
!19/12/2014!	9,79	12,46	6,00 !	15173,13	15936,20	11584,00!	3224,16	3987,16	2053,00!		!
!20/12/2014!	9,96	12,35	6,00 !	15183,10	15948,55	11590,00!	3234,12	3999,51	2059,00!		!
!21/12/2014!	9,32	11,08	5,00 !	15192,42	15959,63	11595,00!	3243,44	4010,59	2064,00!		!
!22/12/2014!	8,78	11,18	6,00 !	15201,20	15970,82	11601,00!	3252,22	4021,77	2070,00!		!
!23/12/2014!	7,74	10,90	9,00 !	15208,95	15981,71	11610,00!	3259,96	4032,67	2079,00!		!
!24/12/2014!	5,80	7,86	6,00 !	15214,75	15989,57	11616,00!	3265,76	4040,53	2085,00!		!
!25/12/2014!	4,54	5,49	3,00 !	15219,29	15995,06	11619,00!	3270,30	4046,02	2088,00!		!
!26/12/2014!	7,87	9,76	6,00 !	15227,16	16004,82	11625,00!	3278,17	4055,78	2094,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!27/12/2014!	7,77	10,74	6,00 !	15234,94	16015,56	11631,00!	3285,94	4066,52	2100,00!		!
!28/12/2014!	4,10	4,72	2,00 !	15239,04	16020,28	11633,00!	3290,04	4071,24	2102,00!		!
!30/12/2014!	1,38	2,37	2,00 !	15240,42	16022,65	11635,00!	3291,42	4073,61	2104,00!		!
!31/12/2014!	6,05	7,59	4,00 !	15246,47	16030,24	11639,00!	3297,47	4081,20	2108,00!		!
!01/01/2015!	2,12	2,78	2,00 !	15248,59	16033,02	11641,00!	3299,59	4083,98	2110,00!		!
!02/01/2015!	8,17	11,30	8,00 !	15256,76	16044,33	11649,00!	3307,76	4095,28	2118,00!		!
!03/01/2015!	12,36	14,38	5,00 !	15269,12	16058,71	11654,00!	3320,12	4109,66	2123,00!		!
!04/01/2015!	11,73	13,11	4,00 !	15280,85	16071,81	11658,00!	3331,85	4122,77	2127,00!		!
!05/01/2015!	9,56	11,97	7,00 !	15290,41	16083,77	11665,00!	3341,41	4134,74	2134,00!		!
!06/01/2015!	8,27	9,64	4,00 !	15298,68	16093,40	11669,00!	3349,68	4144,38	2138,00!		!
!07/01/2015!	7,43	9,29	4,00 !	15306,11	16102,70	11673,00!	3357,11	4153,67	2142,00!		!
!08/01/2015!	8,35	10,23	6,00 !	15314,46	16112,93	11679,00!	3365,46	4163,90	2148,00!		!
!09/01/2015!	7,82	10,14	6,00 !	15322,30	16123,06	11685,00!	3373,28	4174,04	2154,00!		!
!10/01/2015!	7,96	10,16	6,00 !	15330,26	16133,23	11691,00!	3381,24	4184,20	2160,00!		!
!11/01/2015!	4,42	6,12	4,00 !	15334,67	16139,35	11695,00!	3385,66	4190,32	2164,00!		!
!12/01/2015!	5,51	7,12	4,00 !	15340,18	16146,47	11699,00!	3391,17	4197,44	2168,00!		!
!13/01/2015!	8,39	10,55	6,00 !	15348,57	16157,02	11705,00!	3399,56	4207,99	2174,00!		!
!14/01/2015!	6,62	8,27	5,00 !	15355,19	16165,28	11710,00!	3406,18	4216,26	2179,00!		!
!15/01/2015!	8,32	10,12	5,00 !	15363,50	16175,41	11715,00!	3414,50	4226,38	2184,00!		!
!16/01/2015!	8,22	9,76	4,00 !	15371,72	16185,18	11719,00!	3422,72	4236,14	2188,00!		!
!17/01/2015!	8,20	9,55	4,00 !	15379,92	16194,73	11723,00!	3430,92	4245,69	2192,00!		!
!18/01/2015!	8,24	9,67	4,00 !	15388,16	16204,40	11727,00!	3439,16	4255,36	2196,00!		!
!19/01/2015!	12,19	14,21	5,00 !	15400,35	16218,62	11732,00!	3451,35	4269,57	2201,00!		!
!20/01/2015!	11,84	13,63	4,00 !	15412,19	16232,25	11736,00!	3463,19	4283,20	2205,00!		!
!21/01/2015!	11,83	13,41	4,00 !	15424,02	16245,66	11740,00!	3475,02	4296,61	2209,00!		!
!22/01/2015!	9,88	13,09	7,00 !	15433,90	16258,74	11747,00!	3484,90	4309,70	2216,00!		!
!23/01/2015!	5,82	7,89	5,00 !	15439,72	16266,62	11752,00!	3490,72	4317,59	2221,00!		!
!24/01/2015!	6,08	8,41	8,00 !	15445,80	16275,03	11760,00!	3496,80	4326,00	2229,00!		!
!25/01/2015!	8,40	10,64	6,00 !	15454,20	16285,68	11766,00!	3505,20	4336,64	2235,00!		!
!26/01/2015!	9,26	11,95	7,00 !	15463,46	16297,63	11773,00!	3514,46	4348,59	2242,00!		!
!27/01/2015!	9,95	12,92	8,00 !	15473,41	16310,57	11781,00!	3524,41	4361,51	2250,00!		!
!28/01/2015!	10,15	13,20	8,00 !	15483,57	16323,77	11789,00!	3534,56	4374,71	2258,00!		!
!29/01/2015!	7,55	9,50	5,00 !	15491,11	16333,27	11794,00!	3542,11	4384,21	2263,00!		!
!30/01/2015!	8,35	11,11	7,00 !	15499,44	16344,37	11801,00!	3550,46	4395,32	2270,00!		!
!31/01/2015!	7,02	8,70	5,00 !	15506,46	16353,07	11806,00!	3557,48	4404,02	2275,00!		!
!01/02/2015!	9,04	11,31	7,00 !	15515,49	16364,39	11813,00!	3566,52	4415,33	2282,00!		!
!02/02/2015!	5,67	7,27	6,00 !	15521,15	16371,65	11819,00!	3572,19	4422,60	2288,00!		!
!03/02/2015!	6,09	7,88	6,00 !	15527,24	16379,53	11825,00!	3578,28	4430,48	2294,00!		!
!04/02/2015!	4,34	6,27	6,00 !	15531,57	16385,80	11831,00!	3582,62	4436,75	2300,00!		!
!05/02/2015!	8,32	11,19	8,00 !	15539,89	16397,00	11839,00!	3590,94	4447,94	2308,00!		!
!06/02/2015!	9,91	12,82	7,00 !	15549,81	16409,82	11846,00!	3600,85	4460,76	2315,00!		!
!07/02/2015!	10,70	12,83	6,00 !	15560,51	16422,65	11852,00!	3611,55	4473,59	2321,00!		!
!08/02/2015!	9,35	11,50	6,00 !	15569,85	16434,16	11858,00!	3620,90	4485,09	2327,00!		!
!09/02/2015!	10,52	12,94	6,00 !	15580,37	16447,10	11864,00!	3631,42	4498,03	2333,00!		!
!10/02/2015!	10,15	13,56	8,00 !	15590,51	16460,66	11872,00!	3641,57	4511,59	2341,00!		!
!11/02/2015!	9,63	12,29	8,00 !	15600,13	16472,95	11880,00!	3651,20	4523,88	2349,00!		!
!12/02/2015!	9,02	10,78	4,00 !	15609,15	16483,73	11884,00!	3660,22	4534,66	2353,00!		!
!13/02/2015!	11,40	13,94	6,00 !	15620,55	16497,68	11890,00!	3671,62	4548,60	2359,00!		!
!14/02/2015!	8,73	12,08	8,00 !	15629,28	16509,76	11898,00!	3680,35	4560,68	2367,00!		!
!15/02/2015!	8,27	9,60	4,00 !	15637,55	16519,36	11902,00!	3688,62	4570,28	2371,00!		!
!16/02/2015!	8,27	9,48	4,00 !	15645,82	16528,84	11906,00!	3696,89	4579,76	2375,00!		!
!17/02/2015!	6,73	8,44	4,00 !	15652,55	16537,28	11910,00!	3703,62	4588,20	2379,00!		!
!18/02/2015!	5,81	8,74	8,00 !	15658,36	16546,01	11918,00!	3709,43	4596,94	2387,00!		!
!19/02/2015!	8,95	11,44	7,00 !	15667,31	16557,46	11925,00!	3718,38	4608,38	2394,00!		!
!20/02/2015!	14,27	16,73	8,00 !	15681,58	16574,19	11933,00!	3732,65	4625,11	2402,00!		!
!21/02/2015!	8,33	10,21	6,00 !	15689,92	16584,40	11939,00!	3740,98	4635,32	2408,00!		!
!22/02/2015!	5,27	7,35	7,00 !	15695,19	16591,75	11946,00!	3746,25	4642,67	2415,00!		!
!23/02/2015!	2,68	4,58	4,00 !	15697,88	16596,33	11950,00!	3748,93	4647,25	2419,00!		!
!24/02/2015!	10,68	13,25	7,00 !	15708,56	16609,58	11957,00!	3759,61	4660,50	2426,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!25/02/2015!	9,46	11,96	7,00 !	15718,02	16621,54	11964,00!	3769,07	4672,46	2433,00!		!
!26/02/2015!	6,50	7,99	5,00 !	15724,52	16629,53	11969,00!	3775,57	4680,45	2438,00!		!
!27/02/2015!	3,36	4,95	4,00 !	15727,88	16634,48	11973,00!	3778,93	4685,40	2442,00!		!
!28/02/2015!	7,57	10,41	6,00 !	15735,45	16644,90	11979,00!	3786,50	4695,81	2448,00!		!
!01/03/2015!	4,03	4,63	2,00 !	15739,48	16649,53	11981,00!	3790,53	4700,44	2450,00!		!
!02/03/2015!	8,08	9,95	5,00 !	15747,56	16659,48	11986,00!	3798,61	4710,39	2455,00!		!
!03/03/2015!	4,49	5,91	3,00 !	15752,05	16665,39	11989,00!	3803,10	4716,30	2458,00!		!
!04/03/2015!	10,02	13,30	9,00 !	15762,08	16678,68	11998,00!	3813,12	4729,60	2467,00!		!
!05/03/2015!	3,58	4,96	3,00 !	15765,66	16683,64	12001,00!	3816,70	4734,56	2470,00!		!
!06/03/2015!	3,15	5,28	4,00 !	15768,82	16688,92	12005,00!	3819,85	4739,84	2474,00!		!
!07/03/2015!	5,88	7,76	6,00 !	15774,69	16696,67	12011,00!	3825,73	4747,60	2480,00!		!
!08/03/2015!	1,37	2,15	2,00 !	15776,06	16698,82	12013,00!	3827,10	4749,75	2482,00!		!
!09/03/2015!	8,76	12,18	8,00 !	15784,81	16711,00	12021,00!	3835,86	4761,93	2490,00!		!
!10/03/2015!	6,35	7,35	3,00 !	15791,16	16718,35	12024,00!	3842,21	4769,28	2493,00!		!
!11/03/2015!	13,94	16,60	8,00 !	15805,11	16734,94	12032,00!	3856,15	4785,88	2501,00!		!
!12/03/2015!	13,96	16,35	8,00 !	15819,08	16751,29	12040,00!	3870,11	4802,23	2509,00!		!
!13/03/2015!	13,94	16,31	8,00 !	15833,01	16767,61	12048,00!	3884,05	4818,54	2517,00!		!
!14/03/2015!	13,99	16,18	8,00 !	15846,99	16783,79	12056,00!	3898,04	4834,72	2525,00!		!
!15/03/2015!	14,11	16,41	8,00 !	15861,09	16800,20	12064,00!	3912,15	4851,13	2533,00!		!
!16/03/2015!	6,73	9,23	7,00 !	15867,80	16809,43	12071,00!	3918,88	4860,36	2540,00!		!
!17/03/2015!	8,32	12,38	11,00 !	15876,12	16821,81	12082,00!	3927,20	4872,74	2551,00!		!
!18/03/2015!	8,73	11,11	6,00 !	15884,86	16832,93	12088,00!	3935,93	4883,85	2557,00!		!
!19/03/2015!	4,81	7,14	5,00 !	15889,67	16840,06	12093,00!	3940,74	4890,99	2562,00!		!
!20/03/2015!	2,93	4,74	4,00 !	15892,60	16844,81	12097,00!	3943,67	4895,73	2566,00!		!
!21/03/2015!	6,56	9,37	8,00 !	15899,16	16854,17	12105,00!	3950,23	4905,10	2574,00!		!
!22/03/2015!	3,45	4,53	2,00 !	15902,61	16858,70	12107,00!	3953,68	4909,63	2576,00!		!
!23/03/2015!	10,11	12,87	9,00 !	15912,73	16871,59	12116,00!	3963,79	4922,50	2585,00!		!
!24/03/2015!	5,99	8,18	6,00 !	15918,72	16879,78	12122,00!	3969,78	4930,68	2591,00!		!
!25/03/2015!	9,47	12,17	6,00 !	15928,18	16891,95	12128,00!	3979,25	4942,85	2597,00!		!
!26/03/2015!	5,49	7,56	6,00 !	15933,67	16899,52	12134,00!	3984,74	4950,41	2603,00!		!
!27/03/2015!	8,19	10,46	7,00 !	15941,86	16909,99	12141,00!	3992,93	4960,87	2610,00!		!
!28/03/2015!	8,04	10,21	6,00 !	15949,90	16920,21	12147,00!	4000,97	4971,08	2616,00!		!
!29/03/2015!	9,09	11,94	8,00 !	15958,99	16932,15	12155,00!	4010,06	4983,02	2624,00!		!
!30/03/2015!	8,10	9,58	4,00 !	15967,09	16941,73	12159,00!	4018,16	4992,60	2628,00!		!
!31/03/2015!	8,30	9,63	4,00 !	15975,39	16951,36	12163,00!	4026,46	5002,23	2632,00!		!
!01/04/2015!	8,10	9,55	4,00 !	15983,49	16960,91	12167,00!	4034,56	5011,78	2636,00!		!
!02/04/2015!	6,66	7,99	4,00 !	15990,15	16968,89	12171,00!	4041,22	5019,77	2640,00!		!
!05/04/2015!	5,18	7,94	7,00 !	15995,31	16976,83	12178,00!	4046,40	5027,71	2647,00!		!
!06/04/2015!	7,44	9,20	5,00 !	16002,74	16986,03	12183,00!	4053,84	5036,91	2652,00!		!
!07/04/2015!	6,55	8,67	5,00 !	16009,29	16994,70	12188,00!	4060,39	5045,58	2657,00!		!
!08/04/2015!	8,16	11,23	8,00 !	16017,44	17005,93	12196,00!	4068,55	5056,81	2665,00!		!
!09/04/2015!	6,11	8,30	6,00 !	16023,55	17014,23	12202,00!	4074,66	5065,11	2671,00!		!
!10/04/2015!	3,65	5,60	4,00 !	16027,20	17019,83	12206,00!	4078,31	5070,71	2675,00!		!
!11/04/2015!	9,05	11,84	8,00 !	16036,25	17031,66	12214,00!	4087,36	5082,55	2683,00!		!
!12/04/2015!	12,66	15,71	10,00 !	16048,90	17047,37	12224,00!	4100,02	5098,26	2693,00!		!
!13/04/2015!	9,71	12,05	7,00 !	16058,61	17059,41	12231,00!	4109,73	5110,31	2700,00!		!
!14/04/2015!	3,60	4,46	3,00 !	16062,21	17063,87	12234,00!	4113,33	5114,77	2703,00!		!
!15/04/2015!	11,38	13,91	7,00 !	16073,59	17077,78	12241,00!	4124,71	5128,68	2710,00!		!
!16/04/2015!	6,93	9,93	6,00 !	16080,52	17087,73	12247,00!	4131,64	5138,61	2716,00!		!
!17/04/2015!	5,85	7,86	6,00 !	16086,37	17095,59	12253,00!	4137,49	5146,47	2722,00!		!
!18/04/2015!	7,27	9,62	7,00 !	16093,63	17105,19	12260,00!	4144,76	5156,09	2729,00!		!
!28/04/2015!	1,60	2,63	1,00 !	16095,23	17107,82	12261,00!	4146,36	5158,72	2730,00!		!
!29/04/2015!	1,47	2,48	2,00 !	16096,70	17110,30	12263,00!	4147,83	5161,20	2732,00!		!
!30/04/2015!	1,60	2,66	2,00 !	16098,30	17112,96	12265,00!	4149,43	5163,86	2734,00!		!
!01/05/2015!	1,47	2,18	2,00 !	16099,77	17115,14	12267,00!	4150,90	5166,04	2736,00!		!
!03/05/2015!	1,35	2,13	2,00 !	16101,12	17117,27	12269,00!	4152,25	5168,17	2738,00!		!
!06/05/2015!	5,96	8,08	6,00 !	16107,08	17125,34	12275,00!	4158,21	5176,25	2744,00!		!
!07/05/2015!	8,21	9,57	4,00 !	16115,30	17134,90	12279,00!	4166,42	5185,82	2748,00!		!
!08/05/2015!	4,42	5,35	2,00 !	16119,72	17140,25	12281,00!	4170,84	5191,17	2750,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!09/05/2015!	3,76	6,73	4,00 !	16123,48	17146,98	12285,00!	4174,60	5197,90	2754,00!		!
!10/05/2015!	9,68	12,16	7,00 !	16133,14	17159,15	12292,00!	4184,28	5210,06	2761,00!		!
!11/05/2015!	13,33	17,25	9,00 !	16146,48	17176,40	12301,00!	4197,61	5227,31	2770,00!		!
!12/05/2015!	9,67	11,81	6,00 !	16156,15	17188,21	12307,00!	4207,28	5239,12	2776,00!		!
!13/05/2015!	7,37	8,94	4,00 !	16163,52	17197,16	12311,00!	4214,65	5248,06	2780,00!		!
!14/05/2015!	8,21	9,40	4,00 !	16171,73	17206,56	12315,00!	4222,86	5257,46	2784,00!		!
!15/05/2015!	8,15	9,58	4,00 !	16179,88	17216,14	12319,00!	4231,01	5267,04	2788,00!		!
!16/05/2015!	8,13	9,47	4,00 !	16188,01	17225,62	12323,00!	4239,14	5276,51	2792,00!		!
!17/05/2015!	8,45	9,71	4,00 !	16196,46	17235,33	12327,00!	4247,59	5286,22	2796,00!		!
!18/05/2015!	6,72	8,29	4,00 !	16203,17	17243,63	12331,00!	4254,31	5294,51	2800,00!		!
!19/05/2015!	5,09	7,83	6,00 !	16208,25	17251,46	12337,00!	4259,40	5302,34	2806,00!		!
!20/05/2015!	4,67	5,68	3,00 !	16212,92	17257,14	12340,00!	4264,07	5308,02	2809,00!		!
!21/05/2015!	12,39	14,98	6,00 !	16225,31	17272,12	12346,00!	4276,46	5323,00	2815,00!		!
!22/05/2015!	8,48	10,69	6,00 !	16233,79	17282,80	12352,00!	4284,94	5333,69	2821,00!		!
!23/05/2015!	4,77	7,06	5,00 !	16238,57	17289,86	12357,00!	4289,71	5340,75	2826,00!		!
!28/05/2015!	0,00	1,00	0,00 !	16238,57	17290,86	12357,00!	4289,71	5341,75	2826,00!		!
!29/05/2015!	4,76	6,12	4,00 !	16243,34	17296,97	12361,00!	4294,47	5347,87	2830,00!		!
!30/05/2015!	14,00	16,79	8,00 !	16257,34	17313,76	12369,00!	4308,47	5364,66	2838,00!		!
!31/05/2015!	10,09	12,45	8,00 !	16267,42	17326,21	12377,00!	4318,56	5377,11	2846,00!		!
!01/06/2015!	11,82	14,77	9,00 !	16279,24	17340,97	12386,00!	4330,38	5391,88	2855,00!		!
!02/06/2015!	8,74	10,96	6,00 !	16287,97	17351,92	12392,00!	4339,12	5402,84	2861,00!		!
!03/06/2015!	10,22	13,29	8,00 !	16298,19	17365,20	12400,00!	4349,34	5416,13	2869,00!		!
!04/06/2015!	9,90	12,18	5,00 !	16308,09	17377,38	12405,00!	4359,24	5428,31	2874,00!		!
!05/06/2015!	6,37	7,63	4,00 !	16314,45	17385,00	12409,00!	4365,61	5435,94	2878,00!		!
!06/06/2015!	6,46	8,13	4,00 !	16320,91	17393,13	12413,00!	4372,07	5444,07	2882,00!		!
!07/06/2015!	6,64	8,26	4,00 !	16327,55	17401,39	12417,00!	4378,71	5452,33	2886,00!		!
!08/06/2015!	9,92	12,88	8,00 !	16337,47	17414,27	12425,00!	4388,63	5465,21	2894,00!		!
!09/06/2015!	8,52	11,02	6,00 !	16345,98	17425,29	12431,00!	4397,15	5476,23	2900,00!		!
!10/06/2015!	9,90	12,86	8,00 !	16355,88	17438,14	12439,00!	4407,05	5489,09	2908,00!		!
!11/06/2015!	10,24	13,77	8,00 !	16366,12	17451,91	12447,00!	4417,29	5502,86	2916,00!		!
!12/06/2015!	9,03	11,42	7,00 !	16375,14	17463,33	12454,00!	4426,32	5514,28	2923,00!		!
!13/06/2015!	9,98	12,24	7,00 !	16385,13	17475,57	12461,00!	4436,30	5526,52	2930,00!		!
!14/06/2015!	10,83	12,74	7,00 !	16395,94	17488,31	12468,00!	4447,13	5539,26	2937,00!		!
!15/06/2015!	13,89	16,28	8,00 !	16409,83	17504,60	12476,00!	4461,02	5555,54	2945,00!		!
!16/06/2015!	10,72	13,74	8,00 !	16420,55	17518,34	12484,00!	4471,74	5569,28	2953,00!		!
!17/06/2015!	7,46	9,49	5,00 !	16428,01	17527,83	12489,00!	4479,20	5578,77	2958,00!		!
!18/06/2015!	7,85	9,66	5,00 !	16435,86	17537,49	12494,00!	4487,05	5588,43	2963,00!		!
!19/06/2015!	10,41	13,28	8,00 !	16446,27	17550,77	12502,00!	4497,46	5601,71	2971,00!		!
!20/06/2015!	9,42	12,42	8,00 !	16455,68	17563,18	12510,00!	4506,88	5614,13	2979,00!		!
!21/06/2015!	2,30	3,39	3,00 !	16457,98	17566,56	12513,00!	4509,18	5617,52	2982,00!		!
!22/06/2015!	6,75	8,52	6,00 !	16464,74	17575,08	12519,00!	4515,93	5626,04	2988,00!		!
!23/06/2015!	9,37	12,22	8,00 !	16474,10	17587,29	12527,00!	4525,30	5638,26	2996,00!		!
!24/06/2015!	10,39	13,17	8,00 !	16484,49	17600,46	12535,00!	4535,69	5651,43	3004,00!		!
!25/06/2015!	9,82	12,13	8,00 !	16494,30	17612,58	12543,00!	4545,51	5663,56	3012,00!		!
!26/06/2015!	7,83	9,91	6,00 !	16502,13	17622,49	12549,00!	4553,34	5673,47	3018,00!		!
!27/06/2015!	1,35	2,34	1,00 !	16503,48	17624,83	12550,00!	4554,69	5675,81	3019,00!		!
!04/07/2015!	8,18	10,24	5,00 !	16511,66	17635,07	12555,00!	4562,87	5686,05	3024,00!		!
!05/07/2015!	11,80	14,38	8,00 !	16523,46	17649,45	12563,00!	4574,67	5700,43	3032,00!		!
!06/07/2015!	14,03	16,57	8,00 !	16537,49	17666,02	12571,00!	4588,70	5717,00	3040,00!		!
!07/07/2015!	13,39	16,16	7,00 !	16550,87	17682,17	12578,00!	4602,09	5733,16	3047,00!		!
!08/07/2015!	10,59	13,15	7,00 !	16561,45	17695,32	12585,00!	4612,68	5746,31	3054,00!		!
!09/07/2015!	2,85	3,59	2,00 !	16564,30	17698,91	12587,00!	4615,53	5749,90	3056,00!		!
!10/07/2015!	5,33	6,86	4,00 !	16569,64	17705,77	12591,00!	4620,86	5756,76	3060,00!		!
!11/07/2015!	6,95	8,94	6,00 !	16576,59	17714,70	12597,00!	4627,81	5765,70	3066,00!		!
!12/07/2015!	4,21	5,88	6,00 !	16580,80	17720,59	12603,00!	4632,02	5771,58	3072,00!		!
!13/07/2015!	6,26	8,93	7,00 !	16587,05	17729,51	12610,00!	4638,28	5780,51	3079,00!		!
!14/07/2015!	7,65	9,54	5,00 !	16594,70	17739,05	12615,00!	4645,93	5790,05	3084,00!		!
!15/07/2015!	6,64	9,01	6,00 !	16601,34	17748,05	12621,00!	4652,57	5799,06	3090,00!		!
!16/07/2015!	10,85	12,82	5,00 !	16612,17	17760,87	12626,00!	4663,42	5811,88	3095,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!17/07/2015!	12,33	14,05	4,00 !	16624,50	17774,92	12630,00!	4675,75	5825,93	3099,00!		!
!18/07/2015!	9,38	10,68	4,00 !	16633,88	17785,60	12634,00!	4685,13	5836,61	3103,00!		!
!19/07/2015!	8,14	9,98	6,00 !	16642,03	17795,58	12640,00!	4693,27	5846,59	3109,00!		!
!20/07/2015!	8,05	10,01	6,00 !	16650,08	17805,60	12646,00!	4701,32	5856,60	3115,00!		!
!21/07/2015!	9,74	12,07	8,00 !	16659,81	17817,66	12654,00!	4711,06	5868,67	3123,00!		!
!22/07/2015!	9,06	11,15	6,00 !	16668,87	17828,81	12660,00!	4720,12	5879,82	3129,00!		!
!23/07/2015!	8,01	13,12	6,00 !	16676,88	17841,92	12666,00!	4728,13	5892,94	3135,00!		!
!24/07/2015!	5,13	6,66	5,00 !	16682,01	17848,58	12671,00!	4733,26	5899,60	3140,00!		!
!25/07/2015!	7,97	10,44	6,00 !	16689,98	17859,02	12677,00!	4741,23	5910,04	3146,00!		!
!26/07/2015!	8,39	10,52	6,00 !	16698,37	17869,55	12683,00!	4749,62	5920,56	3152,00!		!
!27/07/2015!	8,28	10,82	8,00 !	16706,65	17880,36	12691,00!	4757,90	5931,38	3160,00!		!
!28/07/2015!	8,85	11,17	7,00 !	16715,50	17891,52	12698,00!	4766,75	5942,55	3167,00!		!
!29/07/2015!	7,88	10,36	7,00 !	16723,38	17901,88	12705,00!	4774,63	5952,91	3174,00!		!
!30/07/2015!	7,56	9,30	4,00 !	16730,94	17911,18	12709,00!	4782,19	5962,21	3178,00!		!
!31/07/2015!	8,21	10,00	6,00 !	16739,14	17921,18	12715,00!	4790,40	5972,21	3184,00!		!
!01/08/2015!	6,50	8,48	6,00 !	16745,64	17929,66	12721,00!	4796,90	5980,69	3190,00!		!
!02/08/2015!	9,53	11,63	6,00 !	16755,17	17941,30	12727,00!	4806,43	5992,32	3196,00!		!
!03/08/2015!	9,91	12,24	7,00 !	16765,07	17953,54	12734,00!	4816,34	6004,56	3203,00!		!
!04/08/2015!	8,53	10,47	5,00 !	16773,60	17964,01	12739,00!	4824,87	6015,03	3208,00!		!
!05/08/2015!	8,38	10,51	6,00 !	16781,96	17974,53	12745,00!	4833,25	6025,54	3214,00!		!
!06/08/2015!	9,20	11,08	7,00 !	16791,16	17985,61	12752,00!	4842,45	6036,62	3221,00!		!
!07/08/2015!	7,51	10,41	5,00 !	16798,66	17996,02	12757,00!	4849,96	6047,03	3226,00!		!
!08/08/2015!	7,32	9,73	6,00 !	16805,97	18005,76	12763,00!	4857,28	6056,76	3232,00!		!
!09/08/2015!	10,12	11,93	5,00 !	16816,09	18017,70	12768,00!	4867,40	6068,69	3237,00!		!
!10/08/2015!	10,14	12,20	6,00 !	16826,22	18029,90	12774,00!	4877,54	6080,89	3243,00!		!
!11/08/2015!	10,70	12,56	7,00 !	16836,92	18042,45	12781,00!	4888,24	6093,45	3250,00!		!
!12/08/2015!	14,01	16,29	8,00 !	16850,93	18058,74	12789,00!	4902,25	6109,74	3258,00!		!
!13/08/2015!	11,29	13,63	6,00 !	16862,22	18072,37	12795,00!	4913,54	6123,37	3264,00!		!
!14/08/2015!	5,97	8,11	6,00 !	16868,18	18080,49	12801,00!	4919,51	6131,48	3270,00!		!
!15/08/2015!	5,21	7,77	8,00 !	16873,40	18088,26	12809,00!	4924,72	6139,25	3278,00!		!
!16/08/2015!	7,24	9,70	5,00 !	16880,63	18097,97	12814,00!	4931,96	6148,95	3283,00!		!
!17/08/2015!	8,32	10,50	6,00 !	16888,95	18108,47	12820,00!	4940,28	6159,45	3289,00!		!
!18/08/2015!	7,87	9,99	6,00 !	16896,82	18118,47	12826,00!	4948,15	6169,44	3295,00!		!
!19/08/2015!	8,09	10,23	6,00 !	16904,91	18128,70	12832,00!	4956,24	6179,67	3301,00!		!
!20/08/2015!	8,03	10,07	6,00 !	16912,94	18138,77	12838,00!	4964,27	6189,74	3307,00!		!
!21/08/2015!	8,00	9,85	6,00 !	16920,94	18148,63	12844,00!	4972,27	6199,59	3313,00!		!
!22/08/2015!	6,90	8,90	6,00 !	16927,86	18157,53	12850,00!	4979,17	6208,49	3319,00!		!
!23/08/2015!	8,81	10,76	7,00 !	16936,67	18168,30	12857,00!	4987,98	6219,25	3326,00!		!
!24/08/2015!	8,40	10,54	6,00 !	16945,07	18178,85	12863,00!	4996,38	6229,79	3332,00!		!
!25/08/2015!	10,17	12,39	7,00 !	16955,24	18191,23	12870,00!	5006,55	6242,18	3339,00!		!
!26/08/2015!	10,79	12,69	7,00 !	16966,03	18203,91	12877,00!	5017,34	6254,87	3346,00!		!
!27/08/2015!	5,60	6,42	2,00 !	16971,63	18210,33	12879,00!	5022,94	6261,29	3348,00!		!
!28/08/2015!	12,97	15,99	7,00 !	16984,60	18226,31	12886,00!	5035,91	6277,28	3355,00!		!
!29/08/2015!	12,90	15,06	7,00 !	16997,50	18241,37	12893,00!	5048,81	6292,34	3362,00!		!
!30/08/2015!	7,89	10,11	6,00 !	17005,39	18251,48	12899,00!	5056,70	6302,45	3368,00!		!
!31/08/2015!	14,07	16,43	8,00 !	17019,46	18267,92	12907,00!	5070,77	6318,88	3376,00!		!
!01/09/2015!	13,41	15,93	7,00 !	17032,88	18283,85	12914,00!	5084,18	6334,81	3383,00!		!
!02/09/2015!	13,86	16,17	8,00 !	17046,74	18300,02	12922,00!	5098,04	6350,98	3391,00!		!
!03/09/2015!	10,25	12,13	6,00 !	17056,99	18312,15	12928,00!	5108,29	6363,11	3397,00!		!
!04/09/2015!	11,00	12,89	7,00 !	17067,99	18325,03	12935,00!	5119,29	6376,00	3404,00!		!
!05/09/2015!	17,02	19,69	9,00 !	17085,01	18344,71	12944,00!	5136,31	6395,69	3413,00!		!
!06/09/2015!	5,97	6,97	3,00 !	17090,98	18351,68	12947,00!	5142,28	6402,66	3416,00!		!
!07/09/2015!	5,46	6,75	4,00 !	17096,44	18358,43	12951,00!	5147,74	6409,41	3420,00!		!
!08/09/2015!	11,65	14,20	8,00 !	17108,09	18372,64	12959,00!	5159,39	6423,61	3428,00!		!
!09/09/2015!	9,18	12,12	8,00 !	17117,26	18384,76	12967,00!	5168,57	6435,73	3436,00!		!
!10/09/2015!	6,83	8,31	4,00 !	17124,09	18393,07	12971,00!	5175,40	6444,04	3440,00!		!
!11/09/2015!	8,28	11,72	7,00 !	17132,38	18404,79	12978,00!	5183,68	6455,76	3447,00!		!
!12/09/2015!	10,91	14,27	9,00 !	17143,30	18419,06	12987,00!	5194,59	6470,03	3456,00!		!
!13/09/2015!	6,87	9,19	8,00 !	17150,16	18428,25	12995,00!	5201,46	6479,22	3464,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!14/09/2015!	8,42	10,56	6,00 !	17158,57	18438,80	13001,00!	5209,88	6489,78	3470,00!		!
!15/09/2015!	6,30	7,87	4,00 !	17164,87	18446,67	13005,00!	5216,18	6497,65	3474,00!		!
!16/09/2015!	8,61	10,60	6,00 !	17173,48	18457,28	13011,00!	5224,79	6508,25	3480,00!		!
!17/09/2015!	7,04	9,21	6,00 !	17180,52	18466,48	13017,00!	5231,83	6517,46	3486,00!		!
!18/09/2015!	8,61	10,81	6,00 !	17189,14	18477,28	13023,00!	5240,44	6528,27	3492,00!		!
!19/09/2015!	4,66	6,83	7,00 !	17193,79	18484,11	13030,00!	5245,10	6535,10	3499,00!		!
!20/09/2015!	11,09	13,44	7,00 !	17204,88	18497,55	13037,00!	5256,19	6548,54	3506,00!		!
!21/09/2015!	10,20	12,62	7,00 !	17215,07	18510,16	13044,00!	5266,39	6561,16	3513,00!		!
!22/09/2015!	10,80	12,92	7,00 !	17225,87	18523,08	13051,00!	5277,19	6574,08	3520,00!		!
!23/09/2015!	12,22	15,11	7,00 !	17238,08	18538,19	13058,00!	5289,41	6589,19	3527,00!		!
!24/09/2015!	8,41	10,87	6,00 !	17246,50	18549,06	13064,00!	5297,82	6600,06	3533,00!		!
!25/09/2015!	6,69	11,11	9,00 !	17253,20	18560,16	13073,00!	5304,51	6611,17	3542,00!		!
!26/09/2015!	5,57	8,00	8,00 !	17258,77	18568,16	13081,00!	5310,08	6619,17	3550,00!		!
!27/09/2015!	8,43	11,20	7,00 !	17267,20	18579,36	13088,00!	5318,51	6630,37	3557,00!		!
!28/09/2015!	7,57	10,11	6,00 !	17274,76	18589,47	13094,00!	5326,08	6640,48	3563,00!		!
!29/09/2015!	11,25	15,62	9,00 !	17286,01	18605,08	13103,00!	5337,33	6656,10	3572,00!		!
!30/09/2015!	9,09	11,55	7,00 !	17295,09	18616,63	13110,00!	5346,42	6667,65	3579,00!		!
!01/10/2015!	10,12	12,35	7,00 !	17305,21	18628,96	13117,00!	5356,54	6680,00	3586,00!		!
!02/10/2015!	8,57	11,13	7,00 !	17313,77	18640,09	13124,00!	5365,11	6691,13	3593,00!		!
!03/10/2015!	8,13	10,28	6,00 !	17321,90	18650,37	13130,00!	5373,24	6701,41	3599,00!		!
!04/10/2015!	6,11	8,29	6,00 !	17328,02	18658,68	13136,00!	5379,35	6709,70	3605,00!		!
!05/10/2015!	7,96	11,19	6,00 !	17335,98	18669,87	13142,00!	5387,31	6720,89	3611,00!		!
!06/10/2015!	7,50	9,89	5,00 !	17343,47	18679,76	13147,00!	5394,81	6730,78	3616,00!		!
!07/10/2015!	8,94	11,45	7,00 !	17352,42	18691,21	13154,00!	5403,75	6742,23	3623,00!		!
!08/10/2015!	8,32	10,65	6,00 !	17360,73	18701,86	13160,00!	5412,07	6752,88	3629,00!		!
!09/10/2015!	8,34	10,80	6,00 !	17369,06	18712,66	13166,00!	5420,41	6763,68	3635,00!		!
!10/10/2015!	2,10	3,11	3,00 !	17371,16	18715,77	13169,00!	5422,51	6766,79	3638,00!		!
!13/10/2015!	1,49	2,09	2,00 !	17372,64	18717,85	13171,00!	5424,00	6768,88	3640,00!		!
!14/10/2015!	7,20	9,95	7,00 !	17379,84	18727,81	13178,00!	5431,20	6778,83	3647,00!		!
!15/10/2015!	3,19	5,93	5,00 !	17383,05	18733,74	13183,00!	5434,39	6784,76	3652,00!		!
!16/10/2015!	8,81	11,86	7,00 !	17391,86	18745,59	13190,00!	5443,20	6796,62	3659,00!		!
!17/10/2015!	8,16	10,30	6,00 !	17400,02	18755,89	13196,00!	5451,36	6806,92	3665,00!		!
!18/10/2015!	7,38	9,70	5,00 !	17407,40	18765,59	13201,00!	5458,74	6816,62	3670,00!		!
!19/10/2015!	5,96	9,16	8,00 !	17413,37	18774,75	13209,00!	5464,70	6825,78	3678,00!		!
!20/10/2015!	9,27	11,32	6,00 !	17422,64	18786,07	13215,00!	5473,97	6837,10	3684,00!		!
!21/10/2015!	7,96	10,50	8,00 !	17430,60	18796,57	13223,00!	5481,93	6847,60	3692,00!		!
!22/10/2015!	7,40	9,91	6,00 !	17438,00	18806,47	13229,00!	5489,33	6857,51	3698,00!		!
!23/10/2015!	6,99	10,52	10,00 !	17444,99	18816,99	13239,00!	5496,32	6868,03	3708,00!		!
!24/10/2015!	8,23	10,20	6,00 !	17453,22	18827,19	13245,00!	5504,55	6878,23	3714,00!		!
!25/10/2015!	6,52	8,67	7,00 !	17459,73	18835,86	13252,00!	5511,07	6886,90	3721,00!		!
!26/10/2015!	7,28	9,89	7,00 !	17467,01	18845,75	13259,00!	5518,35	6896,79	3728,00!		!
!27/10/2015!	9,42	11,79	6,00 !	17476,43	18857,55	13265,00!	5527,77	6908,58	3734,00!		!
!28/10/2015!	10,28	12,54	6,00 !	17486,70	18870,10	13271,00!	5538,05	6921,12	3740,00!		!
!29/10/2015!	6,55	7,85	4,00 !	17493,25	18877,95	13275,00!	5544,60	6928,97	3744,00!		!
!30/10/2015!	7,69	9,15	5,00 !	17500,93	18887,11	13280,00!	5552,29	6938,12	3749,00!		!
!31/10/2015!	6,08	7,64	5,00 !	17507,01	18894,75	13285,00!	5558,37	6945,76	3754,00!		!
!01/11/2015!	15,62	18,40	7,00 !	17522,63	18913,15	13292,00!	5573,99	6964,16	3761,00!		!
!02/11/2015!	13,88	16,18	8,00 !	17536,51	18929,33	13300,00!	5587,87	6980,34	3769,00!		!
!03/11/2015!	10,12	12,30	6,00 !	17546,63	18941,63	13306,00!	5597,99	6992,64	3775,00!		!
!04/11/2015!	10,86	13,02	7,00 !	17557,49	18954,65	13313,00!	5608,85	7005,66	3782,00!		!
!05/11/2015!	13,77	16,14	7,00 !	17571,25	18970,80	13320,00!	5622,62	7021,80	3789,00!		!
!06/11/2015!	11,14	13,37	7,00 !	17582,38	18984,16	13327,00!	5633,76	7035,17	3796,00!		!
!07/11/2015!	8,76	10,80	5,00 !	17591,14	18994,96	13332,00!	5642,52	7045,97	3801,00!		!
!08/11/2015!	6,68	8,31	5,00 !	17597,80	19003,27	13337,00!	5649,20	7054,28	3806,00!		!
!09/11/2015!	1,33	2,05	2,00 !	17599,13	19005,32	13339,00!	5650,53	7056,33	3808,00!		!
!10/12/2016!	3,14	3,97	2,00 !	17602,27	19009,28	13341,00!	5653,67	7060,30	3810,00!		!
!11/12/2016!	5,35	6,16	3,00 !	17607,62	19015,45	13344,00!	5659,02	7066,46	3813,00!		!
!12/12/2016!	9,30	11,69	7,00 !	17616,92	19027,13	13351,00!	5668,32	7078,15	3820,00!		!
!13/12/2016!	7,23	9,83	7,00 !	17624,15	19036,97	13358,00!	5675,55	7087,98	3827,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!14/12/2016!	8,03	10,45	6,00 !	17632,18	19047,42	13364,00!	5683,58	7098,43	3833,00!		!
!15/12/2016!	1,40	2,06	2,00 !	17633,58	19049,48	13366,00!	5684,98	7100,49	3835,00!		!
!17/12/2016!	13,57	16,61	9,00 !	17647,15	19066,10	13375,00!	5698,55	7117,10	3844,00!		!
!18/12/2016!	13,11	15,31	8,00 !	17660,27	19081,40	13383,00!	5711,66	7132,41	3852,00!		!
!19/12/2016!	13,26	15,65	8,00 !	17673,53	19097,05	13391,00!	5724,92	7148,06	3860,00!		!
!20/12/2016!	9,49	10,96	5,00 !	17683,02	19108,00	13396,00!	5734,41	7159,02	3865,00!		!
!21/12/2016!	7,73	10,32	7,00 !	17690,75	19118,33	13403,00!	5742,14	7169,34	3872,00!		!
!22/12/2016!	6,85	10,12	7,00 !	17697,60	19128,45	13410,00!	5748,99	7179,46	3879,00!		!
!23/12/2016!	12,96	15,30	8,00 !	17710,57	19143,75	13418,00!	5761,95	7194,76	3887,00!		!
!24/12/2016!	13,04	15,06	8,00 !	17723,62	19158,82	13426,00!	5774,99	7209,82	3895,00!		!
!25/12/2016!	13,01	15,05	8,00 !	17736,62	19173,88	13434,00!	5788,00	7224,87	3903,00!		!
!26/12/2016!	12,89	15,60	8,00 !	17749,50	19189,48	13442,00!	5800,89	7240,47	3911,00!		!
!27/12/2016!	13,05	15,92	8,00 !	17762,55	19205,40	13450,00!	5813,94	7256,39	3919,00!		!
!28/12/2016!	13,24	15,69	8,00 !	17775,78	19221,08	13458,00!	5827,18	7272,08	3927,00!		!
!29/12/2016!	13,13	15,35	8,00 !	17788,90	19236,43	13466,00!	5840,31	7287,43	3935,00!		!
!30/12/2016!	13,02	15,73	8,00 !	17801,92	19252,15	13474,00!	5853,33	7303,16	3943,00!		!
!31/12/2016!	9,84	11,74	6,00 !	17811,77	19263,90	13480,00!	5863,17	7314,90	3949,00!		!
!01/01/2017!	9,73	11,63	6,00 !	17821,50	19275,53	13486,00!	5872,90	7326,53	3955,00!		!
!02/01/2017!	12,91	15,43	8,00 !	17834,40	19290,95	13494,00!	5885,81	7341,96	3963,00!		!
!03/01/2017!	6,40	9,21	7,00 !	17840,80	19300,17	13501,00!	5892,21	7351,17	3970,00!		!
!04/01/2017!	6,86	10,01	8,00 !	17847,67	19310,17	13509,00!	5899,07	7361,18	3978,00!		!
!05/01/2017!	6,84	9,83	7,00 !	17854,52	19319,98	13516,00!	5905,91	7371,01	3985,00!		!
!06/01/2017!	13,17	15,34	8,00 !	17867,68	19335,32	13524,00!	5919,08	7386,35	3993,00!		!
!07/01/2017!	13,03	15,11	8,00 !	17880,72	19350,43	13532,00!	5932,11	7401,46	4001,00!		!
!08/01/2017!	13,33	15,57	8,00 !	17894,05	19365,98	13540,00!	5945,44	7417,03	4009,00!		!
!09/01/2017!	12,98	15,39	8,00 !	17907,03	19381,37	13548,00!	5958,42	7432,42	4017,00!		!
!10/01/2017!	13,08	15,38	8,00 !	17920,12	19396,73	13556,00!	5971,50	7447,80	4025,00!		!
!11/01/2017!	13,09	15,35	8,00 !	17933,20	19412,08	13564,00!	5984,59	7463,15	4033,00!		!
!12/01/2017!	13,08	15,37	8,00 !	17946,28	19427,45	13572,00!	5997,67	7478,52	4041,00!		!
!13/01/2017!	9,77	11,77	6,00 !	17956,05	19439,22	13578,00!	6007,44	7490,29	4047,00!		!
!14/01/2017!	16,65	19,58	10,00 !	17972,70	19458,80	13588,00!	6024,09	7509,87	4057,00!		!
!15/01/2017!	13,19	15,50	8,00 !	17985,88	19474,28	13596,00!	6037,28	7525,37	4065,00!		!
!16/01/2017!	8,05	10,61	7,00 !	17993,93	19484,90	13603,00!	6045,33	7535,98	4072,00!		!
!17/01/2017!	7,48	9,70	7,00 !	18001,42	19494,60	13610,00!	6052,81	7545,68	4079,00!		!
!18/01/2017!	5,01	7,15	5,00 !	18006,43	19501,77	13615,00!	6057,82	7552,83	4084,00!		!
!19/01/2017!	7,01	9,78	9,00 !	18013,43	19511,55	13624,00!	6064,83	7562,61	4093,00!		!
!20/01/2017!	0,00	0,25	0,00 !	18013,43	19511,80	13624,00!	6064,83	7562,86	4093,00!		!
!21/01/2017!	5,02	7,16	7,00 !	18018,47	19518,95	13631,00!	6069,85	7570,02	4100,00!		!
!22/01/2017!	8,64	10,38	6,00 !	18027,10	19529,33	13637,00!	6078,49	7580,40	4106,00!		!
!23/01/2017!	7,15	9,17	5,00 !	18034,25	19538,52	13642,00!	6085,64	7589,57	4111,00!		!
!24/01/2017!	5,59	8,28	6,00 !	18039,85	19546,80	13648,00!	6091,23	7597,85	4117,00!		!
!25/01/2017!	4,33	6,16	6,00 !	18044,17	19552,97	13654,00!	6095,56	7604,01	4123,00!		!
!26/01/2017!	2,20	3,47	3,00 !	18046,37	19556,43	13657,00!	6097,76	7607,48	4126,00!		!
!27/01/2017!	4,77	6,72	5,00 !	18051,15	19563,15	13662,00!	6102,53	7614,20	4131,00!		!
!28/01/2017!	6,68	9,11	7,00 !	18057,83	19572,27	13669,00!	6109,21	7623,31	4138,00!		!
!29/01/2017!	6,94	9,86	8,00 !	18064,77	19582,13	13677,00!	6116,15	7633,17	4146,00!		!
!30/01/2017!	7,37	10,36	8,00 !	18072,13	19592,50	13685,00!	6123,52	7643,53	4154,00!		!
!31/01/2017!	7,25	10,30	8,00 !	18079,40	19602,80	13693,00!	6130,77	7653,83	4162,00!		!
!01/02/2017!	7,04	10,75	8,00 !	18086,43	19613,55	13701,00!	6137,81	7664,58	4170,00!		!
!02/02/2017!	6,20	8,83	7,00 !	18092,63	19622,38	13708,00!	6144,01	7673,41	4177,00!		!
!03/02/2017!	10,26	12,24	7,00 !	18102,90	19634,63	13715,00!	6154,27	7685,65	4184,00!		!
!04/02/2017!	15,51	18,13	9,00 !	18118,42	19652,77	13724,00!	6169,78	7703,78	4193,00!		!
!05/02/2017!	12,80	14,85	8,00 !	18131,22	19667,62	13732,00!	6182,58	7718,63	4201,00!		!
!06/02/2017!	13,10	15,44	8,00 !	18144,32	19683,05	13740,00!	6195,68	7734,07	4209,00!		!
!07/02/2017!	12,95	15,15	8,00 !	18157,27	19698,22	13748,00!	6208,63	7749,22	4217,00!		!
!08/02/2017!	12,90	15,00	8,00 !	18170,17	19713,22	13756,00!	6221,53	7764,22	4225,00!		!
!09/02/2017!	13,20	15,58	8,00 !	18183,37	19728,80	13764,00!	6234,73	7779,80	4233,00!		!
!10/02/2017!	10,47	12,44	7,00 !	18193,83	19741,25	13771,00!	6245,20	7792,24	4240,00!		!
!11/02/2017!	15,73	18,75	9,00 !	18209,55	19760,00	13780,00!	6260,93	7810,99	4249,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!12/02/2017!	12,84	15,11	8,00 !	18222,38	19775,12	13788,00!	6273,77	7826,10	4257,00!		!
!13/02/2017!	13,23	15,59	8,00 !	18235,60	19790,72	13796,00!	6287,00	7841,69	4265,00!		!
!14/02/2017!	13,13	15,43	8,00 !	18248,73	19806,15	13804,00!	6300,13	7857,12	4273,00!		!
!15/02/2017!	13,35	15,63	8,00 !	18262,08	19821,78	13812,00!	6313,48	7872,75	4281,00!		!
!16/02/2017!	13,09	15,66	8,00 !	18275,18	19837,43	13820,00!	6326,57	7888,41	4289,00!		!
!17/02/2017!	8,76	10,96	6,00 !	18283,93	19848,40	13826,00!	6335,33	7899,37	4295,00!		!
!18/02/2017!	6,70	9,68	10,00 !	18290,63	19858,08	13836,00!	6342,03	7909,05	4305,00!		!
!19/02/2017!	3,19	4,62	5,00 !	18293,82	19862,70	13841,00!	6345,22	7913,67	4310,00!		!
!21/02/2017!	2,70	3,30	2,00 !	18296,52	19866,00	13843,00!	6347,92	7916,97	4312,00!		!
!22/02/2017!	3,21	5,15	4,00 !	18299,72	19871,15	13847,00!	6351,13	7922,12	4316,00!		!
!23/02/2017!	9,03	11,67	7,00 !	18308,75	19882,82	13854,00!	6360,16	7933,79	4323,00!		!
!24/02/2017!	7,08	8,81	5,00 !	18315,83	19891,63	13859,00!	6367,24	7942,60	4328,00!		!
!25/02/2017!	6,13	8,31	6,00 !	18321,97	19899,95	13865,00!	6373,37	7950,91	4334,00!		!
!26/02/2017!	5,28	6,73	4,00 !	18327,25	19906,68	13869,00!	6378,65	7957,64	4338,00!		!
!27/02/2017!	5,49	6,84	4,00 !	18332,73	19913,52	13873,00!	6384,14	7964,48	4342,00!		!
!28/02/2017!	8,38	10,41	6,00 !	18341,12	19923,93	13879,00!	6392,52	7974,89	4348,00!		!
!01/03/2017!	7,02	8,64	5,00 !	18348,13	19932,57	13884,00!	6399,54	7983,53	4353,00!		!
!02/03/2017!	7,72	9,75	6,00 !	18355,85	19942,32	13890,00!	6407,26	7993,28	4359,00!		!
!03/03/2017!	7,46	8,92	4,00 !	18363,32	19951,23	13894,00!	6414,72	8002,20	4363,00!		!
!04/03/2017!	7,56	9,64	6,00 !	18370,88	19960,87	13900,00!	6422,28	8011,84	4369,00!		!
!05/03/2017!	4,50	6,72	6,00 !	18375,38	19967,58	13906,00!	6426,78	8018,56	4375,00!		!
!06/03/2017!	8,40	11,49	8,00 !	18383,78	19979,08	13914,00!	6435,18	8030,05	4383,00!		!
!07/03/2017!	5,68	8,16	6,00 !	18389,47	19987,23	13920,00!	6440,86	8038,21	4389,00!		!
!09/03/2017!	7,97	11,22	7,00 !	18397,43	19998,45	13927,00!	6448,83	8049,43	4396,00!		!
!10/03/2017!	4,82	6,80	6,00 !	18402,25	20005,25	13933,00!	6453,65	8056,23	4402,00!		!
!11/03/2017!	7,00	8,63	5,00 !	18409,25	20013,88	13938,00!	6460,65	8064,86	4407,00!		!
!12/03/2017!	6,40	9,09	9,00 !	18415,63	20022,98	13947,00!	6467,05	8073,95	4416,00!		!
!13/03/2017!	6,63	9,60	10,00 !	18422,27	20032,60	13957,00!	6473,68	8083,55	4426,00!		!
!14/03/2017!	4,58	6,71	6,00 !	18426,85	20039,30	13963,00!	6478,26	8090,26	4432,00!		!
!15/03/2017!	5,47	7,70	6,00 !	18432,32	20047,00	13969,00!	6483,73	8097,96	4438,00!		!
!16/03/2017!	6,02	9,34	8,00 !	18438,33	20056,33	13977,00!	6489,75	8107,30	4446,00!		!
!17/03/2017!	6,74	10,69	9,00 !	18445,07	20067,03	13986,00!	6496,49	8117,99	4455,00!		!
!18/03/2017!	5,19	7,79	7,00 !	18450,25	20074,83	13993,00!	6501,68	8125,78	4462,00!		!
!19/03/2017!	6,20	9,03	8,00 !	18456,43	20083,87	14001,00!	6507,88	8134,81	4470,00!		!
!20/03/2017!	4,51	6,98	6,00 !	18460,95	20090,85	14007,00!	6512,39	8141,79	4476,00!		!
!21/03/2017!	5,93	9,08	8,00 !	18466,88	20099,93	14015,00!	6518,32	8150,87	4484,00!		!
!22/03/2017!	6,83	10,29	8,00 !	18473,70	20110,22	14023,00!	6525,15	8161,16	4492,00!		!
!23/03/2017!	7,96	10,47	5,00 !	18481,67	20120,68	14028,00!	6533,11	8171,63	4497,00!		!
!24/03/2017!	8,18	9,98	6,00 !	18489,85	20130,65	14034,00!	6541,29	8181,61	4503,00!		!
!25/03/2017!	6,17	8,37	7,00 !	18496,03	20139,02	14041,00!	6547,46	8189,98	4510,00!		!
!26/03/2017!	7,02	9,89	10,00 !	18503,05	20148,92	14051,00!	6554,48	8199,87	4520,00!		!
!27/03/2017!	5,61	8,94	8,00 !	18508,65	20157,87	14059,00!	6560,09	8208,81	4528,00!		!
!28/03/2017!	5,69	8,15	8,00 !	18514,35	20166,02	14067,00!	6565,78	8216,96	4536,00!		!
!29/03/2017!	7,67	9,59	6,00 !	18522,02	20175,60	14073,00!	6573,45	8226,55	4542,00!		!
!30/03/2017!	5,94	8,68	8,00 !	18527,97	20184,28	14081,00!	6579,39	8235,23	4550,00!		!
!31/03/2017!	5,84	6,97	4,00 !	18533,82	20191,25	14085,00!	6585,23	8242,20	4554,00!		!
!01/04/2017!	7,60	9,42	5,00 !	18541,42	20200,67	14090,00!	6592,83	8251,62	4559,00!		!
!02/04/2017!	6,70	8,31	4,00 !	18548,12	20208,98	14094,00!	6599,53	8259,93	4563,00!		!
!03/04/2017!	5,42	7,77	8,00 !	18553,52	20216,75	14102,00!	6604,95	8267,70	4571,00!		!
!04/04/2017!	3,58	5,03	5,00 !	18557,10	20221,78	14107,00!	6608,53	8272,73	4576,00!		!
!05/04/2017!	4,96	5,87	3,00 !	18562,05	20227,65	14110,00!	6613,49	8278,60	4579,00!		!
!06/04/2017!	1,55	1,88	1,00 !	18563,60	20229,53	14111,00!	6615,04	8280,48	4580,00!		!
!07/04/2017!	6,49	10,42	9,00 !	18570,10	20239,95	14120,00!	6621,53	8290,90	4589,00!		!
!08/04/2017!	2,17	3,35	3,00 !	18572,27	20243,30	14123,00!	6623,70	8294,25	4592,00!		!
!09/04/2017!	2,80	4,30	4,00 !	18575,07	20247,60	14127,00!	6626,50	8298,55	4596,00!		!
!10/04/2017!	5,65	8,32	8,00 !	18580,72	20255,93	14135,00!	6632,15	8306,87	4604,00!		!
!11/04/2017!	6,74	10,12	10,00 !	18587,45	20266,05	14145,00!	6638,89	8316,99	4614,00!		!
!12/04/2017!	4,44	7,28	6,00 !	18591,88	20273,33	14151,00!	6643,33	8324,27	4620,00!		!
!13/04/2017!	7,72	11,73	10,00 !	18599,62	20285,08	14161,00!	6651,05	8336,00	4630,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!14/04/2017!	5,83	7,21	4,00 !	18605,45	20292,28	14165,00!	6656,88	8343,21	4634,00!		!
!15/04/2017!	3,88	5,26	5,00 !	18609,33	20297,55	14170,00!	6660,76	8348,47	4639,00!		!
!16/04/2017!	7,14	8,51	5,00 !	18616,48	20306,07	14175,00!	6667,90	8356,98	4644,00!		!
!17/04/2017!	5,51	7,82	7,00 !	18621,98	20313,90	14182,00!	6673,41	8364,80	4651,00!		!
!18/04/2017!	3,91	5,72	5,00 !	18625,90	20319,62	14187,00!	6677,32	8370,52	4656,00!		!
!19/04/2017!	3,50	6,08	5,00 !	18629,40	20325,70	14192,00!	6680,82	8376,60	4661,00!		!
!20/04/2017!	5,96	9,04	7,00 !	18635,35	20334,73	14199,00!	6686,78	8385,64	4668,00!		!
!21/04/2017!	6,08	8,13	6,00 !	18641,43	20342,87	14205,00!	6692,86	8393,77	4674,00!		!
!22/04/2017!	3,84	4,70	3,00 !	18645,27	20347,57	14208,00!	6696,70	8398,47	4677,00!		!
!23/04/2017!	6,35	8,19	4,00 !	18651,62	20355,75	14212,00!	6703,05	8406,66	4681,00!		!
!24/04/2017!	7,92	10,53	8,00 !	18659,55	20366,28	14220,00!	6710,97	8417,19	4689,00!		!
!25/04/2017!	7,72	11,37	11,00 !	18667,27	20377,67	14231,00!	6718,69	8428,56	4700,00!		!
!26/04/2017!	6,77	10,02	9,00 !	18674,05	20387,68	14240,00!	6725,46	8438,58	4709,00!		!
!27/04/2017!	4,93	7,01	7,00 !	18678,97	20394,70	14247,00!	6730,39	8445,59	4716,00!		!
!28/04/2017!	7,84	10,27	6,00 !	18686,80	20404,97	14253,00!	6738,23	8455,86	4722,00!		!
!29/04/2017!	2,81	4,10	3,00 !	18689,60	20409,07	14256,00!	6741,04	8459,96	4725,00!		!
!02/05/2017!	5,74	7,06	4,00 !	18695,33	20416,13	14260,00!	6746,78	8467,02	4729,00!		!
!03/05/2017!	5,81	8,49	8,00 !	18701,15	20424,62	14268,00!	6752,59	8475,51	4737,00!		!
!04/05/2017!	7,18	9,24	6,00 !	18708,33	20433,85	14274,00!	6759,77	8484,75	4743,00!		!
!05/05/2017!	11,35	14,00	7,00 !	18719,68	20447,85	14281,00!	6771,12	8498,75	4750,00!		!
!06/05/2017!	7,03	8,61	5,00 !	18726,72	20456,47	14286,00!	6778,15	8507,36	4755,00!		!
!07/05/2017!	6,56	9,17	9,00 !	18733,27	20465,63	14295,00!	6784,71	8516,53	4764,00!		!
!08/05/2017!	3,55	4,54	3,00 !	18736,82	20470,17	14298,00!	6788,26	8521,07	4767,00!		!
!09/05/2017!	7,61	11,01	11,00 !	18744,42	20481,18	14309,00!	6795,87	8532,08	4778,00!		!
!10/05/2017!	4,23	6,39	6,00 !	18748,65	20487,57	14315,00!	6800,10	8538,47	4784,00!		!
!11/05/2017!	4,44	6,52	5,00 !	18753,08	20494,10	14320,00!	6804,54	8544,99	4789,00!		!
!12/05/2017!	4,52	7,37	7,00 !	18757,60	20501,48	14327,00!	6809,06	8552,36	4796,00!		!
!13/05/2017!	7,47	9,82	6,00 !	18765,07	20511,30	14333,00!	6816,53	8562,18	4802,00!		!
!14/05/2017!	5,33	6,83	5,00 !	18770,40	20518,13	14338,00!	6821,86	8569,01	4807,00!		!
!15/05/2017!	5,22	7,06	5,00 !	18775,63	20525,18	14343,00!	6827,08	8576,07	4812,00!		!
!16/05/2017!	4,52	5,57	5,00 !	18780,15	20530,77	14348,00!	6831,60	8581,64	4817,00!		!
!17/05/2017!	1,53	2,42	2,00 !	18781,68	20533,18	14350,00!	6833,13	8584,06	4819,00!		!
!18/05/2017!	5,66	8,38	8,00 !	18787,35	20541,57	14358,00!	6838,79	8592,44	4827,00!		!
!19/05/2017!	5,92	8,64	6,00 !	18793,27	20550,22	14364,00!	6844,71	8601,08	4833,00!		!
!20/05/2017!	10,17	12,78	7,00 !	18803,43	20563,00	14371,00!	6854,88	8613,86	4840,00!		!
!21/05/2017!	6,02	7,32	4,00 !	18809,45	20570,32	14375,00!	6860,90	8621,18	4844,00!		!
!22/05/2017!	6,74	9,48	9,00 !	18816,20	20579,78	14384,00!	6867,64	8630,66	4853,00!		!
!23/05/2017!	10,55	12,60	6,00 !	18826,75	20592,38	14390,00!	6878,19	8643,26	4859,00!		!
!24/05/2017!	4,30	6,07	6,00 !	18831,05	20598,45	14396,00!	6882,49	8649,33	4865,00!		!
!25/05/2017!	6,92	8,93	5,00 !	18837,97	20607,38	14401,00!	6889,41	8658,26	4870,00!		!
!26/05/2017!	6,53	9,88	9,00 !	18844,50	20617,25	14410,00!	6895,94	8668,14	4879,00!		!
!27/05/2017!	7,03	9,99	10,00 !	18851,53	20627,23	14420,00!	6902,97	8678,13	4889,00!		!
!28/05/2017!	4,88	6,47	5,00 !	18856,42	20633,70	14425,00!	6907,85	8684,60	4894,00!		!
!29/05/2017!	7,82	10,11	6,00 !	18864,23	20643,82	14431,00!	6915,67	8694,71	4900,00!		!
!30/05/2017!	5,54	7,25	4,00 !	18869,77	20651,07	14435,00!	6921,21	8701,96	4904,00!		!
!31/05/2017!	4,44	6,08	6,00 !	18874,22	20657,15	14441,00!	6925,65	8708,04	4910,00!		!
!01/06/2017!	3,81	5,74	3,00 !	18878,03	20662,90	14444,00!	6929,46	8713,78	4913,00!		!
!02/06/2017!	5,56	8,12	8,00 !	18883,58	20671,03	14452,00!	6935,02	8721,90	4921,00!		!
!03/06/2017!	1,40	1,99	2,00 !	18884,98	20673,02	14454,00!	6936,42	8723,89	4923,00!		!
!05/06/2017!	4,09	6,23	4,00 !	18889,07	20679,25	14458,00!	6940,51	8730,12	4927,00!		!
!06/06/2017!	8,40	10,70	6,00 !	18897,47	20689,95	14464,00!	6948,91	8740,82	4933,00!		!
!07/06/2017!	6,79	10,81	9,00 !	18904,25	20700,77	14473,00!	6955,70	8751,63	4942,00!		!
!08/06/2017!	6,11	7,70	4,00 !	18910,37	20708,47	14477,00!	6961,81	8759,33	4946,00!		!
!09/06/2017!	5,76	9,23	8,00 !	18916,13	20717,70	14485,00!	6967,57	8768,56	4954,00!		!
!10/06/2017!	7,74	9,60	5,00 !	18923,87	20727,30	14490,00!	6975,31	8778,16	4959,00!		!
!11/06/2017!	6,61	7,96	4,00 !	18930,48	20735,27	14494,00!	6981,92	8786,12	4963,00!		!
!12/06/2017!	6,22	9,37	9,00 !	18936,70	20744,65	14503,00!	6988,14	8795,49	4972,00!		!
!13/06/2017!	0,75	1,23	1,00 !	18937,45	20745,88	14504,00!	6988,89	8796,72	4973,00!		!
!14/06/2017!	7,93	10,04	6,00 !	18945,38	20755,92	14510,00!	6996,82	8806,76	4979,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!15/06/2017!	9,06	10,43	5,00 !	18954,45	20766,35	14515,00!	7005,88	8817,19	4984,00!		!
!16/06/2017!	6,56	7,71	4,00 !	18961,02	20774,07	14519,00!	7012,44	8824,90	4988,00!		!
!17/06/2017!	6,50	7,80	4,00 !	18967,52	20781,87	14523,00!	7018,94	8832,70	4992,00!		!
!18/06/2017!	13,17	16,16	8,00 !	18980,68	20798,02	14531,00!	7032,11	8848,86	5000,00!		!
!19/06/2017!	12,75	14,99	8,00 !	18993,42	20813,00	14539,00!	7044,86	8863,85	5008,00!		!
!20/06/2017!	12,94	15,20	8,00 !	19006,37	20828,20	14547,00!	7057,80	8879,05	5016,00!		!
!21/06/2017!	15,92	18,53	9,00 !	19022,28	20846,73	14556,00!	7073,72	8897,58	5025,00!		!
!22/06/2017!	10,44	12,43	7,00 !	19032,73	20859,17	14563,00!	7084,16	8910,01	5032,00!		!
!23/06/2017!	8,16	9,69	5,00 !	19040,88	20868,85	14568,00!	7092,32	8919,70	5037,00!		!
!24/06/2017!	2,78	3,97	3,00 !	19043,67	20872,82	14571,00!	7095,10	8923,67	5040,00!		!
!26/06/2017!	4,23	6,19	5,00 !	19047,90	20879,02	14576,00!	7099,33	8929,86	5045,00!		!
!27/06/2017!	5,00	6,16	3,00 !	19052,90	20885,18	14579,00!	7104,33	8936,02	5048,00!		!
!28/06/2017!	7,14	8,87	5,00 !	19060,05	20894,05	14584,00!	7111,47	8944,89	5053,00!		!
!29/06/2017!	5,06	6,08	3,00 !	19065,12	20900,13	14587,00!	7116,53	8950,97	5056,00!		!
!30/06/2017!	7,60	11,51	11,00 !	19072,72	20911,65	14598,00!	7124,13	8962,48	5067,00!		!
!01/07/2017!	7,66	10,26	8,00 !	19080,37	20921,92	14606,00!	7131,79	8972,74	5075,00!		!
!02/07/2017!	8,28	10,43	6,00 !	19088,65	20932,35	14612,00!	7140,07	8983,17	5081,00!		!
!03/07/2017!	6,75	9,42	9,00 !	19095,40	20941,77	14621,00!	7146,82	8992,59	5090,00!		!
!04/07/2017!	7,33	9,28	5,00 !	19102,73	20951,05	14626,00!	7154,15	9001,87	5095,00!		!
!05/07/2017!	4,81	6,43	5,00 !	19107,55	20957,48	14631,00!	7158,96	9008,30	5100,00!		!
!06/07/2017!	5,07	6,18	3,00 !	19112,62	20963,67	14634,00!	7164,03	9014,48	5103,00!		!
!07/07/2017!	7,97	10,03	6,00 !	19120,58	20973,68	14640,00!	7172,00	9024,51	5109,00!		!
!08/07/2017!	3,35	4,78	5,00 !	19123,93	20978,47	14645,00!	7175,35	9029,29	5114,00!		!
!09/07/2017!	5,87	8,28	8,00 !	19129,80	20986,75	14653,00!	7181,22	9037,57	5122,00!		!
!10/07/2017!	3,27	4,47	4,00 !	19133,08	20991,23	14657,00!	7184,49	9042,04	5126,00!		!
!11/07/2017!	1,62	1,87	1,00 !	19134,70	20993,10	14658,00!	7186,11	9043,91	5127,00!		!
!12/07/2017!	7,64	9,89	6,00 !	19142,33	21003,00	14664,00!	7193,75	9053,80	5133,00!		!
!13/07/2017!	4,55	6,37	3,00 !	19146,88	21009,37	14667,00!	7198,30	9060,17	5136,00!		!
!14/07/2017!	6,76	9,79	10,00 !	19153,65	21019,17	14677,00!	7205,06	9069,96	5146,00!		!
!15/07/2017!	6,88	8,06	4,00 !	19160,53	21027,23	14681,00!	7211,94	9078,02	5150,00!		!
!16/07/2017!	6,63	8,58	4,00 !	19167,17	21035,82	14685,00!	7218,57	9086,60	5154,00!		!
!17/07/2017!	5,50	7,84	7,00 !	19172,67	21043,65	14692,00!	7224,07	9094,44	5161,00!		!
!18/07/2017!	5,85	8,66	7,00 !	19178,52	21052,32	14699,00!	7229,92	9103,10	5168,00!		!
!19/07/2017!	4,30	6,94	6,00 !	19182,82	21059,25	14705,00!	7234,22	9110,04	5174,00!		!
!20/07/2017!	9,38	11,53	5,00 !	19192,20	21070,78	14710,00!	7243,60	9121,57	5179,00!		!
!21/07/2017!	10,71	13,22	6,00 !	19202,92	21084,00	14716,00!	7254,31	9134,79	5185,00!		!
!22/07/2017!	4,64	6,49	5,00 !	19207,55	21090,48	14721,00!	7258,95	9141,28	5190,00!		!
!23/07/2017!	3,67	4,90	4,00 !	19211,22	21095,38	14725,00!	7262,62	9146,18	5194,00!		!
!24/07/2017!	5,58	7,95	7,00 !	19216,80	21103,33	14732,00!	7268,20	9154,13	5201,00!		!
!25/07/2017!	3,89	5,80	5,00 !	19220,68	21109,13	14737,00!	7272,09	9159,93	5206,00!		!
!26/07/2017!	4,13	6,21	5,00 !	19224,82	21115,33	14742,00!	7276,22	9166,14	5211,00!		!
!27/07/2017!	0,85	1,33	1,00 !	19225,67	21116,67	14743,00!	7277,07	9167,47	5212,00!		!
!28/07/2017!	1,47	2,23	2,00 !	19227,13	21118,90	14745,00!	7278,54	9169,70	5214,00!		!
!29/07/2017!	7,73	9,45	6,00 !	19234,87	21128,35	14751,00!	7286,27	9179,15	5220,00!		!
!30/07/2017!	6,23	9,63	8,00 !	19241,10	21137,98	14759,00!	7292,50	9188,78	5228,00!		!
!31/07/2017!	10,57	12,82	7,00 !	19251,68	21150,80	14766,00!	7303,07	9201,60	5235,00!		!
!01/08/2017!	13,45	16,29	8,00 !	19265,13	21167,10	14774,00!	7316,52	9217,89	5243,00!		!
!02/08/2017!	16,06	19,07	9,00 !	19281,18	21186,17	14783,00!	7332,58	9236,96	5252,00!		!
!03/08/2017!	10,46	12,54	7,00 !	19291,65	21198,70	14790,00!	7343,04	9249,50	5259,00!		!
!04/08/2017!	3,78	6,29	6,00 !	19295,43	21205,00	14796,00!	7346,82	9255,79	5265,00!		!
!05/08/2017!	11,35	13,40	7,00 !	19306,80	21218,40	14803,00!	7358,17	9269,19	5272,00!		!
!06/08/2017!	7,01	8,45	5,00 !	19313,82	21226,85	14808,00!	7365,18	9277,64	5277,00!		!
!07/08/2017!	6,85	10,37	10,00 !	19320,67	21237,22	14818,00!	7372,03	9288,01	5287,00!		!
!08/08/2017!	5,64	9,27	8,00 !	19326,30	21246,50	14826,00!	7377,67	9297,28	5295,00!		!
!09/08/2017!	6,61	10,15	10,00 !	19332,92	21256,65	14836,00!	7384,28	9307,43	5305,00!		!
!10/08/2017!	7,86	9,97	6,00 !	19340,78	21266,62	14842,00!	7392,14	9317,40	5311,00!		!
!11/08/2017!	7,03	9,42	5,00 !	19347,82	21276,03	14847,00!	7399,17	9326,82	5316,00!		!
!12/08/2017!	4,90	6,80	7,00 !	19352,72	21282,83	14854,00!	7404,07	9333,62	5323,00!		!
!13/08/2017!	9,08	11,85	9,00 !	19361,80	21294,68	14863,00!	7413,15	9345,47	5332,00!		!

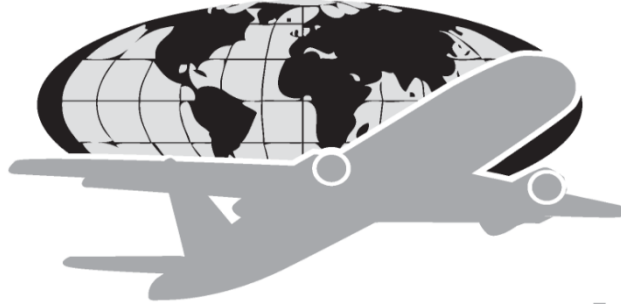
! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!14/08/2017!	6,95	8,88	5,00 !	19368,75	21303,57	14868,00!	7420,10	9354,35	5337,00!		!
!15/08/2017!	9,28	11,73	7,00 !	19378,03	21315,32	14875,00!	7429,38	9366,08	5344,00!		!
!16/08/2017!	8,69	11,09	6,00 !	19386,72	21326,40	14881,00!	7438,07	9377,17	5350,00!		!
!17/08/2017!	6,78	9,86	8,00 !	19393,50	21336,27	14889,00!	7444,85	9387,03	5358,00!		!
!18/08/2017!	8,64	11,79	8,00 !	19402,15	21348,07	14897,00!	7453,49	9398,82	5366,00!		!
!19/08/2017!	7,14	8,65	5,00 !	19409,28	21356,72	14902,00!	7460,63	9407,47	5371,00!		!
!20/08/2017!	6,86	8,43	5,00 !	19416,15	21365,15	14907,00!	7467,49	9415,90	5376,00!		!
!21/08/2017!	6,36	9,71	9,00 !	19422,52	21374,87	14916,00!	7473,85	9425,61	5385,00!		!
!22/08/2017!	4,74	6,82	7,00 !	19427,25	21381,68	14923,00!	7478,59	9432,43	5392,00!		!
!23/08/2017!	11,25	13,71	7,00 !	19438,50	21395,40	14930,00!	7489,84	9446,14	5399,00!		!
!24/08/2017!	8,03	10,08	6,00 !	19446,53	21405,48	14936,00!	7497,87	9456,22	5405,00!		!
!25/08/2017!	7,93	11,46	11,00 !	19454,45	21416,95	14947,00!	7505,80	9467,68	5416,00!		!
!26/08/2017!	5,39	7,49	8,00 !	19459,83	21424,45	14955,00!	7511,19	9475,17	5424,00!		!
!27/08/2017!	5,91	7,96	7,00 !	19465,73	21432,40	14962,00!	7517,10	9483,13	5431,00!		!
!28/08/2017!	7,54	8,93	5,00 !	19473,27	21441,33	14967,00!	7524,64	9492,06	5436,00!		!
!29/08/2017!	7,82	10,73	8,00 !	19481,08	21452,07	14975,00!	7532,46	9502,79	5444,00!		!
!30/08/2017!	8,98	11,38	7,00 !	19490,07	21463,45	14982,00!	7541,44	9514,17	5451,00!		!
!31/08/2017!	8,60	10,72	6,00 !	19498,67	21474,17	14988,00!	7550,04	9524,89	5457,00!		!
!01/09/2017!	7,12	8,75	5,00 !	19505,80	21482,92	14993,00!	7557,16	9533,64	5462,00!		!
!02/09/2017!	7,16	9,04	5,00 !	19512,97	21491,97	14998,00!	7564,32	9542,68	5467,00!		!
!03/09/2017!	4,38	6,67	6,00 !	19517,35	21498,63	15004,00!	7568,70	9549,35	5473,00!		!
!04/09/2017!	7,87	11,21	8,00 !	19525,22	21509,83	15012,00!	7576,57	9560,56	5481,00!		!
!05/09/2017!	6,46	9,35	6,00 !	19531,67	21519,18	15018,00!	7583,03	9569,91	5487,00!		!
!06/09/2017!	4,96	7,47	7,00 !	19536,63	21526,67	15025,00!	7587,99	9577,38	5494,00!		!
!07/09/2017!	8,21	10,54	5,00 !	19544,85	21537,20	15030,00!	7596,20	9587,92	5499,00!		!
!08/09/2017!	6,25	7,79	4,00 !	19551,10	21544,98	15034,00!	7602,45	9595,71	5503,00!		!
!09/09/2017!	6,18	7,50	4,00 !	19557,28	21552,48	15038,00!	7608,63	9603,21	5507,00!		!
!10/09/2017!	6,35	7,64	4,00 !	19563,63	21560,12	15042,00!	7614,98	9610,85	5511,00!		!
!11/09/2017!	6,70	9,35	6,00 !	19570,33	21569,47	15048,00!	7621,68	9620,20	5517,00!		!
!12/09/2017!	10,38	12,68	7,00 !	19580,70	21582,15	15055,00!	7632,06	9632,88	5524,00!		!
!13/09/2017!	13,05	16,44	8,00 !	19593,77	21598,58	15063,00!	7645,11	9649,32	5532,00!		!
!14/09/2017!	13,03	15,43	8,00 !	19606,78	21614,02	15071,00!	7658,14	9664,75	5540,00!		!
!15/09/2017!	12,88	15,26	8,00 !	19619,67	21629,28	15079,00!	7671,02	9680,01	5548,00!		!
!16/09/2017!	13,08	15,39	8,00 !	19632,73	21644,68	15087,00!	7684,10	9695,40	5556,00!		!
!17/09/2017!	13,35	15,91	8,00 !	19646,08	21660,58	15095,00!	7697,45	9711,31	5564,00!		!
!18/09/2017!	13,00	15,65	8,00 !	19659,08	21676,25	15103,00!	7710,45	9726,96	5572,00!		!
!19/09/2017!	13,20	15,49	8,00 !	19672,27	21691,75	15111,00!	7723,65	9742,45	5580,00!		!
!20/09/2017!	13,27	15,88	8,00 !	19685,55	21707,63	15119,00!	7736,92	9758,33	5588,00!		!
!21/09/2017!	13,37	16,02	8,00 !	19698,92	21723,65	15127,00!	7750,29	9774,35	5596,00!		!
!22/09/2017!	13,23	15,57	8,00 !	19712,13	21739,22	15135,00!	7763,52	9789,92	5604,00!		!
!23/09/2017!	13,20	15,72	8,00 !	19725,32	21754,93	15143,00!	7776,72	9805,64	5612,00!		!
!24/09/2017!	13,09	15,70	8,00 !	19738,42	21770,63	15151,00!	7789,81	9821,34	5620,00!		!
!25/09/2017!	13,18	15,52	8,00 !	19751,62	21786,15	15159,00!	7802,99	9836,86	5628,00!		!
!26/09/2017!	6,30	7,23	3,00 !	19757,92	21793,38	15162,00!	7809,29	9844,09	5631,00!		!
!01/10/2017!	5,49	8,10	7,00 !	19763,40	21801,48	15169,00!	7814,78	9852,19	5638,00!		!
!02/10/2017!	8,60	11,67	9,00 !	19772,00	21813,15	15178,00!	7823,38	9863,86	5647,00!		!
!03/10/2017!	7,34	10,74	10,00 !	19779,35	21823,90	15188,00!	7830,72	9874,60	5657,00!		!
!04/10/2017!	8,06	10,64	6,00 !	19787,42	21834,53	15194,00!	7838,78	9885,24	5663,00!		!
!05/10/2017!	5,21	7,66	6,00 !	19792,63	21842,18	15200,00!	7843,99	9892,90	5669,00!		!
!06/10/2017!	6,63	8,85	9,00 !	19799,25	21851,03	15209,00!	7850,62	9901,75	5678,00!		!
!07/10/2017!	4,30	5,35	3,00 !	19803,55	21856,38	15212,00!	7854,92	9907,10	5681,00!		!
!08/10/2017!	12,57	15,63	9,00 !	19816,13	21872,02	15221,00!	7867,49	9922,73	5690,00!		!
!09/10/2017!	8,43	10,69	6,00 !	19824,57	21882,70	15227,00!	7875,92	9933,42	5696,00!		!
!10/10/2017!	5,70	7,74	6,00 !	19830,27	21890,45	15233,00!	7881,62	9941,16	5702,00!		!
!11/10/2017!	8,43	10,51	6,00 !	19838,70	21900,95	15239,00!	7890,05	9951,67	5708,00!		!
!12/10/2017!	8,34	10,91	6,00 !	19847,03	21911,87	15245,00!	7898,39	9962,58	5714,00!		!
!13/10/2017!	9,08	10,53	5,00 !	19856,12	21922,40	15250,00!	7907,47	9973,11	5719,00!		!
!14/10/2017!	5,78	6,76	3,00 !	19861,90	21929,15	15253,00!	7913,25	9979,87	5722,00!		!
!15/10/2017!	3,61	4,35	3,00 !	19865,52	21933,50	15256,00!	7916,86	9984,22	5725,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!18/10/2017!	0,33	1,18	1,00 !	19865,85	21934,68	15257,00!	7917,19	9985,40	5726,00!		!
!19/10/2017!	0,67	1,10	1,00 !	19866,52	21935,78	15258,00!	7917,86	9986,50	5727,00!		!
!20/10/2017!	4,31	5,94	6,00 !	19870,83	21941,73	15264,00!	7922,17	9992,44	5733,00!		!
!21/10/2017!	5,93	8,53	9,00 !	19876,77	21950,28	15273,00!	7928,10	10000,97	5742,00!		!
!22/10/2017!	8,01	9,89	6,00 !	19884,77	21960,18	15279,00!	7936,11	10010,86	5748,00!		!
!23/10/2017!	7,57	9,75	5,00 !	19892,33	21969,93	15284,00!	7943,68	10020,61	5753,00!		!
!24/10/2017!	12,86	14,76	8,00 !	19905,20	21984,70	15292,00!	7956,54	10035,37	5761,00!		!
!25/10/2017!	13,21	15,41	8,00 !	19918,42	22000,10	15300,00!	7969,75	10050,78	5769,00!		!
!26/10/2017!	13,20	15,95	8,00 !	19931,62	22016,07	15308,00!	7982,95	10066,73	5777,00!		!
!27/10/2017!	13,24	15,49	8,00 !	19944,87	22031,57	15316,00!	7996,19	10082,22	5785,00!		!
!28/10/2017!	12,99	15,26	8,00 !	19957,85	22046,82	15324,00!	8009,18	10097,48	5793,00!		!
!29/10/2017!	13,07	15,43	8,00 !	19970,92	22062,25	15332,00!	8022,25	10112,91	5801,00!		!
!30/10/2017!	13,11	15,61	8,00 !	19984,03	22077,85	15340,00!	8035,36	10128,52	5809,00!		!
!31/10/2017!	12,90	15,08	8,00 !	19996,93	22092,93	15348,00!	8048,26	10143,60	5817,00!		!
!01/11/2017!	13,18	15,24	8,00 !	20010,12	22108,17	15356,00!	8061,44	10158,84	5825,00!		!
!02/11/2017!	13,06	15,43	8,00 !	20023,18	22123,60	15364,00!	8074,50	10174,27	5833,00!		!
!03/11/2017!	9,04	10,39	5,00 !	20032,22	22133,98	15369,00!	8083,54	10184,66	5838,00!		!
!04/11/2017!	9,00	10,43	5,00 !	20041,22	22144,42	15374,00!	8092,54	10195,09	5843,00!		!
!05/11/2017!	12,80	15,08	8,00 !	20054,03	22159,48	15382,00!	8105,34	10210,17	5851,00!		!
!06/11/2017!	13,17	15,63	8,00 !	20067,20	22175,10	15390,00!	8118,51	10225,80	5859,00!		!
!07/11/2017!	10,26	12,21	7,00 !	20077,45	22187,30	15397,00!	8128,77	10238,01	5866,00!		!
!08/11/2017!	8,68	10,35	5,00 !	20086,13	22197,65	15402,00!	8137,45	10248,36	5871,00!		!
!09/11/2017!	3,93	5,38	4,00 !	20090,07	22203,03	15406,00!	8141,38	10253,74	5875,00!		!
!10/11/2017!	5,08	8,60	7,00 !	20095,15	22211,63	15413,00!	8146,46	10262,34	5882,00!		!
!11/11/2017!	7,06	10,25	10,00 !	20102,22	22221,88	15423,00!	8153,52	10272,59	5892,00!		!
!12/11/2017!	6,45	9,16	9,00 !	20108,65	22231,03	15432,00!	8159,97	10281,75	5901,00!		!
!13/11/2017!	7,07	9,05	5,00 !	20115,72	22240,08	15437,00!	8167,04	10290,80	5906,00!		!
!14/11/2017!	4,02	5,65	5,00 !	20119,73	22245,73	15442,00!	8171,06	10296,45	5911,00!		!
!16/11/2017!	7,18	10,90	8,00 !	20126,90	22256,63	15450,00!	8178,24	10307,35	5919,00!		!
!17/11/2017!	7,72	10,15	6,00 !	20134,62	22266,78	15456,00!	8185,96	10317,50	5925,00!		!
!18/11/2017!	5,14	7,61	8,00 !	20139,75	22274,40	15464,00!	8191,10	10325,11	5933,00!		!
!19/11/2017!	1,13	1,48	1,00 !	20140,88	22275,88	15465,00!	8192,23	10326,59	5934,00!		!
!20/11/2017!	3,59	5,24	4,00 !	20144,47	22281,12	15469,00!	8195,82	10331,83	5938,00!		!
!21/11/2017!	6,85	10,39	9,00 !	20151,32	22291,52	15478,00!	8202,67	10342,22	5947,00!		!
!22/11/2017!	2,09	3,53	3,00 !	20153,40	22295,05	15481,00!	8204,76	10345,75	5950,00!		!
!23/11/2017!	7,21	10,30	10,00 !	20160,62	22305,35	15491,00!	8211,97	10356,05	5960,00!		!
!24/11/2017!	6,99	10,77	10,00 !	20167,62	22316,12	15501,00!	8218,96	10366,82	5970,00!		!
!25/11/2017!	5,58	8,69	8,00 !	20173,20	22324,82	15509,00!	8224,54	10375,51	5978,00!		!
!26/11/2017!	12,87	15,69	8,00 !	20186,07	22340,50	15517,00!	8237,41	10391,20	5986,00!		!
!27/11/2017!	13,07	15,92	8,00 !	20199,15	22356,42	15525,00!	8250,48	10407,12	5994,00!		!
!28/11/2017!	10,24	12,79	7,00 !	20209,40	22369,20	15532,00!	8260,72	10419,91	6001,00!		!
!29/11/2017!	15,89	18,83	9,00 !	20225,28	22388,03	15541,00!	8276,61	10438,74	6010,00!		!
!30/11/2017!	13,31	15,63	8,00 !	20238,58	22403,67	15549,00!	8289,92	10454,37	6018,00!		!
!01/12/2017!	13,11	15,54	8,00 !	20251,68	22419,22	15557,00!	8303,03	10469,91	6026,00!		!
!02/12/2017!	13,00	15,45	8,00 !	20264,68	22434,67	15565,00!	8316,03	10485,36	6034,00!		!
!03/12/2017!	13,10	15,54	8,00 !	20277,77	22450,22	15573,00!	8329,13	10500,90	6042,00!		!
!04/12/2017!	12,95	15,35	8,00 !	20290,72	22465,57	15581,00!	8342,08	10516,25	6050,00!		!
!05/12/2017!	13,14	15,64	8,00 !	20303,85	22481,20	15589,00!	8355,22	10531,89	6058,00!		!
!06/12/2017!	12,94	15,52	8,00 !	20316,78	22496,73	15597,00!	8368,16	10547,41	6066,00!		!
!07/12/2017!	12,90	15,07	8,00 !	20329,68	22511,82	15605,00!	8381,06	10562,48	6074,00!		!
!08/12/2017!	3,76	4,80	3,00 !	20333,45	22516,62	15608,00!	8384,82	10567,28	6077,00!		!
!09/12/2017!	16,03	18,49	9,00 !	20349,47	22535,10	15617,00!	8400,85	10585,77	6086,00!		!
!10/12/2017!	10,19	12,38	7,00 !	20359,67	22547,48	15624,00!	8411,04	10598,15	6093,00!		!
!11/12/2017!	15,68	18,46	9,00 !	20375,35	22565,93	15633,00!	8426,72	10616,61	6102,00!		!
!12/12/2017!	6,28	8,11	5,00 !	20381,62	22574,05	15638,00!	8433,00	10624,72	6107,00!		!
!13/12/2017!	5,12	7,17	7,00 !	20386,73	22581,23	15645,00!	8438,12	10631,89	6114,00!		!
!14/12/2017!	6,11	9,43	9,00 !	20392,83	22590,67	15654,00!	8444,23	10641,32	6123,00!		!
!15/12/2017!	8,48	10,98	6,00 !	20401,32	22601,65	15660,00!	8452,71	10652,30	6129,00!		!
!16/12/2017!	9,32	11,74	7,00 !	20410,63	22613,40	15667,00!	8462,03	10664,04	6136,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!17/12/2017!	3,82	5,45	4,00 !	20414,45	22618,85	15671,00!	8465,85	10669,49	6140,00!		!
!18/12/2017!	6,09	8,94	6,00 !	20420,53	22627,80	15677,00!	8471,94	10678,43	6146,00!		!
!19/12/2017!	8,56	12,50	8,00 !	20429,10	22640,30	15685,00!	8480,50	10690,93	6154,00!		!
!20/12/2017!	6,91	10,84	8,00 !	20436,02	22651,13	15693,00!	8487,41	10701,77	6162,00!		!
!21/12/2017!	12,92	15,12	8,00 !	20448,95	22666,25	15701,00!	8500,33	10716,89	6170,00!		!
!22/12/2017!	10,46	12,49	7,00 !	20459,42	22678,75	15708,00!	8510,79	10729,38	6177,00!		!
!23/12/2017!	16,13	18,54	9,00 !	20475,55	22697,30	15717,00!	8526,92	10747,92	6186,00!		!
!24/12/2017!	10,45	12,71	7,00 !	20486,00	22710,00	15724,00!	8537,37	10760,63	6193,00!		!
!25/12/2017!	9,28	11,14	5,00 !	20495,28	22721,15	15729,00!	8546,65	10771,77	6198,00!		!
!26/12/2017!	13,19	15,68	8,00 !	20508,50	22736,83	15737,00!	8559,84	10787,45	6206,00!		!
!27/12/2017!	13,22	15,74	8,00 !	20521,72	22752,57	15745,00!	8573,06	10803,19	6214,00!		!
!28/12/2017!	12,98	15,75	8,00 !	20534,70	22768,32	15753,00!	8586,04	10818,94	6222,00!		!
!29/12/2017!	12,92	15,19	8,00 !	20547,62	22783,50	15761,00!	8598,96	10834,13	6230,00!		!
!30/12/2017!	13,03	15,64	8,00 !	20560,65	22799,13	15769,00!	8611,99	10849,77	6238,00!		!
!31/12/2017!	9,59	11,11	6,00 !	20570,23	22810,25	15775,00!	8621,58	10860,88	6244,00!		!
!01/01/2018!	9,79	11,65	6,00 !	20580,03	22821,90	15781,00!	8631,37	10872,53	6250,00!		!
!02/01/2018!	13,00	15,55	8,00 !	20593,03	22837,43	15789,00!	8644,37	10888,08	6258,00!		!
!03/01/2018!	10,26	12,55	7,00 !	20603,30	22849,98	15796,00!	8654,63	10900,63	6265,00!		!
!04/01/2018!	15,65	18,54	9,00 !	20618,95	22868,52	15805,00!	8670,28	10919,17	6274,00!		!
!05/01/2018!	12,93	15,41	8,00 !	20631,88	22883,92	15813,00!	8683,21	10934,58	6282,00!		!
!06/01/2018!	13,09	15,38	8,00 !	20644,97	22899,30	15821,00!	8696,30	10949,96	6290,00!		!
!07/01/2018!	13,18	15,50	8,00 !	20658,17	22914,80	15829,00!	8709,48	10965,46	6298,00!		!
!08/01/2018!	12,92	16,89	8,00 !	20671,07	22931,70	15837,00!	8722,40	10982,35	6306,00!		!
!09/01/2018!	12,92	15,15	8,00 !	20683,98	22946,85	15845,00!	8735,32	10997,50	6314,00!		!
!10/01/2018!	7,44	9,46	6,00 !	20691,43	22956,32	15851,00!	8742,76	11006,96	6320,00!		!
!11/01/2018!	8,22	10,64	6,00 !	20699,65	22966,97	15857,00!	8750,98	11017,60	6326,00!		!
!12/01/2018!	7,15	10,13	8,00 !	20706,82	22977,10	15865,00!	8758,13	11027,73	6334,00!		!
!13/01/2018!	9,08	11,39	6,00 !	20715,90	22988,48	15871,00!	8767,21	11039,12	6340,00!		!
!14/01/2018!	2,81	4,45	4,00 !	20718,72	22992,93	15875,00!	8770,02	11043,57	6344,00!		!
!15/01/2018!	4,45	6,75	6,00 !	20723,17	22999,68	15881,00!	8774,47	11050,32	6350,00!		!
!16/01/2018!	4,84	6,96	7,00 !	20728,00	23006,63	15888,00!	8779,31	11057,28	6357,00!		!
!17/01/2018!	6,52	10,21	9,00 !	20734,53	23016,83	15897,00!	8785,83	11067,49	6366,00!		!
!18/01/2018!	4,30	6,06	6,00 !	20738,83	23022,90	15903,00!	8790,13	11073,55	6372,00!		!
!19/01/2018!	3,42	4,80	5,00 !	20742,25	23027,70	15908,00!	8793,55	11078,35	6377,00!		!
!20/01/2018!	7,77	9,75	6,00 !	20750,02	23037,45	15914,00!	8801,32	11088,10	6383,00!		!
!21/01/2018!	7,85	9,82	6,00 !	20757,87	23047,27	15920,00!	8809,17	11097,92	6389,00!		!
!22/01/2018!	5,48	7,90	8,00 !	20763,37	23055,17	15928,00!	8814,65	11105,82	6397,00!		!
!23/01/2018!	4,94	6,76	6,00 !	20768,30	23061,93	15934,00!	8819,59	11112,58	6403,00!		!
!25/01/2018!	1,50	2,19	2,00 !	20769,80	23064,12	15936,00!	8821,09	11114,77	6405,00!		!
!26/01/2018!	7,79	9,69	6,00 !	20777,60	23073,82	15942,00!	8828,88	11124,46	6411,00!		!
!27/01/2018!	8,35	10,70	6,00 !	20785,95	23084,50	15948,00!	8837,23	11135,16	6417,00!		!
!28/01/2018!	4,12	6,17	6,00 !	20790,07	23090,67	15954,00!	8841,35	11141,33	6423,00!		!
!29/01/2018!	7,43	9,13	5,00 !	20797,48	23099,80	15959,00!	8848,78	11150,46	6428,00!		!
!30/01/2018!	6,42	9,28	8,00 !	20803,92	23109,07	15967,00!	8855,20	11159,74	6436,00!		!
!31/01/2018!	8,22	11,00	6,00 !	20812,13	23120,07	15973,00!	8863,42	11170,74	6442,00!		!
!01/02/2018!	7,42	9,74	5,00 !	20819,55	23129,80	15978,00!	8870,84	11180,48	6447,00!		!
!02/02/2018!	7,76	10,18	6,00 !	20827,32	23139,98	15984,00!	8878,60	11190,66	6453,00!		!
!03/02/2018!	5,55	7,86	8,00 !	20832,87	23147,85	15992,00!	8884,15	11198,52	6461,00!		!
!04/02/2018!	4,24	5,87	6,00 !	20837,12	23153,72	15998,00!	8888,39	11204,39	6467,00!		!
!05/02/2018!	7,26	9,04	5,00 !	20844,38	23162,75	16003,00!	8895,65	11213,43	6472,00!		!
!06/02/2018!	4,32	5,88	6,00 !	20848,70	23168,63	16009,00!	8899,97	11219,31	6478,00!		!
!07/02/2018!	6,39	9,53	9,00 !	20855,08	23178,17	16018,00!	8906,36	11228,84	6487,00!		!
!08/02/2018!	2,86	4,24	4,00 !	20857,93	23182,42	16022,00!	8909,22	11233,08	6491,00!		!
!09/02/2018!	2,84	4,41	4,00 !	20860,78	23186,82	16026,00!	8912,06	11237,49	6495,00!		!
!10/02/2018!	4,18	6,20	6,00 !	20864,97	23193,02	16032,00!	8916,24	11243,69	6501,00!		!
!11/02/2018!	2,76	3,91	4,00 !	20867,73	23196,92	16036,00!	8919,00	11247,60	6505,00!		!
!13/02/2018!	2,85	4,04	4,00 !	20870,58	23200,95	16040,00!	8921,85	11251,64	6509,00!		!
!14/02/2018!	8,58	10,54	6,00 !	20879,17	23211,48	16046,00!	8930,43	11262,18	6515,00!		!
!15/02/2018!	6,16	7,99	7,00 !	20885,33	23219,48	16053,00!	8936,59	11270,17	6522,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!16/02/2018!	8,71	10,85	6,00 !	20894,05	23230,33	16059,00!	8945,30	11281,02	6528,00!		!
!17/02/2018!	7,47	9,28	6,00 !	20901,52	23239,62	16065,00!	8952,77	11290,30	6534,00!		!
!18/02/2018!	1,32	1,95	2,00 !	20902,83	23241,57	16067,00!	8954,09	11292,25	6536,00!		!
!19/02/2018!	3,85	4,63	3,00 !	20906,68	23246,20	16070,00!	8957,94	11296,88	6539,00!		!
!20/02/2018!	12,97	15,72	8,00 !	20919,65	23261,92	16078,00!	8970,91	11312,60	6547,00!		!
!21/02/2018!	13,06	15,56	8,00 !	20932,72	23277,48	16086,00!	8983,97	11328,16	6555,00!		!
!22/02/2018!	12,99	15,33	8,00 !	20945,72	23292,82	16094,00!	8996,96	11343,49	6563,00!		!
!23/02/2018!	12,88	15,07	8,00 !	20958,58	23307,88	16102,00!	9009,84	11358,56	6571,00!		!
!24/02/2018!	12,96	14,89	8,00 !	20971,55	23322,78	16110,00!	9022,80	11373,45	6579,00!		!
!25/02/2018!	5,11	6,15	3,00 !	20976,67	23328,93	16113,00!	9027,91	11379,60	6582,00!		!
!26/02/2018!	3,65	5,95	5,00 !	20980,32	23334,88	16118,00!	9031,56	11385,55	6587,00!		!
!27/02/2018!	8,68	11,41	7,00 !	20989,00	23346,28	16125,00!	9040,24	11396,96	6594,00!		!
!28/02/2018!	6,87	10,05	10,00 !	20995,88	23356,32	16135,00!	9047,11	11407,01	6604,00!		!
!01/03/2018!	6,92	10,20	10,00 !	21002,80	23366,52	16145,00!	9054,03	11417,21	6614,00!		!
!03/03/2018!	5,62	8,45	9,00 !	21008,42	23374,97	16154,00!	9059,65	11425,66	6623,00!		!
!04/03/2018!	3,50	5,31	5,00 !	21011,92	23380,28	16159,00!	9063,15	11430,97	6628,00!		!
!05/03/2018!	10,55	12,47	7,00 !	21022,47	23392,75	16166,00!	9073,70	11443,44	6635,00!		!
!06/03/2018!	12,96	15,03	8,00 !	21035,43	23407,77	16174,00!	9086,66	11458,47	6643,00!		!
!07/03/2018!	13,02	15,14	8,00 !	21048,45	23422,92	16182,00!	9099,68	11473,61	6651,00!		!
!08/03/2018!	13,04	15,54	8,00 !	21061,50	23438,45	16190,00!	9112,72	11489,15	6659,00!		!
!09/03/2018!	12,89	15,20	8,00 !	21074,38	23453,65	16198,00!	9125,61	11504,35	6667,00!		!
!10/03/2018!	13,01	15,07	8,00 !	21087,40	23468,72	16206,00!	9138,62	11519,42	6675,00!		!
!11/03/2018!	2,63	3,55	1,00 !	21090,03	23472,27	16207,00!	9141,25	11522,97	6676,00!		!
!14/03/2018!	3,23	3,81	2,00 !	21093,27	23476,08	16209,00!	9144,48	11526,78	6678,00!		!
!15/03/2018!	0,00	0,45	0,00 !	21093,27	23476,53	16209,00!	9144,48	11527,23	6678,00!		!
!16/03/2018!	9,75	11,40	5,00 !	21103,02	23487,93	16214,00!	9154,23	11538,63	6683,00!		!
!17/03/2018!	10,26	12,39	7,00 !	21113,28	23500,32	16221,00!	9164,49	11551,02	6690,00!		!
!18/03/2018!	13,00	15,30	8,00 !	21126,28	23515,62	16229,00!	9177,49	11566,32	6698,00!		!
!19/03/2018!	13,05	15,36	8,00 !	21139,33	23530,98	16237,00!	9190,54	11581,68	6706,00!		!
!20/03/2018!	12,90	15,23	8,00 !	21152,23	23546,22	16245,00!	9203,44	11596,91	6714,00!		!
!21/03/2018!	12,94	15,44	8,00 !	21165,17	23561,65	16253,00!	9216,38	11612,35	6722,00!		!
!22/03/2018!	13,10	15,28	8,00 !	21178,27	23576,92	16261,00!	9229,48	11627,63	6730,00!		!
!23/03/2018!	12,98	15,42	8,00 !	21191,25	23592,32	16269,00!	9242,46	11643,05	6738,00!		!
!24/03/2018!	15,72	18,63	9,00 !	21206,97	23610,95	16278,00!	9258,18	11661,68	6747,00!		!
!25/03/2018!	13,29	15,80	8,00 !	21220,25	23626,75	16286,00!	9271,47	11677,48	6755,00!		!
!26/03/2018!	10,66	12,69	7,00 !	21230,92	23639,43	16293,00!	9282,13	11690,17	6762,00!		!
!27/03/2018!	13,18	15,36	8,00 !	21244,10	23654,80	16301,00!	9295,31	11705,53	6770,00!		!
!28/03/2018!	13,09	15,50	8,00 !	21257,18	23670,30	16309,00!	9308,40	11721,03	6778,00!		!
!29/03/2018!	13,08	15,82	8,00 !	21270,27	23686,13	16317,00!	9321,48	11736,85	6786,00!		!
!30/03/2018!	12,43	14,72	7,00 !	21282,70	23700,85	16324,00!	9333,91	11751,57	6793,00!		!
!31/03/2018!	6,59	7,72	4,00 !	21289,28	23708,57	16328,00!	9340,50	11759,29	6797,00!		!
!01/04/2018!	9,79	11,67	6,00 !	21299,08	23720,23	16334,00!	9350,29	11770,96	6803,00!		!
!02/04/2018!	8,14	10,79	7,00 !	21307,22	23731,02	16341,00!	9358,43	11781,75	6810,00!		!
!03/04/2018!	5,98	8,57	8,00 !	21313,20	23739,58	16349,00!	9364,41	11790,32	6818,00!		!
!04/04/2018!	6,64	9,30	9,00 !	21319,83	23748,88	16358,00!	9371,05	11799,62	6827,00!		!
!05/04/2018!	8,15	11,15	6,00 !	21327,98	23760,03	16364,00!	9379,20	11810,77	6833,00!		!
!06/04/2018!	8,50	10,37	6,00 !	21336,48	23770,40	16370,00!	9387,70	11821,14	6839,00!		!
!07/04/2018!	4,98	6,85	7,00 !	21341,47	23777,25	16377,00!	9392,68	11827,99	6846,00!		!
!08/04/2018!	4,21	5,96	6,00 !	21345,67	23783,20	16383,00!	9396,89	11833,95	6852,00!		!
!09/04/2018!	7,65	10,38	7,00 !	21353,32	23793,58	16390,00!	9404,54	11844,33	6859,00!		!
!10/04/2018!	7,99	10,92	6,00 !	21361,30	23804,50	16396,00!	9412,53	11855,25	6865,00!		!
!11/04/2018!	7,64	9,73	6,00 !	21368,95	23814,23	16402,00!	9420,17	11864,98	6871,00!		!
!12/04/2018!	5,12	6,14	3,00 !	21374,07	23820,38	16405,00!	9425,29	11871,12	6874,00!		!
!13/04/2018!	7,84	10,25	6,00 !	21381,90	23830,63	16411,00!	9433,13	11881,37	6880,00!		!
!14/04/2018!	9,08	11,12	7,00 !	21390,98	23841,75	16418,00!	9442,21	11892,49	6887,00!		!
!15/04/2018!	9,50	11,60	6,00 !	21400,48	23853,35	16424,00!	9451,71	11904,09	6893,00!		!
!16/04/2018!	8,24	10,57	6,00 !	21408,73	23863,93	16430,00!	9459,95	11914,66	6899,00!		!
!17/04/2018!	8,55	10,74	6,00 !	21417,28	23874,67	16436,00!	9468,50	11925,40	6905,00!		!
!18/04/2018!	8,58	11,07	6,00 !	21425,87	23885,73	16442,00!	9477,08	11936,47	6911,00!		!

! Date !	A/B H	B/B H	Cycles !	A/B TSN	B/B TSN	CSN!	TOTAL A/B	TOTAL B/B	TOTAL CY	! COMMENTS	!
!19/04/2018!	8,99	11,86	7,00 !	21434,85	23897,60	16449,00!	9486,07	11948,33	6918,00!		!
!20/04/2018!	7,06	9,04	5,00 !	21441,90	23906,63	16454,00!	9493,13	11957,37	6923,00!		!
!21/04/2018!	7,79	10,14	6,00 !	21449,68	23916,78	16460,00!	9500,92	11967,51	6929,00!		!
!22/04/2018!	8,40	10,77	6,00 !	21458,08	23927,55	16466,00!	9509,32	11978,28	6935,00!		!
!23/04/2018!	7,38	9,66	7,00 !	21465,45	23937,20	16473,00!	9516,70	11987,94	6942,00!		!
!24/04/2018!	8,04	10,37	8,00 !	21473,48	23947,57	16481,00!	9524,74	11998,31	6950,00!		!
!25/04/2018!	6,34	9,71	9,00 !	21479,82	23957,28	16490,00!	9531,08	12008,02	6959,00!		!
!26/04/2018!	7,38	9,26	5,00 !	21487,20	23966,55	16495,00!	9538,46	12017,28	6964,00!		!
!27/04/2018!	1,45	2,44	2,00 !	21488,65	23968,98	16497,00!	9539,91	12019,72	6966,00!		!
!28/04/2018!	5,82	7,13	4,00 !	21494,47	23976,12	16501,00!	9545,73	12026,85	6970,00!		!
!29/04/2018!	4,67	5,36	3,00 !	21499,13	23981,48	16504,00!	9550,40	12032,21	6973,00!		!
!30/04/2018!	1,33	2,08	2,00 !	21500,47	23983,57	16506,00!	9551,73	12034,29	6975,00!		!
!01/05/2018!	3,91	4,69	3,00 !	21504,37	23988,27	16509,00!	9555,64	12038,98	6978,00!		!
!02/05/2018!	8,72	10,99	6,00 !	21513,08	23999,27	16515,00!	9564,36	12049,97	6984,00!		!
!03/05/2018!	11,00	13,82	8,00 !	21524,08	24013,08	16523,00!	9575,36	12063,79	6992,00!		!
!04/05/2018!	13,12	16,02	8,00 !	21537,22	24029,10	16531,00!	9588,48	12079,81	7000,00!		!
!05/05/2018!	10,60	13,00	7,00 !	21547,82	24042,10	16538,00!	9599,08	12092,81	7007,00!		!
!06/05/2018!	7,81	10,03	7,00 !	21555,63	24052,13	16545,00!	9606,89	12102,84	7014,00!		!
!09/05/2018!	7,01	10,44	7,00 !	21562,65	24062,57	16552,00!	9613,90	12113,28	7021,00!		!
!10/05/2018!	9,33	12,18	7,00 !	21571,98	24074,75	16559,00!	9623,23	12125,46	7028,00!		!
!11/05/2018!	12,94	15,42	8,00 !	21584,93	24090,18	16567,00!	9636,17	12140,88	7036,00!		!
!12/05/2018!	10,41	13,51	7,00 !	21595,35	24103,70	16574,00!	9646,58	12154,39	7043,00!		!
!13/05/2018!	0,75	1,08	1,00 !	21596,10	24104,78	16575,00!	9647,33	12155,47	7044,00!		!
!14/05/2018!	2,13	3,40	3,00 !	21598,23	24108,18	16578,00!	9649,46	12158,87	7047,00!		!
!15/05/2018!	4,45	7,60	7,00 !	21602,68	24115,78	16585,00!	9653,91	12166,47	7054,00!		!
!16/05/2018!	8,50	11,18	6,00 !	21611,18	24126,97	16591,00!	9662,41	12177,65	7060,00!		!
!17/05/2018!	6,59	9,57	9,00 !	21617,77	24136,53	16600,00!	9669,00	12187,22	7069,00!		!
!18/05/2018!	4,10	5,74	4,00 !	21621,87	24142,27	16604,00!	9673,10	12192,96	7073,00!		!
!19/05/2018!	11,05	14,78	10,00 !	21632,92	24157,05	16614,00!	9684,15	12207,74	7083,00!		!
!20/05/2018!	7,40	9,74	7,00 !	21640,32	24166,78	16621,00!	9691,55	12217,48	7090,00!		!
!21/05/2018!	7,53	9,75	6,00 !	21647,85	24176,55	16627,00!	9699,08	12227,23	7096,00!		!
!22/05/2018!	8,38	12,14	10,00 !	21656,22	24188,68	16637,00!	9707,46	12239,37	7106,00!		!
!23/05/2018!	5,09	7,63	7,00 !	21661,30	24196,32	16644,00!	9712,55	12247,00	7113,00!		!
!24/05/2018!	6,85	10,33	10,00 !	21668,15	24206,65	16654,00!	9719,40	12257,33	7123,00!		!
!27/05/2018!	3,37	4,32	3,00 !	21671,52	24210,97	16657,00!	9722,77	12261,65	7126,00!		!
!28/05/2018!	4,11	5,57	4,00 !	21675,63	24216,53	16661,00!	9726,88	12267,22	7130,00!		!
!29/05/2018!	1,38	2,07	2,00 !	21677,02	24218,60	16663,00!	9728,26	12269,29	7132,00!		!
!30/05/2018!	8,94	11,43	7,00 !	21685,97	24230,02	16670,00!	9737,20	12280,72	7139,00!		!
!31/05/2018!	3,97	5,20	4,00 !	21689,93	24235,22	16674,00!	9741,17	12285,92	7143,00!		!
!03/06/2018!	0,88	1,35	1,00 !	21690,82	24236,57	16675,00!	9742,05	12287,27	7144,00!		!
!04/06/2018!	4,41	6,75	6,00 !	21695,22	24243,32	16681,00!	9746,46	12294,02	7150,00!		!
!05/06/2018!	9,22	11,95	7,00 !	21704,45	24255,27	16688,00!	9755,68	12305,97	7157,00!		!
!06/06/2018!	8,54	10,84	6,00 !	21712,98	24266,12	16694,00!	9764,22	12316,81	7163,00!		!
!07/06/2018!	6,21	8,75	8,00 !	21719,20	24274,87	16702,00!	9770,43	12325,56	7171,00!		!
!08/06/2018!	3,48	5,06	5,00 !	21722,68	24279,93	16707,00!	9773,91	12330,62	7176,00!		!
!09/06/2018!	7,57	10,09	8,00 !	21730,25	24290,03	16715,00!	9781,48	12340,71	7184,00!		!
!10/06/2018!	5,85	8,39	9,00 !	21736,10	24298,42	16724,00!	9787,33	12349,10	7193,00!		!
!11/06/2018!	7,32	8,83	5,00 !	21743,42	24307,25	16729,00!	9794,65	12357,93	7198,00!		!
!12/06/2018!	7,06	10,72	10,00 !	21750,48	24317,98	16739,00!	9801,71	12368,65	7208,00!		!
!13/06/2018!	5,55	8,49	8,00 !	21756,03	24326,47	16747,00!	9807,26	12377,14	7216,00!		!
!14/06/2018!	4,89	7,19	6,00 !	21760,92	24333,67	16753,00!	9812,15	12384,33	7222,00!		!
!21/06/2018!	1,35	1,92	2,00 !	21762,27	24335,58	16755,00!	9813,50	12386,25	7224,00!		!
!25/06/2018!	1,37	2,08	2,00 !	21763,63	24337,67	16757,00!	9814,87	12388,33	7226,00!		!
!27/06/2018!	1,33	1,92	1,00 !	21764,97	24339,58	16758,00!	9816,20	12390,25	7227,00!		!



Component Overhaul Services

FAA Repair Station # 8COR883B

EASA 145.6404

Landing Gear Overhaul Report

W/O: 12614

Description: NLG LEG ASSY

P/N: D23757500-7

S/N: B3137

DEC. 2024

SECTION 1. AIRWORTHINESS APPROVAL TAGS

1. Approving Civil Aviation Authority/Country: FAA/UNITED STATES	AUTHORIZED RELEASE CERTIFICATE FAA FORM 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: 219353
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4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74th St. Suite B, Medley, FL 33166 TEL: (305) 406-3885 FAX: (786) 400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12614 / 00023395
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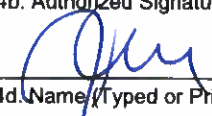
6. Item	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:
1	NOSE LANDING GEAR	D23757500-7	1	B3137	OVERHAULED

12. Remarks:

OVERHAULED I.A.W. SAFRAN CMM 32-27-38 REV. 6 DATED JAN 18/2018. REFER TO COMPONENT OVERHAUL SERVICES WORK STATEMENT REPORT NUMBER 12614 FOR INSPECTION FINDINGS, SERIALIZED COMPONENT LIST, TRACEABILITY, C.S.N., S.B. AND A.D. MODIFICATION LIST.
CSO: -0-

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): DOUGLAS ARAUZ	14e. Date (dd/mmm/yyyy): 10/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.


Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1. It is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statement in Blocks 13a and 14a do not constitute installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	3. Form Tracking Number: COS7494
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AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12614-58 00023395
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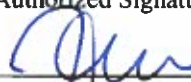
6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11. Status/Work:
1	BARREL	D67583	1	08B0300X8772	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-27-32 Rev. 10 Rev. date 31/Mar/2022.

CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 10/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7495	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  Component Overhaul Services				5. Work Order/Contract/Invoice Number: 12614-158 00023395		
COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B						
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	HINGE PIN (BARREL/STRUCTURE)	D65588-1	1	07JL113X78	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-32 Rev. 10 Rev. date 31/Mar/2022. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/ Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7496	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12614-159 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	HINGE PIN (BARREL/STRUCTURE)	D65588-1	1	07JL113X64	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-32 Rev. 10 Rev. date 31/Mar/2022. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin: 0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin: 0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>				3. Form Tracking Number: COS7497
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B					5. Work Order/Contract/Invoice Number: 12614-160 00023395	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	PIN (RETRACTION ACTUATOR/LOCK LINK BARREL)	D65589-1	1	07JL120X05	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-32 Rev. 10 Rev. date 31/Mar/2022. <div style="float: right; text-align: right;"> CSN: 16,758 CSO: 0 Previous Operator: AVIANCA </div>						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		
				 8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14c. Approval/Certificate No.:		
				14d. Name (Typed or Printed): Douglas Arauz 14e. Date (dd/mmm/yyyy): 10/Dec/2024		
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: COS7498
4. Organization Name and Address:  Component Overhaul Services COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12614-164 00023395	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	PIN (FORESTAY/BARREL)	D65599-1	1	07JL121X17	Overhauled
12. Remarks: Overhauled I.A.W Safran CMM 32-27-32 Rev. 10 Rev. date 31/Mar/2022. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA					
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024
User/Installer Responsibilities					
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.					
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.					
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.					

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7499	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12614 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	SHOCK ABSORBER	D23592030	1	B443	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-33 Rev. 11 Rev. date 15/Mar/2022. CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7500	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  Component Overhaul Services				5. Work Order/Contract/Invoice Number: 12614-29 00023395		
COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B						
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	CYLINDER	D65519	1	07TOF3816P9X003	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-33 Rev. 11 Rev. date 15/Mar/2022. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7501	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  Component Overhaul Services				5. Work Order/Contract/Invoice Number: 12614-34 00023395		
COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B						
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	SLIDING TUBE	D66679	1	08B0713X3712	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-33 Rev. 11 Rev. date 15/Mar/2022. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7502	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 13367 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	AXLE	D66680	1	L952	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-33 Rev. 11 Rev. date 15/Mar/2022. CSN: 27,647 CSO: 0 Previous Operator: AERO NOMAD						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7503	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12614 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	FORESTAY ASSEMBLY	D23596000-3	1	WIA283	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-22 Rev. 11 Rev. date 01/Apr/2024. CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7504	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12614-88 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	FORESTAY LOWER LINK	D67802	1	08B0060X11	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-23 Rev. 17 Rev. date 01/Apr/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/ Authorization No.:	14b. Authorized Signature: 		14c. Approval/ Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: COS7505		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12614-84 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	PIN (LOWER LINK / U-JOINT)	D65773	1	WIA024-289	Overhauled
12. Remarks: <div style="display: flex; justify-content: space-between;"> <div>Overhauled I.A.W Safran CMM 32-27-23 Rev. 17 Rev. date 01/Apr/2024.</div> <div>CSN: 16,758 CSO: 0 Previous Operator: AVIANCA</div> </div> <p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/ Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 10/Dec/2024	
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States		AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: COS7506	
4. Organization Name and Address: 					COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	PIN (LOCK LINK LOWER)	D67928	1	WIA022-381	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-23 Rev. 17 Rev. date 01/Apr/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: <h1 style="margin:0;">COS7507</h1>		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12614-90 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	CROSS PIN	D67932	1	WIA011-168	Overhauled
12. Remarks: Overhauled I.A.W Safran CMM 32-27-23 Rev. 17 Rev. date 01/Apr/2024. <div style="float: right; text-align: right;"> CSN: 16,758 CSO: 0 Previous Operator: AVIANCA </div>					
<p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature:		14c. Approval/Certificate No.:
					8COR883B
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):
			Douglas Arauz		10/Dec/2024
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7508	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12614-60 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11 Status/Work:	
1	UNIVERSAL JOINT	D59638	1	WIA025-244	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-23 Rev. 17 Rev. date 01/Apr/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7509	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12614-85 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	PIN (UPPER PANEL-U-JOINT)	D65887	1	WIA023-325	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-23 Rev. 17 Rev. date 01/Apr/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7510	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12614-59 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	FORESTAY UPPER PANEL	D65237	1	WIA033-291	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-23 Rev. 17 Rev. date 01/Apr/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7511	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12614-74 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	PIN (UPPER PANEL/STRUCTURE)	D65616	1	WIA029-465	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-23 Rev. 17 Rev. date 01/Apr/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Afauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7512	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12614-75 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	PIN (UPPER PANEL/STRUCTURE)	D65616	1	WIA029-464	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-23 Rev. 17 Rev. date 01/Apr/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/ Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7514	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12614-97 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	LOCK LINK LOWER LINK	D59635	1	WIA034-318	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-24 Rev. 11 Rev. date 01/Apr/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:
						8COR883B
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):
				Douglas Arauz		10/Dec/2024
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7515	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12614-101 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	LOCK LINK CENTER PIN	D60054	1	WIA017-234	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-27-24 Rev. 11 Rev. date 01/Apr/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/ Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 10/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7516			
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B		5. Work Order/Contract/Invoice Number: 12614-98 00023395			
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	LOCK LINK UPPER LINK	D59636-1	1	WIA036-338	Overhauled
12. Remarks: Overhauled I.A.W Safran CMM 32-27-24 Rev. 11 Rev. date 01/Apr/2024. <div style="float: right; text-align: right;"> CSN: 16,758 CSO: 0 Previous Operator: AVIANCA </div>					
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 10/Dec/2024	
User/Installer Responsibilities					
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.					
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.					
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.					

SECTION 2. LIFE LIMITED/SERIALIZED PARTS LIST



**Component Overhaul
Services**

**A320 Leg Assembly
LIFE LIMITED/SERIALIZED PARTS LIST
FAA REPAIR STATION # 8COR883B / EASA 145.6404**

**Customer: CAVU AEROSPACE INC.
Purchase Order: 00023395
Work Order: 12614**

INT. ITEM #	DESCRIPTION	PART NUMBER	S/N	WORK ORDER	LAST OPERATOR	LIFE LIMIT	CSN	COMMENTS
	NLG LEG ASSY	D23757500-7	B3137	12614	AVIANCA			
1	BARREL	D67583	08B0300X8772	12614-58	AVIANCA	60,000	16,758	
2	HINGE PIN (BARREL/STRUCTURE)	D65588-1	07JL113X78	12614-158	AVIANCA	60,000	16,758	
3	HINGE PIN (BARREL/STRUCTURE)	D65588-1	07JL113X64	12614-159	AVIANCA	60,000	16,758	
4	PIN (RETRACTION ACTUATOR/LOCK LINK BA	D65589-1	07JL120X05	12614-160	AVIANCA	60,000	16,758	
5	PIN (FORESTAY/BARREL)	D65599-1	07JL121X17	12614-164	AVIANCA	60,000	16,758	
	SHOCK ABSORBER	D23592030	B443	12614	AVIANCA			
6	CYLINDER	D65519	07TOF3816P9X003	12614-29	AVIANCA	60,000	16,758	
7	SLIDING TUBE	D66679	08B0713X3712	12614-34	AVIANCA	60,000	16,758	
8	AXLE	D66680	L952	13367	AERO NOMAD	60,000	27,647	
	FORESTAY ASSEMBLY	D23596000-3	WIA283	12614	AVIANCA			
2.1	FORESTAY LOWER LINK	D67802	08B0060X11	12614-88	AVIANCA	60,000	16,758	
2.2	PIN (LOWER LINK / U-JOINT)	D65773	WIA024-289	12614-84	AVIANCA	60,000	16,758	
2.3	PIN (LOCK LINK LOWER)	D67928	WIA022-381	12614-89	AVIANCA	60,000	16,758	
2.4	CROSS PIN	D67932	WIA011-168	12614-90	AVIANCA	60,000	16,758	
2.5	UNIVERSAL JOINT	D59638	WIA025-244	12614-60	AVIANCA	60,000	16,758	
2.6	PIN (UPPER PANEL-U-JOINT)	D65887	WIA023-325	12614-85	AVIANCA	60,000	16,758	
2.7	FORESTAY UPPER PANEL	D65237	WIA033-291	12614-59	AVIANCA	60,000	16,758	
2.8.1	PIN (UPPER PANEL/STRUCTURE	D65616	WIA029-465	12614-74	AVIANCA	60,000	16,758	
2.8.2	PIN (UPPER PANEL/STRUCTURE	D65616	WIA029-464	12614-75	AVIANCA	60,000	16,758	
2.9	LOCK LINK LOWER LINK	D59635	WIA034-318	12614-97	AVIANCA	60,000	16,758	
2.10	LOCK LINK CENTER PIN	D60054	WIA017-234	12614-101	AVIANCA	60,000	16,758	
2.11	LOCK LINK UPPER LINK	D59636-1	WIA036-338	12614-98	AVIANCA	60,000	16,758	

Asell

12/10/24

Statement nº 1049 /2019

Date: São Paulo, June 07th, 2019

Incident/Accident Clearance Statement – Airframe

Aircraft: Airbus A318-100 | MSN 3642 (PR-ONR) | TSN 21764,97 | CSN 16758
Engine: PW6000 | ESN | TSN | CSN
Engine: PW6000 | ESN P318123 | TSN 27722,97 | CSN 20883
APU: 4500001B | SN 2924 | TSN 14569 | CSN 14920
NLG: NA28008-015 | SN B3137 | TSN 21764,97 | CSN 16758
MLG LH: 201581001 | SN MDL3642 | TSN 21764,97 | CSN 16758
MLG RH: 201581002 | SN MDL3642 | TSN 21764,97 | CSN 16758

To whom it may concern:

This letter is to certify that the referred aircraft has been operated by Oceanair Linhas Aéreas S.A. during the period stated below and to the best of my knowledge:

Operation period: from December 3, 2013 thru June 7, 2019.

1. Neither the aircraft, nor any part installed have been, unless its airworthiness status was re-established by an approved maintenance organization in accordance with the instructions of the type certificate holder and/or OEM of the part, and supported by an authorized release certificate:

a. Damaged during a reportable incident or accident as defined by Brazilian Civil Aviation Authority regulation RBAC 121.703 – Service Difficulty Reports, equivalent to FAA Title 14 - Chapter I - Subchapter G - Part 121 - Subpart V - §121.703, or

b. Subjected to severe stress or heat (such as in a major engine failure, accident or fire) or has been submersed in salt water.

2. No part has been installed on the aircraft which was obtained from military source or was previously fitted to a state aircraft.



Alberto Ottavio Spelta
Chief Inspector

Rua Tamoios 579 Jardim Aeroporto – São Paulo – SP – CEP 04630-001
Telefone: (11) 3475 8200

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description										AMM PN Description	F.I.N	Maint. level Specific Reference	Zone	Phasing	Tolerance	
	Installed Equipment (PN/SN)																
	WO number	WP	Scheduled check														
Inspection				Inspection 2				Overhaul				Life Limit					
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline		
32LMG79 28/10/2008	PIN (RETAINING LOWER BEARING) PN : 201383608 SN : 08BEV81759011										RETAINIG PIN						TCI
LLT																	
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00			
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918			
32LMG83 28/10/2008	BASIC SIDE STAY PN : 201163004-015 SN : AP1269										STAY-SIDE					TCI	
LLT																	
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00			
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918			
32LMG84 28/10/2008	LEG AND DRESSINGS-MLG PN : 201582001-040 SN : MDG4845										LEG AND DRESSINGS-MLG					TCI	
LLT																	
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00			
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918			
32NG00 28/10/2008	HINGE PIN PN : D65588-1 SN : 07JL113X78										PIN, HINGE						
LLT																	
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918			

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance									
	Installed Equipment (PN/SN)		PN Description		Specific Reference												
	WO number	WP	Scheduled check														
	Inspection				Inspection 2				Overhaul				Life Limit				
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32NG01 28/10/2008	HINGE PIN PN : D65588-1		SN : 07JL113X64	PIN, HINGE													
LLT CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			
32NG02 28/10/2008	BARREL PN : D67583		SN : 08B0300X8772	BARREL													
LLT CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			
32NG03 28/10/2008	PIN PN : D65589-1		SN : 07JL120X05	PIN													
LLT CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			
32NG04 28/10/2008	PIN PN : D65599-1		SN : 07JL121X17	PIN													
LLT CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description Installed Equipment (PN/SN) WO number WP Scheduled check				AMM PN Description				F.I.N		Maint. level Specific Reference		Zone		Phasing		Tolerance			
	Inspection				Inspection 2				Overhaul				Life Limit							
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline				
32NG05 28/10/2008	NGL SHOCK ABSORBER CYLINDER PN : D65519 SN : 07TOF3816P9X003				CYLINDER															
LLT CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918						
32NG06 28/10/2008	NGL SLIDING TUBE PN : D66679 SN : 08B0713X3712				SLIDING TUBE															
LLT CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918						
32NG07 28/10/2008	NGL WHEEL AXLE PN : D66680 SN : 08MPS13036X6				WHEEL AXLE															
LLT CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918						
32NG08 28/10/2008	FORESTAY LOWER LINK PN : D67802 SN : 08B0060X11				FORESTAY LOWER LINK															
LLT CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918						

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance										
	Installed Equipment (PN/SN)		PN Description		Specific Reference													
	WO number	WP	Scheduled check															
		Inspection				Inspection 2				Overhaul				Life Limit				
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32NG09 28/10/2008	PIN PN : D65773		SN : WIA024-289		PIN													
	LLT																	
	CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
	Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		
32NG10 28/10/2008	LOCK LINK LOWER PIN PN : D67928		SN : WIA022-381		PIN, LOCK LINK LOWER													
	LLT																	
	CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
	Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		
32NG11 28/10/2008	CROSS PIN PN : D67932		SN : WIA011-168		PIN CROSS													
	LLT																	
	CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
	Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		
32NG12 28/10/2008	UNIVERSAL JOINT PN : D59638		SN : WIA025-244		UNIVERSAL JOINT													
	LLT																	
	CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	31400	16758,00	14642,00	31400,00
	Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance										
	Installed Equipment (PN/SN)		PN Description		Specific Reference													
	WO number	WP	Scheduled check															
		Inspection				Inspection 2				Overhaul				Life Limit				
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32NG13 28/10/2008	PIN PN : D65887		SN : WIA023-325		PIN													
	LLT																	
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	31400	16758,00	14642,00	31400,00	
	Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			
32NG14 28/10/2008	FORESTAY UPPER PANEL PN : D65237		SN : WIA033-291		FORESTAY UPPER PANEL													
	LLT																	
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	31400	16758,00	14642,00	31400,00	
	Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			
32NG15 28/10/2008	PIN PN : D65616		SN : WIA029-465		PIN													
	LLT																	
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	31400	16758,00	14642,00	31400,00	
	Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			
32NG16 28/10/2008	PIN PN : D65616		SN : WIA029-464		PIN													
	LLT																	
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	31400	16758,00	14642,00	31400,00	
	Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32NG17 28/10/2008	LOCK LINK LOWER LINK PN : D59635		SN : WIA034-318	LOCK LINK LOWER LINK												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32NG18 28/10/2008	LOCK LINK CENTER PIN PN : D60054		SN : WIA017-234	PIN, LOCK LINK CENTER												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32NG19 28/10/2008	LOCK LINK UPPER LINK PN : D59636-1		SN : WIA036-338	LOCK LINK UPPER LINK												
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32NG21 28/10/2008	STEERING PIN PN : D59954		SN : 07CZ82036X22	PIN STEERING											TCI	
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)		PN Description		Specific Reference										
	WO number	WP	Scheduled check												
		Inspection		Inspection 2		Overhaul		Life Limit							
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32NG22 28/10/2008	TORNING TUBE PN : D65520		SN : 07MGP70395X2469	TORNING TUBE										TCI	
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG23 28/10/2008	TORQUE LINK PN : D65234		SN : 08FGA239526X472	TORQUE LINK										TCI	
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG24 28/10/2008	TORQUE LINK PN : D65234		SN : 08FGA234441X385	TORQUE LINK										TCI	
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG25 28/10/2008	TORQUE LINK PIN PN : D65566-1		SN : 08JL779X85	TORQUE LINK PIN										TCI	
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)		PN Description		Specific Reference										
	WO number	WP	Scheduled check												
Inspection				Inspection 2				Overhaul				Life Limit			
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32NG26 28/10/2008	TORQUE LINK PIN PN : D65566-1		SN : 08JL779X39		TORQUE LINK PIN				TCI						
LLT												OVH			
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG27 28/10/2008	TORQUE LINK APEX PIN ASSY PN : D64433-1		SN : 07JL249X84		NGL AFT TORQUE LINKS CNTR				TCI						
LLT												OVH			
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG28 28/10/2008	PLUG AXE ASSY PN : D64125		SN : 07COU45617X0021		PLUG AXE ASSY				TCI						
LLT												OVH			
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG30 28/10/2008	PIN AT NLG PN : D60739		SN : WIA021-304		PIN				TCI						
LLT												OVH			
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32NG31 28/10/2008	UNLOCKING SPRING PN : D23611000		SN : WIA031-565	LOCK-SPRING		TCI										
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32NG32 28/10/2008	DOWNLOCK SPRING PN : D59800-0004		SN : 05810	DOWNLOCK SPRING		TCI										
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32NG33 28/10/2008	UNLOCKING SPRING PN : D23611000		SN : WIA031-566	LOCK-SPRING		TCI										
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32NG34 28/10/2008	DOWNLOCK SPRING PN : D59800-0004		SN : 05566	DOWNLOCK SPRING		TCI										
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance					
	Installed Equipment (PN/SN)				PN Description		Specific Reference								
	WO number	WP	Scheduled check												
Inspection				Inspection 2				Overhaul				Life Limit			
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32NG35 28/10/2008	NGL SHOCK ABSORBER PN : D23592030				SN : B443		ABSORBER-SHOCK				TCI				
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG36 28/10/2008	STRUT COMPLETE PN : D23596000-3				SN : WIA283		FORESTAY ASSY				TCI				
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG37 28/10/2008	RACK PN : D66677				SN : 07CZ79057X07		RACK				TCI				
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32RMA00 28/10/2008	MAIN LAINDING GEAR RH PN : 201581002				SN : MDL3642		MAIN LANDING GEAR RH				TCI				
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	

Date 30.12.2022

NON-INCIDENT STATEMENT

To Whom It May Concern:

The undersigned hereby certifies that from available records, the material described below:

Description: Nose Landing Gear

Part number: D23589520-12

Serial number: B1160 Quantity: 1

TSN: UNKNOWN FH; CSN: 27647 TSR: UNKNOWN FH CSR: UNKNOWN FC

from the following aircraft:

Type: AIRBUS A320

Serial #: 01895 Registration: EX-32011

and shipped against the following purchase/sale order:

was originally manufactured by Messier

or one of their authorized licensees (if applicable); and it was not obtained from any Government or Military Source, and throughout its service in Aero Nomad it has not been involved in any accident or incident as defined in ICAO Annex 13, Chapter 1.

Andrei Savinov

Technical Director
Aero Nomad Airlines





LANDING GEAR LIFE LIMITED ITEMS (NLG)

Date as of	25-Oct-2022
AC total HRS	63660 18
AC total Cyc	28496
Date of last OVHL	30-Aug-17
ALS Part 1	Rev. B
ALS Part 4	Rev. 8

A320-232
EX-32011
MSN 1895



DESCRIPTION	PN	SN	At Installation			CSN	CSO	LLC	Remaining Cycles	Next OVH Calendar	Days Since Inst.
			Date	ACFT CYC	Part CYC						
NOSE LANDING GEAR	NA20202-202	B1160	23-Dec-14								
NOSE LANDING GEAR STRUCTURE	D23589520-12	B1160	23-Dec-14								
Hinge pin (Barrel/Structure)	D65588-1	L2553	23-Dec-14	20310	19461	27647	8186	60000	32353	30-Aug-22	2863
Hinge pin (Barrel/Structure)	D65588-1	L2552	23-Dec-14	20310	19461	27647	8186	60000	32353	30-Aug-22	2863
Barrel	D67583	14MDC11911	23-Dec-14	20310	0	8186	8186	60000	51814	30-Aug-22	2863
Pin (Retraction actuator/ Lock link/Barrel)	D65589-1	L1487	23-Dec-14	20310	19461	27647	8186	60000	32353	30-Aug-22	2863
Pin (Forestay/Barrel)	D65599-1	L00902	23-Dec-14	20310	19461	27647	8186	60000	32353	30-Aug-22	2863
SHOCK ABSORBER	D23592020/-2	L2175	23-Dec-14								
Cylinder	D65519	L1099	23-Dec-14	20310	15821	24007	8186	60000	35993	30-Aug-22	2863
Sliding tube (3) Sliding tube with a separate wheel axle.	D66679	L0848	23-Dec-14	20310	19461	27647	8186	60000	32353	30-Aug-22	2863
Wheel axle	D66680	L952	23-Dec-14	20310	19461	27647	8186	60000	32353	30-Aug-22	2863
FORESTAY ASSEMBLY	D23599000-3	B1161	23-Dec-14								
Forestay lower link	D67802	14MDC00432	23-Dec-14	20310	0	8186	8186	60000	51814	30-Aug-22	2863
Pin (Lower link/U-joint)	D65773	VI9914-26	23-Dec-14	20310	19461	27647	8186	60000	32353	30-Aug-22	2863
Lock link lower pin	D65631-1	B1161	23-Dec-14	20310	19461	27647	8186	60000	32353	30-Aug-22	2863
Cross pin	D65628-1	B1161	23-Dec-14	20310	19461	27647	8186	60000	32353	30-Aug-22	2863
Universal joint	D59638	14CZ1384331	23-Dec-14	20310	0	8186	8186	60000	51814	30-Aug-22	2863
Pin (Upper panel/U-joint)	D65887	DCR39961-43	23-Dec-14	20310	12763	20949	8186	60000	39051	30-Aug-22	2863
Forestay upper panel	D65237	SAD065-2661	23-Dec-14	20310	19170	27356	8186	60000	32644	30-Aug-22	2863
Pin (Upper panel/ Structure)	D65616	12BA7925955	23-Dec-14	20310	0	8186	8186	60000	51814	30-Aug-22	2863
Pin (Upper panel/ Structure)	D65616	DCR34826-22	23-Dec-14	20310	0	8186	8186	60000	51814	30-Aug-22	2863
LOCK LINK	D23612000	N/A	23-Dec-14								
Lock link lower link	D59635	DCR27131-2601	23-Dec-14	20310	20526	28712	8186	60000	31288	30-Aug-22	2863
Lock link center pin	D60054	C237694-2	23-Dec-14	20310	19461	27647	8186	60000	32353	30-Aug-22	2863
Lock link upper link	D59636-1	SRA/006-1895	23-Dec-14	20310	19461	27647	8186	60000	32353	30-Aug-22	2863

THIS IS TO CERTIFY THAT ABOVE INFORMATION IS TRUE AND CORRECT AS PER THE RECORDS

*** END ***

SECTION 3. REPAIR APPROVALS

SECTION 4. MODIFICATION STATUS

Component Overhaul Services, Corp.	CAVU AEROSPACE	WORK ORDER: 12614	PURCHASE ORDER: 00023395
FAA Repair Station: #8COR883B EASA 145.6404	PART NUMBER: D23757500-7 SERIAL NUMBER: B3137		
MODIFICATION	DESCRIPTION	STATUS	
SB 580-32-3153	LANDING GEAR NLG BARREL. INTRODUCTION OF A REINFORCED BARREL.	PCW	
SB 580-32-3155	NOSE LANDING GEAR LEG. IMPROVEMENT OF THE DETECTION SYSTEM.	PCW	
SB 580-32-3156, REV. 1	LANDING GEAR -NOSE LANDING GEAR LEG. IMPROVEMENT OF THE UPPER SUPPORT FOR THE SHOCK ABSORBER CYLINDER INSTALLATION.	PCW	
SB 580-32-3157	NOSE LANDING GEAR LEG. IMPROVEMENT OF THE EXTENDED LANDING GEAR DETECTION DEVICE	PCW	
SB 580-32-3158, REV. 1	LANDING GEAR NLG LEG. INTRODUCTION OF A NEW STOP RING.	PCW	
SB 580-32-3160	RECOMMENDED OVERHAUL OF THE ACTUATING CYLINDER. NO CHANGE IN P/N'S.	CW	
SB 580-32-3164, REV. 1	LANDING GEAR-NLG 2M UPPER ELECTRICAL HARNESS. REPLACEMENT OF THE EYELET BY A SLEEVE.	PCW	
SB 580-32-3166	LANDING GEAR NLG SHOCK ABSORBER INSPECTION OF CYLINDER LUGS.	CW	
SB 580-32-3168, REV. 1	LANDING GEAR NLG LEG- INSPECTION OF THE BARREL.	CW	
SB 580-32-3171	INTRODUCTION OF A STANDARD TARGET AND A LEVER WITH THE PREVIOUS ADJUSTMENT INSTEAD OF THE CUSTOMIZED "FOOL PROOFED" TARGET AND LEVER WITH A SPECIFIC ADJUSTMENT	PCW	
ALS PART 1	SAFE LIFE AIRWORTHINESS LIMITATION ITEMS (GEAR)	CW	
EASA AD 2021-0236	Landing Gear – Landing Gear Sliding Tube – Inspection / Replacement	S/N NOT AFFECTED	
MODIFICATION STATUS LEGEND CW: COMPLIED WITH PCW: PREVIOUSLY COMPLIED WITH NC: NOT COMPLIED N/A: NOT APPLICABLE		QUALITY ENGINEERING APPROVAL  December 10, 2024	

SECTION 5. INSPECTION STATUS REPORT

A/C: A320

CUSTOMER: CAVU AEROSPACE

P/NAME: NLG BARREL

W/ORDER: 12614

P/NUMB.: D67583

OHM CHAP.: 32-27-02

INSPECTION STATUS REPORT

S/NUMB.: 08B0300X8772

SUB W/O: 12614-58



LIFE LIMIT: 60, 000

C.S.N.: 16758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	5.9859	6.1653	5.9860	5.9860	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.9859	D52315-204EB	0.0030	5.989
	5.9875	NOTE 2					5.9875			
B			COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
C	6.5350	6.6259	6.5370	6.5370	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	NOTE 1	D59835EB	0.0110	6.548
	REF.									
C1	4.7250	4.6732	4.7260	4.7260	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	4.8190	D59835EB		
	REF.						REF.			
C1A	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.0940	D59835EB		0.093
			ONLY							
D			COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
1A	2.3640	2.4448	2.3620	2.3620	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.1653	D65580EB	0.0020	2.364
1B	REF.		2.3620	2.3620	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.1661	D65580EB	0.0020	2.364
2A	36.4800	36.4082	36.4700	36.4700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	36.6900	D65580EB		
	REF.						± .003			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65580EB		0.110
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65580EB		0.110
3A	0.4744		0.4755	0.4755	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4744	D52315-201EB	0.0015	0.477
3B	0.4754		0.4755	0.4755	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4754	D52315-201EB	0.0015	0.477
3C	0.4744		0.4755	0.4755	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4744	D52315-201EB	0.0015	0.477
3D	0.4754		0.4755	0.4755	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4754	D52315-201EB	0.0015	0.477
3A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
3B1			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
3C1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
3D1			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
4A	3.7250		3.7180	3.7180	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
4B	REF.		3.7180	3.7180	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
5A	1.0650	1.1377	1.0650	1.0650	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.9448	D65578EB	0.0020	1.067
	REF.						0.9457			
5B	0.2559	0.3500	0.2600	0.2600	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.2559	B8-0065H9-130EB	0.0010	0.261
	0.2638	NOTE 4					0.2573			
6A	0.3140	0.4094	0.3160	0.3160	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3140	B8-0080H9-190EB	0.0015	0.318
6B	0.3163		0.3160	0.3160	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3163	B8-0080H9-190EB	0.0015	0.318
7A	0.3140	0.4094	0.3160	0.3160	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3140	B8-0080H9-190EB	0.0015	0.318
7B	0.3163		0.3160	0.3160	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3163	B8-0080H9-190EB	0.0015	0.318
7C	0.3140	0.4094	0.3160	0.3160	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3140	B8-0080H9-190EB	0.0015	0.318
7D	0.3163		0.3160	0.3160	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3163	B8-0080H9-190EB	0.0015	0.318
7E	0.3140	0.4094	0.3160	0.3160	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3140	B8-0080H9-190EB	0.0015	0.318

A/C: A320

CUSTOMER: CAVU AEROSPACE

P/NAME: NLG BARREL

W/ORDER: 12614

P/NUMB.: D67583

OHM CHAP.: 32-27-02

INSPECTION STATUS REPORT

S/NUMB.: 08B0300X8772

SUB W/O: 12614-58



LIFE LIMIT: 60, 000

C.S.N.: 16758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
7F	0.3163		0.3160	0.3160	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3163	B8-0080H9-190EB	0.0015	0.318
7G	0.3140	0.4094	0.3160	0.3160	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3140	B8-0080H9-190EB	0.0015	0.318
	0.3163						0.3163			
8A	1.7720	1.8464	1.7740	1.7740	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.5748	D65996EB	0.0020	1.776
8B	REF.		1.7740	1.7740	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.5757	D65655EB	0.0020	1.776
9A	7.5560	7.4802	7.5590	7.5590	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	8.1467	D65996EB		
	REF.						8.1496	D65655EB		
9A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1940	D65996EB		0.196
9A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3940	D65655EB		0.392
10A	5.0430	5.1180	5.0400	5.0400	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	4.7440	D65996EB		
	REF.						4.7465	D65656EB		
10A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1980	D65996EB		0.197
10A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.0980	D65656EB		0.099
11A	0.8640	0.7874	0.8650	0.8650	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
11B	1.6500	1.5748	1.6540	1.6540	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
12A	1.1040	1.1771	1.1040	1.1040	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.9055	D65576EB	0.0020	1.106
12B	REF.		1.1040	1.1040	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.9063	D65576EB	0.0020	1.106
13A	3.3050	3.2248	3.3080	3.3080	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.5018	D65576EB		
	REF.						3.5039			
13A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.0960	D65576EB		0.097
13A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.0960	D65576EB		0.097
13B	3.3050	3.2248	3.3080	3.3080	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.5018	D65576EB		
	REF.						3.5039			
13B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.0960	D65576EB		0.097
13B2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.0960	D65576EB		0.097
14A	0.4750	0.5393	0.4750	0.4750	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3125	D65579EB	0.0015	0.477
14B	REF.		0.4750	0.4750	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3134	D65579EB	0.0015	0.477
15A	3.3060	3.2228	3.3070	3.3070	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.4590	D65579EB		
	REF.						3.4645			
15A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65579EB		0.076
15A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65579EB		0.076
16A	0.4750	0.5393	0.4750	0.4750	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3125	D65573EB	0.0015	0.477
16B	REF.		0.4750	0.4750	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3134	D65573EB	0.0015	0.477
17A	3.3060	3.2228	3.3070	3.3070	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.6220	D65573EB		
	REF.						3.6199			
17A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65573EB		0.158
17A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65573EB		0.158

A/C: A320

CUSTOMER: CAVU AEROSPACE

P/NAME: NLG BARREL

W/ORDER: 12614

P/NUMB.: D67583

OHM CHAP.: 32-27-02

INSPECTION STATUS REPORT

S/NUMB.: 08B0300X8772

SUB W/O: 12614-58



LIFE LIMIT: 60, 000

C.S.N.: 16758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
18A	0.5910	0.6574	0.5910	0.5910	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4724	D65517EB	0.0015	0.593
18B	REF.		0.5910	0.5910	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4731	D65517EB	0.0015	0.593
19A	3.4250	3.3346	3.4230	3.4230	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.5377	D65517EB		
	REF.						3.5433			
19A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65517EB		0.057
19A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65517EB		0.057
20A	1.9900	2.0708	1.9890	1.9890	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.7716	D59943EB	0.0020	1.991
20B	REF.		1.9890	1.9890	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.7726	D59943EB	0.0020	1.991
21A			3.4750	3.4750	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.7030			
							3.7086			
21A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
21A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
22A	1.5800	1.6574	1.5760	1.5760	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.3779	D59943EB		
	REF.						1.3818			
22A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D59943EB		0.099
22A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D59943EB		0.099
23A	0.4620	0.5787	0.4620	0.4620	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3125	D61189EB	0.0020	0.464
	REF.						0.3134	D65574EB	0.0015	
24A	1.6500	1.6063	1.6480	1.6480	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.3036	D65574EB		
	REF.						3.3070	D61189EB		
24A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65574EB		0.828
24A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65575EB		0.828
25A	3.4645	3.5905	3.4650	3.4650	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.4659	D52315-203EB	0.0020	3.467
25B	3.4659		3.4650	3.4650	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.4645	D52315-203EB	0.0020	3.467
26A	9.0500		9.0520	9.0520	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
26A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D52315-203EB		4.530
26A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D52315-203EB		4.530
27A	1.3400	1.4133	1.3400	1.3400	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.0669	D65577EB	0.0015	1.342
27B	REF.		1.3400	1.3400	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.0677	D65577EB	0.0015	1.342
27C	REF.		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.1653	D65577EB	0.0015	0.002
	ONLY		ONLY				2.1671			
28A	6.0866		6.0980	6.0980	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	6.5787			
	6.1180						6.5826			
28A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1530	D65577EB		0.240
28A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1610	D65577EB		0.240
29A	0.1968	0.2913	0.1970	0.1970	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1968	B8-0050H9-195EB	0.0015	0.199
29B	0.1980		0.1970	0.1970	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1980	B8-0050H9-195EB	0.0015	0.199
29C	0.1968	0.2913	0.1970	0.1970	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1968	B8-0050H9-195EB	0.0015	0.199

A/C: A320

CUSTOMER: CAVU AEROSPACE

P/NAME: NLG BARREL

W/ORDER: 12614

P/NUMB.: D67583

OHM CHAP.: 32-27-02

INSPECTION STATUS REPORT

S/NUMB.: 08B0300X8772

SUB W/O: 12614-58



LIFE LIMIT: 60, 000

C.S.N.: 16758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
29D	0.1980		0.1970	0.1970	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1980	B8-0050H9-195EB	0.0015	0.199
30A	0.3818	0.4763	0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3818	B8-0097H9-195EB	0.0015	0.385
30B	0.3833		0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3833	B8-0097H9-195EB	0.0015	0.385
30C	0.3818	0.4763	0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3818	B8-0097H9-195EB	0.0015	0.385
30D	0.3833		0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3833	B8-0097H9-195EB	0.0015	0.385
30E	0.3818	0.4763	0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3818	B8-0097H9-195EB	0.0015	0.385
30F	0.3833		0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3833	B8-0097H9-195EB	0.0015	0.385
30G	0.3818	0.4763	0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3818	B8-0097H9-195EB	0.0015	0.385
30H	0.3833		0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3833	B8-0097H9-195EB	0.0015	0.385
30I	0.3818	0.4763	0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3818	B8-0097H9-195EB	0.0015	0.385
	0.3833						0.3833			
31A	0.3818	0.4763	0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3818	B8-0097H9-195EB	0.0015	0.385
31B	0.3833		0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3833	B8-0097H9-195EB	0.0015	0.385
31C	0.3818	0.4763	0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3818	B8-0097H9-195EB	0.0015	0.385
31D	0.3833		0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3833	B8-0097H9-195EB	0.0015	0.385
31E	0.3818	0.4763	0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3818	B8-0097H9-195EB	0.0015	0.385
31F	0.3833		0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3833	B8-0097H9-195EB	0.0015	0.385
31G	0.3818	0.4763	0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3818	B8-0097H9-195EB	0.0015	0.385
31H	0.3833		0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3833	B8-0097H9-195EB	0.0015	0.385
31I	0.3818	0.4763	0.3830	0.3830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3818	B8-0097H9-195EB	0.0015	0.385
	0.3833									
32A	1.6535	1.8031	1.6540	1.6540	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.6535	D52315-227EB	0.0015	1.656
32B	1.6545		1.6540	1.6540	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.6545	D52315-227EB	0.0015	1.656
33A	INTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
34A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
35A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REGJECTION		ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE

P/NAME: NLG BARREL

W/ORDER: 12614

P/NUMB.: D67583

OHM CHAP.: 32-27-02

S/NUMB.: 08B0300X8772

SUB W/O: 12614-58

LIF. LIMIT: 60,000

C.S.N.: 16758

INSPECTION STATUS REPORT



NOTE 1: ► DESIGN DIMENSION OF I/D IS: 6.2643 / 6.2659 TO A DEPTH OF: 1.940.
**** DO NOT MACHINE THE SMALLER I/D LAND END OF BUSHING ****
DESING DIMENSION FOR THIS SMALLER I/D IS 6.2086 / 6.2323

NOTE 2: ► MACHINE TO A MINIMUM OF: 6.104 / 6.105 FOR A REPAIR BUSHING
***** P/N: D52315-204EB (REWORK LIMITS IS: 6.165 *****

NOTE 3: ► BORE MUST BE MACHINED TO A MINIMUM OF 1.7716 FOR A REPAIR BUSHING P/N: D52315-227EB

NOTE 4: ► BORE MUST BE MACHINED TO A MINIMUM OF 0.3351 FOR A REPAIR SLEEVE

A/C: A320

CUSTOMER: CAVU AEROSPACE

P/NAME: NLG BARREL

W/ORDER: 12614

P/NUMB.: D67583

OHM CHAP.: 32-27-02

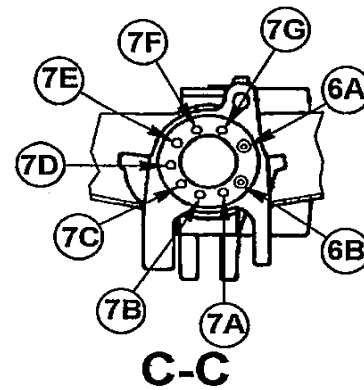
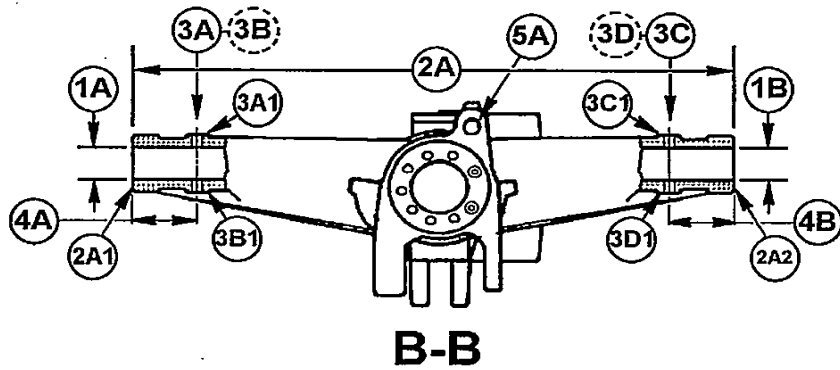
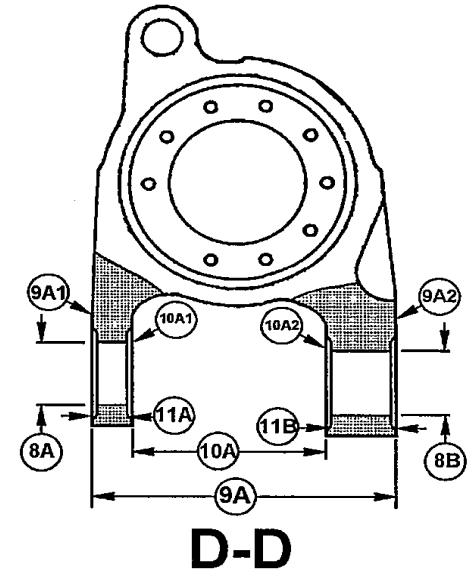
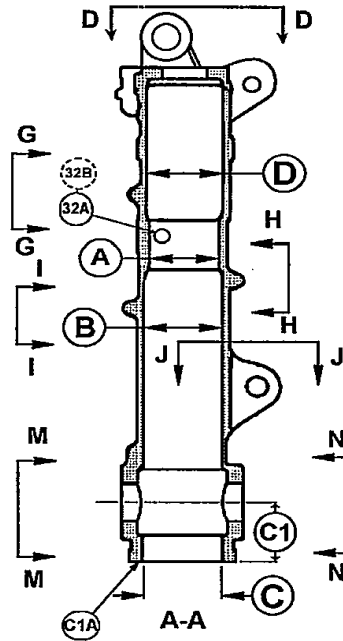
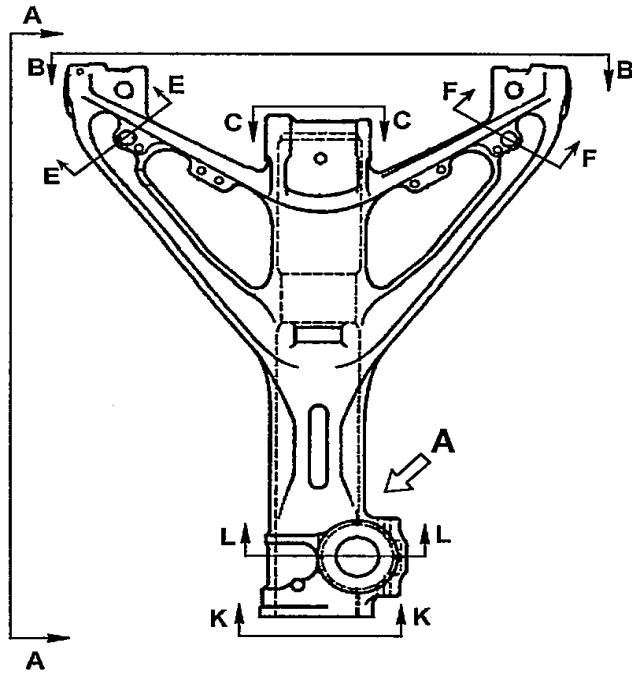
S/NUMB.: 08B0300X8772

SUB W/O: 12614-58

LIF. LIMIT: 60,000

C.S.N.: 16758

INSPECTION STATUS REPORT



A/C: A320

P/NAME: NLG BARREL

P/NUMB.: D67583

S/NUMB.: 08B0300X8772

LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE

W/ORDER: 12614

OHM CHAP.: 32-27-02

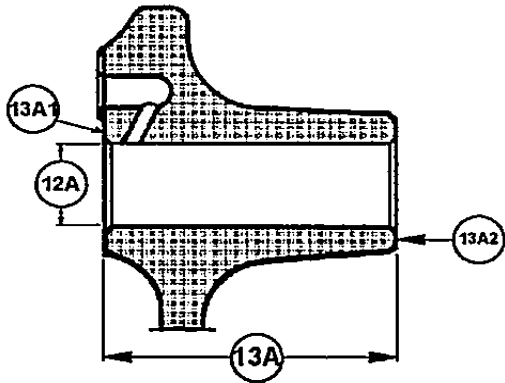
SUB W/O: 12614-58

C.S.N.: 16758

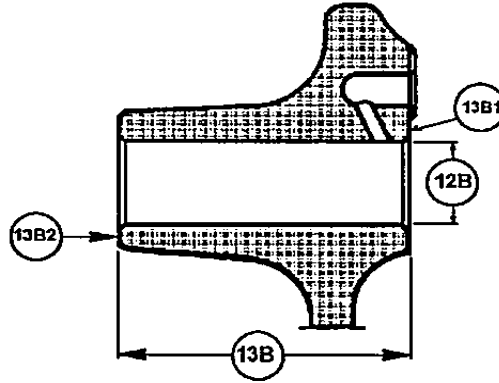
INSPECTION STATUS REPORT



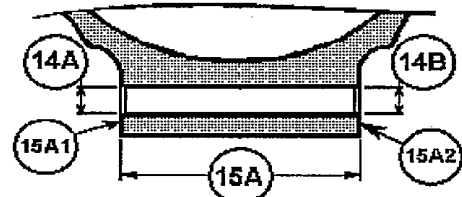
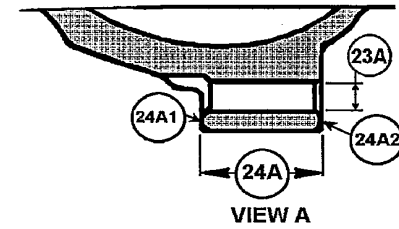
Component Overhaul Services



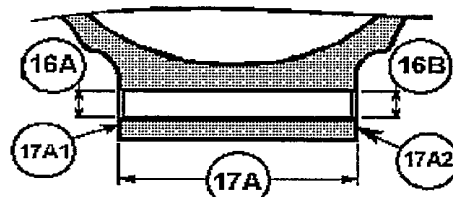
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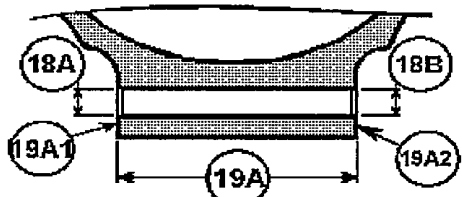
F-F



G-G



H-H



H-H

A/C: A320

P/NAME: NLG BARREL

P/NUMB.: D67583

S/NUMB.: 08B0300X8772

LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE

W/ORDER: 12614

OHM CHAP.: 32-27-02

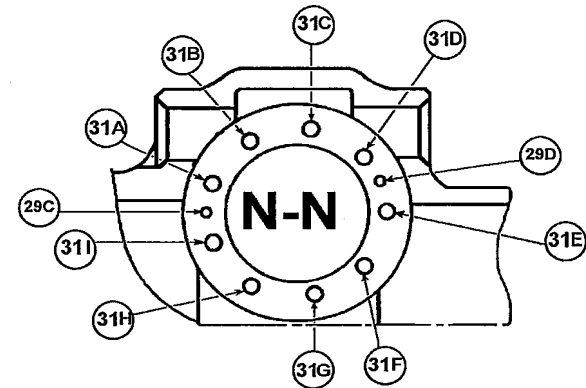
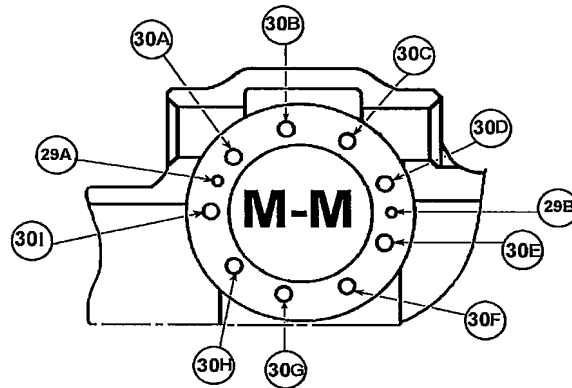
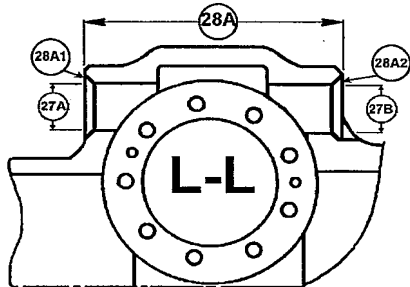
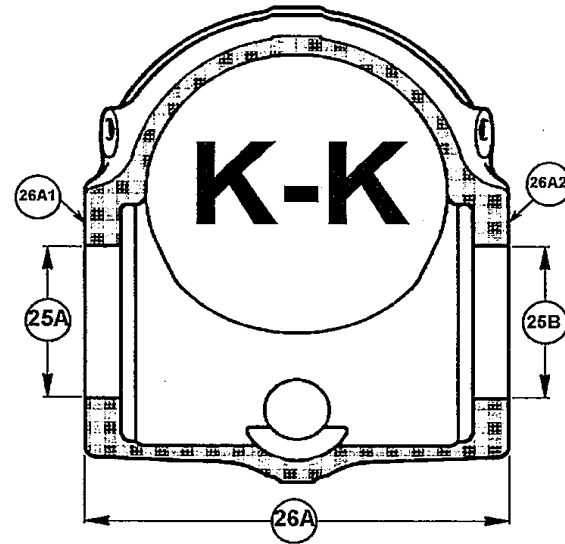
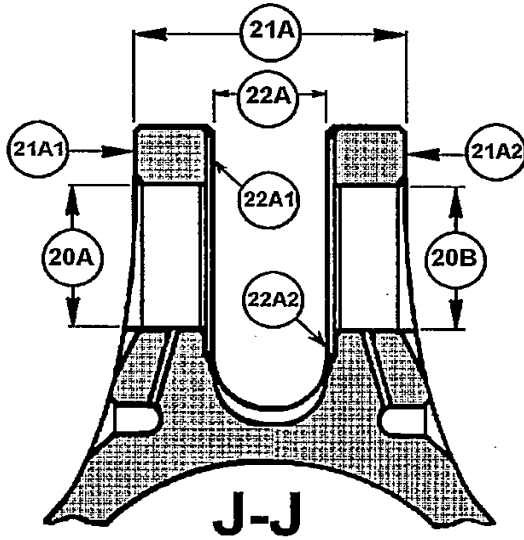
SUB W/O: 12614-58

C.S.N.: 16758

INSPECTION STATUS REPORT



Component Overhaul Services



A/C: A320 / A321

CUSTOMER: CAVU AEROSPACE

P/NAME: LOWER LINK

W/O: 12614

P/N: D67802

CMM: 32-37-02

S/N: 08B0060X11

SUB W/O: 12614-88

LIFE LIMIT: 60,000

C.S.N.: 16758

INSPECTION STATUS REPORT



LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A		1.9331	1.8510	1.8510	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.6535	D60091EB	0.002	1.853
1B			1.8510	1.8510	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.6545	D60091EB	0.002	1.853
2A		2.4063	2.3250	2.3250	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.1250	D60091EB		
							2.1277			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.0900	D60091EB		0.100
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.0900	D60091EB		0.100
2B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
2B2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
3A	2.9520	NOTE 1	2.9530	2.9530	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		NOTE 1		
	2.9530	NOTE 2								
4A	1.1024		1.1090	1.1090	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		NOTE 1		
	1.1102									
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
5A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
6A	INTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
7A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTION		ONLY							

A/C: A320 / A321
P/NAME: LOWER LINK
P/N: D67802
S/N: 08B0060X11
LIFE LIMIT: 60000

CUSTOMER: CAVU AEROSPACE
W/O: 12614
CMM: 32-37-02
SUB W/O: 12614-88
C.S.N.: 16758

INSPECTION STATUS REPORT



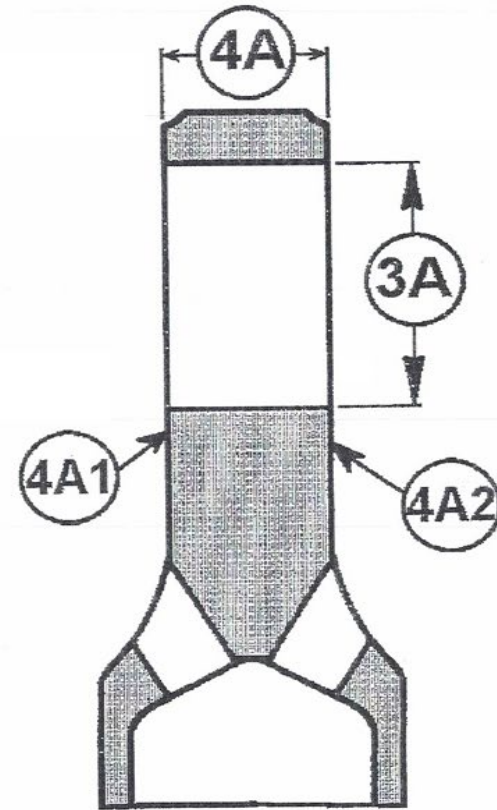
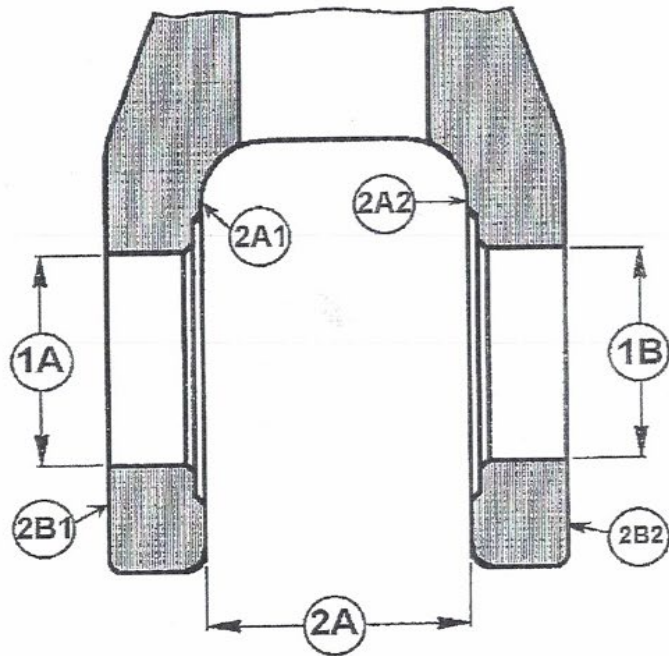
- NOTE 1:**
- ▶ BEARING P/N: GA66033 IS STANDARD DIMENSION: 2.9525 / 2.9539
 - ▶ BEARING P/N: GA66033-1R60 IS O/S: 0.024 (MACHINE TO 2.9764 / 2.9776)
 - ▶ BEARING P/N: GA66033-1R80 IS O/S: 0.032 (MACHINE TO 2.9842 / 2.9854)
 - ▶ BEARING P/N: GA66033-1R100 IS O/S: 0.040 (MACHINE TO 2.9921 / 2.9933)

NOTE 2: MINIMUM WALL THICKNESS ALLOWED IS: 0.3779

A/C: A320 / A321
P/NAME: LOWER LINK
P/N: D67802
S/N: 08B0060X11
LIFE LIMIT: 60000

CUSTOMER: CAVU AEROSPACE
W/O: 12614
CMM: 32-37-02
SUB W/O: 12614-88
C.S.N.: 16758

INSPECTION STATUS REPORT



A/C: A320 / A321

CUSTOMER: CAVU AEROSPACE

P/NAME: JOINT UNIVERSAL

W/ORDER: 12614

P/NUMB.: D59368

OHM CHAP.: 32-27-23

INSPECTION STATUS REPORT

S/NUMB.: WIA025-244

SUB W/O: 12614-60

Component Overhaul
Services

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	1.8500	1.9330	1.8510	1.8510	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.6535	D60088EB	0.0020	1.853
1B	REF.		1.8510	1.8510	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.6545	D60088EB	0.0020	1.853
2A		3.6973	3.7780	3.7780	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.9728 3.9749	D60088EB		
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D60088EB		0.097
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D60088EB		0.097
2B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
2B2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
3A	0.5512	0.6456	0.5519	0.5519	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5512	B1-0140H7-075EB	0.0020	0.554
3B	0.5519		0.5519	0.5519	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5519	B1-0140H7-075EB	0.0020	0.554
4A		1.9330	1.8510	1.8510	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.6533 1.6545	D60363EB	0.0020	1.853
5A		1.7692	1.8500	1.8500	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.0442 2.0460	D60363EB		
5A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D60363EB		0.097
5A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D60363EB		0.097
6A	EXTERNAL SURFACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7A	OTHER REJECTIONS		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				

A/C: A320 / A321

CUSTOMER: CAVU AEROSPACE

P/NAME: JOINT UNIVERSAL

W/ORDER: 12614

P/NUMB.: D59368

OHM CHAP.: 32-27-23

INSPECTION STATUS REPORT



Component Overhaul
Services

S/NUMB.: WIA025-244

SUB W/O: 12614-60

LIF. LIMIT: 60,000

C.S.N.: 16758

NOTE 1: IF MACHINING BORES -6A-, -6B-, -6C- OR -6D- IS REQUIRED THEN REFER TO THE CMM 32-37-23 FIGURE 602-2 (SHEETS 1 & 2) PAGE 620 FOR APPROPRIATE REPAIR

A/C: A320 / A321

P/NAME: JOINT UNIVERSAL

P/NUMB.: D59368

S/NUMB.: WIA025-244

LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE

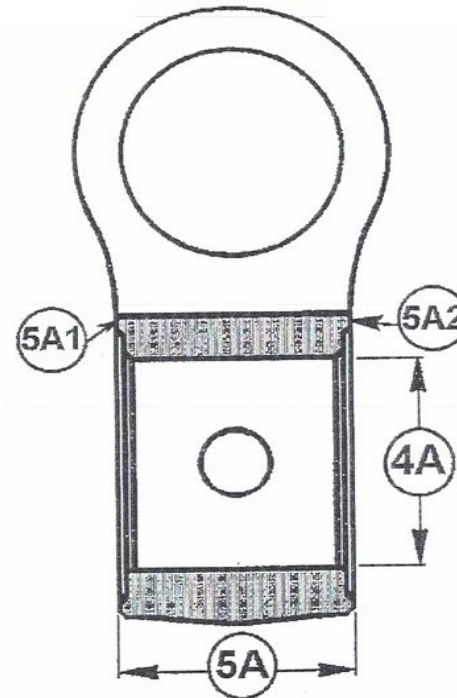
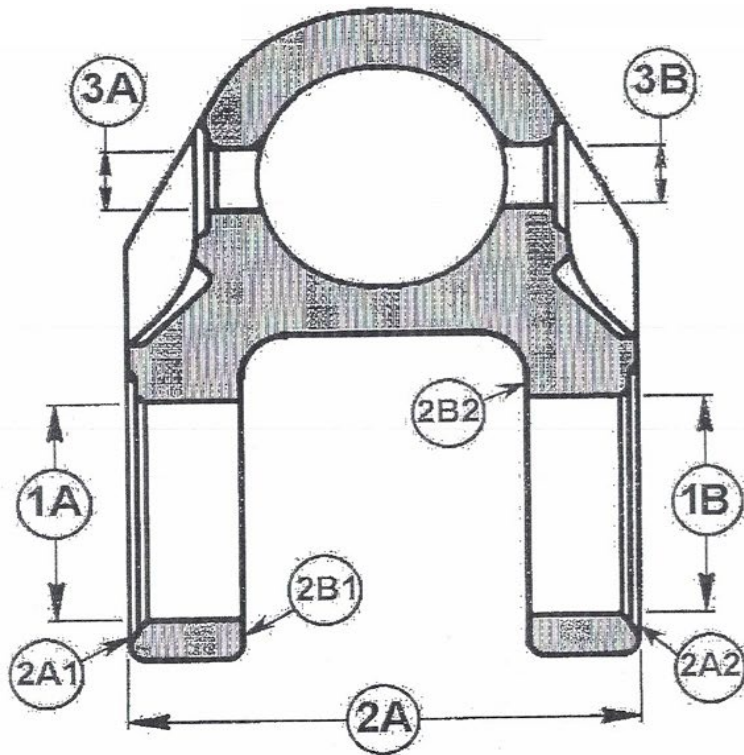
W/ORDER: 12614

OHM CHAP.: 32-27-23

SUB W/O: 12614-60

C.S.N.: 16758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE

P/NAME: UPPER PANEL

W/ORDER: 12614

P/NUMB.: D65237

OHM CHAP.: 32-27-23

INSPECTION STATUS REPORT

S/NUMB.: WIA033-291

SUB W/O: 12614-59



LIFE LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A		2.2480	2.1660	2.1660	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.9685	D65617ER	0.0015	2.168
1B			2.1550	2.1550	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.9694	D65617ER	0.0015	2.157
2A		36.4564	36.5350	36.5350	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	36.7681	D65617ER		
		NOTE 5					36.7716			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65617ER		0.117
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65617ER		0.117
3A		1.9330	1.8530	1.8530	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.6535	D65618EB	0.0015	1.855
3B			1.8530	1.8530	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.6545	D65618EB	0.0015	1.855
4A	6.2935		6.2950	6.2950	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	6.2975									
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
5A		4.3362	4.2520	4.2520	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	4.0551	D65618EB		
							4.0572			
5A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65618EB		0.098
5A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65618EB		0.098
6A	0.3937	0.4889	0.3950	0.3950	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3941	D52315-259EB	0.0015	0.397
6B	0.3942		0.3950	0.3950	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3941	D52315-259EB	0.0020	0.397
6C	0.3937	0.4889	0.3950	0.3950	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3941	D52315-259EB	0.0015	0.397
6D	0.3942		0.3950	0.3950	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3941	D52315-259EB	0.0020	0.397
7A	3.1500	3.0236	3.1500	3.1500	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.1417	D52315-259EB		
7B			3.1500	3.1500	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.1574	D52315-259EB		
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D52315-259EB		-0.004
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D52315-259EB		-0.004
7B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D52315-259EB		-0.004
7B2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D52315-259EB		-0.004
8A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
9A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE

P/NAME: UPPER PANEL

W/ORDER: 12614

P/NUMB.: D65237

OHM CHAP.: 32-27-23

INSPECTION STATUS REPORT

S/NUMB.: WIA033-291

SUB W/O: 12614-59



LIF. LIMIT: 60,000

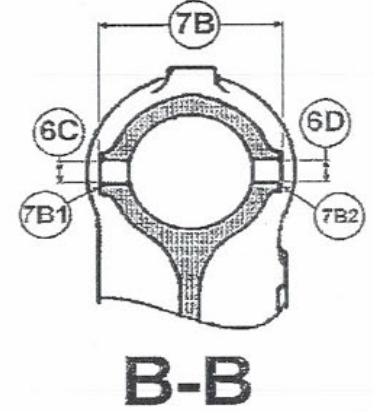
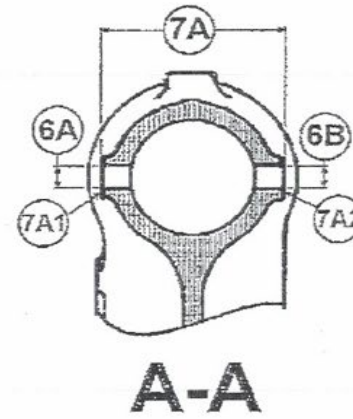
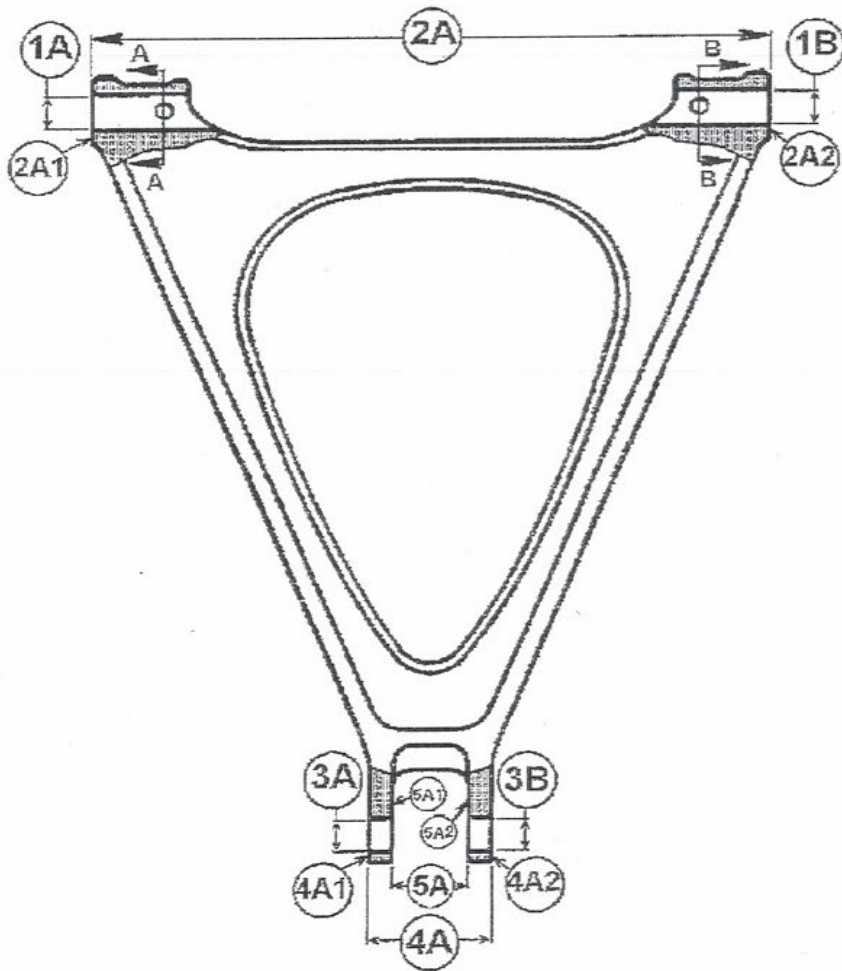
C.S.N.: 16758

NOTE 1: IF MACHINING BORES -6A-, -6B-, -6C- OR -6D- IS REQUIRED THEN REFER TO THE CMM 32-37-23 FIGURE 602-2 (SHEETS 1 & 2) PAGE 620 FOR APPROPRIATE REPAIR

A/C: A320
P/NAME: UPPER PANEL
P/NUMB.: D65237
S/NUMB.: WIA033-291
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE
W/ORDER: 12614
OHM CHAP.: 32-27-23
SUB W/O: 12614-59
C.S.N.: 16758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE

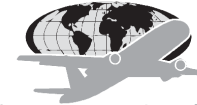
P/NAME: LINK, LOWER

W/ORDER: 12614

P/NUMB.: D59635

OHM CHAP.: 32-27-24

INSPECTION STATUS REPORT



Component Overhaul Services

S/NUMB.: WIA034-318

SUB W/O: 12614-97

LIFE LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A		1.1819	1.1819	1.1819	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		00-200-1525	0.0010	
B		1.1819	1.1819	1.1819	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		00-200-1525	0.0010	
3A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	OTHER REJECTIONS		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				

A/C: A320

CUSTOMER: CAVU AEROSPACE

P/NAME: LINK, LOWER

W/ORDER: 12614

P/NUMB.: D59635

OHM CHAP.: 32-27-24

S/NUMB.: WIA034-318

SUB W/O: 12614-97

LIF. LIMIT: 60,000

C.S.N.: 16758

INSPECTION STATUS REPORT

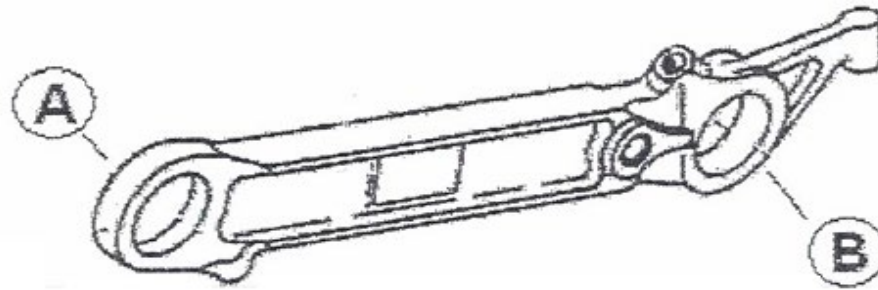


NOTE 1: REQUIRES BEARING P/N: 00-200-1525

A/C: A320
P/NAME: LINK, LOWER
P/NUMB.: D59635
S/NUMB.: WIA034-318
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE
W/ORDER: 12614
OHM CHAP.: 32-27-24
SUB W/O: 12614-97
C.S.N.: 16758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE

P/NAME: UPPER LINK

W/ORDER: 12614

INSPECTION STATUS REPORT

P/NUMB.: D59636- 1

OHM CHAP.: 32-27-24



S/NUMB.: WIA36-338

SUB W/O: 12614-98

LIF. LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A		1.8474	1.7720	1.7720	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.5748	D65637	0.0015	1.774
1B			1.7720	1.7720	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.5757	D65638	0.0015	1.774
2A		3.2249	3.3050	3.3050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.5011			
		MINIMUM					3.5025			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65637		0.098
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65638		0.098
3A		0.8937	0.8290	0.8290	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7086	D65634	0.0015	0.831
3B		NOTE 1	0.8290	0.8290	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7094	D65635	0.0015	0.831
4A	1.8032	SUBMITTL	1.8120	1.8120	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.8030	D65634		
	1.8189	NOTE 1						D65635		
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65634		-0.005
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65635		-0.005
5A		0.6574	0.5920	0.5920	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4738	D65636	0.0015	0.594
5B			0.5920	0.5920	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4731	D65636	0.0015	0.594
6A		3.3385	3.4200	3.4200	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.5354	D65636		
							3.5511			
6A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65636		0.058
6A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65636		0.058
7A		0.7755	0.7100	0.7100	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5511	D65633	0.0015	0.712
7B			0.7100	0.7100	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5518	D65633	0.0015	0.712
8A		0.5000	0.4340	0.4340	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3307	D65632	0.0015	0.436
8B			0.4340	0.4340	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3385	D65632	0.0015	0.436
9A		1.3031	1.2240	1.2240	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.1023			
							1.1036			
9A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65633		0.061
9A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65633		0.061
10A	PER REF. ONLY		1.8860	1.8860	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.8850	D65633		
10A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65632		
10A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D65632		
11A	0.314 REF		0.3140	0.3140	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		D60053	0.001	0.315
12A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
13A	OTHER REJECTION		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				

A/C: A320
P/NAME: UPPER LINK
P/NUMB.: D59636- 1
S/NUMB.: WIA36-338
LIF. LIMIT: 60000

CUSTOMER: CAVU AEROSPACE
W/ORDER: 12614
OHM CHAP.: 32-27-24
SUB W/O: 12614-98
C.S.N.: 16,758

INSPECTION STATUS REPORT



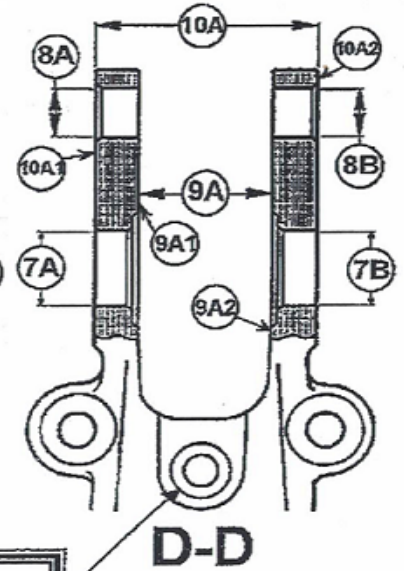
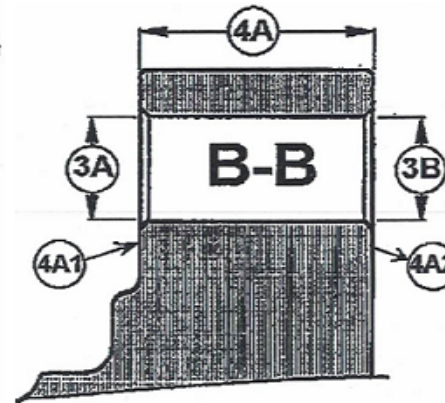
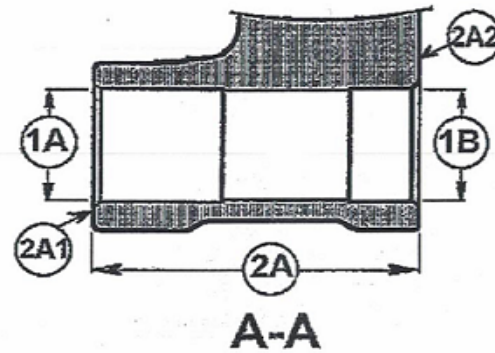
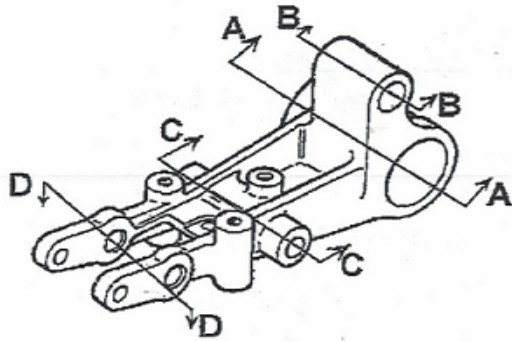
NOTE 1: ► IF BORE IS MACHINED BEYOND A DIMANEION OF: 0.892 AND A FACE TO FACE BELOW 1.784, SCRAP UNIT REFERENCE SUBMITTAL RESPONSE

NOTE 2: ► PERFORM AN EDDY CURRENT INSPECTION ON BORES AND ADJECENT AREAS (SEE DRAWING) WITH THE FREQUENCY 5MHz (NO CRACKS ARE ACCEPTABLE)

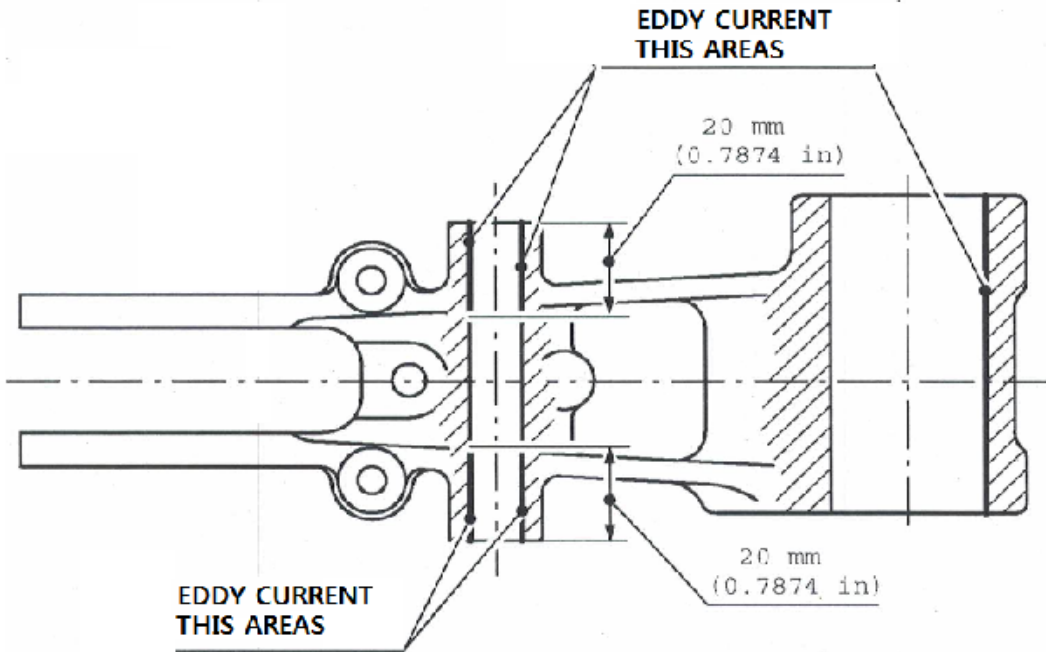
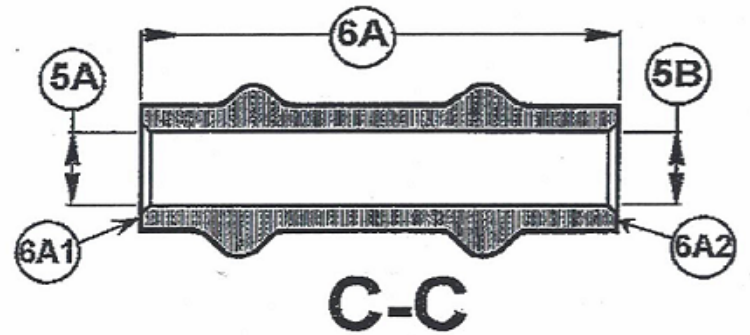
A/C: A320
P/NAME: UPPER LINK
P/NUMB.: D59636- 1
S/NUMB.: WIA36-338
LIF. LIMIT: 60000

CUSTOMER: CAVU AEROSPACE
W/ORDER: 12614
OHM CHAP.: 32-27-24
SUB W/O: 12614-98
C.S.N.: 16,758

INSPECTION STATUS REPORT



Requires bushing
P/N: D60053



A/C: A320 / A321

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: CYLINDER

W/ORDER: 12614

P/NUMB.: D65519

OHM CHAP.: 32-27-21

S/NUMB.: 07TOF3816P9X003

SUB W/O: 12614-29

LIFE LIMIT: 60, 000

C.S.N.: 16758

INSPECTION STATUS REPORT



LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	1.5728		1.5730	1.5730	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	1.5738									
B	THREADS		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
C	THREADS		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
D	3.7454	3.7355 NOTE 1	3.7463	3.7463	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	3.7468									
E	THREADS		COND. COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
F			COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
1A	0.3050		0.2980	0.2980	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
1B	REF.		0.2980	0.2980	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
2A	0.5472	0.5315	0.5472	0.5472	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
2B	0.5486	NOTE 1	0.5472	0.5472	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
3A	2.7680		2.7690	2.7690	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.									
4A	INTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
5A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
6A	OTHER REJECTIONS		COND. ONLY							

A/C: A320 / A321

CUSTOMER: CAVU AEROSPACE

P/NAME: CYLINDER

W/ORDER: 12614

P/NUMB.: D65519

OHM CHAP.: 32-27-21

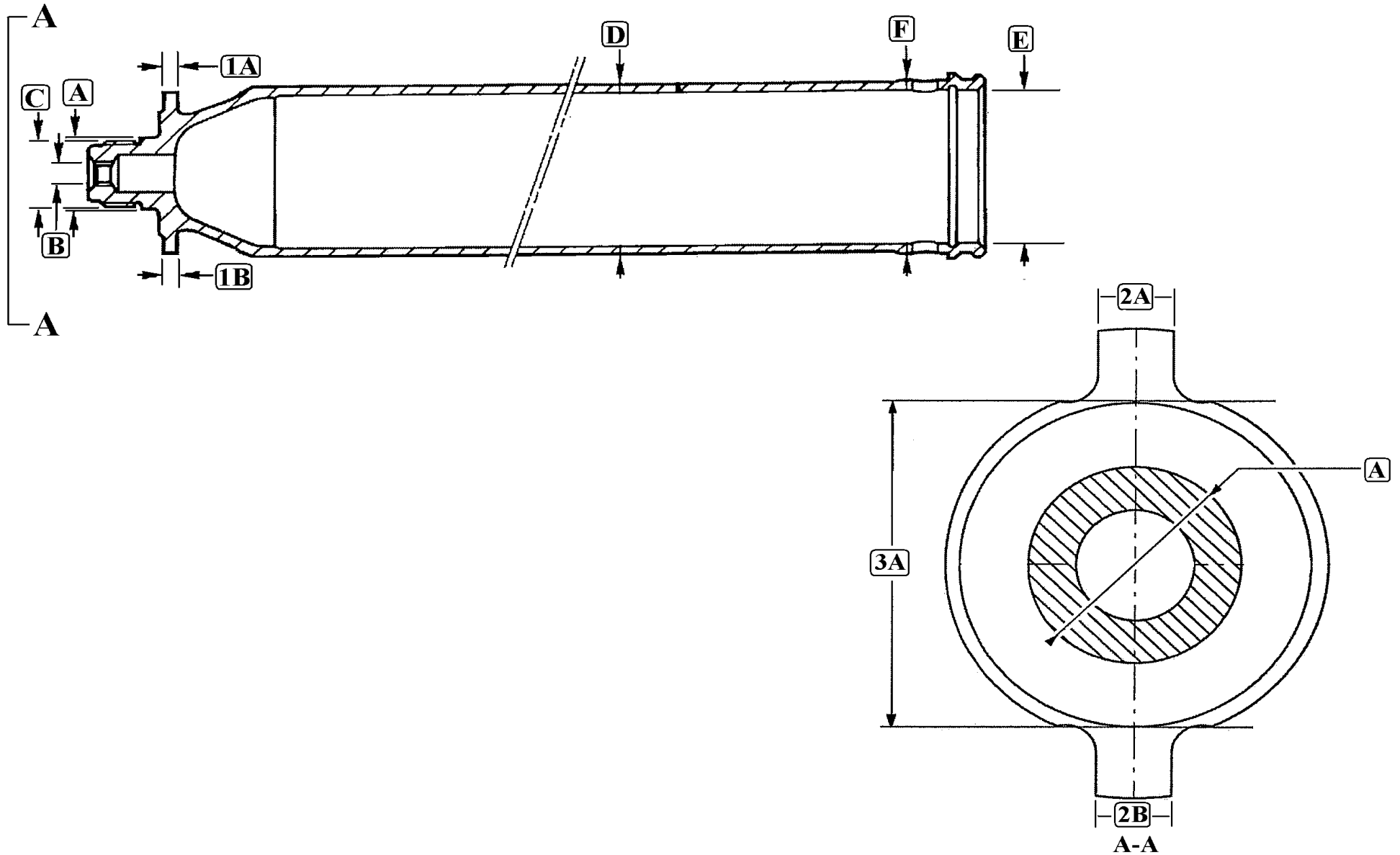
S/NUMB.: 07TOF3816P9X003

SUB W/O: 12614-29

LIF. LIMIT: 60, 000

C.S.N.: 16758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: NLG SLIDING TUBE

W/ORDER: 12614

INSPECTION STATUS REPORT

P/NUMB.: D66679

OHM CHAP.: 32-27-21

Component Overhaul
Services

S/NUMB.: 08B0713X3712

SUB W/O: 12614-34

LIFE LIMIT: 60, 000

C.S.N.: 16758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	4.6822	4.6723	4.6825	4.6825	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	4.6836	CHROME								
B	4.3681	4.3737	4.3691	4.3691	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	4.3702									
C	FITTING		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	END BORE		ONLY							
D	NON WORK		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	I/D		ONLY							
1A	2.9496	3.1982	2.9539	2.9539	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.9496	D52315-0465EB	0.0024	2.9563
1B	2.9539	O/S BUS	2.9539	2.9539	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.9514	D52315-0465EB	0.0024	2.9563
2A	9.4500	NOTE 2	9.4510	9.4510	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	9.4550 REF.									
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				-4.726
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				-4.726
3A	1.770 REF.	1.8474	1.7740	1.7740	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	1.5748	D65569EB	0.0016	1.7756
3B	1.772 REF.	MAXIMUM	1.7740	1.7740	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	1.5757	D65569EB	0.0016	1.7756
4A	4.0900	4.0123	4.0940	4.0940	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	4.2885	D65569EB		
	REF.	MAXIMUM					4.2899			
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		D65569EB		0.097
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		D65569EB		0.097
5A	0.3937	0.4890	0.3940	0.3940	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.3925	D52315-268EB	0.0013	0.3953
5B	0.3945		0.3940	0.3940	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.3938	D52315-268EB	0.0013	0.3953
6A	0.7500	0.8158	0.7490	0.7490	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.5905	D65570EB	0.0010	0.7500
6B	REF.	MAXIMUM	0.7490	0.7490	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.5915	D65570EB	0.0010	0.7500
6C	0.7500	0.8158	0.7490	0.7490	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.5905	D65570EB	0.0010	0.7500
6D	REF.	MAXIMUM	0.7490	0.7490	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.5915	D65570EB	0.0010	0.7500
7A	0.4050	SUBMIT	0.4150	0.4150	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7B	REF.		0.4150	0.4150	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7C	0.4050	SUBMIT	0.4150	0.4150	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7D	REF.		0.4150	0.4150	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7B2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: NLG SLIDING TUBE

W/ORDER: 12614

INSPECTION STATUS REPORT

P/NUMB.: D66679

OHM CHAP.: 32-27-21



S/NUMB.: 08B0713X3712

SUB W/O: 12614-34

LIFE LIMIT: 60, 000

C.S.N.: 16758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
7C1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7C2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7D1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7D2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
8A	1.0591	1.2135	1.0650	1.0650	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	1.0590	D52315-0470EB	0.0015	1.0665
	1.0663	NOTE 3					1.0604			
9A	4.3276	4.5525	4.3320	4.3320	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	4.3276	D52315-0466EB	0.0030	4.3350
	4.3328						4.3310			
10A	0.9645	SUBMIT NOTE 1	0.9650	0.9650	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	1.0039									
11A	INTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
12A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
13A	OTHER REJECTIONS		COND. ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: NLG SLIDING TUBE

W/ORDER: 12614

P/NUMB.: D66679

OHM CHAP.: 32-27-21

S/NUMB.: 08B0713X3712

SUB W/O: 12614-34

LIF. LIMIT: 60,000

C.S.N.: 16758

INSPECTION STATUS REPORT



Component Overhaul
Services

NOTE 1: IF REPAIR IS REQUIRED DO NOT EXCEED A MAXIMUM LENGTH OF: 0.7086.
SEE FIGURE 602-8 (SHEET 1 / 2 FOR LENGTH AND RADIUS CONSIDERATION).

NOTE 2: IF CORROSION IS FOUND ON FACE -2A- MUST BE REMOVED AS FOLLOW: FACES -2A1- & -2A2- MAY BE MACHINED A MAXIMUM OF 0.010.
PER FACE TO REMOVE THE CORROSION. REMAINING CORROSION MAY BE LIGHTLY POLISH TO REOVE ON EITHER FACE AS LONG AS
THE DESIGN DIMENSION FROM NOTE 2 IS MAINTAINED, IF UNDER DIMENSION FROM NOTE 2, APPLY FOR A CONCESSION.

AFTER MACHINING THE FACES, MESURE THE LENGTH BETWEEN THE SLIDING ROD MAIN AXIS AND THE EDGE OF THE SPACER TO READ A
DIMENSION OF: 6.5674 / 6.5826. IF IT MESURES BELOW THIS DIMENSION A CONCESSION MUST BE SUBMITTED TO MESSIER DOWTY.

NOTE 3: IF BORE REQUIRES MACHINING IT MUST BE MACHINED TO A MINIMUM OF: 1.1811 FOR A REPAIR SLEEVE

NOTE 4: IF THE AXLE BORES -1A- AND -1B- NEED TO BE MACHINED FOR A REPAIR BUSHINGS
MACHINE BOTH BORES IN LINE TO A MINIMUM OF 3.1496 FOR A REPAIR BUSHING (THE RWK. LIMITS IS: 3.1982)
MAINTAIN A CHAMFER OR REDIUS OF: 0.0787 / 0.1181 ON BOTH SIDES OF THE ATTACH LUG

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: NLG SLIDING TUBE

W/ORDER: 12614

P/NUMB.: D66679

OHM CHAP.: 32-27-21

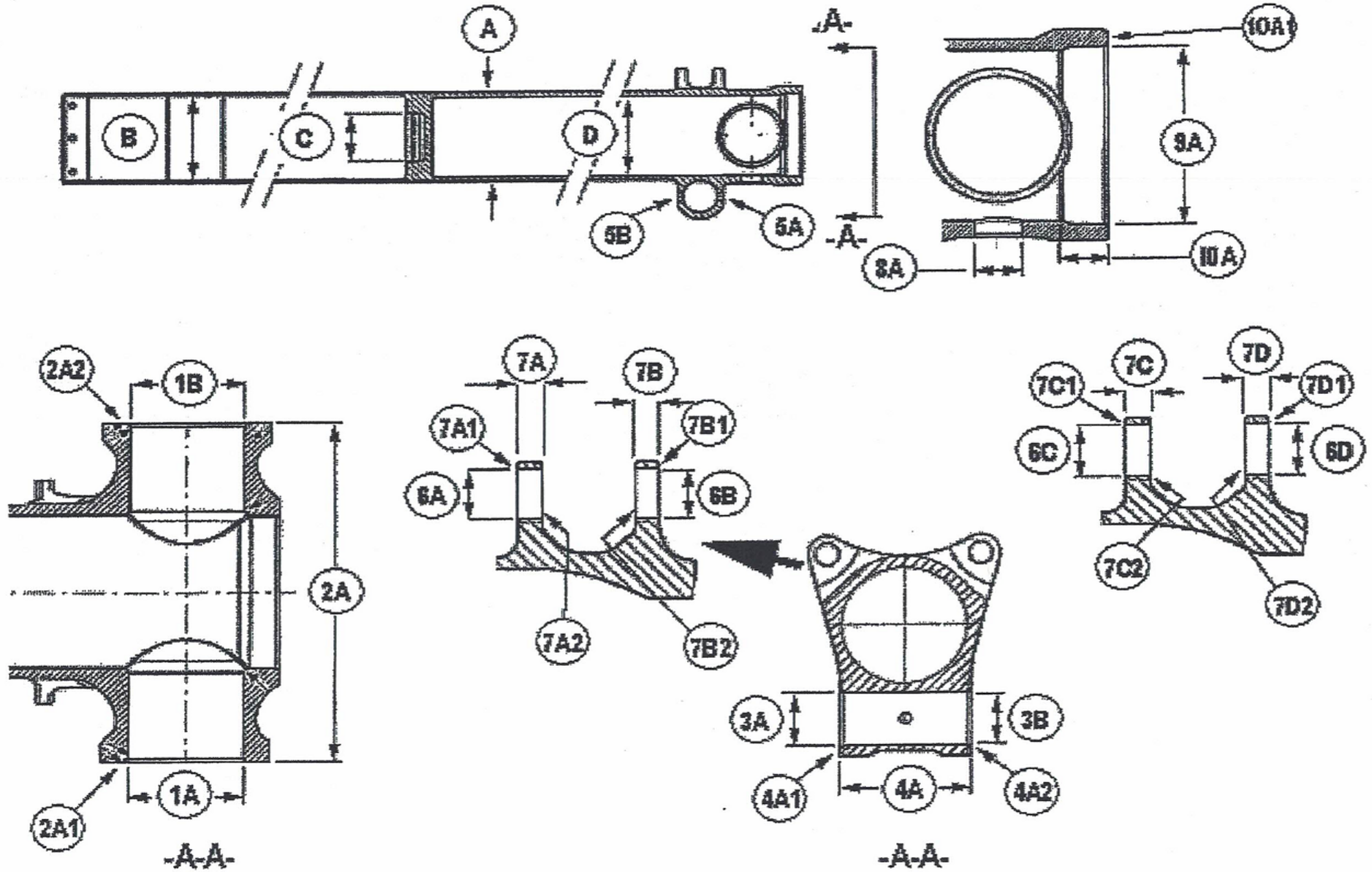
S/NUMB.: 08B0713X3712

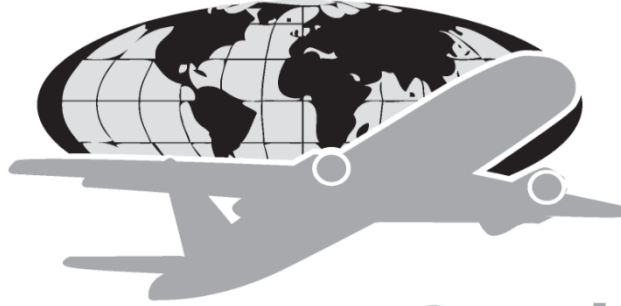
SUB W/O: 12614-34

LIF. LIMIT: 60, 000

C.S.N.: 16758

INSPECTION STATUS REPORT





Component Overhaul Services

FAA Repair Station # 8COR883B

EASA 145.6404

Landing Gear Overhaul Report

W/O: 12615

Description: A320 RH MLG

LEG & DRESSING

P/N: 201582002-040

S/N: MDG4846

DEC. 2024

SECTION 1. AIRWORTHINESS APPROVAL TAGS

1. Approving Civil Aviation Authority/Country: FAA/UNITED STATES	2. AUTHORIZED RELEASE CERTIFICATE FAA FORM 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: 219354
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4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74th St. Suite B, Medley, FL 33166 TEL: (305) 406-3885 FAX: (786) 400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615
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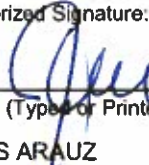
6. Item	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:
1	LEG AND DRESSINGS, MLG, RH	201582002-040	1	MDG4846	OVERHAULED

12. Remarks:

OVERHAULED I.A.W. SAFRAN CMM 32-12-24 REV. 43 DATED MAR 15/2024. REFER TO COMPONENT OVERHAUL SERVICES WORK STATEMENT REPORT NUMBER 12615 FOR INSPECTION FINDINGS, SERIALIZED COMPONENT LIST, TRACEABILITY, C.S.N., S.B. AND A.D. MODIFICATION LIST.
 CSO: -0-

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): DOUGLAS ARAUZ	14e. Date (dd/mmm/yyyy): 19/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.


Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1. It is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statement in Blocks 13a and 14a do not constitute installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	3. Form Tracking Number: COS7771
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AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-147
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	FWD. PINTLE PIN STUD BOLT	201540616	1	PA79539X444	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.

CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 19/Dec/2024


User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7772
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4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-148
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	FWD. PINTLE PIN STUD BOLT	201540616	1	PA79539X438	Overhauled

12. Remarks:
 Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.

CSN: 16,758
 CSO: 0
 Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
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13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 19/Dec/2024
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User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.


Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	3. Form Tracking Number: COS7773
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AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-2
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6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11 Status/Work:
1	MAIN FITTING	201540301	1	08MDM2048	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.
CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
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13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 19/Dec/2024
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User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.


Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7774	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12615-145		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	FWD. PINTLE PIN	201173600	1	08MDG6980	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			<h2 style="text-align: center;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="text-align: center;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>		3. Form Tracking Number: COS7775	
4. Organization Name and Address:  <p style="text-align: center;">COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B</p>							5. Work Order/Contract/Invoice Number: 12615-84	
6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11 Status/Work:			
1	AFT PINTLE PIN	D3215303000800	1	F07647-35	Overhauled			
12. Remarks: Overhauled I.A.W Airbus CMM 32-11-07 Rev. 33 Rev. date 01/Jul/2023. <div style="text-align: right;"> CSN: 16,758 CSO: 0 Previous Operator: AVIANCA </div> <p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>								
13a. Certifies the items identified above were manufactured in conformity to:			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12					
<input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:		
						8COR883B		
13d Name (Typed or Printed):		13e Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):		
				Douglas Arauz		19/Dec/2024		
User/Installer Responsibilities								
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>								

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7776
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4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-139
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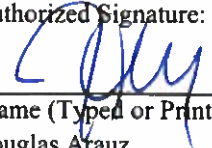
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	LOCKSTAY CARDAN	201058306	1	08COU46164X5769	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.

CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 19/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.


Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	3. Form Tracking Number: COS7777
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AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG


4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-17
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6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11 Status/Work:
1	UPPER TORQUE LINK	201540303	1	08BEL0061	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.
CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 19/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: <h1 style="margin:0;">COS7778</h1>		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-14		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	UPPER TORQUE LINK PIN	201160603	1	08MDG8380	Overhauled
12. Remarks: <div style="display: flex; justify-content: space-between;"> <div> <p>Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.</p> </div> <div style="text-align: right;"> <p>CSN: 16,758 CSO: 0 Previous Operator: AVIANCA</p> </div> </div> <p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature:	14c. Approval/Certificate No.:	
				8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed):	14e. Date (dd/mmm/yyyy):	
			Douglas Arauz	19/Dec/2024	
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					


1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7779	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12615-105		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	TORQUE LINK APEX PIN	201383606	1	08MDG5306	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:
						8COR883B
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):
				Douglas Arauz		19/Dec/2024
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: COS7780	
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B							5. Work Order/Contract/Invoice Number: 12615-113	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:			
1	TORQUE LINK APEX PIN NUT	201587612	1	08MSL5902352	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to:			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12					
<input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:		
						8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):		
				Douglas Arauz		19/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.								
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.								
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.								

1. Approving Civil Aviation Authority/Country: FAA/United States	2	3. Form Tracking Number: COS7781
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AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

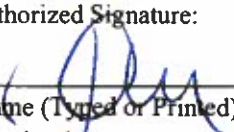
4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-22
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6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11 Status/Work:
1	LOWER TORQUE LINK	201540302	1	08BEL0129	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.
CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 19/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States		2			<h1>AUTHORIZED RELEASE CERTIFICATE</h1> <p>FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>		3. Form Tracking Number: COS7782	
4. Organization Name and Address:  <p>COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B</p>							5. Work Order/Contract/Invoice Number: 12615-20	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:			
1	LOWER TORQUE LINK PIN	201160602	1	08MDG6742	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to:			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12					
<input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:		
						8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):		
				Douglas Arauz		19/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.								
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.								
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.								

1. Approving Civil Aviation Authority/Country: FAA/United States		2			AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: COS7783	
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B		5. Work Order/Contract/Invoice Number: 12615						
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:			
1	SHOCK ABSORBER	201371281	1	08B5178X9131	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.								

1. Approving Civil Aviation Authority/Country: FAA/United States		2			AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: COS7784	
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B							5. Work Order/Contract/Invoice Number: 12615-1	
6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11 Status/Work:			
1	SLIDING TUBE	201371304	1	08B5178X9131	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.								
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.								
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1. Approving Civil Aviation Authority/Country: FAA/United States		2			AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: COS7785	
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B							5. Work Order/Contract/Invoice Number: 12615-132	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:			
1	UPPER DIAPHRAGM TUBE	201371615	1	08B0314X2	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to:					14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12			
<input type="checkbox"/> Approved design data and are in a condition for safe operation.					Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
<input type="checkbox"/> Non-approved design data specified in Block 12.								
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:		
						8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):		
				Douglas Arauz		19/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.								
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.								
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.								

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7786	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12615		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	SIDE STAY COMPLETE	201166014-025	1	AP1270	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-12-26 Rev. 30 Rev. date 17/Sep/2021. CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Atauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7787	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12615-58		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	LOWER CARDAN	201163307	1	08B0893X1767	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7788	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12615-62		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	LOWER CARDAN NUT	201163945	1	2763/08	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7789
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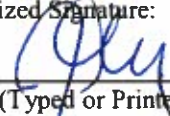
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-59
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	SIDE STAY LOWER CARDAN PIN	201163619	1	07AP0268	Overhauled

12. Remarks:
 Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.
 CSN: 16,758
 CSO: 0
 Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Araúz	14e. Date (dd/mmm/yyyy): 19/Dec/2024

User/Installer Responsibilities


It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: <h1 style="margin:0;">COS7790</h1>		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-57		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	LOWER SIDE STAY	201163301	1	07AP0223	Overhauled
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. <div style="float: right; text-align: right;"> CSN: 16,758 CSO: 0 Previous Operator: AVIANCA </div>					
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 19/Dec/2024	
User/Installer Responsibilities					
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.					
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.					
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.					

1. Approving Civil Aviation Authority/Country: FAA/United States	2 <h1 style="margin: 0;">AUTHORIZED RELEASE CERTIFICATE</h1> <p style="margin: 0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: COS7791
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4. Organization Name and Address:  Component Overhaul Services COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-176
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	SIDE STAY PIVOT PIN	201163609	1	08AP0124	Overhauled

12. Remarks:

Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.

CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/ Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
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13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 19/Dec/2024
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User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7792	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12615-76		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	UPPER SIDE STAY	201163300	1	AP042106	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7793	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12615-77		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	SIDE STAY UPPER CARDAN PIN	201163618	1	07AP0048	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7794	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12615-78		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	UPPER CARDAN	201163620	1	08B0563X3	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7795	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12615-80		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	UPPER CARDAN NUT	201163944	1	2185/07	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input checked="" type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: COS7796	
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B							5. Work Order/Contract/Invoice Number: 12615-81	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:			
1	UPPER PIN SPRING	201661607	1	08AP0140	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.								
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.								
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.								

1. Approving Civil Aviation Authority/Country: FAA/United States		2			AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: COS7797	
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B							5. Work Order/Contract/Invoice Number: 12615-183	
6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11 Status/Work:			
1	LOWER PIN SPRING	201661606	1	08AP0118	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.								
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.								
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.								

1. Approving Civil Aviation Authority/Country: FAA/United States		2			AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: COS7798	
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B							5. Work Order/Contract/Invoice Number: 12615-179	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:			
1	BANANA LINK	201661306	1	08AP0114	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to:			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12					
<input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:		
						8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):		
				Douglas Arauz		19/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.								
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.								
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.								

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7799	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12615-180		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	BANANA LINK	201661306	1	08AP0121	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: <h2 style="margin:0;">COS7800</h2>		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-185		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	CUFF PIN	201661612	1	08AP0172	Overhauled
12. Remarks: <div style="display: flex; justify-content: space-between;"> <div> <p>Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.</p> </div> <div style="text-align: right;"> <p>CSN: 16,758 CSO: 0 Previous Operator: AVIANCA</p> </div> </div> <p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.		14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 19/Dec/2024	
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: <h1 style="margin:0;">COS7801</h1>		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-186		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	CUFF PIN	201661612	1	08AP0170	Overhauled
12. Remarks: <div style="display: flex; justify-content: space-between;"> <div style="width: 60%;"> <p>Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.</p> </div> <div style="width: 35%;"> <p>CSN: 16,758 CSO: 0 Previous Operator: AVIANCA</p> </div> </div> <p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 19/Dec/2024	
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: <h1 style="margin:0;">COS7802</h1>		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-187		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	CUFF	201661613	1	07AP0218	Overhauled
12. Remarks: <div style="display: flex; justify-content: space-between;"> <div style="width: 60%;"> <p>Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.</p> </div> <div style="width: 35%;"> <p>CSN: 16,758 CSO: 0 Previous Operator: AVIANCA</p> </div> </div> <p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 19/Dec/2024	
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: COS7803
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12615-182	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	PIN LOCK LINK / BANANA LINK	201661605	1	08AP0054	Overhauled
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. <div style="float: right; text-align: right;"> CSN: 16,758 CSO: 0 Previous Operator: AVIANCA </div>					
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 19/Dec/2024	
User/Installer Responsibilities					
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.					
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.					
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.					

1. Approving Civil Aviation Authority/Country: FAA/United States		2			AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: COS7804	
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B							5. Work Order/Contract/Invoice Number: 12615-163	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:			
1	LOCK LINK LOWER PIN	201057627	1	07AP0220	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.								

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: <h1 style="margin:0;">COS7805</h1>		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-181		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	TRIANGULATION LINK	201661307	1	08AP0195	Overhauled
12. Remarks: <div style="display: flex; justify-content: space-between;"> <div> <p>Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.</p> </div> <div style="text-align: right;"> <p>CSN: 16,758 CSO: 0 Previous Operator: AVIANCA</p> </div> </div> <p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/ Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 19/Dec/2024	
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7806	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12615-184		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	TRIANGULATION LINK PIN	201661609	1	08AP0129	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: COS7807		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	LOCK STAY	201058003	1	AP1316	Overhauled
12. Remarks: Overhauled I.A.W Safran CMM 32-12-27 Rev. 26 Rev. date 15/Mar/2024. <div style="text-align: right;"> CSO: 0 Previous Operator: AVIANCA </div> <p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 19/Dec/2024	
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					


1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: <h1 style="margin:0;">COS7808</h1>		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-38		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	LOWER LOCK LINK	201058305	1	08AP0051	Overhauled
12. Remarks: Overhauled I.A.W Safran CMM 32-12-27 Rev. 26 Rev. date 15/Mar/2024. <div style="float: right; text-align: right;"> CSN: 16,758 CSO: 0 Previous Operator: AVIANCA </div>					
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 Douglas Arauz	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14c. Approval/Certificate No.:	
				8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				19/Dec/2024	
User/Installer Responsibilities					
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.					
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.					
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.					

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7809	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG					5. Work Order/Contract/Invoice Number: 12615-9	
4. Organization Name and Address:  Component Overhaul Services		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	LOCK LINK CENTER PIN	201058608	1	07AP0224	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-12-27 Rev. 26 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 19/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: <h1 style="margin:0;">COS7810</h1>		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-8		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	UPPER LOCK LINK	201058310	1	08AP0006	Overhauled
12. Remarks: Overhauled I.A.W Safran CMM 32-12-27 Rev. 26 Rev. date 15/Mar/2024. <div style="float: right; text-align: right;"> CSN: 16,758 CSO: 0 Previous Operator: AVIANCA </div> <p style="margin-top: 20px;">Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 19/Dec/2024	
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: <h1 style="margin:0;">COS7811</h1>		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-11		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	LOCK LINK UPPER PIN	201058611	1	07AP0073	Overhauled
12. Remarks: Overhauled I.A.W Safran CMM 32-12-27 Rev. 26 Rev. date 15/Mar/2024. <div style="float: right; text-align: right;"> CSN: 16,758 CSO: 0 Previous Operator: AVIANCA . </div> <p style="margin-top: 20px;">Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 19/Dec/2024	
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7812
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4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12615-85
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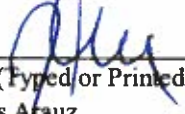
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	AFT PINTLE PIN NUT	SLN41193	1	M337550-12801	Overhauled

12. Remarks:
Overhauled I.A.W Airbus CMM 32-11-04 Rev. 6 Rev. date 01/Jul/2023.

CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/ Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 19/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

SECTION 2. LIFE LIMITED/SERIALIZED PARTS LIST



**Component Overhaul
Services**

**A320 R/H MLG
LIFE LIMITED/SERIALIZED PARTS LIST
FAA REPAIR STATION # 8COR883B / EASA 145.6404**

**Customer: CAVU AEROSPACE
Purchase Order:
Work Order: 12615**

INT. ITEM #	DESCRIPTION	PART NUMBER	S/N	WORK ORDER	LAST OPERATOR	LIFE LIMIT	CSN	COMMENTS
	RH MLG LEG AND DRESSING	201582002-040	MDG4846	12615	AVIANCA			
1	FWD. PINTLE PIN STUD BOLT	201540616	PA79539X444	12615-147	AVIANCA	60,000	16,758	
2	FWD. PINTLE PIN STUD BOLT	201540616	PA79539X438	12615-148	AVIANCA	60,000	16,758	
3	MAIN FITTING	201540301	08MDM2048	12615-2	AVIANCA	60,000	16,758	
4	FWD. PINTLE PIN	201173600	08MDG6980	12615-145	AVIANCA	60,000	16,758	
5	AFT PINTLE PIN	D3215303000800	F07647-35	12615-84	AVIANCA	60,000	16,758	
6	LOCKSTAY CARDAN	201058306	08COU46164X5769	12615-139	AVIANCA	60,000	16,758	
7	LOCKSTAY CARDAN PIN	201056909	R0290196000	N/A	NEW	60,000	0	P33699
8	UPPER TORQUE LINK	201540303	08BEL0061	12615-17	AVIANCA	60,000	16,758	
9	UPPER TORQUE LINK PIN	201160603	08MDG8380	12615-14	AVIANCA	60,000	16,758	
10	TORQUE LINK APEX PIN	201383606	08MDG5306	12615-105	AVIANCA	60,000	16,758	
11	TORQUE LINK APEX PIN NUT	201587612	08MSL5902352	12615-113	AVIANCA	60,000	16,758	
12	LOWER TORQUE LINK	201540302	08BEL0129	12615-22	AVIANCA	60,000	16,758	
13	LOWER TORQUE LINK PIN	201160602	08MDG6742	12615-20	AVIANCA	60,000	16,758	
	SHOCK ABSORBER	201371281	08B5178X9131	12615	AVIANCA			
14	SLIDING TUBE	201371304	08B5178X9131	12615-1	AVIANCA	60,000	16,758	
15	UPPER DIAPHRAGM TUBE	201371615	08B0314X2	12615-132	AVIANCA	60,000	16,758	
	SIDE STAY COMPLETE	201166014-025	AP1270	12615	AVIANCA			
16	LOWER CARDAN	201163307	08B0893X1767	12615-58	AVIANCA	60,000	16,758	
17	LOWER CARDAN NUT	201163945	2763/08	12615-62	AVIANCA	60,000	16,758	
18	SIDE STAY LOWER CARDAN PIN	201163619	07AP0268	12615-59	AVIANCA	60,000	16,758	
19	LOWER SIDE STAY	201163301	07AP0223	12615-57	AVIANCA	60,000	16,758	
20	SIDE STAY PIVOT PIN	201163609	08AP0124	12615-176	AVIANCA	60,000	16,758	
21	UPPER SIDE STAY	201163300	AP042106	12615-76	AVIANCA	60,000	16,758	
22	SIDE STAY UPPER CARDAN PIN	201163618	07AP0048	12615-77	AVIANCA	60,000	16,758	
23	UPPER CARDAN	201163620	08B0563X3	12615-78	AVIANCA	60,000	16,758	
24	UPPER CARDAN NUT	201163944	2185/07	12615-80	AVIANCA	60,000	16,758	
25	UPPER PIN SPRING	201661607	08AP0140	12615-81	AVIANCA	60,000	16,758	
26	LOWER PIN SPRING	201661606	08AP0118	12615-183	AVIANCA	60,000	16,758	
	BANANA LINK	201661306	08AP0114	12615-179	AVIANCA	60,000	16,758	
27	BANANA LINK	201661306	08AP0121	12615-180	AVIANCA	60,000	16,758	
28	CUFF PIN	201661612	08AP0172	12615-185	AVIANCA	60,000	16,758	
29	CUFF PIN	201661612	08AP0170	12615-186	AVIANCA	60,000	16,758	
30	CUFF	201661613	07AP0218	12615-187	AVIANCA	60,000	16,758	
31	PIN LOCK LINK / BANANA LINK	201661605	08AP0054	12615-182	AVIANCA	60,000	16,758	
32	LOCK LINK LOWER PIN	201057627	07AP0220	12615-163	AVIANCA	60,000	16,758	
33	TRIANGULATION LINK	201661307	08AP0195	12615-181	AVIANCA	60,000	16,758	
34	TRIANGULATION LINK PIN	201661609	08AP0129	12615-184	AVIANCA	60,000	16,758	
	LOCK STAY	201058003	AP1316	12615	AVIANCA			



**Component Overhaul
Services**

**A320 R/H MLG
LIFE LIMITED/SERIALIZED PARTS LIST
FAA REPAIR STATION # 8COR883B / EASA 145.6404**

**Customer: CAVU AEROSPACE
Purchase Order:
Work Order: 12615**

INT. ITEM #	DESCRIPTION	PART NUMBER	S/N	WORK ORDER	LAST OPERATOR	LIFE LIMIT	CSN	COMMENTS
35	LOWER LOCK LINK	201058305	08AP0051	12615-38	AVIANCA	60,000	16,758	
36	LOCK LINK CENTER PIN	201058608	07AP0224	12615-9	AVIANCA	60,000	16,758	
37	UPPER LOCK LINK	201058310	08AP0006	12615-8	AVIANCA	60,000	16,758	
38	LOCK LINK UPPER PIN	201058611	07AP0073	12615-11	AVIANCA	60,000	16,758	
47	AFT PINTLE PIN NUT	SLN41193	M337550-12801	12615-85	AVIANCA	60,000	16,758	

Aissa C.

12/19/24

Statement nº 1049 /2019

Date: São Paulo, June 07th, 2019

Incident/Accident Clearance Statement – Airframe

Aircraft: Airbus A318-100 | MSN 3642 (PR-ONR) | TSN 21764,97 | CSN 16758
Engine: PW6000 | ESN | TSN | CSN
Engine: PW6000 | ESN P318123 | TSN 27722,97 | CSN 20883
APU: 4500001B | SN 2924 | TSN 14569 | CSN 14920
NLG: NA28008-015 | SN B3137 | TSN 21764,97 | CSN 16758
MLG LH: 201581001 | SN MDL3642 | TSN 21764,97 | CSN 16758
MLG RH: 201581002 | SN MDL3642 | TSN 21764,97 | CSN 16758

To whom it may concern:

This letter is to certify that the referred aircraft has been operated by Oceanair Linhas Aéreas S.A. during the period stated below and to the best of my knowledge:

Operation period: from December 3, 2013 thru June 7, 2019.

1. Neither the aircraft, nor any part installed have been, unless its airworthiness status was re-established by an approved maintenance organization in accordance with the instructions of the type certificate holder and/or OEM of the part, and supported by an authorized release certificate:

a. Damaged during a reportable incident or accident as defined by Brazilian Civil Aviation Authority regulation RBAC 121.703 – Service Difficulty Reports, equivalent to FAA Title 14 - Chapter I - Subchapter G - Part 121 - Subpart V - §121.703, or

b. Subjected to severe stress or heat (such as in a major engine failure, accident or fire) or has been submersed in salt water.

2. No part has been installed on the aircraft which was obtained from military source or was previously fitted to a state aircraft.



Alberto Ottavio Spelta
Chief Inspector

Rua Tamoios 579 Jardim Aeroporto – São Paulo – SP – CEP 04630-001
Telefone: (11) 3475 8200

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32NG22 28/10/2008	TORNING TUBE PN : D65520		SN : 07MGP70395X2469	TORNING TUBE						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32NG23 28/10/2008	TORQUE LINK PN : D65234		SN : 08FGA239526X472	TORQUE LINK						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32NG24 28/10/2008	TORQUE LINK PN : D65234		SN : 08FGA234441X385	TORQUE LINK						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32NG25 28/10/2008	TORQUE LINK PIN PN : D65566-1		SN : 08JL779X85	TORQUE LINK PIN						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
		Inspection		Inspection 2		Overhaul		Life Limit								
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32NG26 28/10/2008	TORQUE LINK PIN PN : D65566-1		SN : 08JL779X39	TORQUE LINK PIN												TCI
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32NG27 28/10/2008	TORQUE LINK APEX PIN ASSY PN : D64433-1		SN : 07JL249X84	NGL AFT TORQUE LINKS CNTR												TCI
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32NG28 28/10/2008	PLUG AXE ASSY PN : D64125		SN : 07COU45617X0021	PLUG AXE ASSY												TCI
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32NG30 28/10/2008	PIN AT NLG PN : D60739		SN : WIA021-304	PIN												TCI
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)		PN Description		Specific Reference										
	WO number	WP	Scheduled check												
Inspection				Inspection 2				Overhaul				Life Limit			
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32NG31 28/10/2008	UNLOCKING SPRING PN : D23611000		SN : WIA031-565	LOCK-SPRING											TCI
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG32 28/10/2008	DOWNLOCK SPRING PN : D59800-0004		SN : 05810	DOWNLOCK SPRING											TCI
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG33 28/10/2008	UNLOCKING SPRING PN : D23611000		SN : WIA031-566	LOCK-SPRING											TCI
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32NG34 28/10/2008	DOWNLOCK SPRING PN : D59800-0004		SN : 05566	DOWNLOCK SPRING											TCI
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance					
	Installed Equipment (PN/SN)				PN Description		Specific Reference								
	WO number	WP	Scheduled check												
Inspection				Inspection 2				Overhaul				Life Limit			
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32NG35 28/10/2008	NGL SHOCK ABSORBER PN : D23592030				SN : B443		ABSORBER-SHOCK				TCI				
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00	20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918		
32NG36 28/10/2008	STRUT COMPLETE PN : D23596000-3				SN : WIA283		FORESTAY ASSY				TCI				
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00	20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918		
32NG37 28/10/2008	RACK PN : D66677				SN : 07CZ79057X07		RACK				TCI				
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00	20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918		
32RMA00 28/10/2008	MAIN LAINDING GEAR RH PN : 201581002				SN : MDL3642		MAIN LANDING GEAR RH				TCI				
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00	20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance			
	Installed Equipment (PN/SN)				PN Description				Specific Reference													
	WO number	WP	Scheduled check																			
		Inspection				Inspection 2				Overhaul				Life Limit								
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline					
32RMG00 28/10/2008	MAIN FITTING PN : 201540301				SN : 08MDM2048				MAIN FITTING													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG01 28/10/2008	LOWER CARDAN PIN PN : 201163307				SN : 08B0893X1767				LOWER CARDAN PIN													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG02 28/10/2008	SIDE STAY LOWER CARDAN PIN PN : 201163619				SN : 07AP0268				SIDE STAY LWR CARDAN PIN													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG03 28/10/2008	REAR PINTLE PIN PN : D3215303000800				SN : F07647-35				REAR PINTLE PIN													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)				PN Description		Specific Reference										
	WO number	WP	Scheduled check														
	Inspection				Inspection 2				Overhaul				Life Limit				
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32RMG04 28/10/2008	REAR PINTLE PIN NUT PN : SLN41193				SN : M337550-12801		REAR PINTLE PIN NUT										
LLT																	
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			
32RMG05 28/10/2008	AFT PINTLE SPHERICAL BEARING PN : 201042106				SN : 08NM132600X1179		AFT PINTLE BEARING										
LLT																	
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			
32RMG06 28/10/2008	FORWARD PINTLE PIN PN : 201173600				SN : 08MDG6980		PIN										
LLT																	
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			
32RMG07 28/10/2008	FORWARD PINTLE PIN STUB BOLTS PN : 201540616				SN : 08PA79539X444		CROSS BOLT										
LLT																	
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)		PN Description		Specific Reference										
	WO number	WP	Scheduled check												
Inspection				Inspection 2				Overhaul				Life Limit			
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG08 28/10/2008	FORWARD PINTLE PIN STUB BOLTS PN : 201540616		SN : 08PA79539X438	CROSS BOLT											
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32RMG09 28/10/2008	UPPER TORQUE LINK PIN PN : 201160603		SN : 08MDG8380	PIN											
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32RMG10 28/10/2008	UPPER TORQUE LINK PN : 201540303		SN : 08BEL0061	UPPER TORQUE LINK											
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32RMG11 28/10/2008	TORQUE LINK APEX PIN PN : 201383606		SN : 08MDG5306	MLG TORQUE LINK APEX PIN											
LLT															
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description										AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance				
	Installed Equipment (PN/SN)										PN Description		Specific Reference							
	WO number	WP	Scheduled check																	
Inspection				Inspection 2				Overhaul				Life Limit								
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline					
32RMG12 28/10/2008	TORQUE LINK APEX PIN NUT PN : 201587612										SN : 08MSL5902352		NUT							
LLT												OVH								
CY		16758,00				16758,00		20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918		3650	3918	-268	26/10/2018		3918							
32RMG13 28/10/2008	LOWER TORQUE LINK AT RH MLG PN : 201540302										SN : 08BEL0129		LOWER TORQUE LINK							
LLT												OVH								
CY		16758,00				16758,00		20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918		3650	3918	-268	26/10/2018		3918							
32RMG14 28/10/2008	LOWER TORQUE LINK PIN PN : 201160602										SN : 08MDG6742		PIN							
LLT												OVH								
CY		16758,00				16758,00		20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918		3650	3918	-268	26/10/2018		3918							
32RMG15 28/10/2008	SLIDING TUBE PN : 201371304										SN : 08B5178X9131		SLIDING ROD							
LLT												OVH								
CY		16758,00				16758,00		20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918		3650	3918	-268	26/10/2018		3918							

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)		PN Description		Specific Reference										
	WO number	WP	Scheduled check												
Inspection				Inspection 2				Overhaul				Life Limit			
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG16 28/10/2008	UPPER DIAPHRAGM TUBE PN : 201371615		SN : 08B0314X2	UPPER DIAPHRAGM TUBE											
LLT												OVH			
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32RMG17 28/10/2008	LOWER CARDAN NUT PN : 201163945		SN : 2763/08	LOWER CARDAN NUT											
LLT												OVH			
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32RMG18 28/10/2008	LOWER SIDE STAY PN : 201163301		SN : 07AP0223	LOWER SIDE STAY											
LLT												OVH			
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32RMG19 28/10/2008	SIDE STAY PIVOT PIN PN : 201163609		SN : 08AP0124	SIDE STAY PIVOT PIN											
LLT												OVH			
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG20 28/10/2008	UPPER SIDE STAY PN : 201163300		SN : AP042106	UPPER SIDE STAY												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32RMG21 28/10/2008	SIDE STAY UPPER CARDAN PIN PN : 201163618		SN : 07AP0048	SIDE STAY UPR CARDAN PIN												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32RMG22 28/10/2008	UPPER CARDAN PN : 201163620		SN : 08B0563X3	UPPER CARDAN												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32RMG23 28/10/2008	UPPER CARDAN NUT PN : 201163944		SN : 2185/07	UPPER CARDAN NUT												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance			
	Installed Equipment (PN/SN)				PN Description				Specific Reference													
	WO number	WP	Scheduled check																			
		Inspection				Inspection 2				Overhaul				Life Limit								
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline					
32RMG24 28/10/2008	UPPER PIN PN : 201661607				SN : 08AP0140				UPPER PIN (SPRINGS)													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG25 28/10/2008	LOWER PIN PN : 201661606				SN : 08AP0118				LOWER PIN (SPRINGS)													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG26 28/10/2008	TRIANGULATION LINK PIN PN : 201661609				SN : 08AP0129				PIN													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG27 28/10/2008	TRIANGULATION LINK PN : 201661307				SN : 08AP0195				LINK													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
		Inspection		Inspection 2		Overhaul		Life Limit								
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32RMG28 28/10/2008	BANANA LINK PN : 201661306		SN : 08AP0114	LINK BANANA												
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG29 28/10/2008	BANANA LINK PN : 201661306		SN : 08AP0121	LINK BANANA												
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG30 28/10/2008	CUFF PIN PN : 201661612		SN : 08AP0170	PIN												
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG31 28/10/2008	CUFF PIN PN : 201661612		SN : 08AP0172	PIN												
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance			
	Installed Equipment (PN/SN)				PN Description				Specific Reference													
	WO number	WP	Scheduled check																			
		Inspection				Inspection 2				Overhaul				Life Limit								
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline					
32RMG32 28/10/2008	CUFF PN : 201661613				SN : 07AP0218				CUFF													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG33 28/10/2008	PIN PN : 201661605				SN : 08AP0054				PIN (LOCK/BANANA LINK)													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG34 28/10/2008	LOCK LINK LOWER PIN PN : 201057627				SN : 07AP0220				LOCK LINK LOWER PIN													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG35 28/10/2008	LOWER LOCK LINK PN : 201058305				SN : 08AP0051				LOWER ARM													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32RMG36 28/10/2008	LOCK LINK CENTER PIN PN : 201058608		SN : 07AP0224	LOCK LINK CENTER PIN												
LLT																
CY		16758,00			16758,00				16758,00			60000	16758,00	43242,00	60000,00	
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			
32RMG37 28/10/2008	UPPER LOCK LINK PN : 201058310		SN : 08AP0006	UPPER ARM												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00	
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			
32RMG38 28/10/2008	LOCK LINK UPPER PIN PN : 201058611		SN : 07AP0073	LOCK LINK UPPER PIN												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00	
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			
32RMG39 28/10/2008	LOCK STAY CARDAN PN : 201058306		SN : 08COU46164X5769	COMPLETE CARDAN												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00	
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32RMG40 28/10/2008	LOCK STAY CARDAN PIN															
	PN : 201056909		SN : 08BEV81754025		LOCK STAY CARDAN PIN											
	LLT											OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG46 28/10/2008	SIDESTAY ASSY															TCI
	PN : 201166014-025		SN : AP1270		STAY ASSY-SIDE,MLG											
	LLT											OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG47 28/10/2008	DAMPER TORQUE LINK															O/C
	PN : 201419001-020		SN : MG2204		DAMPER, TORQUE LINK - MLG											
	LLT															
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C
Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C	
32RMG49 28/10/2008	RETRACTION ACTUATOR															TCI
	PN : 201590002-020		SN : B5242		ACTUATOR-RETRACTION											
	LLT											OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance						
	Installed Equipment (PN/SN)				PN Description		Specific Reference									
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG50 28/10/2008	MONITORING UNIT BRK TEMP L PN : 35-1H5-1002				SN : 12325				BTMU-BRAKE TEMPERATURE MO				O/C			
LLT																
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C
Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C	
32RMG51 28/10/2008	HOUSING PN : 201056669				SN : 08BEV81828077				HOUSING				TCI			
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG52 28/10/2008	PIN (UPLOCK) PN : 201383648				SN : 08BEV82156017				PIN				TCI			
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG53 28/10/2008	PIN PN : 201056885				SN : 08BEV81971X219				PIN				TCI			
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32RMG54 28/10/2008	SHOCK ABSORBER PN : 201371281		SN : 08B5178X9131	SHOCK ABSORBER						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG55 28/10/2008	PIN PN : 201160317		SN : 08SOP87792X018	PIN						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG56 28/10/2008	NUT PN : D52041-1		SN : 08CZ85600X172	NUT						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32RMG57 28/10/2008	NUT PN : D52041-1		SN : 08CZ85600X151	NUT						TCI						
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance	
	Installed Equipment (PN/SN)				PN Description				Specific Reference											
	WO number	WP	Scheduled check																	
Inspection				Inspection 2				Overhaul				Life Limit								
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline					
32RMG58 28/10/2008	PISTON PN : 201160693				PISTON SN : 08MG10733								TCI							
	LLT																			
	CY		16758,00			16758,00		20000	16758,00	3242,00	OVH 20000,00		16758,00							
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918							
32RMG59 28/10/2008	LOCKING SPRING ASSY PN : 201059001				LOCK SPRING SN : AP2571								O/C							
	LLT																			
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C			
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C				
Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C					
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918							
32RMG60 28/10/2008	SPRING PN : 201059602				SPRING SN : 017503								TCI							
	LLT																			
	CY		16758,00			16758,00		20000	16758,00	3242,00	OVH 20000,00		16758,00							
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918							

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description																
	Installed Equipment (PN/SN)																
	WO number	WP	Scheduled check														
	Inspection				Inspection 2				Overhaul				Life Limit				
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32RMG61 28/10/2008	LOCKING SPRING ASSY PN : 201059001 SN : AP2572 LOCK SPRING O/C																
	LLT																
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C
	CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C
	Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			
32RMG62 28/10/2008	SPRING PN : 201059602 SN : 017511 SPRING TCI																
	LLT																
	CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00			
	Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			
32RMG63 28/10/2008	LOCK LINK ASSY PN : 201058003 SN : AP1316 STAY-LOCK TCI																
	LLT																
	CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00			
	Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			
32RMG65 28/10/2008	CYLINDER PN : 114095305 SN : 07AP0414 CYLINDER TCI																
	LLT																
	CY		16758,00			16758,00			20000	16758,00	3242,00	OVH 20000,00		16758,00			
	Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918			

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM		F.I.N		Maint. level		Zone		Phasing		Tolerance					
	Installed Equipment (PN/SN)		PN Description		Specific Reference													
	WO number	WP	Scheduled check															
		Inspection				Inspection 2				Overhaul				Life Limit				
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32RMG66 28/10/2008	PISTON ROD PN : 114095667		SN : 07AP0125		PISTON ROD								TCI					
LLT																		
CY		16758,00				16758,00				20000	16758,00	3242,00		OVH 20000,00		16758,00		
Cal Total		3918				3918				3650	3918	-268		26/10/2018		3918		
32RMG67 28/10/2008	EYE END PN : 114095602		SN : 07AP0479		EYE END								TCI					
LLT																		
CY		16758,00				16758,00				20000	16758,00	3242,00		OVH 20000,00		16758,00		
Cal Total		3918				3918				3650	3918	-268		26/10/2018		3918		
32RMG68 28/10/2008	CYLINDER PN : 201590906		SN : 08MDC00080		RTCN ACT CYLINDER													
LLT																		
CY		16758,00				16758,00				20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918				3650	3918	-268		26/10/2018		3918		
32RMG69 28/10/2008	PISTON HEAD PN : 201590909		SN : 08JL985X155		PISTON HEAD													
LLT																		
CY		16758,00				16758,00				20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918				3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance			
	Installed Equipment (PN/SN)				PN Description				Specific Reference													
	WO number	WP	Scheduled check																			
		Inspection				Inspection 2				Overhaul				Life Limit								
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline					
32RMG70 28/10/2008	NUT PN : 201590910				SN : 08JL1090X140				RTCN ACT PISTON HEAD NUT													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG71 28/10/2008	GLAND PN : 201590913				SN : 08JL986X54				RETRACTION ACT GLAND													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG72 28/10/2008	PISTON ROD PN : 201590908				SN : 08MDC00160				PISTON ROD													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								
32RMG73 28/10/2008	PIN PN : 201590901				SN : 07BA58521X25				RTCN ACTUATOR PIN (REAR)													
	LLT												OVH									
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918								

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32RMG74 28/10/2008	PIN PN : 201590900		SN : 08BA59459X02		RTCN ACTUATOR PIN (MAIN)											
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG77 28/10/2008	PIN (RETAINING LOWER BEARING) PN : 201383608		SN : 08BEV81759X012		RETAINIG PIN				TCI							
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG78 28/10/2008	PIN (RETAINING LOWER BEARING) PN : 201383608		SN : 08BEV81759X013		RETAINIG PIN				TCI							
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32RMG79 28/10/2008	PIN (RETAINING LOWER BEARING) PN : 201383608		SN : 08BEV81759X014		RETAINIG PIN				TCI							
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

2. Seller's Name: AvAir	3. Reference #: RLS-221861-RLSL-419019
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4. Organization: AvAir, LLC Address: 6877 W. Frye Rd Chandler AZ 85226 Website: avair.aero	Phone#: 480-763-8200 Fax#: 480-763-8212 SITA/Wire Code: Status: Surplus Parts Supplier
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5A. Seller's Contract #: SO-190942	5B. Buyer's PO #: P33699
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6. Item	7. Description	8. Manufacturer and Part Number	9. App Code	10. Qty	11. Serial/Batch #	12. Status
1	LOCK STAY CARDA	201056909		1	R0290196000	New Surplus

13A. Remarks: AvAir certifies to the best of our knowledge, based on the information and documents available at the time of purchase the above referenced part(s) or material was/were not obtained from any governmental or military source and has/have not been subjected to severe stress or heat or immersed insaltwater (as in major engine failure, accident, incident, or fire) unless otherwise indicated and is/are therefore understood to be non-incident related.

13B. Traceable To: AIR MALTA	13C. Last Certificated Agency:
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14. New Parts/Material Verification: THE FOLLOWING SIGNATURE ATTESTS THAT THE PART(S) OR MATERIAL(S) IDENTIFIED ABOVE WAS (WERE) MANUFACTURED BY A FAA PRODUCTION APPROVAL HOLDER (PAH), OR TO AN INDUSTRY COMMERCIAL STANDARD.	18. New Surplus, Used, Repaired or Overhaul Parts Verification: THE FOLLOWING SIGNATURE ATTESTS THAT THE DOCUMENTATION SPECIFIED ABOVE OR ATTACHED IS ACCURATE WITH REGARD TO THE ITEM(S) DESCRIBED.
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15. Signature	19. Signature 
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16. Name	17. Date	20. Name Robert Munoz	21. Date 12/19/2024
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NOTICE: The above signature binds the seller and the signer to the accuracy of the information provided on the form. Should the information provided in this Form contain inaccuracies or misrepresentations, the signer and seller may be liable for damages and be subject to criminal prosecution under state and federal law.

Air Malta
Shipping Invoice No. 2524

Av-Air Inc.
PO No.: P0015914 MB

14th March 2011

PN	DESCRIPTION	QTY	UOM	Box No.
201056604	BUSH	2	EA	302
201056639	RING WIPER	2	EA	347
201056643	SPACER	4	EA	230
201056648	WASHER	3	EA	49
201056649	DOME JACKING	1	EA	341
201056668	SPACER	2	EA	49
201056688	SPACER	2	EA	341
201056691	SEAL	2	EA	347
201056745	TAB WASHER	5	EA	347
201056840	PLATE	1	EA	347
201056866	BLOCK GUIDE	1	EA	347
201056868	WASHER	4	EA	347
201056909	PIN	1	EA	338
201056910	WASHER	2	EA	347
201056930	WASHER	2	EA	341
201056938	WASHER	5	EA	26
201056963	WASHER TAB	2	EA	26
201056966	CABLE-BONDING	12	EA	313
201056966	CABLE-BONDING	2	EA	313
201057616	SPACER	2	EA	26
201057625	PIN	1	EA	26
201057627	PIN	1	EA	26
201057628	SLEEVE	1	EA	28
201057629	WASHER	2	EA	28
201057638	WASHER	1	EA	28
201057639	WASHER	3	EA	28
201057641	WASHER	2	EA	28
201057643	SPACER	2	EA	28
201057644	PIN	1	EA	28
201057650	PIN	1	EA	28
201058603	WASHER	1	EA	27
201058604	PIN	1	EA	28


This is to certify that the part(s) referenced on Air Malta Shipping Invoice No. 2524 were sold to Av-Air Inc. under Purchase Order number P0015914 MB and was / were not obtained from any Government or Military source and was/ were not subject to severe stress, heat or immersed in salt water (as in a major engine failure, accident or fire).




ANTHONY MUSCAT INC


HEAD OF QUALITY

SECTION 3. REPAIR APPROVALS

SECTION 4. MODIFICATION STATUS

Component Overhaul Services, Corp. FAA Repair Station: #8COR883B EASA 145.6404	CUSTOMER: CAVU AEROSPACE, INC.		WORK ORDER: 12615
	PART NUMBER: 201582002-040 SERIAL NUMBER: MDG4846		
MODIFICATION	DESCRIPTION		
SB 200-32-220	MLG-TORQUE LINK DAMPER UNIT-INTRODUCTION OF A CLAMP.		CW
SB 200-32-230	IDENTIFICATION OF SAFE LIFE OF COMPONENTS AND UNITS.		PCW
SB 200-32-250	INSPECTION OF LOCKING BUSH AND SLIDING TUBE.		CW
SB 200-32-254	MAIN LANDING GEAR - INSTALLATION OF LOCK STAY 201058003 TO 201376 AND 201175 SERIES MAIN LANDING GEARS.		PCW
SB 200-32-261	MLG- INSTALLATION OF TWO STUB BOLTS FOR THE FWD. PINTLE PIN IN PLACE OF THE CROSS BOLT.		PCW
SB 200-32-267	MLG- INTRODUCTION OF A NEW CUFF SUB-ASSEMBLY AND BANANA LINK SUB-ASSEMBLY.		PCW
SB 200-32-268	MLG- TORQUE LINK DAMPER UNIT- INTRODUCTION OF IMPROVED DISK SPRINGS.		PCW
SB 200-32-271	MLG- TO ADD TRACKING NUMBERS TO PARTS LISTED IN AIRBUS DOCUMENT MPD, SECTION 9-1. (RETRACTION ACTUATOR PINS AND FWD. PINTLE PIN CROSS BOLTS). RETRACTION ACTUATOR NOT PROVIDED.		PC
SB 200-32-282	MLG- TO ADD TRACKING NUMBERS TO PARTS LISTED IN AIRBUS DOCUMENT MPD, SECTION 9-1 (TORQUE LINK APEX PIN NUT).		PCW
SB 200-32-283	MLG- TO ADD TRACKING NUMBERS TO PARTS LISTED IN AIRBUS DOCUMENT MPD, SECTION 9-1 (SIDE STAY).		PCW
SB 200-32-285	MLG- INTRODUCTION OF A NEW GLAND IN THE LOCK STAY ACTUATOR.		PCW
SB 200-32-286	MLG- INSPECTION OF THE SLIDING TUBE FOR CRACKS.		S/N NOT AFFECTED
SB 200-32-295	LANDING GEAR-MLG-DEFINITION OF RESTORATION TASKS FOR THE MAIN LANDING GEAR.		CW
SB 200-32-297	MLG- INTRODUCTION OF A NEW LOWER BEARING SUB-ASSY.		PCW
SB 200-32-302	MLG- INTRODUCTION OF NEW CHARGING LABELS.		PCW
SB 200-32-312	MLG- INTRODUCTION OF A NEW UPPER CARDAN AND MODIFICATION OF CURRENT P/N: 201163620. MODIFIED TO 201163948.		CW
SB 200-32-313	MLG- BARKHAUSEN NOISE INSPECTION OF MLG SLIDING TUBE AXLES.		S/N NOT AFFECTED
SB 200-32-315	LANDING GEAR - MAIN LANDING GEAR COMPLETE - SIDE STAY COMPLETE - SIDE STAY BASIC- INTRODUCTION OF NE END CAPS AND BOLTS FOR THE UPPER AND LOWER CARDAN JOINTS.		CW
SB 200-32-321	LANDING GEAR- MAIN GEAR INSPECTION SLIDING TUBE / SLAVE LINK LUG CRACK (REFERENCE AIRBUS SB A320-32-1441).		CW
SB 200-32-335	LANDING GEAR - MAIN LANDING GEAR COMPLETE - MAIN LANDING GEAR LEG - INSPECTION OF MAIN LANDING GEAR SLIDING TUBE AXLES.		S/N NOT AFFECTED
SB A320-32A1223	MLG- INTRODUCTION OF A NEW CUFF SUB-ASSEMBLY AND BANANA LINK SUB-ASSEMBLY.		PCW
SB A320-32-1441	"MANDATORY" LANDING GEAR- MAIN GEAR INSPECTION SLIDING TUBE / SLAVE LINK LUG CRACK (REFERENCE SAFRAN SB 200-32-321).		CW
FAA AD 2000-11-09	INSPECTION OF LOCKING BUSH AND SLIDING TUBE. AGREES WITH SB 200-32-250.		CW
FAA AD 2002-06-01	MLG- INSTALLATION OF TWO STUB BOLTS FOR THE FWD. PINTLE PIN IN PLACE OF THE CROSS BOLT. AGREES WITH SB 200-32-261.		PCW

Component Overhaul Services, Corp. FAA Repair Station: #8COR883B EASA 145.6404	CUSTOMER: CAVU AEROSPACE, INC.	WORK ORDER: 12615	
	PART NUMBER: 201582002-040	SERIAL NUMBER: MDG4846	
MODIFICATION	DESCRIPTION		STATUS
FAA AD 2005-17-07	THIS AD RESULTS FROM A REPORT THAT AN AXLE NUT HAD SEPARATED FROM AN AXLE ON A MLG WHEEL, DUE TO MISSING LOCKING BOLTS. THE FAA IS ISSUING THIS AD TO DETECT AND CORRECT MISSING LOCKING BOLTS ON THE AXLE NUTS OF THE MLG WHEELS. ABSENCE OF THE LOCKING BOLTS COULD RESULT IN SEPARATION OF A WHEEL (S) FROM THE AXLE AND CONSEQUENT REDUCE THE CONTROLLABILITY OF THE AIRPLANE DURING TAKEOFF AND LANDING, AND POSSIBLE INJURY TO PEOPLE ON THE GROUND (REFERENCE AOT A320-32A1303).		CW
FAA AD 2005-10-04	TO PREVENT FAILURE OF THE MLG SIDE-STAY CUFF LUGS OR DOWN-LOCK SPRING ATTACHMENTS, WHICH COULD RESULT IN IMPROPER DOWN-LOCK OF THE MLG DURING A FREEFALL EXTENSION, AND POSSIBLE COLLAPSE OF THE MLG.		PCW
FAA AD 2015-22-08	MLG- INTRODUCTION OF A NEW UPPER CARDAN AND MODIFICATION OF CURRENT P/N: 201163620. MODIFIED TO 201163948. AGREES WITH SB 200-32-312.		CW
FAA AD 2017-18-21	MLG - INTRODUCE DOUBLE LOCKING SIDESTAY CARDAN JOINTS. AGREES WITH SB 200-32-315.		CW
FAA AD 2019-14-06	Landing Gear – Main Landing Gear Sliding Tubes – Inspection		S/N NOT AFFECTED
FAA AD 2020-14-08	Landing Gear – Main Landing Gear Torque Link Apex Pin – Replacement / Inspection		CW
FAA AD 2022-12-05	Landing Gear – Main Landing Gear Sliding Tubes – Inspection		CW
FAA AD 2022-22-10	THE FAA IS ISSUING THIS AD TO ADDRESS THE FAILURE OF CERTAIN LIFE-LIMITED PARTS, WHICH COULD RESULT IN REDUCED STRUCTURAL INTEGRITY OF THE AIRPLANE. SEE THE MCAI FOR ADDITIONAL BACKGROUND INFORMATION.		PCW
FAA AD 2023-01-01	Landing Gear – NLG & MLG Landing Gear Sliding Tube – Inspection / Replacement		S/N NOT AFFECTED
FAA AD 2023-23-11	Landing Gear – Landing Gear Sliding Tube – Inspection / Replacement		S/N NOT AFFECTED
EASA AD 2011-0024R1	LANDING GEAR-MAIN LANDING GEAR SIDE STAY CUFF LUG INSPECTION/REPAIR (REFERENCE AIRBUS SB A320-32A1223 AND 32A1224).		PCW
EASA AD 2016-0018R1	LANDING GEAR - MAIN LANDING GEAR COMPLETE - SIDE STAY COMPLETE - SIDE STAY BASIC- INTRODUCTION OF NE END CAPS AND BOLTS FOR THE UPPER AND LOWER CARDAN JOINTS. AGREES WITH SB 200-32-315.		CW
EASA AD 2019-0151-E	Landing Gear – Main Landing Gear Sliding Tubes – Inspection		S/N NOT AFFECTED
EASA AD 2020-0130	Landing Gear – Main Landing Gear Torque Link Apex Pin – Replacement / Inspection		CW
EASA AD 2020-0193	Landing Gear – Main Landing Gear Sliding Tubes – Inspection / Replacement		CW
EASA AD 2021-0236	Landing Gear – NLG & MLG Landing Gear Sliding Tube – Inspection / Replacement		S/N NOT AFFECTED
EASA AD 2021-0175	Landing Gear – Main Landing Gear Sliding Tubes – Inspection		S/N NOT AFFECTED
EASA 2022-0204R1	Landing Gear – Main Landing Gear Sliding Tube – Inspection / Replacement		CW
EASA 2024-0066	Time Limits / Maintenance Checks – Safe Life Airworthiness Limitations Items – ALS Part 1		CW
COMPONENT OVERHAUL SERVICES, CORP., CERTIFIES THAT TO THE BEST OF OUR KNOWLEDGE THE COMPONENTS THAT MAKE UP THIS ASSEMBLY WERE NOT OBTAINED FROM ANY U.S. GOVERNMENT OR MILITARY SOURCE AND HAVE NOT BEEN SUBJECTED TO EXTREME HEAT OR STRESS (AS IN A MAJOR ENGINE FAILURE, ACCIDENT, OR FIRE.)			
MODIFICATION STATUS LEGEND CW: COMPLIED WITH PCW: PREVIOUSLY COMPLIED WITH NC: NOT COMPLIED PC: PARTIALLY COPLIED WITH N/A: NOT APPLICABLE		QUALITY ENGINEERING APPROVAL  December 19, 2024	

SECTION 5. INSPECTION STATUS REPORT

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: CUFF

W/ORDER: 12615

P/NUMB.: 201661613

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 07AP0218

SUB W/O: 12615-187

Component Overhaul
Services

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A		2.4840	2.4050	2.4050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2050	201057642	0.0010	2.406
1B			2.4050	2.4050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2060	201057642	0.0010	2.406
2A		1.8120	1.8500	1.8500	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.0450	201057642		
							2.0460			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201057642		0.097
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201057642		0.097
3A	0.6319	NOTE 1	0.6323	0.6323	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-08M	0.0010	0.633
3B	0.6323		0.6323	0.6323	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	4.7237	BCA2-12-08M	0.0010	0.633
3C	0.6319	NOTE 1	0.6323	0.6323	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-08M	0.0010	0.633
3D	0.6323		0.6323	0.6323	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4737	BCA2-12-08M	0.0010	0.633
4A			0.8280	0.8280	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7083	BCA2-12-08M		
	REF.						0.7184			
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-08M		0.060
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-08M		0.060
4B			0.8280	0.8280	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7083	BCA2-12-08M		
	REF.						0.7184			
4B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-08M		0.060
4B2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-08M		0.060
5A	INTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
6A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
7A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: CUFF

W/ORDER: 12615

P/NUMB.: 201661613

OHM CHAP.: 32-11-26

S/NUMB.: 07AP0218

SUB W/O: 12615-187

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT



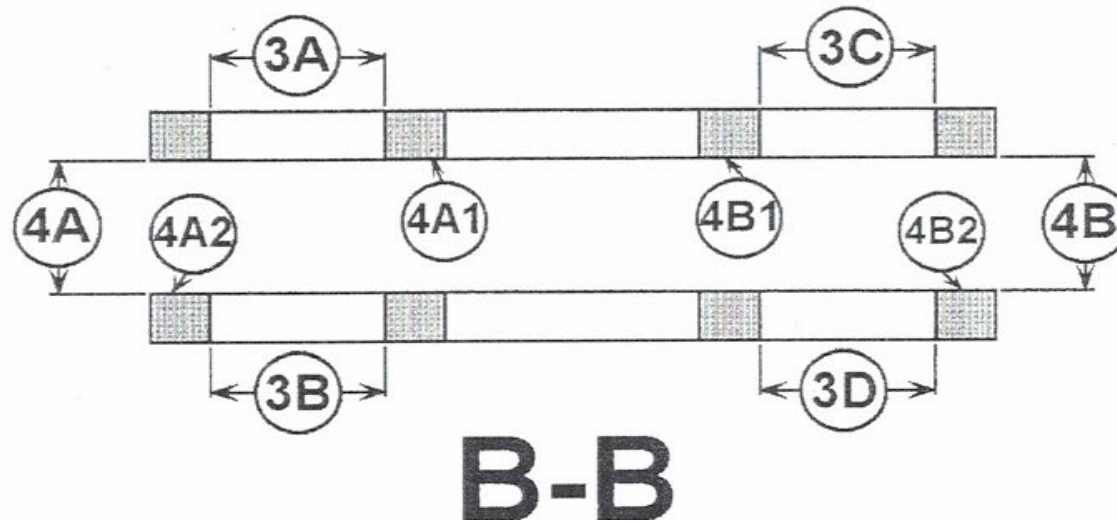
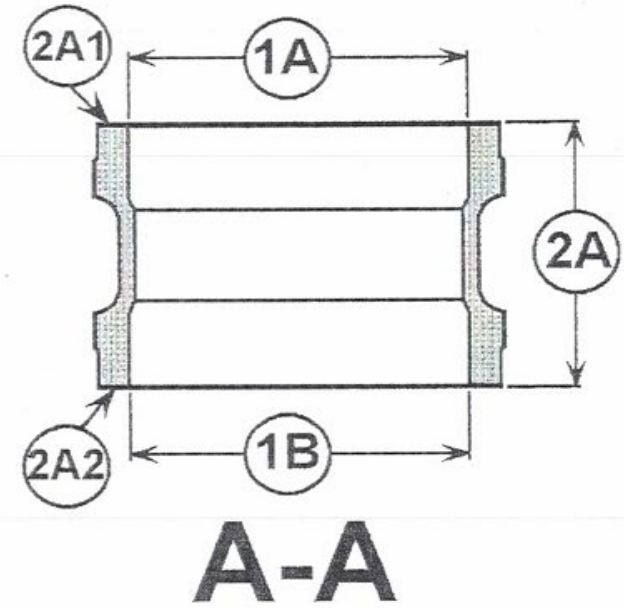
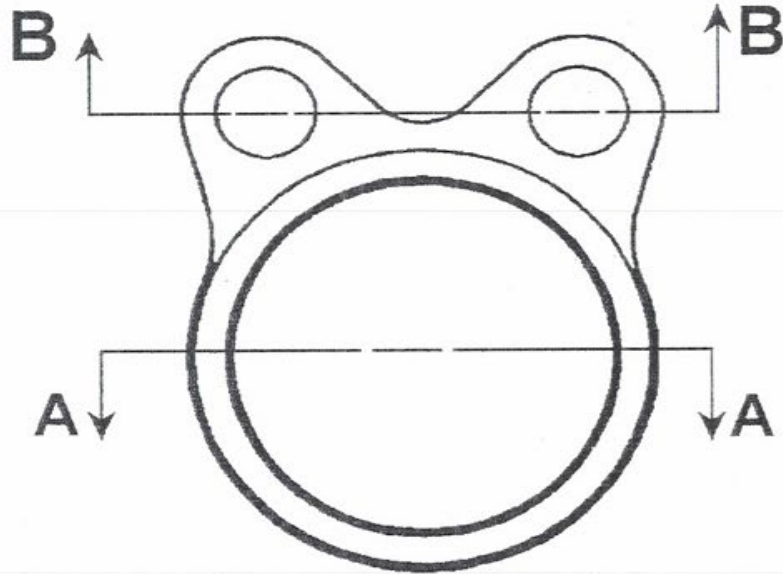
Component Overhaul
Services

NOTE 1: IF BORE IS OVERSIZED OR REQUIRES MACHINING SCRAP THE UNIT.

A/C: A320
P/NAME: CUFF
P/NUMB.: 201661613
S/NUMB.: 07AP0218
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE, INC.
W/ORDER: 12615
OHM CHAP.: 32-11-26
SUB W/O: 12615-187
C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: TRIANGULAR LINK

W/ORDER: 12615

INSPECTION STATUS REPORT

P/NUMB.: 201661307

OHM CHAP.: 32-11-26



S/NUMB.: 08AP0195

SUB W/O: 12615-181

LIFE LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	0.6306	0.6914	0.6311	0.6311	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-06M	0.0015	0.633
	0.6311						0.4737			
2A		0.4696	0.5090	0.5090	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6192	BCA2-12-06M		
							0.6283			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
3A	0.7506		0.7511	0.7511	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-15-12M	0.0015	0.753
3B	0.7511		0.7511	0.7511	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4737	BCA2-15-12M	0.0015	0.753
4A			2.0850	2.0850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.1940	BCA2-15-12M		
							2.2040			
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-15-12M		0.055
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-15-12M		0.055
5A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
6A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: TRIANGULAR LINK

W/ORDER: 12615

P/NUMB.: 201661307

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT



Component Overhaul
Services

S/NUMB.: 08AP0195

SUB W/O: 12615-181

LIF. LIMIT: 60, 000

C.S.N.: 16,758

NOTE 1: IF BORE OVERSIZED OR REQUIRES MACHINING SCRAP THE UNIT

A/C: A320

P/NAME: TRIANGULAR LINK

P/NUMB.: 201661307

S/NUMB.: 08AP0195

LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE, INC.

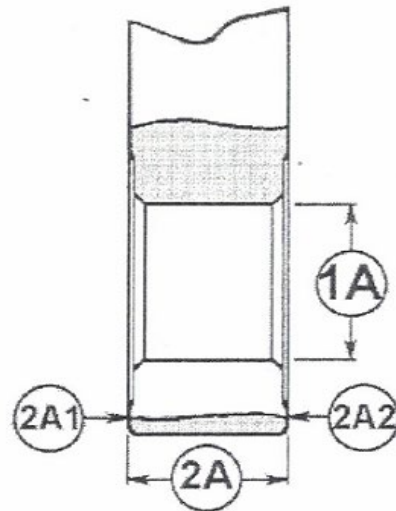
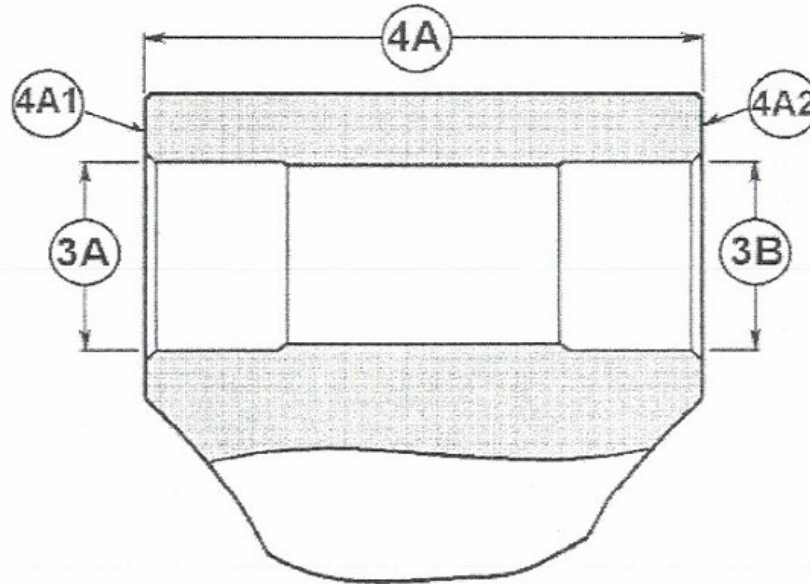
W/ORDER: 12615

OHM CHAP.: 32-11-26

SUB W/O: 12615-181

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: LOWER LOCK LINK

W/ORDER: 12615

P/NUMB.: 201058305

OHM CHAP.: 32-12-27

INSPECTION STATUS REPORT

S/NUMB.: 08AP0051

SUB W/O: 12615-38

Component Overhaul
Services

LIFE LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	0.9843	1.0599	0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058601	0.0010	0.986
1B	0.9851		0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7882	201058601	0.0010	0.986
2A		2.7924	2.8360	2.8360	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.0211	201058601		
							3.0303			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058601		0.093
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058601		0.093
3A	0.6319	0.7240	0.6323	0.6323	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-06M	0.0010	0.633
3B	0.6323				WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4737	BCA2-12-06M	0.0010	0.001
4A		0.4698	0.5100	0.5100	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6196	BCA2-12-06M		
		NOTE 2					0.6283			
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
5A	0.7480	0.8150	0.7489	0.7489	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201058600	0.0010	0.750
5B	0.7489		0.7489	0.7489	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5913	201058600	0.0010	0.750
6A			1.7320	1.7320	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.7717			
							1.7828			
6A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				0.020
6A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				0.020
7A		1.0663	1.0240	1.0240	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.8701	201058600		
							0.8760			
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058600		0.077
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058600		0.077
8A	0.9843	1.0589	0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058612	0.0010	0.986
8B	0.9851		0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7882	201058612	0.0010	0.986
9A			2.0810	2.0810	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
9A1	FACES	NOTE 2	COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				-1.041
9A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				-1.041

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: LOWER LOCK LINK

W/ORDER: 12615

P/NUMB.: 201058305

OHM CHAP.: 32-12-27

INSPECTION STATUS REPORT

S/NUMB.: 08AP0051

SUB W/O: 12615-38



LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
10A		1.3425	1.2970	1.2970	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.1417	201058600		
							1.1430			
10A1	FACES	NOTE 2	COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058600		0.078
10A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058600		0.078
11A	EXTERNAL SURFACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
12A	OTHER REJECTIONS		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: LOWER LOCK LINK

W/ORDER: 12615

P/NUMB.: 201058305

OHM CHAP.: 32-12-27

INSPECTION STATUS REPORT

S/NUMB.: 08AP0051

SUB W/O: 12615-38



LIF. LIMIT: 60, 000

C.S.N.: 16,758

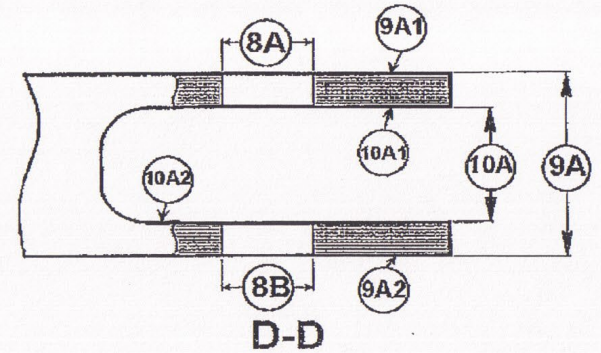
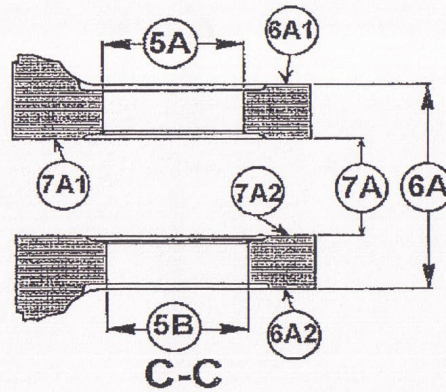
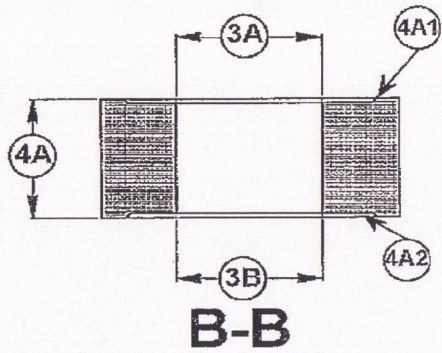
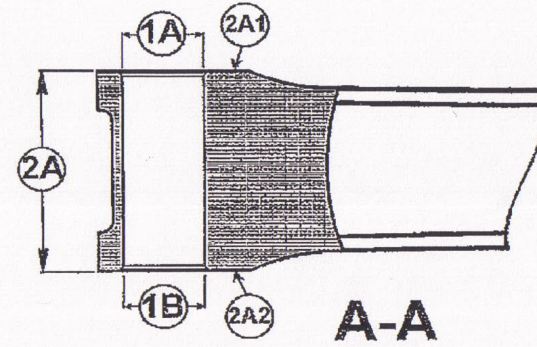
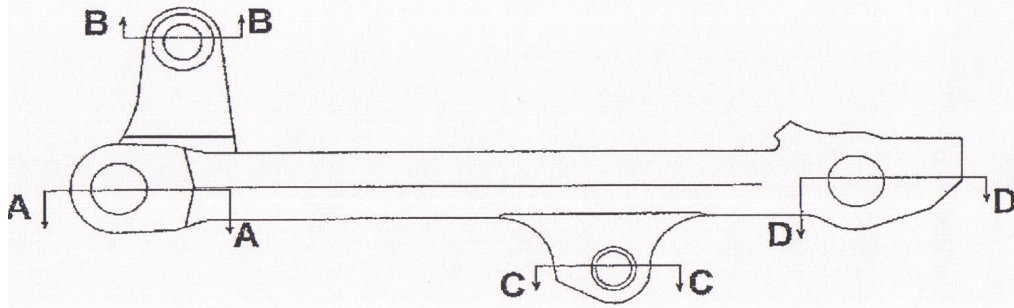
NOTE 1: IF MACHINED IS REQUIRED TO ORDER O/S BUSHING P/N: 899006001 QTY 2

NOTE 2: FACE MACHINING (SPOTFACE DIAMETER OF 1.063)

A/C: A320
P/NAME: LOWER LOCK LINK
P/NUMB.: 201058305
S/NUMB.: 08AP0051
LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE, INC.
W/ORDER: 12615
OHM CHAP.: 32-12-27
SUB W/O: 12615-38
C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE

P/NAME: UPPER LOCK LINK

W/ORDER: 12615

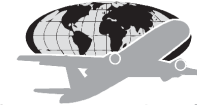
P/NUMB.: 201058310

OHM CHAP.: 32-12-27

INSPECTION STATUS REPORT

S/NUMB.: 08AP0006

SUB W/O: 12615-8

Component Overhaul
Services

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	0.9843	1.0600	0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058613	0.0010	0.986
1B	0.9851	NOTE 1	0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7882	201058613	0.0010	0.986
2A	0.9830	0.9428	0.9830	0.9830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.1401	201058613		
	REF.	NOTE 2					1.1409			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058613		0.079
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058613		0.079
3A	0.1929	0.2913	0.1980	0.1980	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1929	201058310-RS3	0.0010	0.199
3B	0.2008		0.1980	0.1980	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.2008	201058310-RS3	0.0010	0.199
3C	0.1929	0.2913	0.1980	0.1980	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1929	201058310-RS3	0.0010	0.199
3D	0.2008		0.1980	0.1980	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.2008	201058310-RS3	0.0010	0.199
4A			COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
4B			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
5A	0.9843	1.0600	0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058601	0.0010	0.986
5B	0.9851	NOTE 1	0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7882	201058601	0.0010	0.986
6A	2.0768		2.0890	2.0890	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.1180			
	2.0965						2.1300			
6A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				0.015
6A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				0.015
7A	1.3010	1.3430	1.3050	1.3050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.1024	201058601		
	REF.	NOTE 2					1.1037			
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058601		0.101
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058601		0.101
8A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
9A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE

P/NAME: UPPER LOCK LINK

W/ORDER: 12615

P/NUMB.: 201058310

OHM CHAP.: 32-12-27

S/NUMB.: 08AP0006

SUB W/O: 12615-8

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT



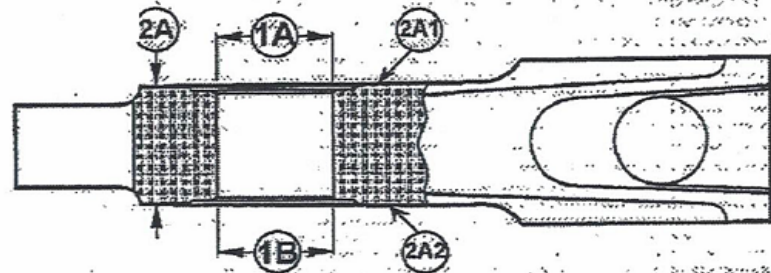
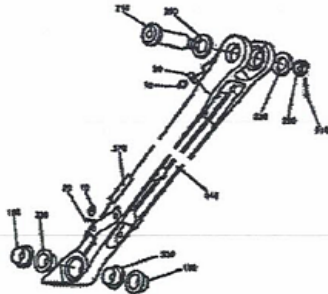
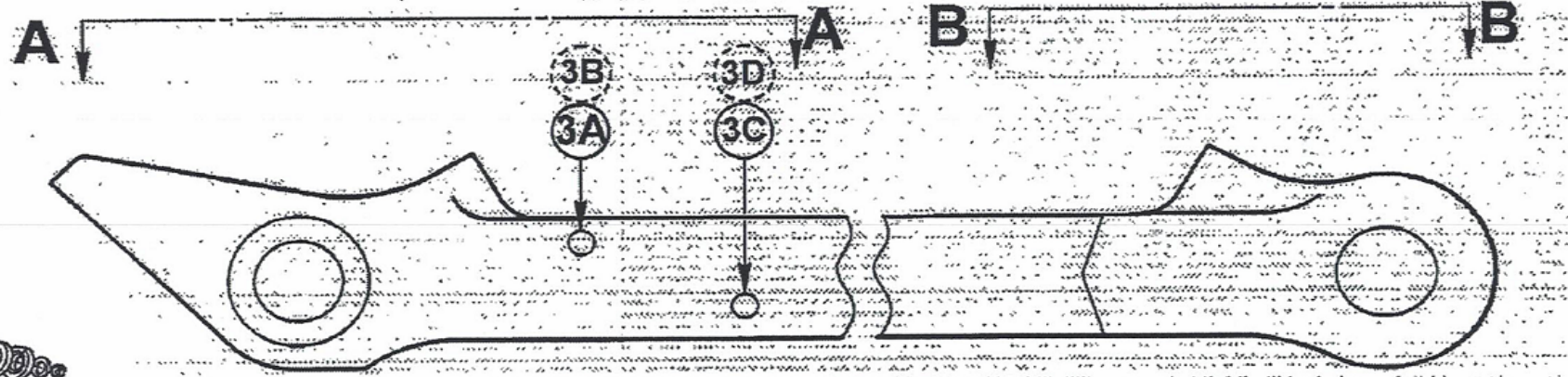
NOTE 1: O/S BUSHINGS

NOTE 2: O/S FLANGE BUSHINGS

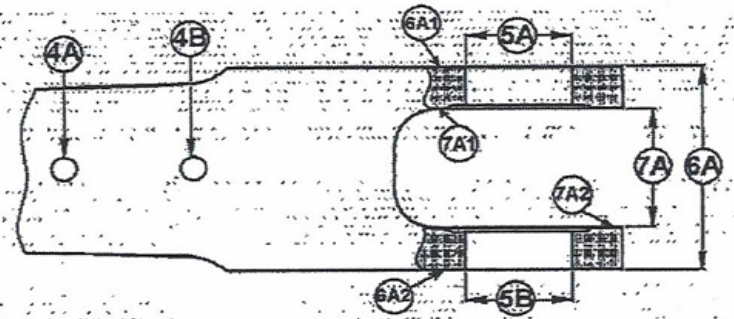
A/C: A320
P/NAME: UPPER LOCK LINK
P/NUMB.: 201058310
S/NUMB.: 08AP0006
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE
W/ORDER: 12615
OHM CHAP.: 32-12-27
SUB W/O: 12615-8
C.S.N.: 16,758

INSPECTION STATUS REPORT



A-A



B-B

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: STAY, UPPER SIDE

W/ORDER: 12615

P/NUMB.: 201163300

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: AP042106

SUB W/O: 12615-76


Component Overhaul
 Services

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	2.4016	2.4850	2.4050	2.4050	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.2048	201163601	0.0015	2.407
1B	2.4028		2.4050	2.4050	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.2059	201163601	0.0015	2.407
2A		5.0354	5.0720	5.0720	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	5.2724	201163601		
		NOTE 5					5.2738			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201163601		0.100
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201163601		0.100
3A		2.4480	2.4010	2.4010	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.2048	201163601		
		NOTE 4					2.2065			
3A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201163601		0.098
3A2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201163601		0.098
4A	0.9843	1.0600	0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.7874	201163637	0.0020	0.987
4B	0.9851		0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.7882	201163637	0.0020	0.987
5A	3.3450		3.3480	3.3480	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	3.1490	201163637		
	REF.						3.1510			
5A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.0950	201163637		0.099
5A2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.0950	201163637		0.099
5B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
5B2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
6A	0.7494	0.8236	0.7490	0.7490	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.5908	BCA2-15-15M	0.0020	0.751
6B	0.7489		0.7490	0.7490	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.5919	BCA2-15-15M	0.0020	0.751
7A			2.4030	2.4030	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.2840	BCA2-15-15M		
							2.2980			
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		BCA2-15-15M		0.060
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		BCA2-15-15M		0.060
7B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7B2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
8A	0.7874	0.8630	0.7880	0.7880	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.5905	201163602	0.0020	0.790
8B	0.7882		0.7880	0.7880	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.5912	201163602	0.0020	0.790
9A		2.0910	2.0490	2.0490	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	1.8543	201163602		
							1.8602			
9A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201163602		0.097
9A2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201163602		0.097

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: STAY, UPPER SIDE

W/ORDER: 12615

P/NUMB.: 201163300

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: AP042106

SUB W/O: 12615-76

Component Overhaul
Services

LIFE LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
10A	0.6299	0.6890	0.6310	0.6310	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.4726	BCA2-12-12M	0.0020	0.633
10B	0.6306				WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.4737	BCA2-12-12M	0.0020	0.002
11A		2.9180	2.8760	2.8760	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.7559 2.7704	BCA2-12-12M		
11A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		BCA2-12-12M		0.060
11A2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		BCA2-12-12M		0.060
11B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
11B2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
12A	2.4016	2.4850	2.4030	2.4030	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.2048	201163600	0.0015	2.405
12B	2.4028		2.4030	2.4030	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.2059	201163600	0.0015	2.405
13A			4.7150	4.7150	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	4.7560 4.7680			
		NOTE 5								
13A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				0.021
13A2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				0.021
14A		2.6810	2.6400	2.6400	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.3623 2.3640			
14A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201163601		0.139
14A2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201163601		0.139
15A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
16A	OTHER REJECTIONS		COND. ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: STAY, UPPER SIDE

W/ORDER: 12615

P/NUMB.: 201163300

OHM CHAP.: 32-11-26

S/NUMB.: AP042106

SUB W/O: 12615-76

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT



NOTE 1: FOR OVERSIZED USE BUSHING P/N: 450217880

NOTE 2: FOR OVERSIZED USE BUSHING P/N: 450237808

NOTE 3 FOR OVERSIZED USE BUSHING P/N: 450237815

NOTE 4: MAXIMUM DIMENSION OF 1.224 IN FROM THE STAY CENTERLINE TO FACE -3A1- -3A2 PER REPAIR 3-4 MUST BE MAINTAINED.

NOTE 5: MAXIMUM DIMENSION OF 2.5157 IN FROM THE STAY CENTERLINE TO FACE -2A1- -2A2 PER REPAIR 3-4 MUST BE MAINTAINED.

A/C: A320

P/NAME: STAY, UPPER SIDE

P/NUMB.: 201163300

S/NUMB.: AP042106

LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.

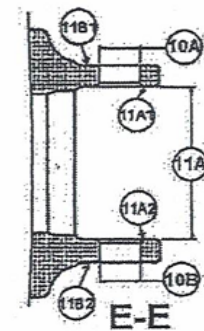
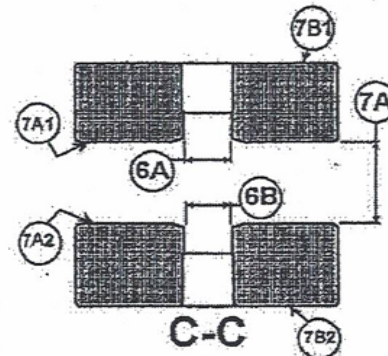
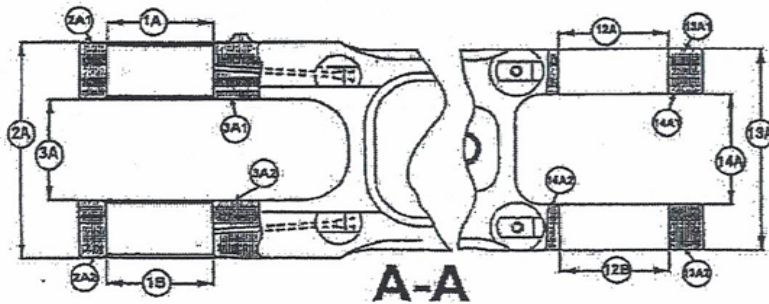
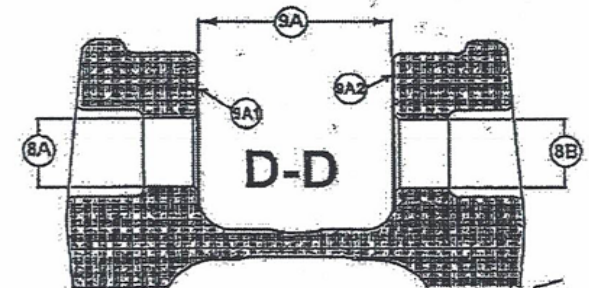
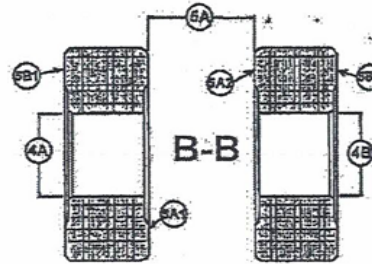
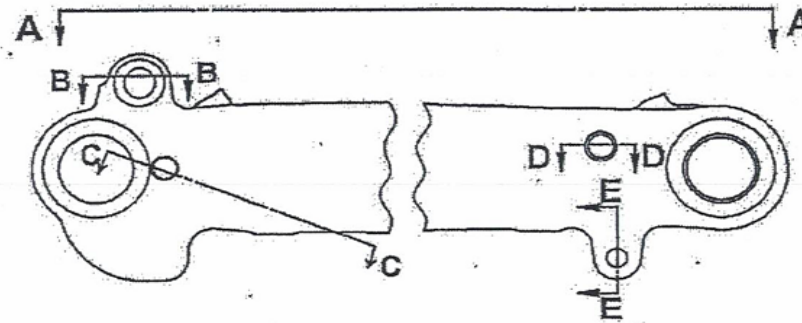
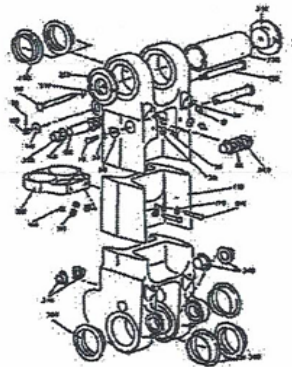
W/ORDER: 12615

OHM CHAP.: 32-11-26

SUB W/O: 12615-76

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: CARDAN, UPPER

W/ORDER: 12615

INSPECTION STATUS REPORT

P/NUMB.: 201163620

OHM CHAP.: 32-11-26



S/NUMB.: 08B0563X3

SUB W/O: 12615-78

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	3.7370	3.7130	3.7390	3.7390	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	3.7390	CHROME								
B	2.9500	2.9260	2.9515	2.9515	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	2.9520	CHROME								
C		2.5200 OHM	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
D	THREADS		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
1A	2.4010	2.4854	2.4020	2.4500	LOCAL CORROSION	MACHINE	2.2050	201057636	0.0030	2.4530
	REF.	NOTE 2					2.2060			
2A	2.0880	2.0060	2.0870	2.0870	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3580	201057636		
	REF.						2.3600			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201057636		0.136
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201057636		0.136
3A	9.9170		9.9200	9.9200	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	9.9250									
4A	2.3250	NOTE 1	2.3250	2.3250	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	2.3280									
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
5A		NOTE 4	0.3840		INCORPORATE S.B	WITHIN OHM LIMITS				
5B			0.3840		INCORPORATE S.B	WITHIN OHM LIMITS				
6A	INTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
8A	A/D	SEE NOTE 4	YES (X)		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	2015-22-08		NO ()							
9A	OTHER REJECTIONS	SEE NOTE 3	COND. ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: CARDAN, UPPER

W/ORDER: 12615

P/NUMB.: 201163620

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 08B0563X3

SUB W/O: 12615-78



LIF. LIMIT: 60,000

C.S.N.: 16,758

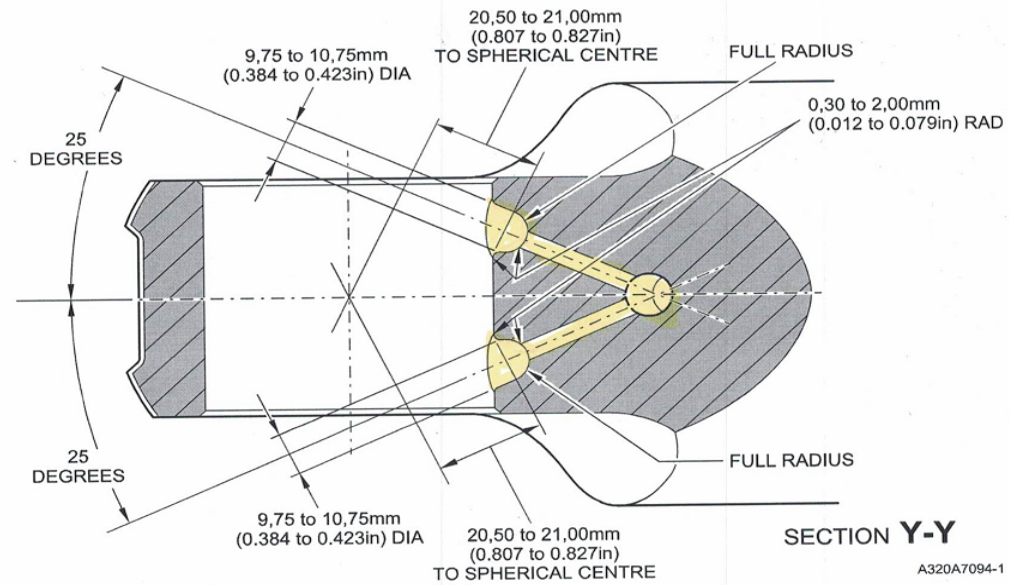
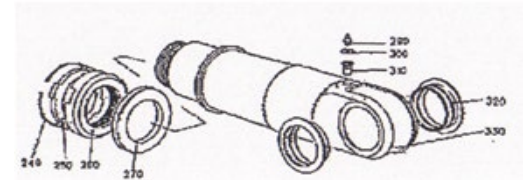
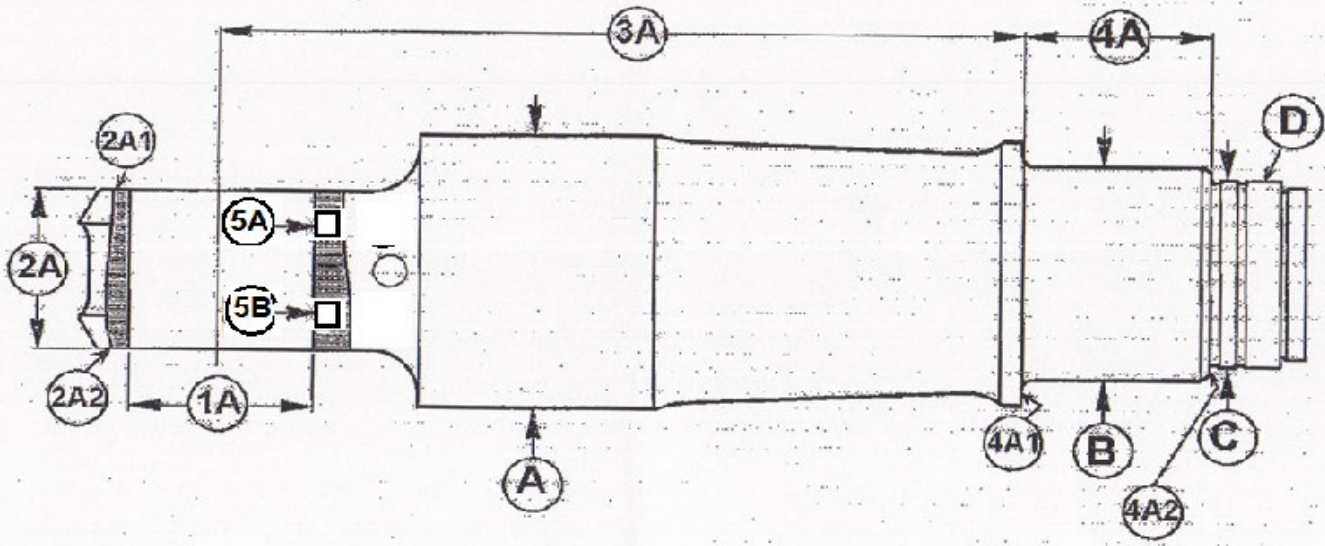
- NOTE 1:** IF LOCATION -4A1- REQUIRES MACHINE, MAINTAIN -4A- DIMENSION TO 2.3318 / 2.336, USE CHROME PLATE PER REPAIR 2-7

IF LOCATION -4A2- REQUIRES MACHINE, MAINTAIN -4A- DIMENSION TO 2.308 MINIMUM, METAL SPRAY (TUNGSTEN CARBIDE, UCAR LW-IN40)
- NOTE 2:** MACHINE DIAMETER -A- JUST SUFFICIENTLY TO REPAIR THE DAMAGE OR WEAR, TO NOT EXCEED A MINIMUM DIAMETER OF 2.323. THE SURFACE FINISH MUST BE 63 MICRO-INCHS.
50 % OF THE SPLINES (94 IN TOTAL, MAY BE REMOVED PROVIDED THAT NOT MORE THAN 33% (61 IN TOTAL) ARE REMOVED FROM ANY ONE 180 DEGREES ARC.
- NOTE 3:** (FOR UAL & TURKISH TECHNIC ONLY) "INCORPORATE S/B 200-32-312 FOR CUSTOMERS REQUIREMENTS MANDATOREY"
- NOTE 4:** INCORPORATE A/D 2015-22-08 IT OPENS AND REDUces THE LUBE HOLES -5A- AND -5B- FROM 0.384 TO 0.423 IT ALSO INCREASES THE DEPTH OF THE HOLE FROM 0.807 TO 0.827 (SEE S/B 200-32-312 ,FAA A/D 2015-22-08 OR INPECTION DRAWING SEE DRAWING FOR INFORMATION)
NEW P/N: 201163948

A/C: A320
 P/NAME: CARDAN, UPPER
 P/NUMB.: 201163620
 S/NUMB.: 08B0563X3
 LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE, INC.
 W/ORDER: 12615
 OHM CHAP.: 32-11-26
 SUB W/O: 12615-78
 C.S.N.: 16,758

INSPECTION STATUS REPORT



Upper Cardan - 201163948 - After Modification

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: LINK, BANANA

W/ORDER: 12615

P/NUMB.: 201661306

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 08AP0114

SUB W/O: 12615-179



LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	0.6319	NOTE 1	0.6323	0.6323	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-06M	0.0015	0.634
	0.6323						0.4737			
2A		2.0480	0.5100	0.5100	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6192	BCA2-12-06M		
		NOTE 2					0.6283			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
3A	0.6319	NOTE 1	0.6323	0.6323	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-06M	0.0015	0.634
3B	0.6323						0.4737			
4A			1.6130	1.6130	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.7220	BCA2-12-06M		
							1.7317			
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
5A			0.8280	0.8280	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7083	BCA2-12-06M		
							0.7184			
5A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.060
5A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.060
6A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7A	OTHER REJECTIONS		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: LINK, BANANA

W/ORDER: 12615

P/NUMB.: 201661306

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 08AP0114

SUB W/O: 12615-179



LIF. LIMIT: 60, 000

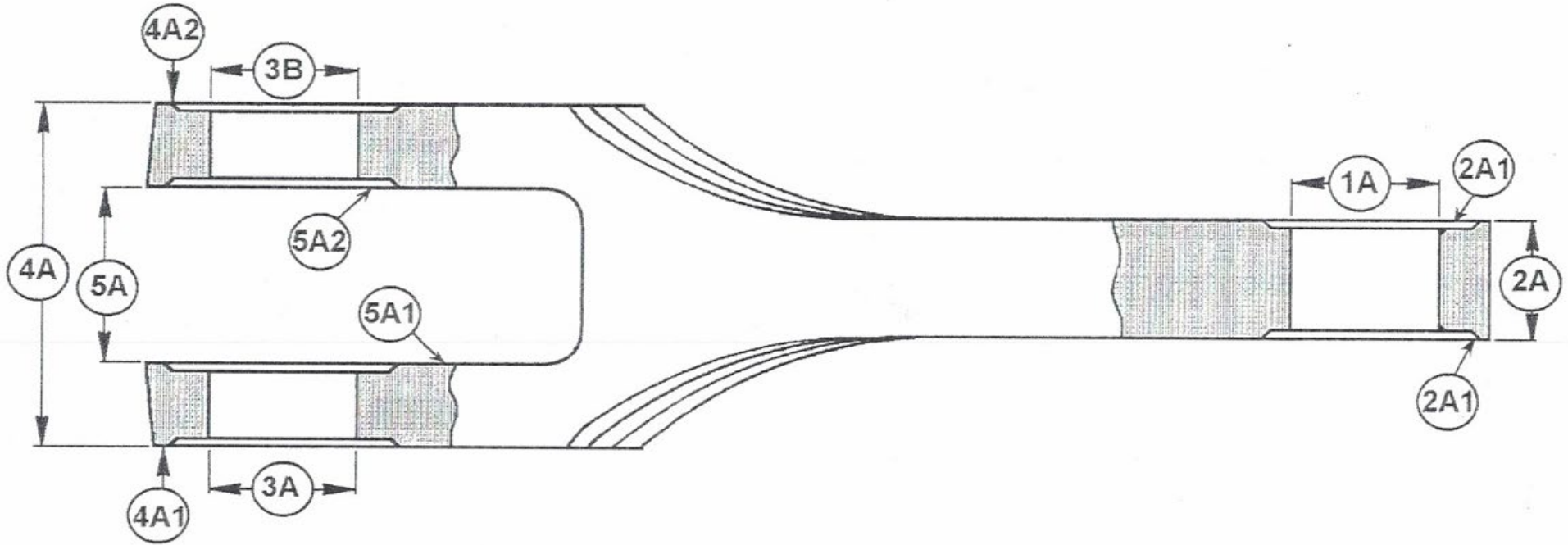
C.S.N.: 16,758

NOTE 1: IF BORE OVERSIZED OR REQUIRES MACHINING SCRAP THE UNIT

A/C: A320
P/NAME: LINK, BANANA
P/NUMB.: 201661306
S/NUMB.: 08AP0114
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE, INC.
W/ORDER: 12615
OHM CHAP.: 32-11-26
SUB W/O: 12615-179
C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: LINK, BANANA

W/ORDER: 12615

P/NUMB.: 201661306

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 08AP0121

SUB W/O: 12615-180



LIFE LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	0.6319	NOTE 1	0.6323	0.6323	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-06M	0.0015	0.634
	0.6323						0.4737			
2A		2.0480	0.5100	0.5100	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6192	BCA2-12-06M		
		NOTE 2					0.6283			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
3A	0.6319	NOTE 1	0.6323	0.6323	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-06M	0.0015	0.634
3B	0.6323						0.4737			
4A			1.6130	1.6130	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.7220	BCA2-12-06M		
							1.7317			
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
5A			0.8280	0.8280	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7083	BCA2-12-06M		
							0.7184			
5A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.060
5A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.060
6A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7A	OTHER REJECTIONS		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: LINK, BANANA

W/ORDER: 12615

P/NUMB.: 201661306

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 08AP0121

SUB W/O: 12615-180



LIF. LIMIT: 60, 000

C.S.N.: 16,758

NOTE 1: IF BORE OVERSIZED OR REQUIRES MACHINING SCRAP THE UNIT

A/C: A320

P/NAME: LINK, BANANA

P/NUMB.: 201661306

S/NUMB.: 08AP0121

LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE, INC.

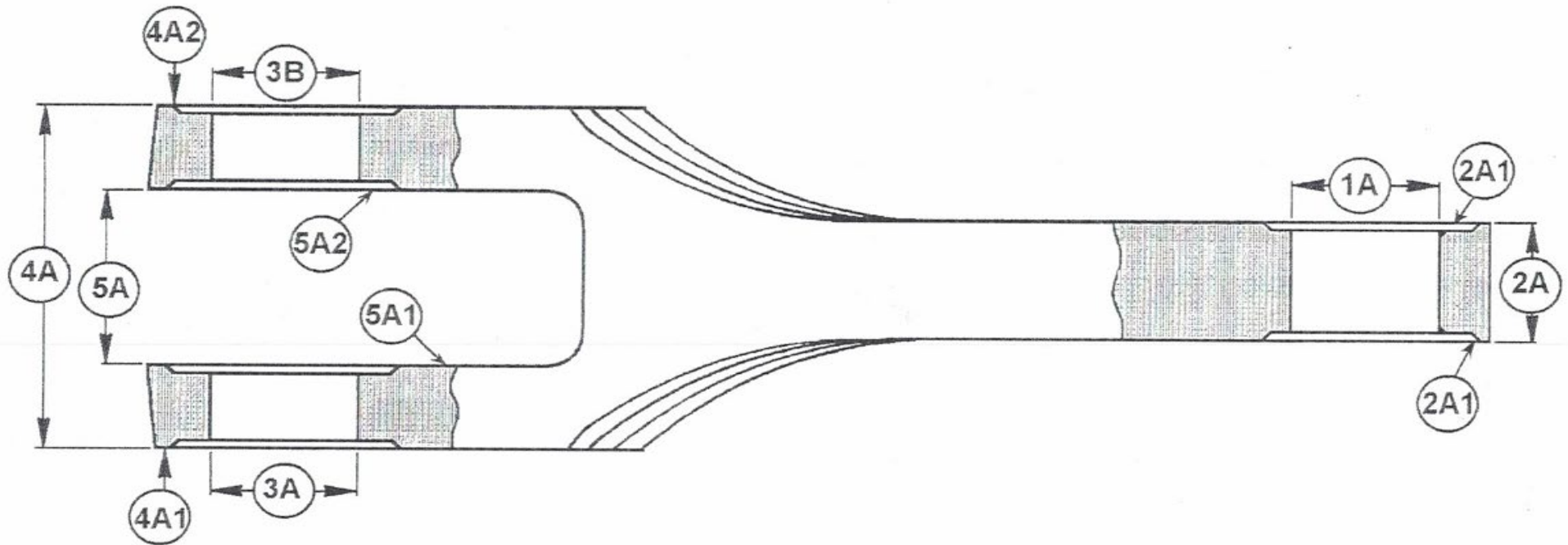
W/ORDER: 12615

OHM CHAP.: 32-11-26

SUB W/O: 12615-180

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: MLG SLIDING TUBE

W/ORDER: 12615

P/NUMB.: 201371304

OHM CHAP.: 32-11-33 / 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08B5178X9131

SUB W/O: 12615-1



LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	6.9950	6.9730	6.9960		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	6.9970		6.9965							
A1	6.8471	SUBMIT	6.8480		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	6.8487									
B	6.5355	REFER OHM FIG 800'S	6.5370		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	6.5379									
C	5.3489	5.3700	5.3500		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	5.3512									
D	NON WORKING ID	REFER OHM	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
E	THREADS	REFER OHM	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
F					WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
G	4.6822	4.6596	4.6830		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
H	4.6836	NOTE 6	4.6830		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
I	4.8390	4.8168	4.9400		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
J	4.8408	NOTE 6	4.9400		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
K	4.9565	4.9349	4.9575		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
L	4.9589	NOTE 6	4.9575		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
M	FACES	REFER OHM	COND.		CHROME BREAK	STRIP				
N			ONLY		CHROME BREAK	STRIP				
1A	AXLE I/D	REFER OHM	COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
1B			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
2A	0.2559	NOTE 5	0.2610	0.2610	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
2B	0.2638		0.2610	0.2610	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
2C	0.2559	NOTE 5	0.2610	0.2610	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
2D	0.2638		0.2610	0.2610	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
3A	1.5770	REFERENCE	1.5770	1.5770	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	1.5800	DIMENSION								
3B	1.3820	REFERENCE	1.3820	1.3820	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	1.3830	DIMENSION								
4A	1.8110	1.8947	1.8120	1.8120	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.3976	201383654	0.0010	1.8130
	1.8120						1.4016	NOTE 3		

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: MLG SLIDING TUBE

W/ORDER: 12615

INSPECTION STATUS REPORT

P/NUMB.: 201371304

OHM CHAP.: 32-11-33 / 32-12-25

Component Overhaul
Services

S/NUMB.: 08B5178X9131

SUB W/O: 12615-1

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
5A	5.7030	5.6650	5.7040	5.7040	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.9416	201383650		
		NOTE 1					5.9432			
5A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201383650		0.1188
5A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201383650		0.1188
5B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
5B2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
6A	2.5984	2.6822	2.5990	2.5990	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3622	201383650	0.0030	2.6020
6B	2.5996		2.5990	2.5990	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3634	201383650	0.0030	2.6020
7A	5.1150	5.0296	5.1120	5.1120	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.3142	201371617		
	REF.						5.3285			
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1020	201371617		0.1011
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1020	201371617		0.1011
8A	1.0650	1.1229	1.0650	1.0650	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.8661	201371617	0.0015	1.0665
8B	REF. DIM.		1.0650	1.0650	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.8674	201371617	0.0015	1.0665
9A	0.6693	0.0731	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6710
9B	0.6704	NOTE 4	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6710
9C	0.7559	0.8294	0.7570	0.7570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6590	201056703	0.0010	0.7580
	0.7572	NOTE 2					0.6791			
9D	0.6693	0.0731	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6710
	0.6704	NOTE 4					0.5916			
9E	0.7559	0.8294	0.7570	0.7570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6590	201056703	0.0010	0.7580
	0.7572	NOTE 2					0.6791			
9F	0.6693	0.0731	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6710
9G	0.6704	NOTE 3	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6710
9H	0.6693	0.0731	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6710
9I	0.6704	NOTE 4	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6710
9J	0.7559	0.8294	0.7570	0.7570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6590	201056703	0.0010	0.7580
	0.7572	NOTE 2					0.6791			
9K	0.6693	0.0731	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6710
9L	0.6704	NOTE 3	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6710
10A	0.6693	0.0731	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6710
10B	0.6704	NOTE 4	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6710

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: MLG SLIDING TUBE

W/ORDER: 12615

INSPECTION STATUS REPORT

P/NUMB.: 201371304

OHM CHAP.: 32-11-33 / 32-12-25



S/NUMB.: 08B5178X9131

SUB W/O: 12615-1

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
10C	0.7559	0.8294	0.7570	0.7570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6590	201056703	0.0010	0.7580
	0.7572	NOTE 2					0.6791			
10D	0.6693	0.0731	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6710
	0.6704	NOTE 4					0.5916			
10E	0.7559	0.8294	0.7570	0.7570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6590	201056703	0.0010	0.7580
	0.7572	NOTE 2					0.6791			
10F	0.6693	0.0731	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6710
10G	0.6704	NOTE 4	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6710
10H	0.6693	0.0731	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6710
10I	0.6704	NOTE 4	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6710
10J	0.7559	0.8294	0.7570	0.7570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6590	201056703	0.0010	0.7580
	0.7572	NOTE 2					0.6791			
10K	0.6693	0.0731	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6710
10L	0.6704	NOTE 4	0.6700	0.6700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6710
11A	VERIFY	S/N NOT LISTED	YES ()		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	NOTE 5	ON S/B	NO ()							
12A	VERIFY	S/N NOT LISTED	YES ()		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	NOTE 6	ON S/B	NO ()							
13A	INTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
14A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
15A	OTHER REJECTIONS		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: MLG SLIDING TUBE

W/ORDER: 12615

P/NUMB.: 201371304

OHM CHAP.: 32-11-33 / 32-12-25

INSPECTION STATUS REPORT



Component Overhaul
Services

S/NUMB.: 08B5178X9131

SUB W/O: 12615-1

LIF. LIMIT: 60,000

C.S.N.: 16,758

NOTE 1: ► DO NOT MACHINE MORE THAN 0.025 FROM EACH FACE. LUG THICKNESS MUST NOT BE LESS THAN 1.415

NOTE 2: ► IF BORE IS IN DESIGN DIMENSION INSTALL BUSHING P/N: 201056703, IF BORE IS O/S OR REQUIRES MACHINING, MACHINE AND INSTALL BUSHINGS IWA CMM 32-12-25 REPAIR 9-9 POAGE 602, AND ORDER THE CORRECT BUSHING

NOTE 3: ► IF NECESSARY, MACHINE FACE TO REMOVE DAMAGE OR WEAR TO A MINIMUM DIMENSION OF: 3.063, SHOWN IN CMM REPORT 9-10 (CENTER OF AXLE BORE TO FACE) WITH A SURFACE FINISH 63 MICROINCHES THEN RESTORE CHAMFER. INSTALL O/S BUSHING AND MACHINE BUSHINGS TO RESTORE DIMENSION SHOWN IN CMM 3.268 / 3.287 (CENTER OF AXLE BORE TO FACE)

NOTE 4: ► IF BORE IS IN DESIGN DIMENSION INSTALL BUSHING P/N: 201056702, IF BORE IS O/S OR REQUIRES MACHINING, MACHINE AND INSTALL BUSHINGS IWA CMM 32-12-25 REPAIR 9-3 PAGE 601, AND ORDER THE CORRECT BUSHING

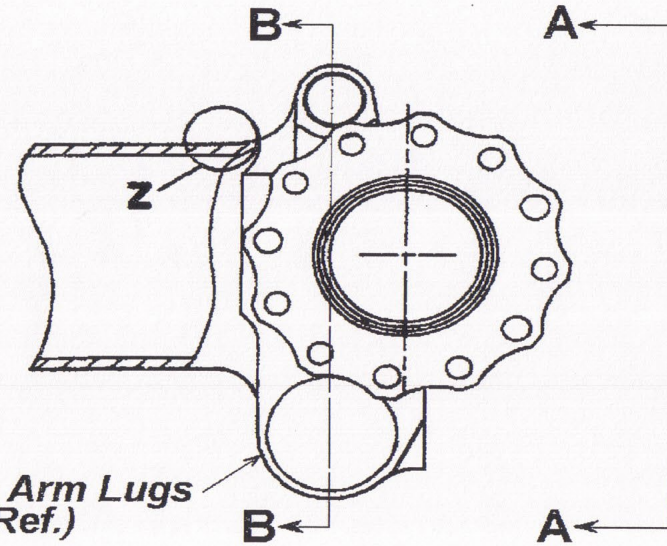
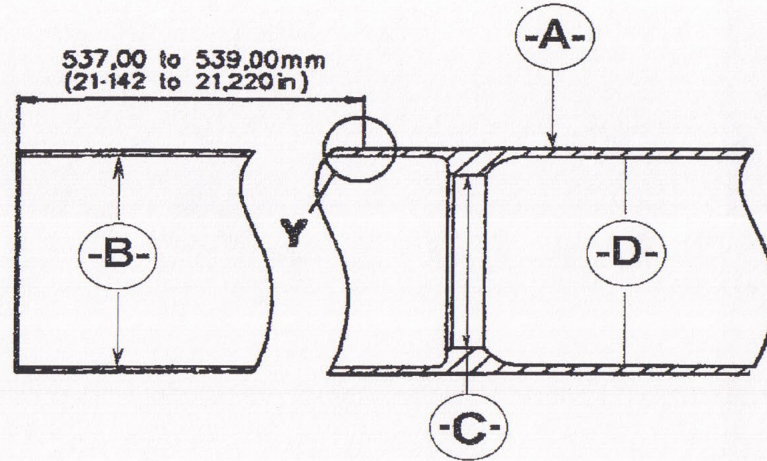
NOTE 5: ► ***** MANDATORY ***** INSPECT COMPONENT AND VERIFY COMPLIANCE WITH (SAFRAN S/B: 200-32-231) AND (AIRBUS S/B A320-32-1441), VERIFY BORES ARE FREE OF DEFECTS (SERVICE LIMITS IS 0.2724)

NOTE 6: ► ***** MANDATORY ***** PER (SAFRAN S/B: 200-32-313) VERIFY PISTON S/N IS NOT LISTED ON THE LIST IN THE S/B 200-32-313
** IF S/N IS LIST ON THE S/B 200-32-313 (CONTACT MESSIER-DOWTY) AND ONLY MESSIER DOWTY QUALIFIED PERSONNEL WILL INSPECT **

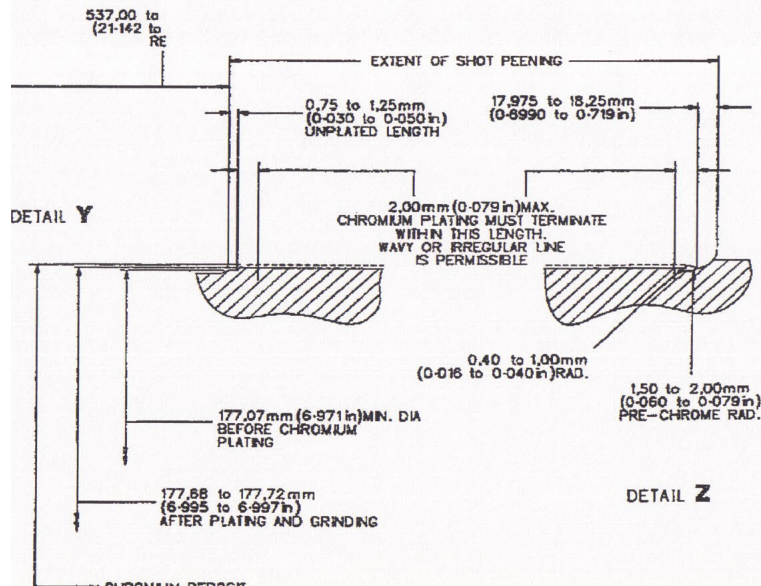
A/C: A320
 P/NAME: MLG SLIDING TUBE
 P/NUMB.: 201371304
 S/NUMB.: 08B5178X9131
 LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE, INC.
 W/ORDER: 12615
 OHM CHAP.: 32-11-33 / 32-12-25
 SUB W/O: 12615-1
 C.S.N.: 16,758

INSPECTION STATUS REPORT



Torque Arm Lugs
 (Ref.)



DRAWING 1 OF 2

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: MLG SLIDING TUBE

W/ORDER: 12615

P/NUMB.: 201371304

OHM CHAP.: 32-11-33 / 32-12-25

INSPECTION STATUS REPORT



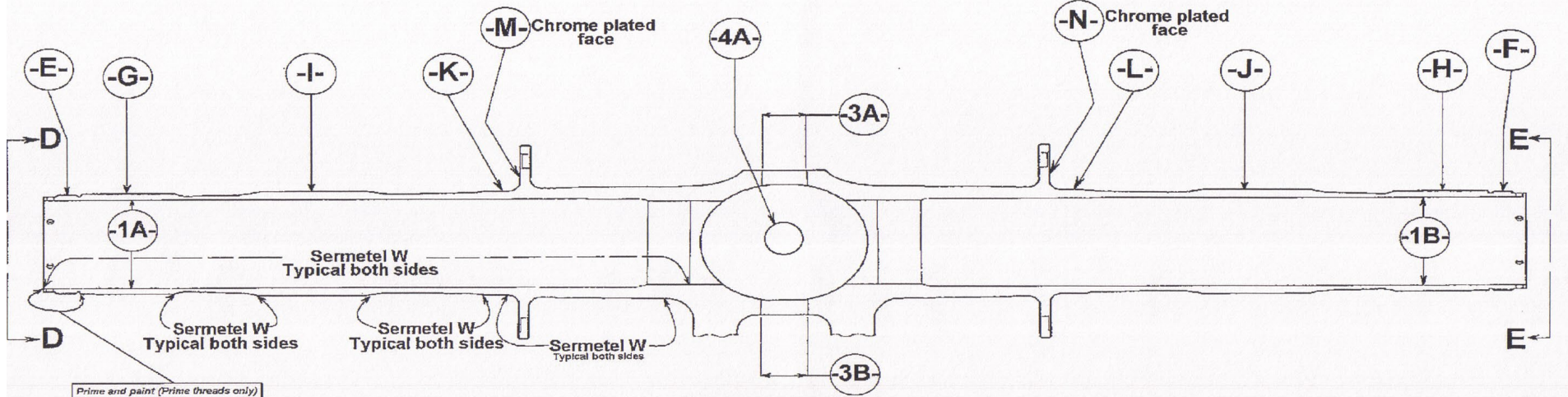
Component Overhaul Services

S/NUMB.: 08B5178X9131

SUB W/O: 12615-1

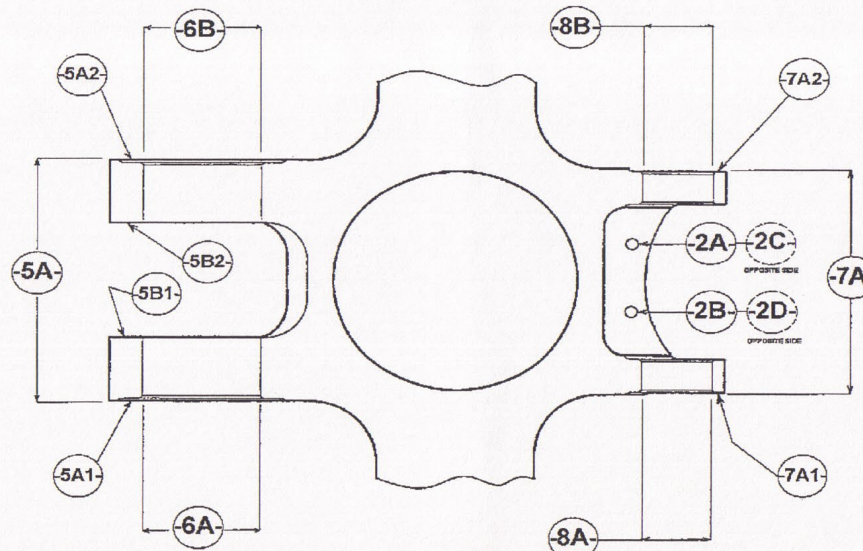
LIF. LIMIT: 60, 000

C.S.N.: 16,758



TORQUE ARM LUGS
DOWN

A - A



R - R

DRAWING 2 OF

A/C: A320

P/NAME: MLG SLIDING TUBE

P/NUMB.: 201371304

S/NUMB.: 08B5178X9131

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE, INC.

W/ORDER: 12615

OHM CHAP.: 32-11-33 / 32-12-25

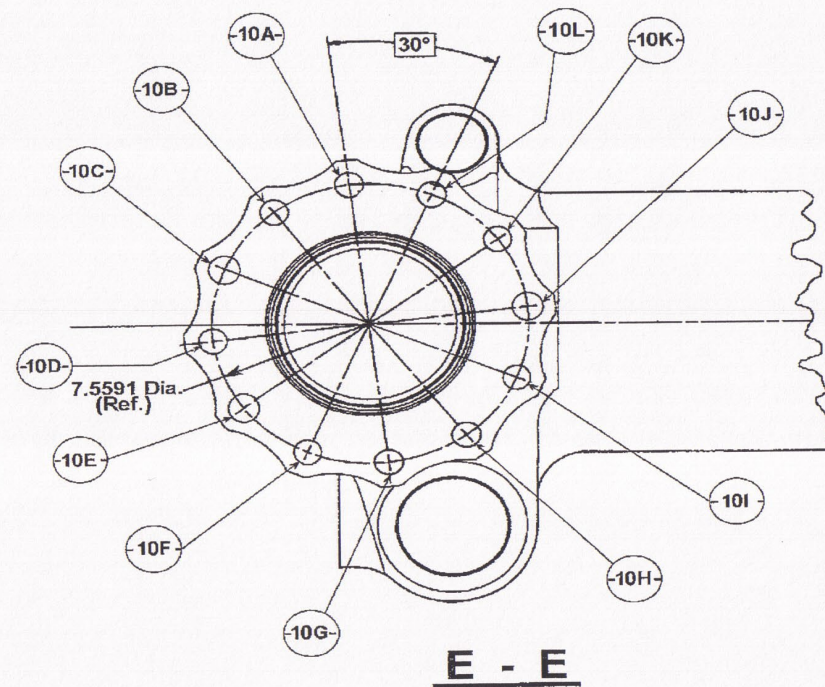
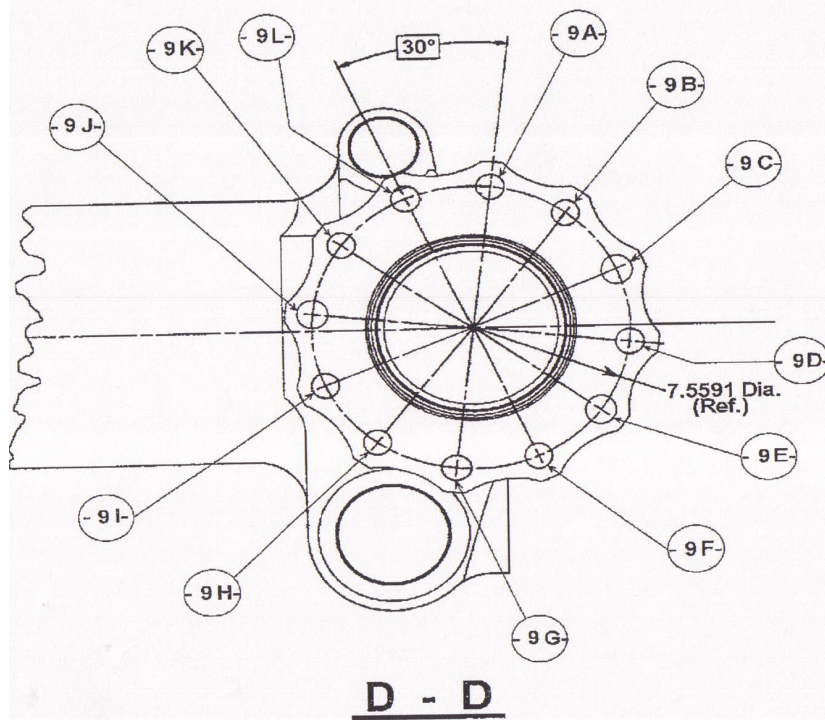
SUB W/O: 12615-1

C.S.N.: 16,758

INSPECTION STATUS REPORT



Component Overhaul
Services



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: TUBE, UPPER DIAPHRAGM

W/ORDER: 12615

INSPECTION STATUS REPORT

P/NUMB.: 201371615

OHM CHAP.: 32-11-33



S/NUMB.: 08B0314X2

SUB W/O: 12615-132

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	S/B 200 32-247		YES (X) NO ()		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	1.4173 1.4183	1.4860	1.4200	1.4200	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	1.2224 1.2234	201522668	0.0010	1.4210
2A	3.3410 REF.	3.3267	3.3410	3.3410	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	3.4803 3.4842	201522668 NOTE 2		
	2A1 2A2	FACES	COND. ONLY		WITHIN DESIGN LIMITS WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS WITHIN DESIGN LIMITS	0.140	201522668		0.139
3A	0.7460 REF.	0.6590	0.7460	0.7460	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
							0.060			0.0000
3A1	FACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.060			0.0000
4A	2.0079 2.0091	2.0926	2.0100	2.0100	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	1.8110 1.8126	201522669	0.0010	2.0110
5A	2.2400 REF.		2.2400	2.2400	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	5A1 5A2	FACES	COND. ONLY		WITHIN DESIGN LIMITS WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS WITHIN DESIGN LIMITS		201522669		-1.1200
6A	2.6340 2.6380	2.6570	2.6340	2.6340	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
6A1	FACE	SEE CMM	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7A	7.2195 7.2209	7.1610	7.2200	7.2200	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
		NICKEL REPAIR								
8A	6.9803 6.9828	6.9230	6.9820	6.9820	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
		NICKEL REPAIR								
9A	7.2202 7.2220	7.2140	7.2200	7.2200	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
		NICKEL REPAIR								
10A	6.7444 6.7468	6.8870	6.7454	6.7454	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
		MINMUM LUG								
11A	7.2202 7.2220	WEAR LIMITS 7.2140	7.2200	7.2200	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
12A	6.9840 REF.	6.9790	6.9840	6.9840	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
		SEE CMM								

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: TUBE, UPPER DIAPHRAGM

W/ORDER: 12615

P/NUMB.: 201371615

OHM CHAP.: 32-11-33

INSPECTION STATUS REPORT

S/NUMB.: 08B0314X2

SUB W/O: 12615-132



LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
13A	0.3974	0.3981	0.3960	0.3960	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	0.3961	WEAR LIMITS								
15A	INTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	SURFACE		ONLY							
16A	EXTERNAL	6.0000	COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	SURFACE		ONLY							
17A	NOTES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
			ONLY							
18A	0.3620	0.3071	0.3600	0.3600	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.									
19A	4.9016		4.9015	4.9015	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF,									
20A	0.3150	0.2750	0.3170	0.3170	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.									
21A	0.3947		0.3960	0.3960	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	.3961 REF									
22A	0.8071		0.8070	0.8070	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	AS REF.									
23A	0.3800	0.3940	0.3850	0.3850	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	RER.	MINIMUM								
24A	0.9940	1.0240	1.0000	1.0000	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	1.0140	MAX								
24A1	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
			ONLY							
25A	1.1940	1.1810	1.1930	1.1930	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.	MINIMUM								
26A	1.4070	1.4370	1.4150	1.4150	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	1.427	MAX								
27A	0.2126		0.2160	0.2160	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	0.2165									
27A1	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
			ONLY							
28A	0..579		0.5850	0.5850	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	0..589									

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: TUBE, UPPER DIAPHRAGM

W/ORDER: 12615

P/NUMB.: 201371615

OHM CHAP.: 32-11-33

INSPECTION STATUS REPORT



S/NUMB.: 08B0314X2

SUB W/O: 12615-132

Component Overhaul
Services

LIF. LIMIT: 60, 000

C.S.N.: 16,758

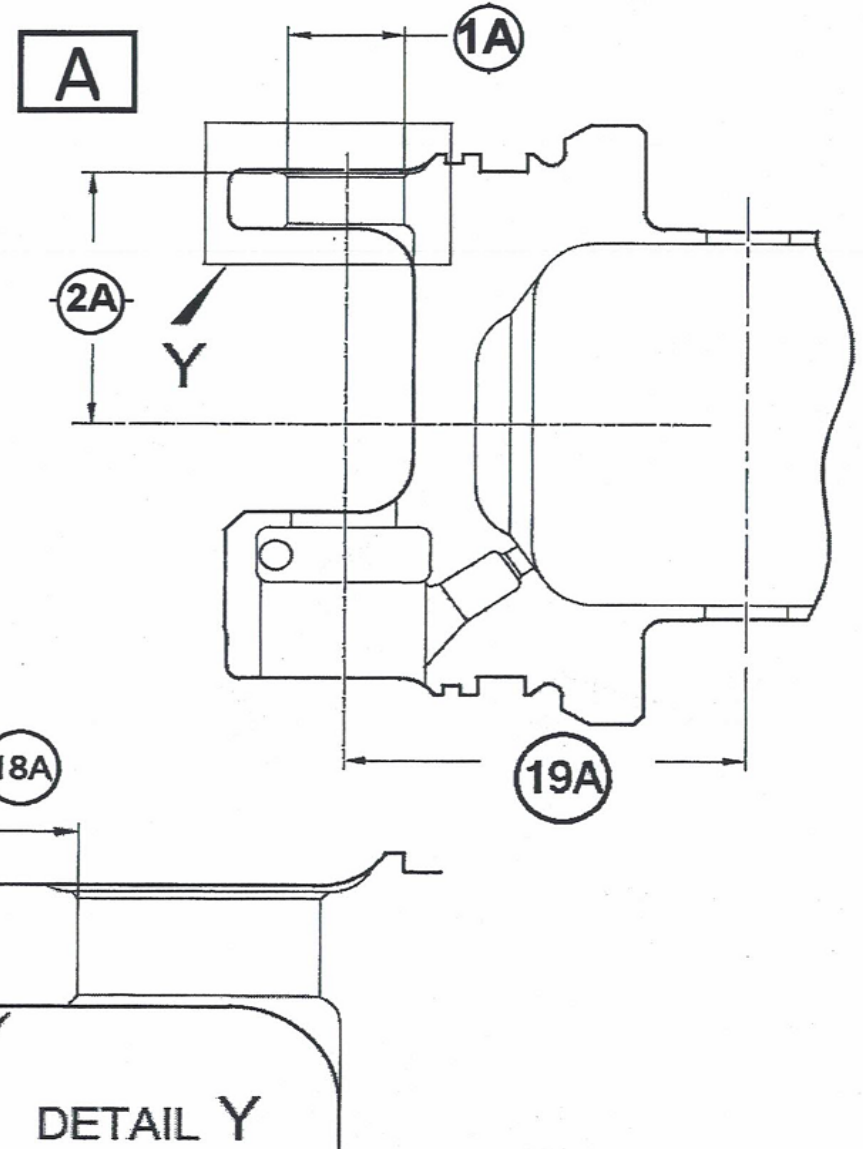
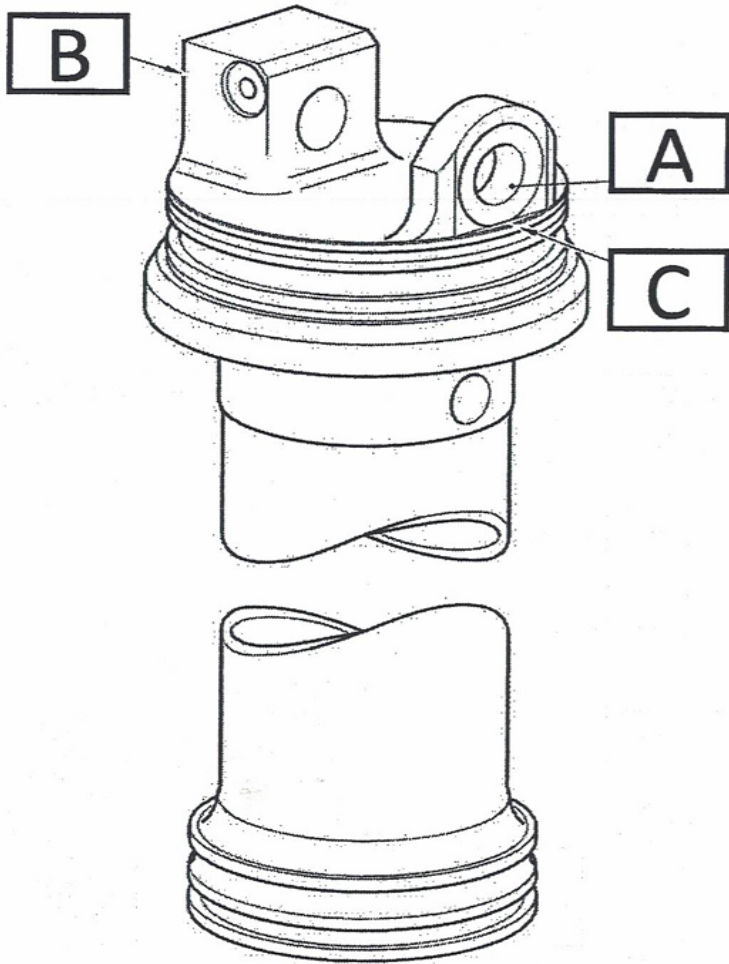
NOTE 1: ▶ PRE S/B 200-32-247 MADE FROM ALUMINUM
▶ POST S/B 200-32-247 MADE FROM STEEL

NOTE 2: ▶ TO OBTAIN DIMENSION -2A- THE PART MUST BE MESURED FROM THE CENTER LINE OF THE TUBE TO THE START OF THE OUTER CHAMFER OF THE BUSHING FLANGE PER CMM 32-11-33 REPAIR NO. 12-1 FIGURE 602 OR CMM 32-12-25 REPAIR NO. 15-1 FIGURE 602

A/C: A320
P/NAME: TUBE, UPPER DIAPHRAGM
P/NUMB.: 201371615
S/NUMB.: 08B0314X2
LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12615
OHM CHAP.: 32-11-33
SUB W/O: 12615-132
C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: TUBE, UPPER DIAPHRAGM

W/ORDER: 12615

P/NUMB.: 201371615

OHM CHAP.: 32-11-33

INSPECTION STATUS REPORT



Component Overhaul
Services

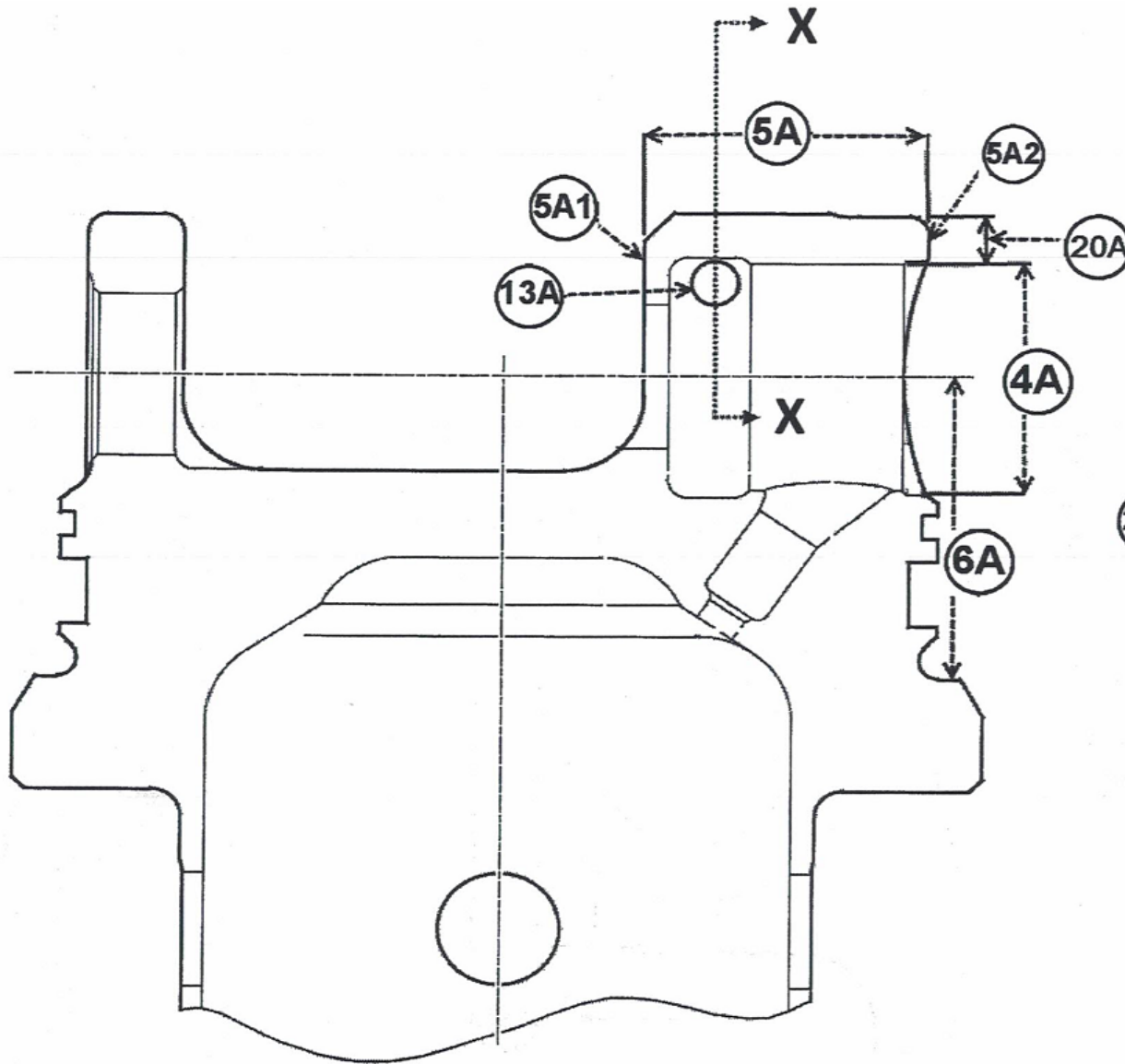
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SUB W/O: 12615-132

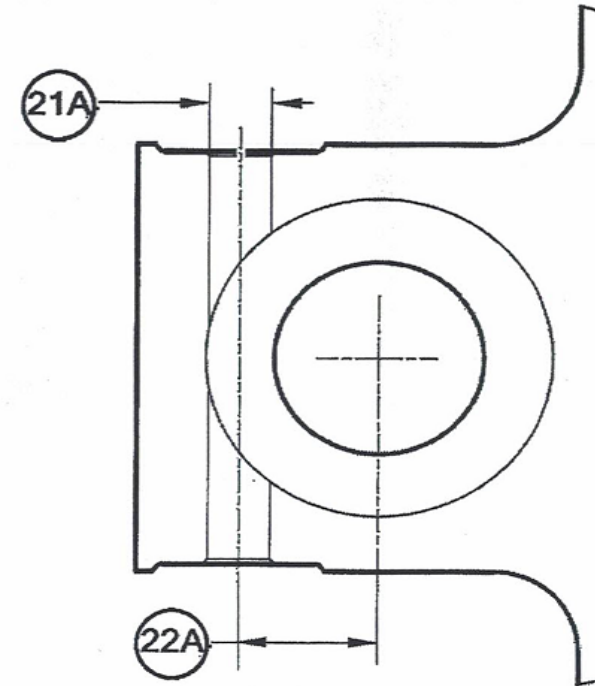
LIF. LIMIT: 60,000

C.S.N.: 16,758

B



SECTION **X - X**



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: TUBE, UPPER DIAPHRAGM

W/ORDER: 12615

P/NUMB.: 201371615

OHM CHAP.: 32-11-33

S/NUMB.: 08B0314X2

SUB W/O: 12615-132

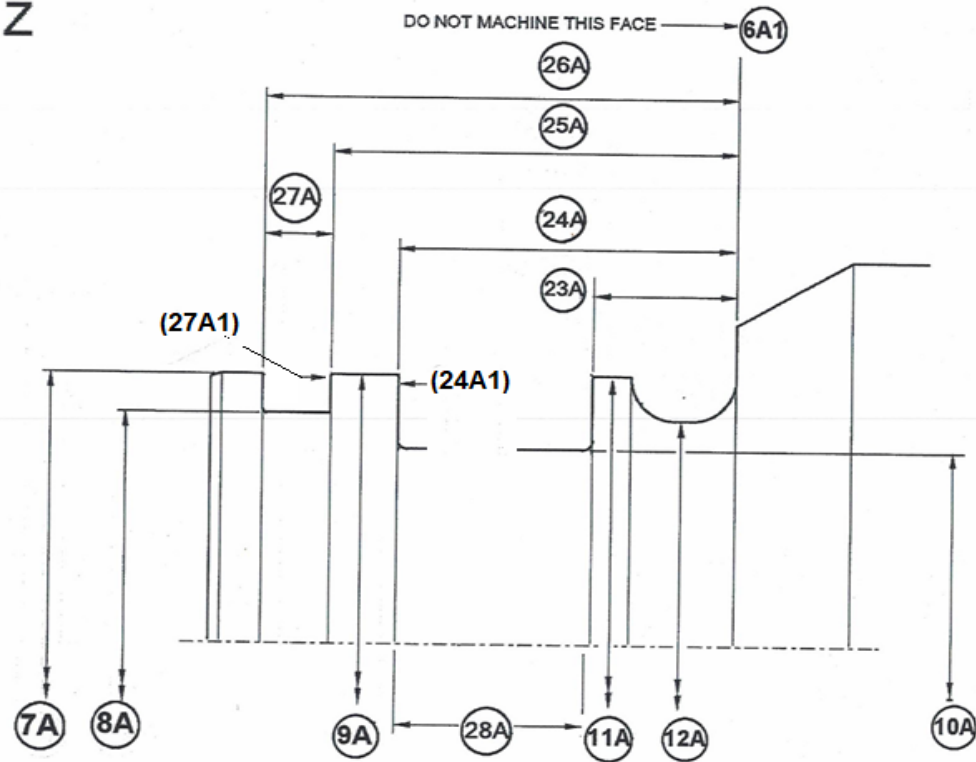
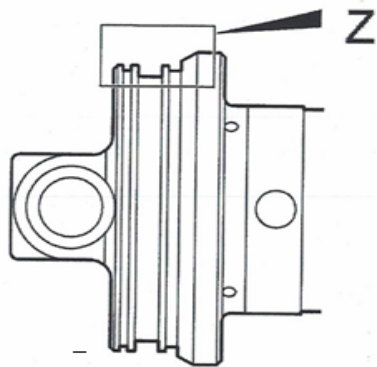
LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT



C



DETAIL Z

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: CARDAN, LOWER

W/ORDER: 12615

P/NUMB.: 201163307

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 08B0893X1767

SUB W/O: 12615-58



LIFE LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	2.7541	2.7300	2.7544	2.7544	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	2.7547	CHROME								
B	2.7530	2.7300	2.7544	2.7544	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	2.7547	CHROME								
C		2.3230 NOTE 4	2.3600	2.3600	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
D	THREADS		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
1A	2.4037	2.4850	2.4040	2.4040	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.0866	201163624	0.0030	2.4070
	2.4044	NOTE 1					2.0878			
2A		2.0480	2.0870	2.0870	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.2810	201163624		
		NOTE 2					2.2823			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201163624		0.0970
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201163624		0.0970
3A	5.4340	NOTE 1	5.4340	5.4340	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	5.4350	NOTE 2								
3A1	FACES	CHROME	COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
3A2		NOTE 3	ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
4A		0.4690	0.4690	0.4690	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
4B		0.4920	0.4920	0.4920	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
5A	INTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
6A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7A	OTHER REJECTIONS		COND. ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: CARDAN, LOWER

W/ORDER: 12615

P/NUMB.: 201163307

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT



S/NUMB.: 08B0893X1767

SUB W/O: 12615-58

LIF. LIMIT: 60, 000

C.S.N.: 16,758

NOTE 1: ► METAL SPRAY REPAIR 1-5 (MAKE SURE TO ONLY MACHINE GRIP FACE -3A2- WHEN REPAIRING THIS DIMENSION, DO NOT EXCEED THE REPAIR LIMITS GIVEN OF: 5.4150)

NOTE 2: ► CHROME PLATE REPAIR 1-4 (MAKE SURE TO ONLY MACHINE GRIP FACE -3A1- WHEN REPAIRING THIS DIMENSION, DO NOT EXCEED THE REPAIR LIMIT GIVEN OF 5.445)

***** FOR
UAL REFERENCE DIMENSION IS TIGHTER (REFERENCE DIMENSION IS: 5.434 / 5.435)

NOTE 3: PARTS WHICH ARE REPAIRED PER CMM REPAIR SCHEME 450237620

ARE LIFE LIMITED TO A FURTHER 18,00 LANDINGS AND MUST NOT EXCEED A TOTAL LIFE LIMITS OF 39,500 LANDINGS

IF THE PART WAS INSTALLED IN A TWIN LUG MLG FITTING PRE SB 200-32-183 ASSIGN LIFE LIMITS PER ALS PQRT 1 AIRWORTHINESS LIMITATIONS ITEMS

NOTE 4: MACHINE DIAMETER -A- JUST SUFFICIENTLY TO REPAIR THE DAMAGE OR WEAR, TO NOT EXCEED A MINIMUM DIAMETER OF 2.323. THE SURFACE FINISH MUST BE 63 MICRO-INCHS.

50 % OF THE SPLINES (94 IN TOTAL, MAY BE REMOVED PROVIDED THAT NOT MORE THAN 33% (61 IN TOTAL) ARE REMOVED FROM ANY ONE 180 DEGREES ARC.

A/C: A320

P/NAME: CARDAN, LOWER

P/NUMB.: 201163307

S/NUMB.: 08B0893X1767

LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.

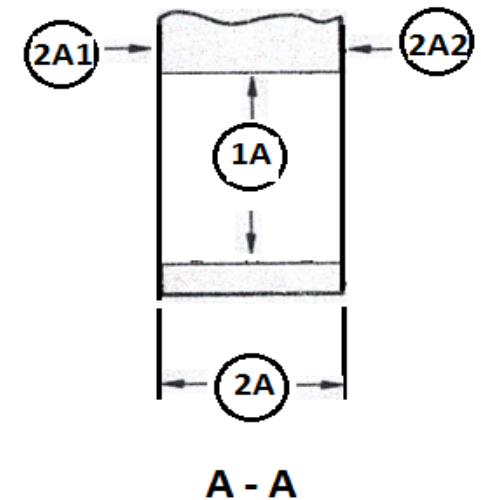
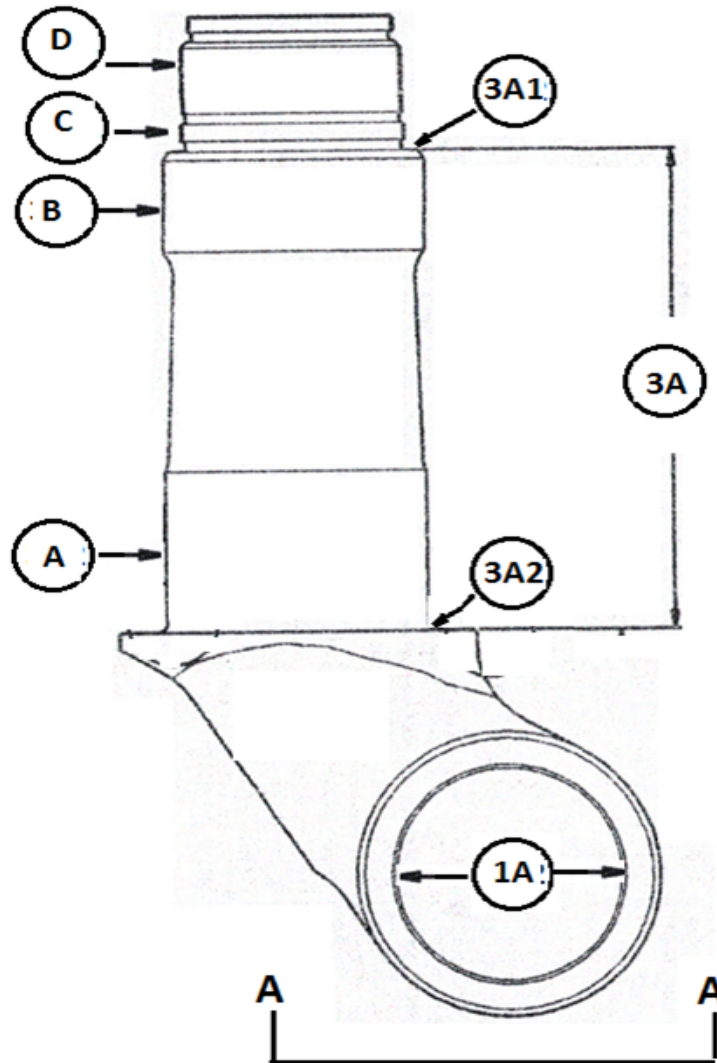
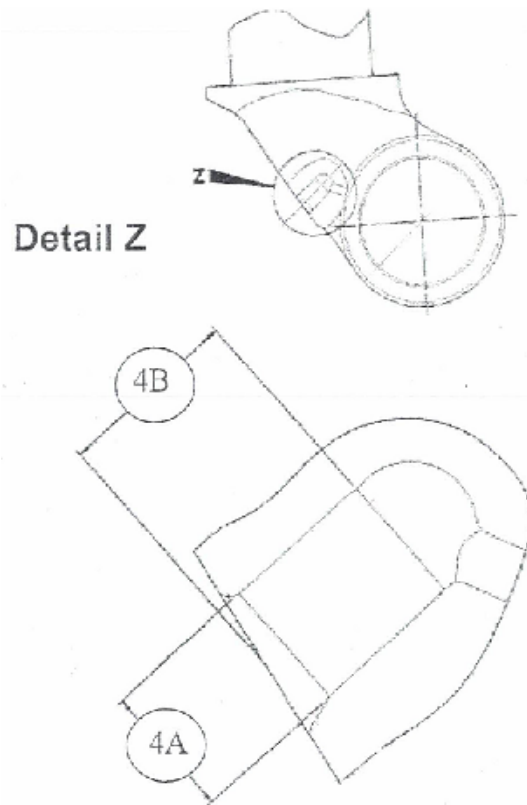
W/ORDER: 12615

OHM CHAP.: 32-11-26

SUB W/O: 12615-58

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: STAY, LOWER SIDE

W/ORDER: 12615

P/NUMB.: 201163301

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 07AP0223

SUB W/O: 12615-57


Component Overhaul
 Services

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	2.4016	2.4850	2.4020	2.4020	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2048	201163603	0.0020	2.404
1B	2.4028	NOTE1	2.4020	2.4020	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2059	201163603	0.0020	2.404
2A		7.7660	7.8230	7.8230	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
		NOTE 2								
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
3A		5.6772	5.6350	5.6350	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.4331	201163603		
		NOTE 2					5.4356			
3A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163603		0.101
3A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163603		0.101
4A		1.0490	1.0940	1.0940	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
4B			1.0940	1.0940	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
5A	2.4016	2.4850	2.4020	2.4020	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.0866	201163621	0.0020	2.404
5B	2.4028	NOTE 1	2.4020	2.4020	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.0878	201163621	0.0020	2.404
6A	2.6400	2.6840	2.6400	2.6400	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.4409	201163621		
	REF.						2.4429			
6A1	FACES	NOTE 2	COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163621		0.100
6A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163621		0.100
7A		1.0150	1.0380	1.0380	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7B			1.0380	1.0380	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
8A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
9A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTION		ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: STAY, LOWER SIDE

W/ORDER: 12615

P/NUMB.: 201163301

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 07AP0223

SUB W/O: 12615-57



LIF. LIMIT: 60,000

C.S.N.: 16,758

NOTE 1: FOR OVERSIZED

NOTE 2: FOR OVERSIZED FLNGE BUSHINGS

A/C: A320

P/NAME: STAY, LOWER SIDE

P/NUMB.: 201163301

S/NUMB.: 07AP0223

LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE, INC.

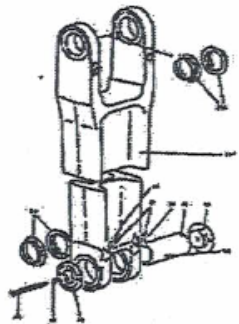
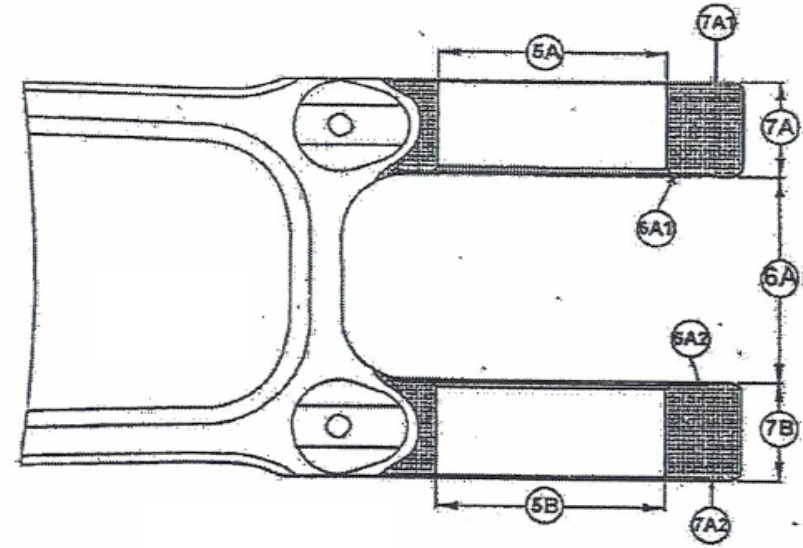
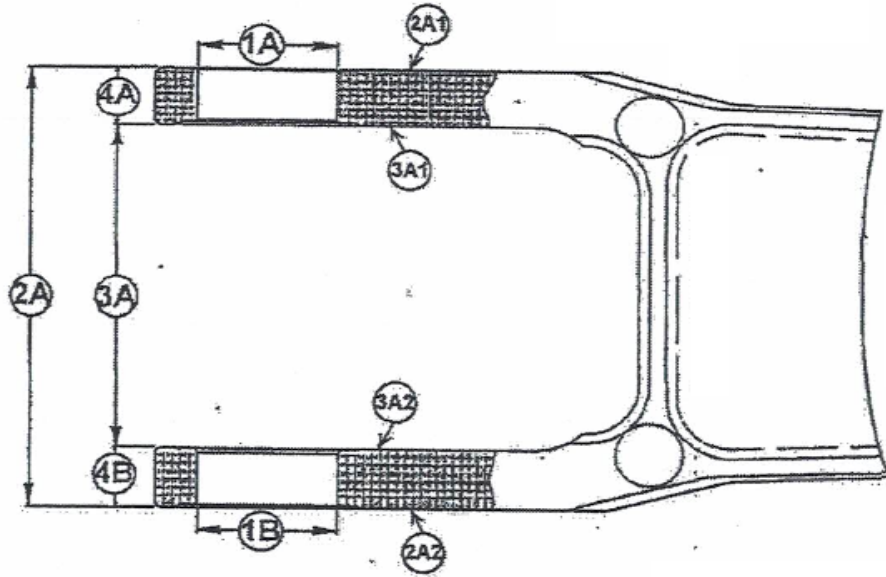
W/ORDER: 12615

OHM CHAP.: 32-11-26

SUB W/O: 12615-57

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: UPPER TORQUE LINK

W/ORDER: 12615

P/NUMB.: 201540303

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08BEL0061

SUB W/O: 12615-17



LIF. LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	1.6535	1.7293	1.6560	1.6560	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	1.4961	201522603	0.0020	1.6580
	1.6546	MAXIMUM					1.4970			
2A	1.6920	1.6500	1.6940	1.6940	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	1.8878	201522603		
	REF.	NOTE 1					1.8880			
2A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201522603		0.097
2A2										WITHIN DESIGN LIMITS
3A	5.1380	5.1870	5.1380	5.1380	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	5.1359			
	REF.						5.1397			
4A	2.5984	2.6820	2.6010	2.6010	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.3622	201056605	0.0030	2.6040
							2.3634			
4B	2.5996	2.6820	2.6010	2.6010	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.3622	201056605	0.0030	2.6040
							2.3634			
5A	6.3386	6.3843	6.3410	6.3410	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	6.1024	201056605		
	6.3448						6.1048			
5A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201056605		0.119
5A2										WITHIN DESIGN LIMITS
6A	1.5340	1.4799	1.5340	1.5340	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.									
6B	1.5340	1.4961	1.5340	1.5340	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.									
6A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
6B1										
7A	0.3850	0.4530	0.3850	0.3850	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.3850	440016734	0.0010	0.3860
7B	AS A REF.		0.3850	0.3850	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.3850	440016734	0.0010	0.3860
7C	0.3880	0.4530	0.3850	0.3850	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.3850	440016734	0.0010	0.3860
7D	AS A REF.		0.3850	0.3850	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.3850	440016734	0.0010	0.3860
8A	LUBE HOLES	NOTE 2	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
9A	EXTERNAL SUFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
10A	NOTES SUFACE		COND. ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: UPPER TORQUE LINK

W/ORDER: 12615

P/NUMB.: 201540303

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT



S/NUMB.: 08BEL0061

SUB W/O: 12615-17

LIF. LIMIT: 60,000

C.S.N.: 16,758

NOTE 1: DO NOT REMOVE MORE THAN 0.020 PER FACE

NOTE 2: ► THREE LUBE FITTING HOLES INSTALL LUBRICATION ADAPTER P/N 899005010 (QTY 3)

A/C: A320

P/NAME: UPPER TORQUE LINK

P/NUMB.: 201540303

S/NUMB.: 08BEL0061

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE, INC.

W/ORDER: 12615

OHM CHAP.: 32-12-25

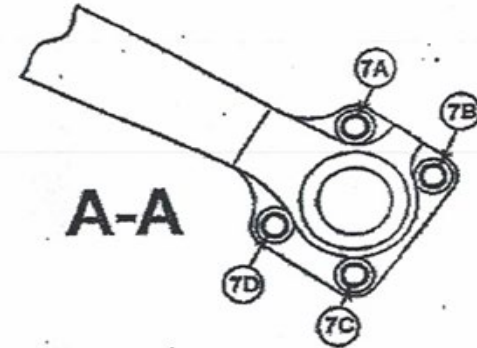
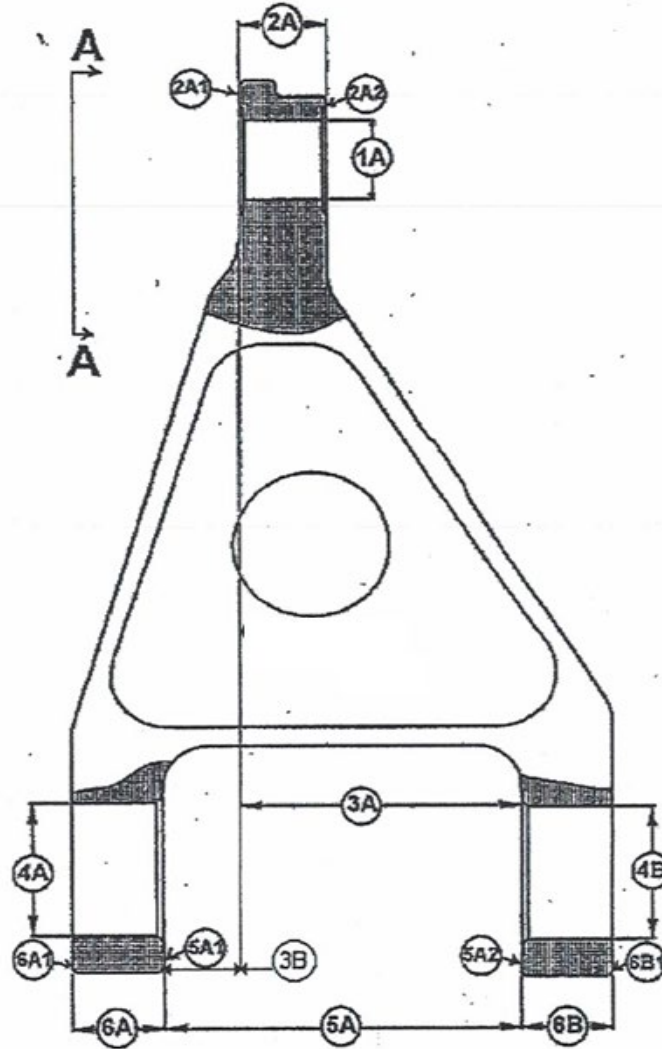
SUB W/O: 12615-17

C.S.N.: 16,758

INSPECTION STATUS REPORT



Component Overhaul
Services



A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: PIN, UPR.

W/ORDER: 12615

P/NUMB.: 201160603

OHM CHAP.: 32-11-33

S/NUMB.: 08MDG8380

SUB W/O: 12615-14

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT



Component Overhaul Services

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	2.3603	2.3374	2.3603	2.3603	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	2.3610	NOTE 1								
B	1.8890		1.8890	1.8890	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
C	1.8890		1.8890	1.8890	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
D	1.5570		1.5570	1.5570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
E	1.5570		1.5570	1.5570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
F	9.6100		6.6100	6.6100	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
G	0.5632	0.626	0.5632	0.5632	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5632	450237501	0.001	0.5642
H	0.5625	0.649	0.5632	0.5632	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5625	450237501	0.001	0.5642
I	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320
P/NAME: PIN, UPR.
P/NUMB.: 201160603
S/NUMB.: 08MDG8380
LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE, INC.
W/ORDER: 12615
OHM CHAP.: 32-11-33
SUB W/O: 12615-14
C.S.N.: 16,758

INSPECTION STATUS REPORT



NOTE 1: CHROME REPAIR

A/C: A320

P/NAME: PIN, UPR.

P/NUMB.: 201160603

S/NUMB.: 08MDG8380

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE, INC.

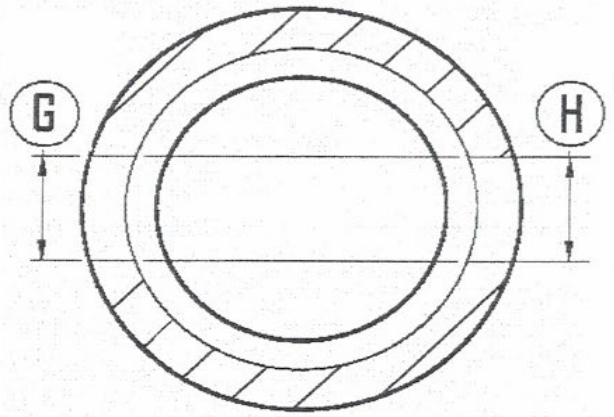
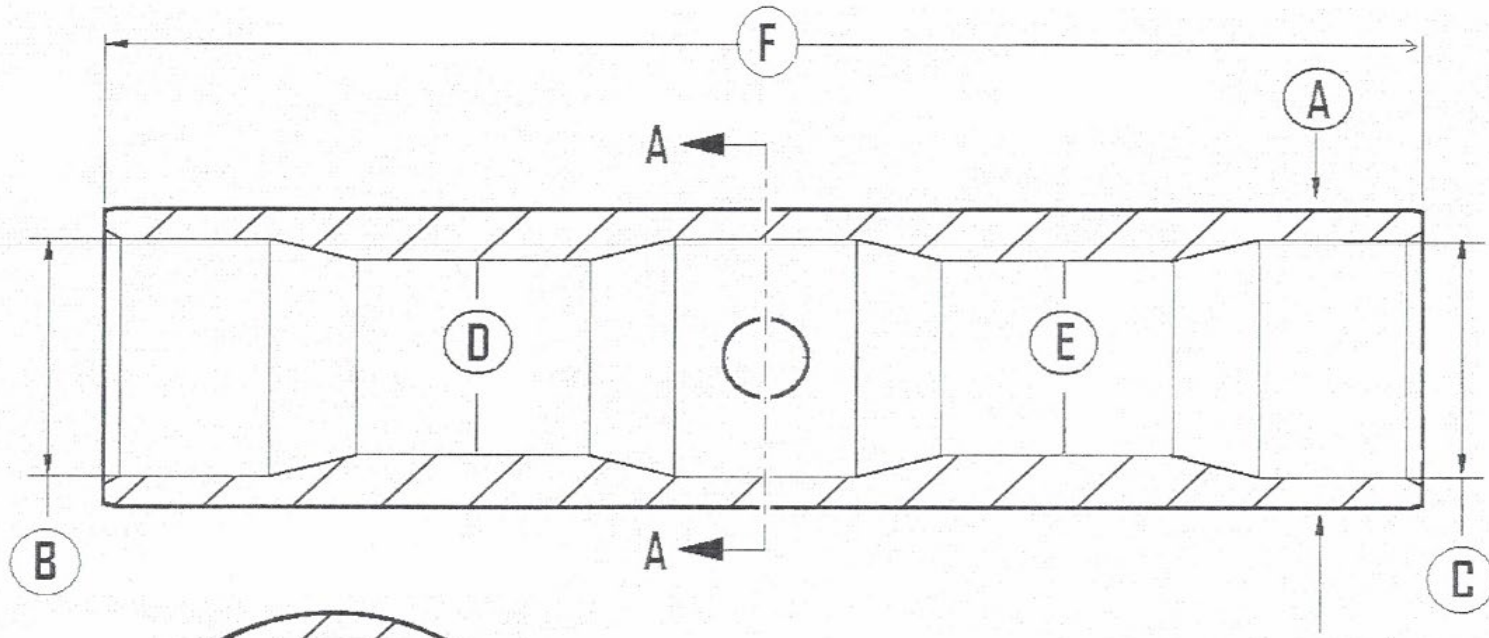
W/ORDER: 12615

OHM CHAP.: 32-11-33

SUB W/O: 12615-14

C.S.N.: 16,758

INSPECTION STATUS REPORT



SECTION A-A
(WITHOUT BUSHES)

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: UPPER TORQUE LINK

W/ORDER: 12615

P/NUMB.: 201540302

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08BEL0129

SUB W/O: 12615-22



LIF. LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	1.6535	1.7293	1.6550	1.6550	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	1.4567	201383609	0.0020	1.6570
1B	1.6545	MAXIMUM	1.6550	1.6550	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	1.4577	201383609	0.0020	1.6570
2A	1.6900 REF.	1.6500 MINIMUM	1.6950	1.6950	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	1.8878			
							1.8888	201383609		
2A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201383609		0.096
2A2					WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201383609		0.096
3A		5.1870 MAXIMUM	5.1380	5.1380	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	5.1358			
							5.1398			
4A	2.5984	2.6823 MAXIMUM	2.6010	2.6010	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.3622	201056605	0.0030	2.6040
							2.3634			
4B	2.5996		2.6010	2.6010	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	2.3622	201056605	0.0030	2.6040
							2.3634			
5A	6.3386	6.3843 MAXIMUM	6.3430	6.3430	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	6.1024	201056605		
	6.3448						6.1048			
5A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201056605		0.120
5A2					WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201056605		0.120
6A		1.4790 MINIMUM	1.5360	1.5360	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
6B		1.4790 MINIMUM	1.5360	1.5360	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
6A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
6B1					WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7A	LUBE HOLES	NOTE 1	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
8A	EXTERNAL SUFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
9A	NOTES SUFACE		COND. ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: UPPER TORQUE LINK

W/ORDER: 12615

P/NUMB.: 201540302

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08BEL0129

SUB W/O: 12615-22



LIF. LIMIT: 60,000

C.S.N.: 16,758

NOTE 1: ► THREE LUBE FITTING HOLES INSTALL LUBRICATION ADAPTER P/NI 899005010 (QTY 3)

A/C: A320

P/NAME: UPPER TORQUE LINK

P/NUMB.: 201540302

S/NUMB.: 08BEL0129

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE, INC.

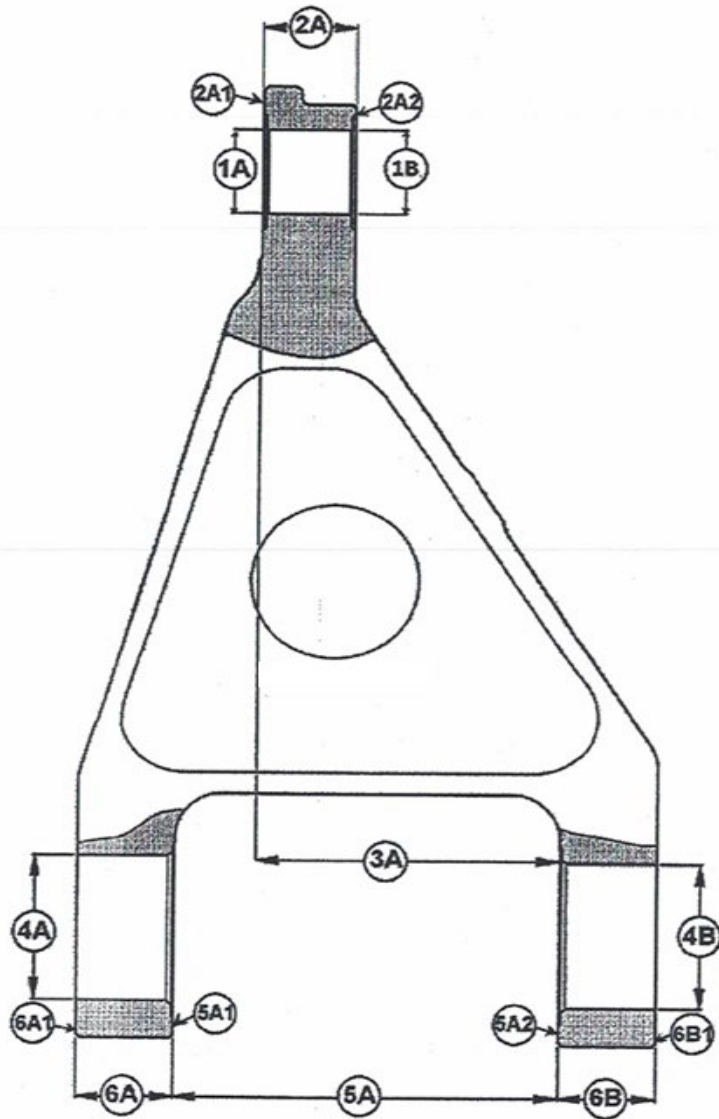
W/ORDER: 12615

OHM CHAP.: 32-12-25

SUB W/O: 12615-22

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: PIN

W/ORDER: 12615

INSPECTION STATUS REPORT



Component Overhaul Services

P/NUMB.: 201160602

OHM CHAP.: 32-12-25

S/NUMB.: 08MDG6742

SUB W/O: 12615-20

LIF. LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	2.3603	2.3374	2.3607	2.3607	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	2.3610	NOTE 1								
B	1.8890		1.8910	1.8910	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.									
C	1.8890		1.8900	1.8900	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.									
D	1.578		1.5780	1.5780	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.									
E	1.578		1.5780	1.5780	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.									
F	9.600		9.6100	9.6100	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.									
G	0.500		0.5010	0.5010	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.5000	450237506	0.001	0.5020
H	0.501		0.5010	0.5010	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.5010	450237506	0.001	0.5020
I	OTHER			COND.	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REJECTIONS			ONLY						

A/C: A320
P/NAME: PIN
P/NUMB.: 201160602
S/NUMB.: 08MDG6742
LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12615
OHM CHAP.: 32-12-25
SUB W/O: 12615-20
C.S.N.: 16,758

INSPECTION STATUS REPORT



NOTE 1: CHROME REPAIR

A/C: A320

P/NAME: PIN

P/NUMB.: 201160602

S/NUMB.: 08MDG6742

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.

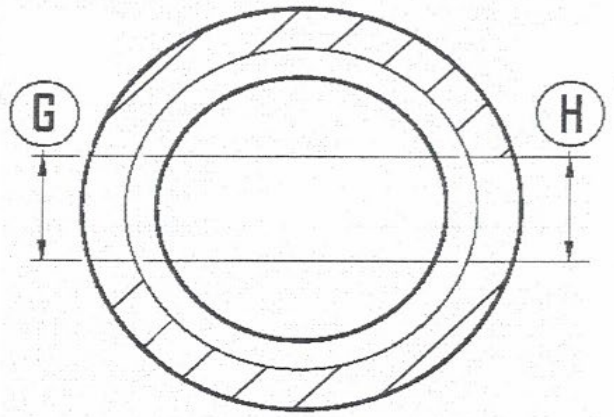
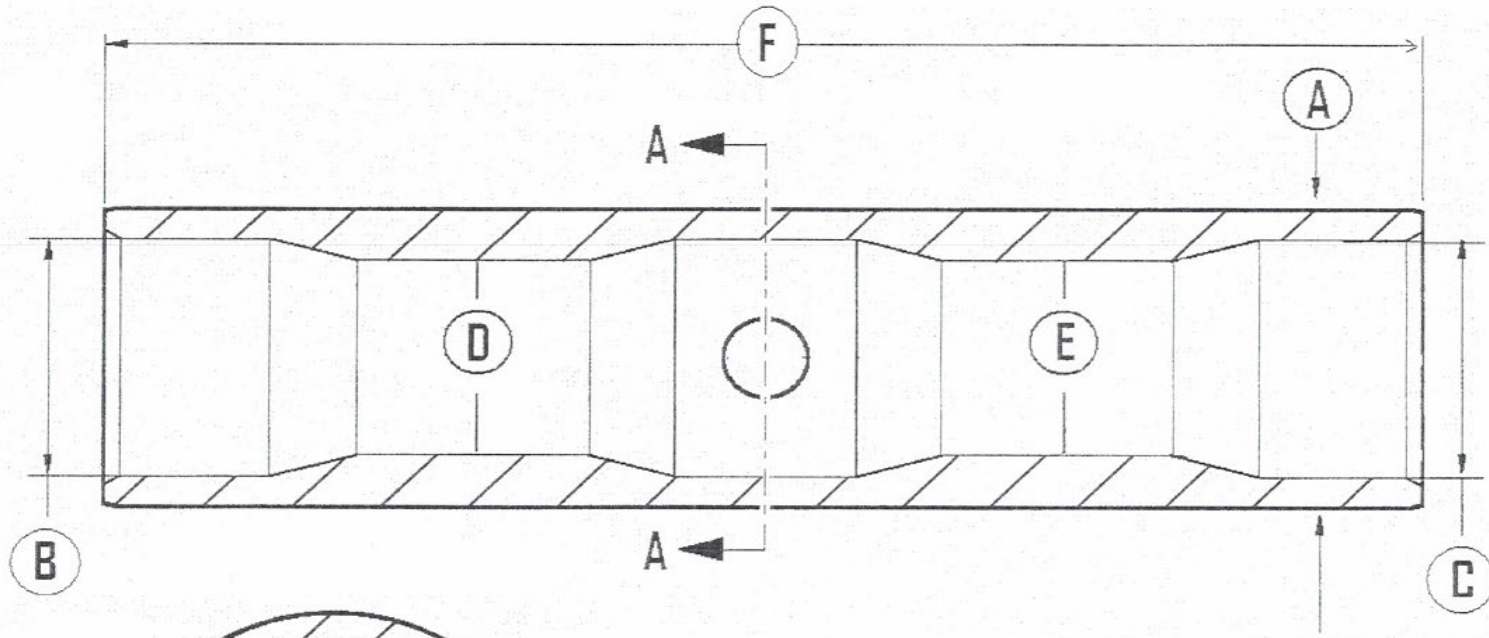
W/ORDER: 12615

OHM CHAP.: 32-12-25

SUB W/O: 12615-20

C.S.N.: 16,758

INSPECTION STATUS REPORT



SECTION A-A
(WITHOUT BUSHES)

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: MLG RH FITTING

W/ORDER: 12615

P/NUMB.: 201540301

OHM CHAP.: 32-12-25

S/NUMB.: 08MDM2048

SUB W/O: 12615-2

LIFE LIMIT: 60, 000

C.S.N.: 16,758

INSPECTION STATUS REPORT



LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	8.3465	8.408 MAX	8.3510	8.3760	PITTING	STRIP				
	8.3493	NOTE 14	#1							
B	8.2270	8.258 MAX	8.2310	8.2530	PITTING	STRIP				
	8.2241	NOTE 14	#1							
C	8.0020	8.0040	8.0040	8.0040	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	8.0048	CHROME								
D	7.2270	7.2825	7.2270	7.2270	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	7.2240	NOTE 11								
E			COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
1A	4.3701		4.3710	4.3710	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	4.1339	201160639	0.0060	4.377
	4.3715						4.1352			
2A	7.5886		7.6000	7.6000	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	7.5886			
	7.6082						7.6082			
2A1	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
2A2	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201160639		
			ONLY							
3A	0.7400	0.8158	0.7489	0.7489	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6299	201056715	0.0020	0.751
3B	0.7489	O/S BUSHING	0.7489	0.7489	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6310	201056715	0.0020	0.751
4A	4.5670		4.5760	4.5760	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	4.5670	201056715		
	4.5767						4.5677			
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056715		-0.004
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056715		-0.004
5A		5.3440	5.4000	5.4000	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.5420	201056715		
		FLANGE BUSHING					5.5760			
5A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056715		0.071
5A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056715		0.071
6A	2.2047	2.2880	2.2050	2.2050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.0080	201160604	0.0020	2.207
	2.2059						2.0090			
6B	2.2441	2.3270	2.2453	2.2453	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.0019	201160605	0.0020	2.247
	2.2453						2.0091			

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: MLG RH FITTING

W/ORDER: 12615

P/NUMB.: 201540301

OHM CHAP.: 32-12-25

S/NUMB.: 08MDM2048

SUB W/O: 12615-2

LIFE LIMIT: 60, 000

C.S.N.: 16,758

INSPECTION STATUS REPORT



LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
7A		2.4400	2.4050	2.4050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2050	201160604		
		O/S BUSHING					2.2080	201160605		
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201160604		0.100
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201160605		0.100
8A	1.1080	1.0620	1.1050	1.1050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		NOTE 6		
		MINIMUM								
8B	1.1080	1.0720	1.1050	1.1050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		NOTE 6		
		MINIMUM								
8A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
8B1			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
9A	6.2992	NOTES	6.3010	6.3010	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201042106	0.0020	6.303
	6.3017	1, 2 AND 3								
10A		0.6100	0.4920	0.4920	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3750	NOTE 7	0.0020	0.494
							0.3772			
11A		2.6680	2.6770	2.6770	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.6680	NOTE 7		
11A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		NOTE 7		
11A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
12A		0.8465	0.7850	0.7850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6398	201160633	0.0020	0.787
		MAXIMUM					0.6594			
13A	2.7461		2.7800	2.7800	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.7461			
	2.8240						2.8240			
13A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				-0.017
13A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				-0.017
14A	2.0472	2.1310	2.0480	2.0480	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.8436	201056686	0.0020	2.050
							1.8450			
14B	2.0484		2.0480	2.0480	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.4173	201056887	0.0020	2.050
							1.4183			
15A		8.4650	8.5700	8.5700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	8.7600	201056686		
							8.7690	201056687		
15A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056686		0.095
15A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056687		0.095

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: MLG RH FITTING

W/ORDER: 12615

P/NUMB.: 201540301

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08MDM2048

SUB W/O: 12615-2



LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
16A	2.9528	3.0370	2.9539	2.9539	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.7559	201056603	0.0020	2.956
16B	2.9539	O/S BUSHING	2.9539	2.9539	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.7571	201160696	0.0020	2.956
17A		5.0790	5.1240	5.1240	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.4298	201056603		
							5.4313	201160696		
17A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056603		0.153
17A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201160696		0.153
18A	2.5984	2.6820	2.5996	2.5996	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3623	201056605	0.0020	2.602
18B	2.5996		2.5996	2.5996	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3633	201056605	0.0020	2.602
19A	5.7050	5.6200	5.7050	5.7050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.9416	201056605		
		FLANGE BUSHING					5.9430			
19A1	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056605		0.118
19A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056605		0.118
20A		1.5110	1.5750	1.5750	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
20A1	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
20B		1.5110	1.5750	1.5750	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
20B1	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
21A	1.2598	1.3356	1.3608	1.3608	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.0236	201383607	0.0020	1.363
	1.2608	MAXIMUM					1.0244			
21B	1.2598	1.3356	1.3608	1.3608	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.0236	201383607	0.0020	1.363
	1.2608	MAXIMUM					1.0244			
21C	1.2598	1.3356	1.3608	1.3608	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.0236	201383607	0.0020	1.363
	1.2608	MAXIMUM					1.0244			
22A	0.5510	0.6192	0.5520	0.5520	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3720	201173669	0.0020	0.554
	0.5522						0.3917			

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: MLG RH FITTING

W/ORDER: 12615

P/NUMB.: 201540301

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08MDM2048

SUB W/O: 12615-2



LIF. LIMIT: 60,000

C.S.N.: 16,758

NOTE 1: ► MACHINE THE BORE TO ONE OF THE FOLLOWING REWORKEDS

- 1- 6.3043 / 6.3066 O/S BEARING P/N: 450239401
- 2- 6.3093 / 6.3116 ----- P/N: 450239402
- 3- 6.3143 / 6.3166 ----- P/N: 450239403
- 4- 6.3193 / 6.3216 ----- P/N: 450239404

NOTE 2 5- 6.3243 / 6.3266 ----- P/N: 450239405

- 6- 6.3293 / 6.3316 ----- P/N: 450239406
- 7- 6.3343 / 6.3366 ----- P/N: 450239407
- 8- 6.3393 / 6.3416 ----- P/N: 450239408

NOTE 3: 9- 6.3443 / 6.3466 ----- P/N: 450239409

- 10- 6.3493 / 6.3516 ----- P/N: 450239410
- 11- 6.3543 / 6.3566 ----- P/N: 450239411
- 12- 6.3593 / 6.3616 ----- P/N: 450239412

NOTE 4: ► -14A- BUSHING IS INSTALLED ON THE INBOARD SIDE OF THE MAIN FITTING, SAME SIDE AS THE LOWER CARDAN LUG

NOTE 5: ► REPAIR IAW MESSIER DOWTY CMM 32-12-25 REPAIR 13-28 PG 604
IF DISCREPANCIES REMAIN AND YOU MUST MACHINE OVER RWK LIMITS. THEN CONTACT AN INSPECTOR FOR FURTHER INSTRUCTION

NOTE 6: REFER TO COCESSION NUMBER RC-GL-0027227 FOR REFERENCE DESIGN DIMENSION FOR THE LUG

NOTE 7: MACHINE 0.0197 MAX PER FACE (REF. REPAIR 13-9 FIG. 602 PAGE 604

NOTE 8: ► IF HOLE DIAMETER IS STANDARD, USE BEARING P/N: 201056606 QTY 1 PER CMM 32-12-25 AS REQUIRED
IF PART REQUIRES MACHINING TO REMOVE DEFECTS , THEN MACHINE A MINIMUM OF 0.060 OVER FOR A REPAIR BEARING P/N: 4502378214
QTY. 1 PER CMM 32-12-25 REPAIR NUMBER 13-19

NOTE 9: IF THE LUBRICATION BORE LOCATION 32G REQUIRES REWORKED FOLLOW REPAIR 13-24 FOR THE CORRECT RWK ADAPTER REQUIRED

NOTE 10: IF THE LUBRICATION BORE LOCATION 32H REQUIRES REWORKED FOLLOW REPAIR 13-24 FOR THE CORRECT RWK ADAPTER REQUIRED

NOTE 11: ► REPAIR PER MESSIER DOWTY, CMM 32-11-33 REPAIR NO 1-26 TABLE 601 PG 603
1- DIAMETER 0.4850 / 0.4860 TRANSFER DOWEL 45023951
2- DIAMETER 0.5500 / F.5520 TRANSFER DOWEL 45029352

NOTE 12: DO NOT MACHINE MORE THAN 0.196 IN FROM FACE AND MAKE FINISH 63 MICRO-INCHES

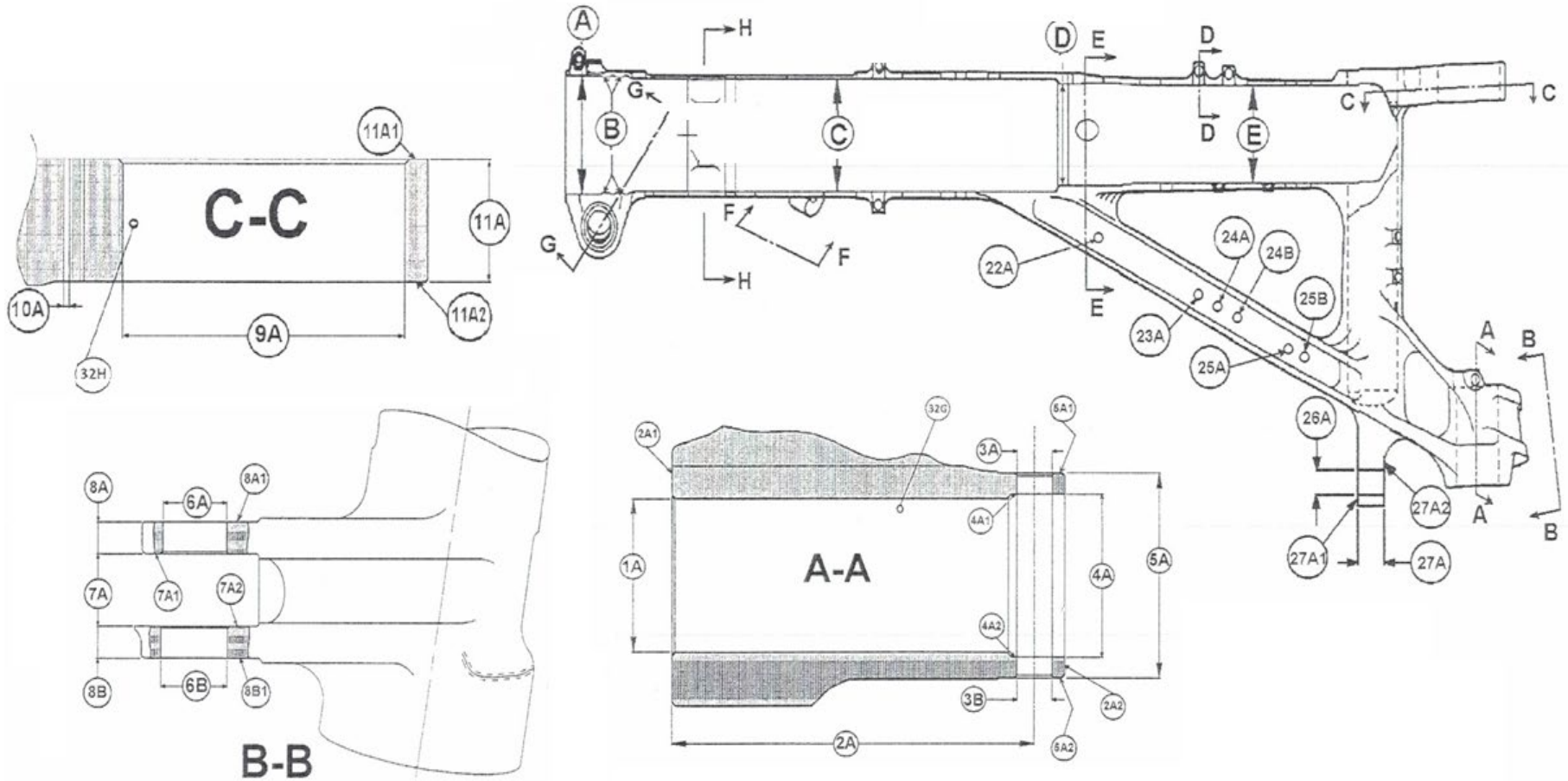
NOTE 13: IF DIAMETER -A- HAS WEARMARKS OR DAMAGE AND REQUIRES REPAIR, DO NOT MACHINE MORE THAN 0.0196 FROM EACH

NOTE 14:

A/C: A320
P/NAME: MLG RH FITTING
P/NUMB.: 201540301
S/NUMB.: 08MDM2048
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE, INC.
W/ORDER: 12615
OHM CHAP.: 32-12-25
SUB W/O: 12615-2
C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

P/NAME: MLG RH FITTING

P/NUMB.: 201540301

S/NUMB.: 08MDM2048

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE, INC.

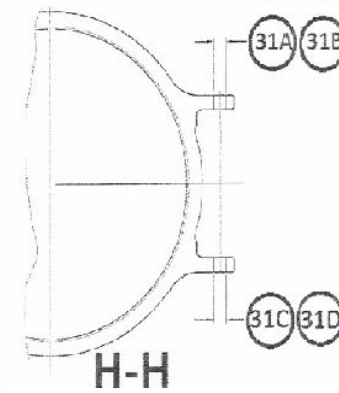
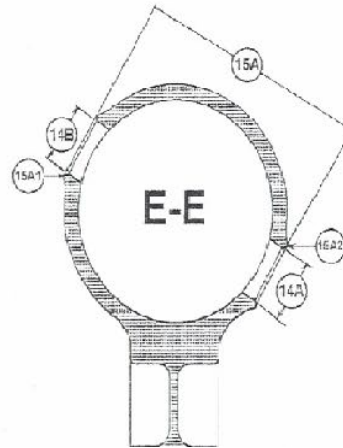
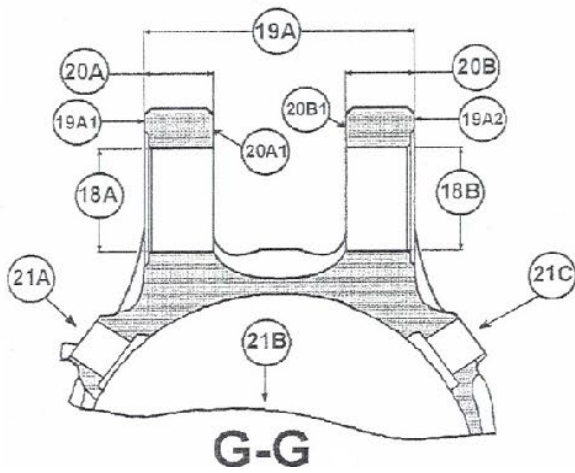
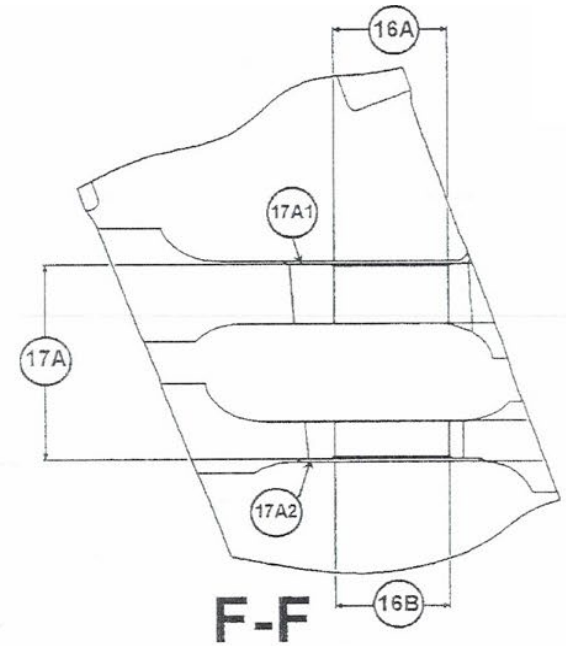
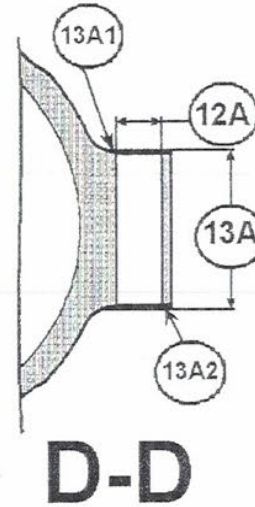
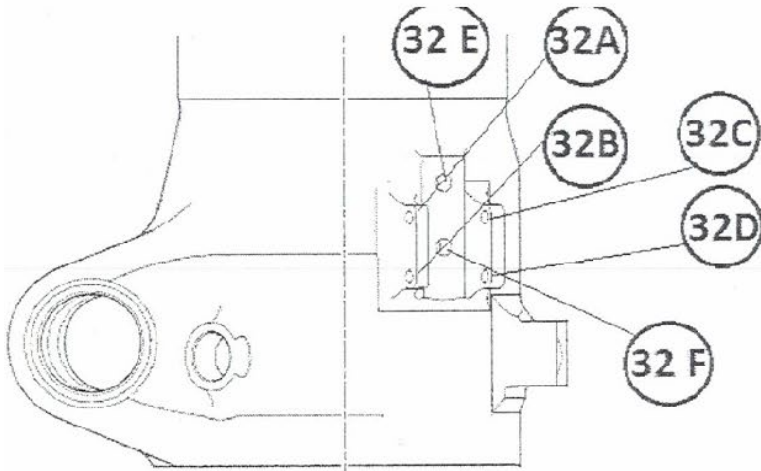
W/ORDER: 12615

OHM CHAP.: 32-12-25

SUB W/O: 12615-2

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: PIN

W/ORDER: 12615

INSPECTION STATUS REPORT

Component Overhaul
Services

P/NUMB.: 201173600

OHM CHAP.: 32-12-25

S/NUMB.: 08MDG6980

SUB W/O: 12615-145

LIF. LIMIT: 60,,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	3.7388	3.7156	3.7391	3.7391	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	3.7397	NOTE 1								
B	4.1325	4.109	4.1326	4.1326	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	4.1334	NOTE 1								
1A	0.630	0.693	0.6310	0.6310	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.6300	450239471	0.0015	0.633
1B	0.631	NOTE 2	0.6310	0.6310	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.6310	450239471	0.0015	0.633
2A		0.275	0.3250	0.3250	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
		MINIMUM	0.3250	0.3250						
3A		0.205	0.2500	0.2500	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
		MINIMUM								
4A	INTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	SURFACE		ONLY							
5A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	SURFACE		ONLY							
6A	OTHER		COND.							
	REJECTIONS		ONLY							

A/C: A320
P/NAME: PIN
P/NUMB.: 201173600
S/NUMB.: 08MDG6980
LIF. LIMIT: 60,,000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12615
OHM CHAP.: 32-12-25
SUB W/O: 12615-145
C.S.N.: 16,758

INSPECTION STATUS REPORT



NOTE 1: CHROME REPAIR, PER CMM 32-11-33 AND 32-12-25 BOTH JOURNALS -A- AND -B- MUST BE REPAIRED AT THE SAME TIME

NOTE 2: TO REPAIR THIS BORES, MACHINE AND INSTALL BUSHINGS P/N: 450239471

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: PIN

W/ORDER: 12615

P/NUMB.: 201173600

OHM CHAP.: 32-12-25

S/NUMB.: 08MDG6980

SUB W/O: 12615-145

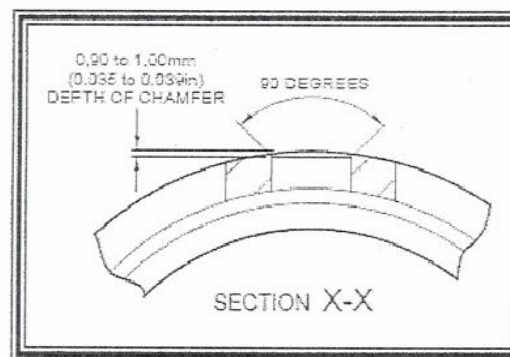
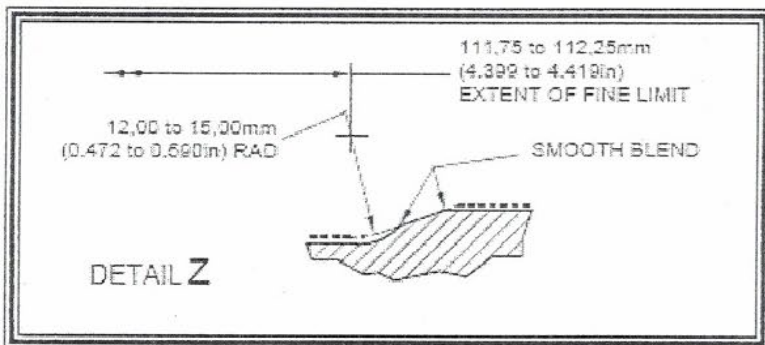
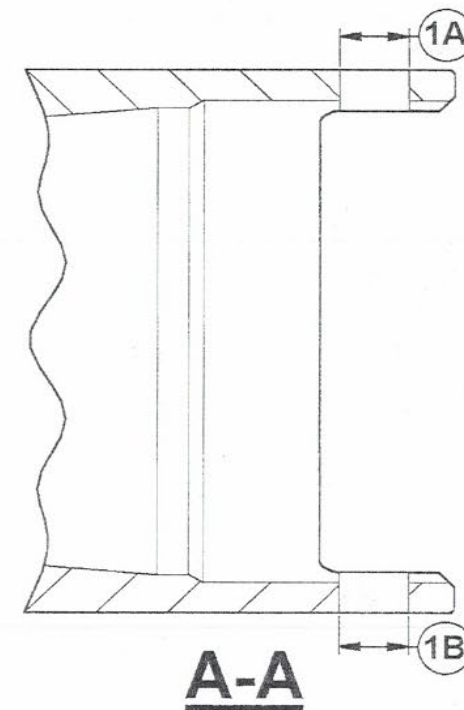
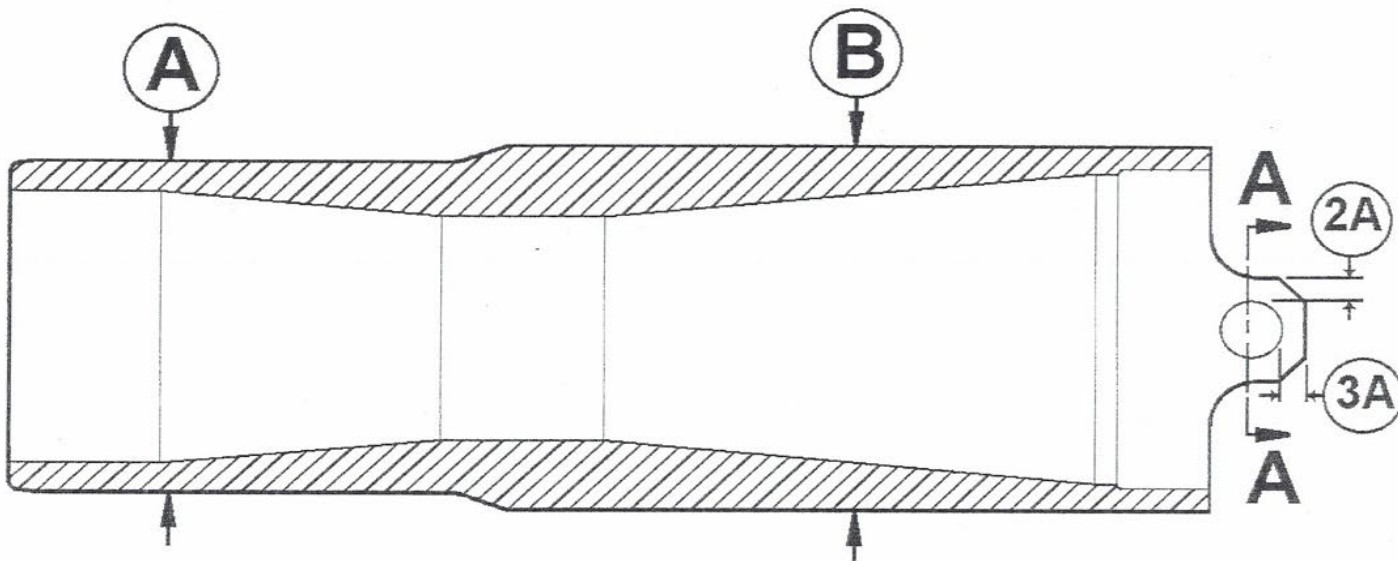
LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT



Component Overhaul Services



BUSHING REPAIR DETAILS

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: CARDAN LOCK STAY

W/ORDER: 12615

P/NUMB.: 201058306

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08COU46164X5769

SUB W/O: 12615-139

Component Overhaul
Services

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	0.9843	1.0599	0.9851	0.9851	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058602	0.0010	0.986
	0.9851						0.7882			
2A			0.7880	0.7880	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.9826	201058602		
							0.9835			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058602		0.097
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058602		0.097
3A			2.0300	2.0300	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.8228	201058620		
							1.8244			
3A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	201058620		0.104
3A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4737	201058620		0.104
4A	0.9873	1.0599	0.9873	0.9873	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058620	0.0020	0.989
	0.9851						0.7882			
4A1	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
4B	0.9873	1.0599	0.9873	0.9873	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058620	0.0020	0.989
	0.9851						0.7882			
4B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
5A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
6A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: CARDAN LOCK STAY

W/ORDER: 12615

P/NUMB.: 201058306

OHM CHAP.: 32-12-25

S/NUMB.: 08COU46164X5769

SUB W/O: 12615-139

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT

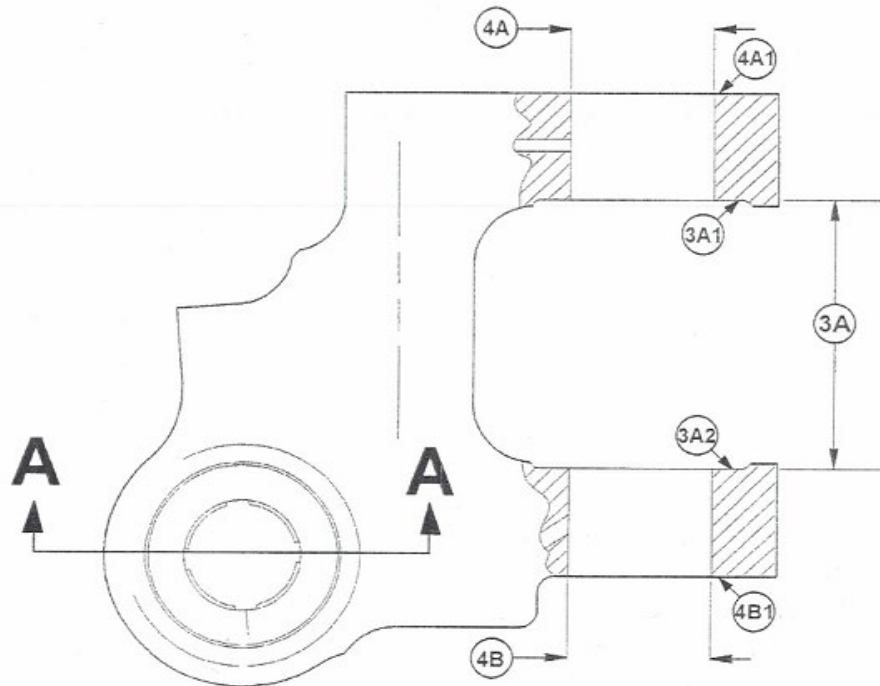
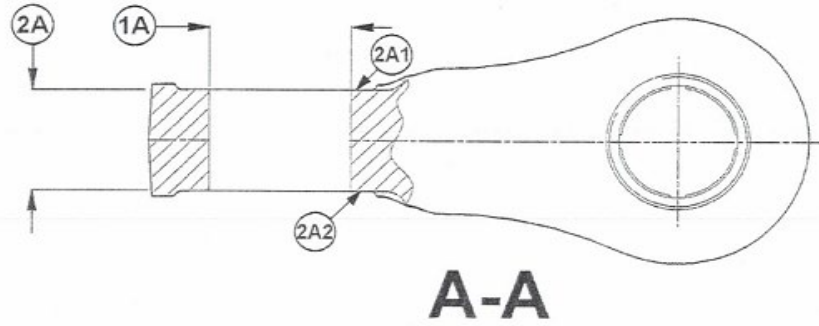


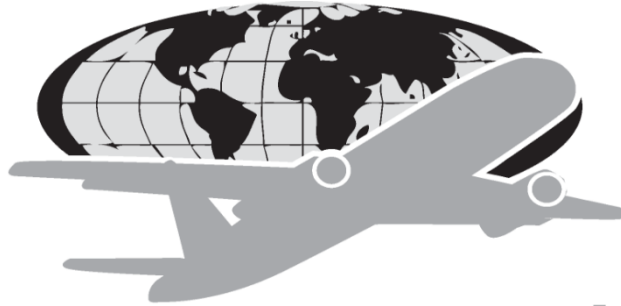
NOTE 1: IF BORE IS OVERSIZED OR REQUIURES MACHINING SCRAP THE UNIT.

A/C: A320
P/NAME: CARDAN LOCK STAY
P/NUMB.: 201058306
S/NUMB.: 08COU46164X5769
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE, INC.
W/ORDER: 12615
OHM CHAP.: 32-12-25
SUB W/O: 12615-139
C.S.N.: 16,758

INSPECTION STATUS REPORT





Component Overhaul Services

FAA Repair Station # 8COR883B

EASA 145.6404

Landing Gear Overhaul Report

W/O: 12616

Description: A320 L/H MLG

BUILDUP ASSY

P/N: 201582001-040

S/N: MDG4845

DEC. 2024

SECTION 1. AIRWORTHINESS APPROVAL TAGS

1. Approving Civil Aviation Authority/Country: FAA/UNITED STATES	AUTHORIZED RELEASE CERTIFICATE FAA FORM 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: 219355
--------------------------------------------------------------------------------	--------------------------------------------------------------------------------------	-----------------------------------------------

4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74th St. Suite B, Medley, FL 33166 TEL: (305) 406-3885 FAX: (786) 400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616 / 00023395
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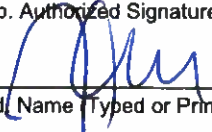
6. Item	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:
1	LEG AND DRESSINGS, MLG, LH	201582001-040	1	MDG4845	OVERHAULED

12. Remarks:

OVERHAULED I.A.W. SAFRAN CMM 32-12-24 REV. 43 DATED MAR 15/2024. REFER TO COMPONENT OVERHAUL SERVICES WORK STATEMENT REPORT NUMBER 12616 FOR INSPECTION FINDINGS, SERIALIZED COMPONENT LIST, TRACEABILITY, C.S.N., S.B. AND A.D. MODIFICATION LIST.
CSO: -0-

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): DOUGLAS ARAUZ	14e. Date (dd/mmm/yyyy): 11/Dec/2024


User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1. It is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statement in Blocks 13a and 14a do not constitute installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7517
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4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-48 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11. Status/Work:
1	FWD. PINTLE PIN STUD BOLT	201540616	1	PA79539X437	Overhauled

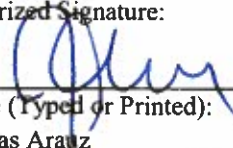
12. Remarks:

Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.

CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024


User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7518
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4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-49 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11 Status/Work:
1	FWD. PINTLE PIN STUD BOLT	201540616	1	PA79539X443	Overhauled

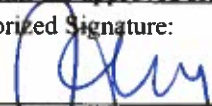
12. Remarks:

Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.

CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, <u>the items are approved for return to service.</u>
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.


Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	3. Form Tracking Number: COS7519
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AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-3 00023395
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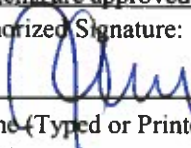
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	MAIN FITTING	201540300	1	08MDM2051	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.

CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024

User/Installer Responsibilities


It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7520	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616-45 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	FWD. PINTLE PIN	201173600	1	08MDG6974	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Atauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7521
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4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-199 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11 Status/Work:
1	AFT PINTLE PIN	D3215303000800	1	F07647-45	Overhauled

12. Remarks:
 Overhauled I.A.W Airbus CMM 32-11-07 Rev. 32 Rev. date 01/Oct/2019. CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
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13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024
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User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.


Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	3. Form Tracking Number: COS7522
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AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-38 00023395
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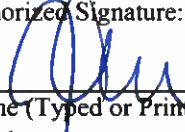
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	LOCKSTAY CARDAN	201058306	1	07COU45295X5723	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.

CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Afauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.


Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7523	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  Component Overhaul Services				5. Work Order/Contract/Invoice Number: 12616-36 00023395		
COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B						
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	LOCKSTAY CARDAN PIN	201056909	1	08BEV81754019	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: COS7524	
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B							5. Work Order/Contract/Invoice Number: 12616-77 00023395	
6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11 Status/Work:			
1	UPPER TORQUE LINK	201540303	1	08BEL0060	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to:			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12					
<input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:		
						8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):		
				Douglas Arauz		11/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.								
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.								
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.								

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7525
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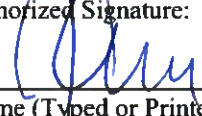
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-74 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	UPPER TORQUE LINK PIN	201160603	1	08MDG8381	Overhauled

12. Remarks:
 Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024


User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7526
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4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-162 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	TORQUE LINK APEX PIN	201383606	1	08MDG5303	Overhauled

12. Remarks:
 Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
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13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024
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User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.


Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	3. Form Tracking Number: COS7527
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AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-170 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	TORQUE LINK APEX PIN NUT	201587612	1	08MSL5902386	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.
CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.


Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States		2			AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: COS7528	
4. Organization Name and Address:  Component Overhaul Services					COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B		5. Work Order/Contract/Invoice Number: 12616-83 00023395	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:			
1	LOWER TORQUE LINK	201540302	1	08BEL0113	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to:					14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12			
<input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.					Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:		
						8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):		
				Douglas Arauz		11/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.								
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.								
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.								

1. Approving Civil Aviation Authority/Country: FAA/United States	2	3. Form Tracking Number: COS7529
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AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

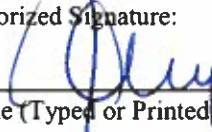
4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-81 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	LOWER TORQUE LINK PIN	201160602	1	08MDG6745	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.
CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024

User/Installer Responsibilities


It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: COS7530		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	SHOCK ABSORBER	201371281	1	08B5151X6200	Overhauled
12. Remarks: <p style="margin-left: 40px;">Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.</p> <p style="text-align: right; margin-right: 100px;">CSO: 0 Previous Operator: AVIANCA</p> <p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 11/Dec/2024	
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7531
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4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-2 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	SLIDING TUBE	201371304	1	08B5151X6200	Overhauled

12. Remarks:
 Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024.
 CSN: 16,758
 CSO: 0
 Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.


Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: COS7532
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616-192 00023395	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	UPPER DIAPHRAGM TUBE	201371615	1	08B0314X1	Overhauled
12. Remarks: Overhauled I.A.W Safran CMM 32-12-25 Rev. 74 Rev. date 15/Mar/2024. <div style="float: right; text-align: right;"> CSN: 16,758 CSO: 0 Previous Operator: AVIANCA </div>					
<p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7533	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	SIDE STAY COMPLETE	201166013-025	1	AP1269	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-12-26 Rev. 30 Rev. date 17/Sep/2021. CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:
						8COR883B
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):
				Douglas Arauz		11/Dec/2024
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7534
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4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-123 00023395
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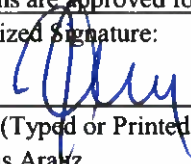
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	LOWER CARDAN	201163307	1	08B0893X1768	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.

CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.


Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	3. Form Tracking Number: COS7535
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AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

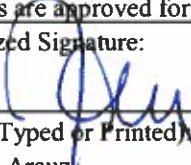
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-127 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	LOWER CARDAN NUT	201163945	1	2762/08	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.
CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024


User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7536
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4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-124 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	SIDE STAY LOWER CARDAN PIN	201163619	1	07AP0267	Overhauled

12. Remarks:
 Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.
 CSN: 16,758
 CSO: 0
 Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
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13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024
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User/Installer Responsibilities


It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States		2			AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: COS7537	
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B							5. Work Order/Contract/Invoice Number: 12616-122 00023395	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:			
1	LOWER SIDE STAY	201163301	1	07AP0228	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to:			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12					
<input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:		
						8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):		
				Douglas Arauz		11/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.								
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.								
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.								

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7538
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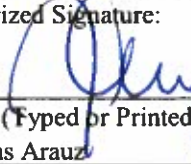
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-21 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	SIDE STAY PIVOT PIN	201163609	1	08AP0110	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.
CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024


User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7539
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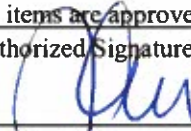
4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-115 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	UPPER SIDE STAY	201163300	1	AP042206	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.
CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024

User/Installer Responsibilities


It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: COS7540		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-116 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	SIDE STAY UPPER CARDAN PIN	201163618	1	07AP0042	Overhauled
12. Remarks: <div style="display: flex; justify-content: space-between;"> <div>Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.</div> <div>CSN: 16,758 CSO: 0 Previous Operator: AVIANCA</div> </div> <p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 11/Dec/2024	
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7541
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4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-142 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	UPPER CARDAN	201163948	1	08B0549X5	Overhauled

12. Remarks:
 Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
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13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024
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User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: <h1 style="margin:0;">COS7542</h1>		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-118 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	UPPER CARDAN NUT	201163944	1	2186/07	Overhauled
12. Remarks: <div style="display: flex; justify-content: space-between;"> <div>Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.</div> <div>CSN: 16,758 CSO: 0 Previous Operator: AVIANCA</div> </div> <p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input checked="" type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 11/Dec/2024	
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7543	
AUTHORIZED RELEASE CERTIFICATE				FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616-29 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	UPPER PIN SPRING	201661607	1	08AP0148	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7544	
AUTHORIZED RELEASE CERTIFICATE				FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616-28 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	LOWER PIN SPRING	201661606	1	08AP0123	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7545	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616-25 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	BANANA LINK (SPRINGS)	201661306	1	08AP0133	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7546	
AUTHORIZED RELEASE CERTIFICATE				FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616-26 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	BANANA LINK (SPRINGS)	201661306	1	08AP0132	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/ Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7547	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  Component Overhaul Services				5. Work Order/Contract/Invoice Number: 12616-31 00023395		
COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B						
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11 Status/Work:	
1	CUFF PIN	201661612	1	08AP0163	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: COS7548	
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B		5. Work Order/Contract/Invoice Number: 12616-32 00023395						
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:			
1	CUFF PIN	201661612	1	08AP0169	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to:			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12					
<input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:		
						8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):		
				Douglas Arauz		11/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.								
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.								
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.								


1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7549	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616-24 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	CUFF	201661613	1	07AP0226	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: COS7550	
4. Organization Name and Address:  Component Overhaul Services					COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B		5. Work Order/Contract/Invoice Number: 12616-27 00023395	
6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11 Status/Work:			
1	PIN LOCK LINK / BANANA LINK	201661605	1	08AP0030	Overhauled			
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA								
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.								
13a. Certifies the items identified above were manufactured in conformity to:			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12					
<input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:		
						8COR883B		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mmm/yyyy):		
				Douglas Arauz		11/Dec/2024		
User/Installer Responsibilities								
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.								
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.								
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.								

1. Approving Civil Aviation Authority/Country: FAA/United States	2	3. Form Tracking Number: COS7551
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AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-8 00023395
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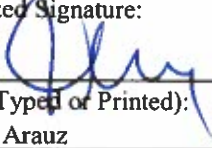
6. Item	7. Description	8. Part Number:	9. Quantity:	10 Serial Number	11 Status/Work:
1	LOCK LINK LOWER PIN	201057627	1	07AP0219	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024.

CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7552	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616-102 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	TRIANGULATION LINK	201661307	1	08AP0202	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7553	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616-30 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	TRIANGULATION LINK PIN	201661609	1	08AP0130	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-11-26 Rev. 63 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/ Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7554	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	LOCK STAY	201058003	1	AP1313	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-12-27 Rev. 26 Rev. date 15/Mar/2024. CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7555	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616-97 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	LOWER LOCK LINK	201058305	1	08AP0041	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-12-27 Rev. 26 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7556	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  Component Overhaul Services				5. Work Order/Contract/Invoice Number: 12616-68 00023395		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	LOCK LINK CENTER PIN	201058608	1	07AP0223	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-12-27 Rev. 26 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						


1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7557	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616-67 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	UPPER LOCK LINK	201058310	1	08AP0005	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-12-27 Rev. 26 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7558	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616-70 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	LOCK LINK UPPER PIN	201058611	1	07AP0071	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-12-27 Rev. 26 Rev. date 15/Mar/2024. CSN: 16,758 CSO: 0 Previous Operator: AVIANCA						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	2	3. Form Tracking Number: COS7559
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AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-1 00023395
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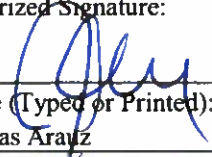
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	RETRACT ACTUATOR ASSY	201590003	1	11B18221	Overhauled

12. Remarks:
Overhauled I.A.W Safran CMM 32-31-86 Rev. 44 Rev. date 15/Mar/2024.

CSO: 0
Previous Operator: ETIHAD

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arayz	14e. Date (dd/mmm/yyyy): 11/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7560	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	RETRACT ACT PIN / MAIN FITTING	201590900	1	11BA7381270	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-31-86 Rev. 44 Rev. date 15/Mar/2024. CSN: 13,245 CSO: 0 Previous Operator: ETIHAD						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/ Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7561	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  Component Overhaul Services				5. Work Order/Contract/Invoice Number: 12616 00023395		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	RETRACT ACT PIN / REAR SPAR	201590901	1	11BA748686	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-31-86 Rev. 44 Rev. date 15/Mar/2024. CSN: 13,245 CSO: 0 Previous Operator: ETIHAD						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						


1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7562	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:	
1	PISTON ROD	201590908	1	11SOP11808912	Overhauled	
12. Remarks: Overhauled I.A.W Safran CMM 32-31-86 Rev. 44 Rev. date 15/Mar/2024. CSN: 13,245 CSO: 0 Previous Operator: ETIHAD						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.: 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz		14e. Date (dd/mmm/yyyy): 11/Dec/2024	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin: 0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin: 0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>			3. Form Tracking Number: COS7563
4. Organization Name and Address:  COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B				5. Work Order/Contract/Invoice Number: 12616 00023395	
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	PISTON HEAD	201590909	1	11SL493116	Overhauled
12. Remarks: Overhauled I.A.W Safran CMM 32-31-86 Rev. 44 Rev. date 15/Mar/2024. <div style="float: right; text-align: right;"> CSN: 13,245 CSO: 0 Previous Operator: ETIHAD </div> Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				 Douglas Arauz	
14e. Date (dd/mmm/yyyy):		14e. Date (dd/mmm/yyyy):			
		11/Dec/2024			
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States	2	<h2 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h2> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: <h1 style="margin:0;">COS7564</h1>		
4. Organization Name and Address: 		COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616 00023395		
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	PISTON HEAD NUT	201590910	1	11JL518484	Overhauled
12. Remarks: <div style="display: flex; justify-content: space-between;"> <div style="width: 60%;"> <p>Overhauled I.A.W Safran CMM 32-31-86 Rev. 44 Rev. date 15/Mar/2024.</p> </div> <div style="width: 35%;"> <p>CSN: 13,245 CSO: 0 Previous Operator: ETIHAD</p> </div> </div> <p>Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.</p>					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:	
				 8COR883B	
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed):	
				Douglas Arauz 11/Dec/2024	
User/Installer Responsibilities					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

1. Approving Civil Aviation Authority/Country: FAA/United States		2			3. Form Tracking Number: COS7565	
4. Organization Name and Address:  Component Overhaul Services				5. Work Order/Contract/Invoice Number: 12616 00023395		
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG						
7. Description		8. Part Number:		9. Quantity:		
6. Item	1	GLAND	201590913	1	10. Serial Number 11JL4932113	
11. Status/Work: Overhauled						
12. Remarks: Overhauled I.A.W Safran CMM 32-31-86 Rev. 44 Rev. date 15/Mar/2024. CSN: 13,245 CSO: 0 Previous Operator: ETIHAD						
Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature: 		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14c. Approval/Certificate No.: 8COR883B		
				14d. Name (Typed or Printed): Douglas Arauz		
				14e. Date (dd/mmm/yyyy): 11/Dec/2024		
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7566
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4. Organization Name and Address: 	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616 00023395
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6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	CYLINDER	201697600	1	11SOP118144017	Overhauled

12. Remarks:
 Overhauled I.A.W Safran CMM 32-31-86 Rev. 44 Rev. date 15/Mar/2024.

CSN: 13,245
 CSO: 0
 Previous Operator: ETIHAD

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024


User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: COS7567
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4. Organization Name and Address:  Component Overhaul Services	COMPONENT OVERHAUL SERVICES CORP. 6891 NW 74TH St. Suite B, Medley FL 33166 TEL: (305) 406-3885 FAX: (786)400-1872 FAA Repair Station Certificate No. 8COR883B	5. Work Order/Contract/Invoice Number: 12616-143 00023395
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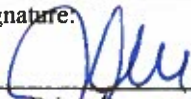
6. Item	7. Description	8. Part Number:	9. Quantity:	10. Serial Number	11. Status/Work:
1	AFT PINTLE PIN NUT	SLN41193	1	M337550-12804	Overhauled

12. Remarks:
Overhauled I.A.W Airbus CMM 32-11-04 Rev. 4 Rev. date 01/Jan/2022.

CSN: 16,758
CSO: 0
Previous Operator: AVIANCA

Component Overhaul Services certifies that the work specified in Block 11 and 12 was carried out in accordance with EASA Part 145 and in respect to that work the component is considered ready for release to service under EASA Part 145 approval number EASA.145.6404.

13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: 8COR883B
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Douglas Arauz	14e. Date (dd/mmm/yyyy): 11/Dec/2024

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

SECTION 2. LIFE LIMITED/SERIALIZED PARTS LIST



**Component Overhaul
Services**

**A320 L/H MLG
LIFE LIMITED/SERIALIZED PARTS LIST
FAA REPAIR STATION # 8COR883B / EASA 145.6404**

**Customer: CAVU AEROSPACE INC.
Purchase Order: 00023395
Work Order: 12616**

INT. ITEM #	DESCRIPTION	PART NUMBER	S/N	WORK ORDER	LAST OPERATOR	LIFE LIMIT	CSN	COMMENTS
	L/H MLG LEG AND DRESSING	201582001-040	MDG4845	12616	AVIANCA			
1	FWD. PINTLE PIN STUD BOLT	201540616	PA79539X437	12616-48	AVIANCA	60,000	16,758	
2	FWD. PINTLE PIN STUD BOLT	201540616	PA79539X443	12616-49	AVIANCA	60,000	16,758	
3	MAIN FITTING	201540300	08MDM2051	12616-3	AVIANCA	60,000	16,758	
4	FWD. PINTLE PIN	201173600	08MDG6974	12616-45	AVIANCA	60,000	16,758	
5	AFT PINTLE PIN	D3215303000800	F07647-45	12616-199	AVIANCA	60,000	16,758	
6	LOCKSTAY CARDAN	201058306	07COU45295X5723	12616-38	AVIANCA	60,000	16,758	
7	LOCKSTAY CARDAN PIN	201056909	08BEV81754019	12616-36	AVIANCA	60,000	16,758	
8	UPPER TORQUE LINK	201540303	08BEL0060	12616-77	AVIANCA	60,000	16,758	
9	UPPER TORQUE LINK PIN	201160603	08MDG8381	12616-74	AVIANCA	60,000	16,758	
10	TORQUE LINK APEX PIN	201383606	08MDG5303	12616-162	AVIANCA	60,000	16,758	
11	TORQUE LINK APEX PIN NUT	201587612	08MSL5902386	12616-170	AVIANCA	60,000	16,758	
12	LOWER TORQUE LINK	201540302	08BEL0113	12616-83	AVIANCA	60,000	16,758	
13	LOWER TORQUE LINK PIN	201160602	08MDG6745	12616-81	AVIANCA	60,000	16,758	
	SHOCK ABSORBER	201371281	08B5151X6200	12616	AVIANCA			
14	SLIDING TUBE	201371304	08B5151X6200	12616-2	AVIANCA	60,000	16,758	
15	UPPER DIAPHRAGM TUBE	201371615	08B0314X1	12616-192	AVIANCA	60,000	16,758	
	SIDE STAY COMPLETE	201166013-025	AP1269	12616	AVIANCA			
16	LOWER CARDAN	201163307	08B0893X1768	12616-123	AVIANCA	60,000	16,758	
17	LOWER CARDAN NUT	201163945	2762/08	12616-127	AVIANCA	60,000	16,758	
18	SIDE STAY LOWER CARDAN PIN	201163619	07AP0267	12616-124	AVIANCA	60,000	16,758	
19	LOWER SIDE STAY	201163301	07AP0228	12616-122	AVIANCA	60,000	16,758	
20	SIDE STAY PIVOT PIN	201163609	08AP0110	12616-21	AVIANCA	60,000	16,758	
21	UPPER SIDE STAY	201163300	AP042206	12616-115	AVIANCA	60,000	16,758	
22	SIDE STAY UPPER CARDAN PIN	201163618	07AP0042	12616-116	AVIANCA	60,000	16,758	
23	UPPER CARDAN	201163948	08B0549X5	12616-142	AVIANCA	60,000	16,758	
24	UPPER CARDAN NUT	201163944	2186/07	12616-118	AVIANCA	60,000	16,758	
25	UPPER PIN SPRING	201661607	08AP0148	12616-29	AVIANCA	60,000	16,758	
26	LOWER PIN SPRING	201661606	08AP0123	12616-28	AVIANCA	60,000	16,758	
	BANANA LINK (SPRINGS)	201661306	08AP0133	12616-25	AVIANCA	60,000	16,758	
27	BANANA LINK (SPRINGS)	201661306	08AP0132	12616-26	AVIANCA	60,000	16,758	
28	CUFF PIN	201661612	08AP0163	12616-31	AVIANCA	60,000	16,758	
29	CUFF PIN	201661612	08AP0169	12616-32	AVIANCA	60,000	16,758	
30	CUFF	201661613	07AP0226	12616-24	AVIANCA	60,000	16,758	
31	PIN LOCK LINK / BANANA LINK	201661605	08AP0030	12616-27	AVIANCA	60,000	16,758	
32	LOCK LINK LOWER PIN	201057627	07AP0219	12616-8	AVIANCA	60,000	16,758	
33	TRIANGULATION LINK	201661307	08AP0202	12616-102	AVIANCA	60,000	16,758	
34	TRIANGULATION LINK PIN	201661609	08AP0130	12616-30	AVIANCA	60,000	16,758	
	LOCK STAY	201058003	AP1313	12616	AVIANCA			

12/11/24



**Component Overhaul
Services**

**A320 L/H MLG
LIFE LIMITED/SERIALIZED PARTS LIST
FAA REPAIR STATION # 8COR883B / EASA 145.6404**

**Customer: CAVU AEROSPACE INC.
Purchase Order: 00023395
Work Order: 12616**

INT. ITEM #	DESCRIPTION	PART NUMBER	S/N	WORK ORDER	LAST OPERATOR	LIFE LIMIT	CSN	COMMENTS
35	LOWER LOCK LINK	201058305	08AP0041	12616-97	AVIANCA	60,000	16,758	
36	LOCK LINK CENTER PIN	201058608	07AP0223	12616-68	AVIANCA	60,000	16,758	
37	UPPER LOCK LINK	201058310	08AP0005	12616-67	AVIANCA	60,000	16,758	
38	LOCK LINK UPPER PIN	201058611	07AP0071	12616-70	AVIANCA	60,000	16,758	
	RETRACT ACTUATOR ASSY	201590003	11B18221	12616-1	ETIHAD			
40	RETRACT ACT PIN / MAIN FITTING	201590900	11BA7381270	12616	ETIHAD	60,000	13,245	
41	RETRACT ACT PIN / REAR SPAR	201590901	11BA748686	12616	ETIHAD	60,000	13,245	
42	PISTON ROD	201590908	11SOP11808912	12616	ETIHAD	60,000	13,245	
43	PISTON HEAD	201590909	11SL493116	12616	ETIHAD	60,000	13,245	
44	PISTON HEAD NUT	201590910	11JL518484	12616	ETIHAD	60,000	13,245	
45	GLAND	201590913	11JL4932113	12616	ETIHAD	60,000	13,245	
46	CYLINDER	201697600	11SOP118144017	12616	ETIHAD	60,000	13,245	
47	AFT PINTLE PIN NUT	SLN41193	M337550-12804	12616-143	AVIANCA	60,000	16,758	

12/11/24

Statement nº 1049 /2019

Date: São Paulo, June 07th, 2019

Incident/Accident Clearance Statement – Airframe

Aircraft: Airbus A318-100 | MSN 3642 (PR-ONR) | TSN 21764,97 | CSN 16758
Engine: PW6000 | ESN | TSN | CSN
Engine: PW6000 | ESN P318123 | TSN 27722,97 | CSN 20883
APU: 4500001B | SN 2924 | TSN 14569 | CSN 14920
NLG: NA28008-015 | SN B3137 | TSN 21764,97 | CSN 16758
MLG LH: 201581001 | SN MDL3642 | TSN 21764,97 | CSN 16758
MLG RH: 201581002 | SN MDL3642 | TSN 21764,97 | CSN 16758

To whom it may concern:

This letter is to certify that the referred aircraft has been operated by Oceanair Linhas Aéreas S.A. during the period stated below and to the best of my knowledge:

Operation period: from December 3, 2013 thru June 7, 2019.

1. Neither the aircraft, nor any part installed have been, unless its airworthiness status was re-established by an approved maintenance organization in accordance with the instructions of the type certificate holder and/or OEM of the part, and supported by an authorized release certificate:

a. Damaged during a reportable incident or accident as defined by Brazilian Civil Aviation Authority regulation RBAC 121.703 – Service Difficulty Reports, equivalent to FAA Title 14 - Chapter I - Subchapter G - Part 121 - Subpart V - §121.703, or

b. Subjected to severe stress or heat (such as in a major engine failure, accident or fire) or has been submersed in salt water.

2. No part has been installed on the aircraft which was obtained from military source or was previously fitted to a state aircraft.



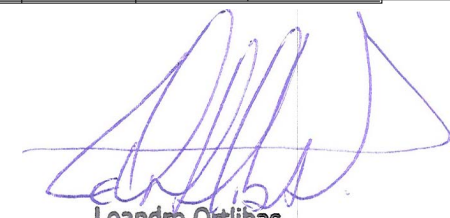
Alberto Ottavio Spelta
Chief Inspector

Rua Tamoios 579 Jardim Aeroporto – São Paulo – SP – CEP 04630-001
Telefone: (11) 3475 8200

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

From Position 32 To 3299999 Protocol Type LLT Part level * Maintenance level All
 Simulation 0 Months 0,00 FH 0 CY
 Ageing on 21/07/2019 21 764,97 FH 16 758 CY

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance						
	Installed Equipment (PN/SN)				PN Description		Specific Reference									
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32LMA00 28/10/2008	MAIN LAINDING GEAR LH PN : 201581001				SN : MDL3642		MAIN LANDING GEAR LH				TCI					
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00		16758,00	
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32LMG00 28/10/2008	MAIN FITTING PN : 201540300				SN : 08MDM2051		MAIN FITTING									
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	
32LMG01 28/10/2008	LOWER CARDAN PIN PN : 201163307				SN : 08B0893X1768		LOWER CARDAN PIN									
LLT													OVH			
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918	



Leandro Ortibas
 Engenheiro de Produção
 CREA 5060921772

Remaining potentials report

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance			
	Installed Equipment (PN/SN)				PN Description				Specific Reference													
	WO number	WP	Scheduled check																			
		Inspection				Inspection 2				Overhaul				Life Limit								
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline					
32LMG02 28/10/2008	SIDE STAY LOWER CARDAN PIN PN : 201163619				SN : 07AP0267				SIDE STAY LWR CARDAN PIN													
LLT													OVH									
CY			16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total			3918				3918			3650	3918	-268	26/10/2018		3918							
32LMG03 28/10/2008	REAR PINTLE PIN PN : D3215303000800				SN : F07647-45				REAR PINTLE PIN													
LLT													OVH									
CY			16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total			3918				3918			3650	3918	-268	26/10/2018		3918							
32LMG04 28/10/2008	REAR PINTLE PIN NUT PN : SLN41193				SN : M337550-12804				REAR PINTLE PIN NUT													
LLT													OVH									
CY			16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total			3918				3918			3650	3918	-268	26/10/2018		3918							
32LMG05 28/10/2008	AFT PINTLE SPHERICAL BEARING PN : 201042106				SN : 08NM132600X1177				AFT PINTLE BEARING													
LLT													OVH									
CY			16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total			3918				3918			3650	3918	-268	26/10/2018		3918							

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance					
	Installed Equipment (PN/SN)				PN Description		Specific Reference								
	WO number	WP	Scheduled check												
Inspection				Inspection 2				Overhaul				Life Limit			
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32LMG06 28/10/2008	FORWARD PINTLE PIN PN : 201173600				SN : 08MDG6974		PIN								
LLT												OVH			
CY		16758,00				16758,00		20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918				3918		3650	3918	-268		26/10/2018		3918	
32LMG07 28/10/2008	FORWARD PINTLE PIN STUB BOLTS PN : 201540616				SN : PA79539X437		CROSS BOLT								
LLT												OVH			
CY		16758,00				16758,00		20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918				3918		3650	3918	-268		26/10/2018		3918	
32LMG08 28/10/2008	FORWARD PINTLE PIN STUB BOLTS AT LH MLG PN : 201540616				SN : PA79539X443		CROSS BOLT								
LLT												OVH			
CY		16758,00				16758,00		20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918				3918		3650	3918	-268		26/10/2018		3918	
32LMG09 28/10/2008	UPPER TORQUE LINK PIN PN : 201160603				SN : 08MDG8381		PIN								
LLT												OVH			
CY		16758,00				16758,00		20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00
Cal Total		3918				3918		3650	3918	-268		26/10/2018		3918	

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance	
	Installed Equipment (PN/SN)				PN Description				Specific Reference											
	WO number	WP	Scheduled check																	
	Inspection				Inspection 2				Overhaul				Life Limit							
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline				
32LMG10 28/10/2008	UPPER TORQUE LINK PN : 201540303				SN : 08BEL0060				UPPER TORQUE LINK											
LLT													OVH							
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00		60000,00		
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					
32LMG11 28/10/2008	TORQUE LINK APEX PIN PN : 201383606				SN : 08MDG5303				MLG TORQUE LINK APEX PIN											
LLT													OVH							
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00		60000,00		
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					
32LMG12 28/10/2008	TORQUE LINK APEX PIN NUT PN : 201587612				SN : 08MSL5902386				NUT											
LLT													OVH							
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00		60000,00		
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					
32LMG13 28/10/2008	LOWER TORQUE LINK PN : 201540302				SN : 08BEL0113				LOWER TORQUE LINK											
LLT													OVH							
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00		60000,00		
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description								AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance				
	Installed Equipment (PN/SN)								PN Description		Specific Reference							
	WO number	WP	Scheduled check															
	Inspection				Inspection 2				Overhaul				Life Limit					
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline		
32LMG14 28/10/2008	LOWER TORQUE LINK PIN PN : 201160602 SN : 08MDG6745								PIN									
LLT												OVH						
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918				
32LMG15 28/10/2008	SLIDING TUBE AT LH MLG PN : 201371304 SN : 08B5151X6200								SLIDING ROD									
LLT												OVH						
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918				
32LMG16 28/10/2008	UPPER DIAPHRAGM TUBE PN : 201371615 SN : 08B0314X1								UPPER DIAPHRAGM TUBE									
LLT												OVH						
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918				
32LMG17 28/10/2008	LOWER CARDAN NUT PN : 201163945 SN : 2762/08								LOWER CARDAN NUT									
LLT												OVH						
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00		
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918				

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description										AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance
	Installed Equipment (PN/SN)										PN Description		Specific Reference			
	WO number	WP	Scheduled check		Inspection		Inspection 2		Overhaul		Life Limit					
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32LMG18 28/10/2008	LOWER SIDE STAY AT LH MLG PN : 201163301										SN : 07AP0228		LOWER SIDE STAY			
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG19 28/10/2008	SIDE STAY PIVOT PIN PN : 201163609										SN : 08AP0110		SIDE STAY PIVOT PIN			
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG20 28/10/2008	UPPER SIDE STAY PN : 201163300										SN : AP042206		UPPER SIDE STAY			
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG21 28/10/2008	SIDE STAY UPPER CARDAN PIN PN : 201163618										SN : 07AP0042		SIDE STAY UPR CARDAN PIN			
LLT												OVH				
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
	Inspection				Inspection 2				Overhaul				Life Limit			
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32LMG22 28/10/2008	UPPER CARDAN PN : 201163620		SN : 08B0549X5	UPPER CARDAN												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG23 28/10/2008	UPPER CARDAN NUT PN : 201163944		SN : 2186/07	UPPER CARDAN NUT												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG24 28/10/2008	UPPER PIN PN : 201661607		SN : 08AP0148	UPPER PIN (SPRINGS)												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		
32LMG25 28/10/2008	LOWER PIN PN : 201661606		SN : 08AP0123	LOWER PIN (SPRINGS)												
LLT																
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance	
	Installed Equipment (PN/SN)				PN Description				Specific Reference											
	WO number	WP	Scheduled check																	
Inspection				Inspection 2				Overhaul				Life Limit								
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline					
32LMG26 28/10/2008	TRIANGULATION LINK PIN				SN : 08AP0130				PIN											
	LLT											OVH								
	CY		16758,00			16758,00		20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918							
32LMG27 28/10/2008	TRIANGULATION LINK				SN : 08AP0202				LINK											
	LLT											OVH								
	CY		16758,00			16758,00		20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918							
32LMG28 28/10/2008	BANANA LINK				SN : 08AP0132				LINK BANANA											
	LLT											OVH								
	CY		16758,00			16758,00		20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918							
32LMG29 28/10/2008	BANANA LINK				SN : 08AP0133				LINK BANANA											
	LLT											OVH								
	CY		16758,00			16758,00		20000	16758,00	3242,00	20000,00	60000	16758,00	43242,00	60000,00					
Cal Total		3918			3918			3650	3918	-268	26/10/2018		3918							

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description															
	Installed Equipment (PN/SN)															
	WO number WP Scheduled check															
AMM																
F.I.N																
Maint. level																
Zone																
Phasing																
Tolerance																
PN Description																
Specific Reference																
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32LMG30 28/10/2008	CUFF PIN PN : 201661612 SN : 08AP0169 PIN															
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG31 28/10/2008	CUFF PIN PN : 201661612 SN : 08AP0163 PIN															
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG32 28/10/2008	CUFF PN : 201661613 SN : 07AP0226 CUFF															
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG33 28/10/2008	PIN PN : 201661605 SN : 08AP0030 PIN (LOCK/BANANA LINK)															
LLT												OVH				
CY		16758,00			16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance					
	Installed Equipment (PN/SN)				PN Description		Specific Reference								
	WO number	WP	Scheduled check												
Inspection				Inspection 2				Overhaul				Life Limit			
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32LMG34 28/10/2008	LOCK LINK LOWER PIN PN : 201057627				SN : 07AP0219		LOCK LINK LOWER PIN								
LLT CY		16758,00				16758,00		20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918		3650	3918	-268	26/10/2018		3918		
32LMG35 28/10/2008	LOWER LOCK LINK PN : 201058305				SN : 08AP0041		LOWER ARM								
LLT CY		16758,00				16758,00		20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918		3650	3918	-268	26/10/2018		3918		
32LMG36 28/10/2008	LOCK LINK CENTER PIN PN : 201058608				SN : 07AP0223		LOCK LINK CENTER PIN								
LLT CY		16758,00				16758,00		20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918		3650	3918	-268	26/10/2018		3918		
32LMG37 28/10/2008	UPPER LOCK LINK PN : 201058310				SN : 08AP0005		UPPER ARM								
LLT CY		16758,00				16758,00		20000	16758,00	3242,00	OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918				3918		3650	3918	-268	26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
Inspection		Inspection 2				Overhaul				Life Limit						
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32LMG38 28/10/2008	LOCK LINK UPPER PIN PN : 201058611		SN : 07AP0071	LOCK LINK UPPER PIN												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG39 28/10/2008	LOCK STAY CARDAN PN : 201058306		SN : 07COU45295X5723	COMPLETE CARDAN												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG40 28/10/2008	LOCK STAY CARDAN PIN PN : 201056909		SN : 08BEV81754019	LOCK STAY CARDAN PIN												
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00	60000,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG46 28/10/2008	SIDESTAY ASSY PN : 201166013-025		SN : AP1269	STAY ASSY-SIDE,MLG											TCI	
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance						
	Installed Equipment (PN/SN)				PN Description		Specific Reference									
	WO number	WP	Scheduled check													
Inspection				Inspection 2				Overhaul				Life Limit				
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32LMG47 28/10/2008	DAMPER TORQUE LINK				DAMPER, TORQUE LINK - MLG										O/C	
	PN : 201419001-020				SN : MG2205											
	LLT															
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C
Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C	O/C	O/C	3918	O/C	O/C	O/C	3918	O/C	
32LMG49 28/10/2008	RETRACTION ACTUATOR				ACTUATOR-RETRACTION		2503GM						731		TCI	
	PN : 201590002-020				SN : B5241											
	LLT															
	CY		16758,00			16758,00		20000	16758,00	3242,00	OVH	20000,00		16758,00		
Cal Total		3918			3918		3650	3918	-268		26/10/2018		3918			
32LMG50 28/10/2008	MONITORING UNIT BRK TEMP L				BTMU-BRAKE TEMPERATURE MO										O/C	
	PN : 35-1H5-1002				SN : 11944											
	LLT															
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C
Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C	O/C	O/C	3918	O/C	O/C	O/C	3918	O/C	
32LMG51 28/10/2008	HOUSING				HOUSING										TCI	
	PN : 201056669				SN : 08BEV81828072											
	LLT															
	CY		16758,00			16758,00		20000	16758,00	3242,00	OVH	20000,00		16758,00		
Cal Total		3918			3918		3650	3918	-268		26/10/2018		3918			

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description										AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance		
	Installed Equipment (PN/SN)										PN Description		Specific Reference					
	WO number	WP	Scheduled check		Inspection		Inspection 2		Overhaul		Life Limit							
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline		
32LMG52 28/10/2008	PIN (UPLOCK) PN : 201383648										SN : 08BEV82156012		PIN				TCI	
LLT																		
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918				
32LMG53 28/10/2008	PIN PN : 201056885										SN : 08BEV81971X214		PIN				TCI	
LLT																		
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918				
32LMG54 28/10/2008	SHOCK ABSORBER PN : 201371281										SN : 08B5151X6200		SHOCK ABSORBER				TCI	
LLT																		
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918				
32LMG55 28/10/2008	PIN PN : 201160317										SN : 08SOP87792X006		PIN				TCI	
LLT																		
CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00				
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918				

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance		
	Installed Equipment (PN/SN)				PN Description				Specific Reference												
	WO number	WP	Scheduled check																		
		Inspection				Inspection 2				Overhaul				Life Limit							
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline				
32LMG56 28/10/2008	NUT PN : D52041-1				SN : 08CZ85600X148				NUT				TCI								
	LLT												OVH								
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00						
Cal Total			3918				3918			3650	3918	-268	26/10/2018		3918						
32LMG57 28/10/2008	NUT PN : D52041-1				SN : 08CZ85600X34				NUT				TCI								
	LLT												OVH								
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00						
Cal Total			3918				3918			3650	3918	-268	26/10/2018		3918						
32LMG58 28/10/2008	PISTON PN : 201160693				SN : 08MG10734				PISTON				TCI								
	LLT												OVH								
	CY		16758,00				16758,00			20000	16758,00	3242,00	20000,00		16758,00						
Cal Total			3918				3918			3650	3918	-268	26/10/2018		3918						
32LMG59 28/10/2008	LOCKING SPRING ASSY PN : 201059001				SN : AP2569				LOCK SPRING				O/C								
	LLT																				
	FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	O/C	
	CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	O/C	
	Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C					
Cal Total			3918				3918			3650	3918	-268	26/10/2018		3918						

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance								
	Installed Equipment (PN/SN)		PN Description		Specific Reference											
	WO number	WP	Scheduled check													
		Inspection		Inspection 2		Overhaul		Life Limit								
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32LMG60 28/10/2008	SPRING PN : 201059602		SN : 017450	SPRING										TCI		
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG61 28/10/2008	LOCKING SPRING ASSY PN : 201059001		SN : AP2570	LOCK SPRING										O/C		
LLT																
FH	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C	O/C	21764,97	O/C	O/C
CY	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C	O/C	16758,00	O/C	O/C
Cal A/C Days	O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C		O/C	3918	O/C	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG62 28/10/2008	SPRING PN : 201059602		SN : 017494	SPRING										TCI		
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		
32LMG63 28/10/2008	LOCK LINK ASSY PN : 201058003		SN : AP1313	STAY-LOCK										TCI		
LLT																
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH		16758,00		
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918		

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance							
	Installed Equipment (PN/SN)		PN Description		Specific Reference										
	WO number	WP	Scheduled check												
Inspection				Inspection 2				Overhaul				Life Limit			
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline
32LMG65 28/10/2008	CYLINDER PN : 114095305		SN : 07AP0407	CYLINDER											TCI
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32LMG66 28/10/2008	PISTON ROD PN : 114095667		SN : 07AP0122	PISTON ROD											
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00	
Cal A/C Days		3918			3918			3653	3918	-265		29/10/2018		3918	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32LMG67 28/10/2008	EYE END PN : 114095602		SN : 07AP0247	EYE END											TCI
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00		16758,00	
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	
32LMG68 28/10/2008	CYLINDER PN : 201590906		SN : 08MDC00078	RTCN ACT CYLINDER											
LLT CY		16758,00			16758,00			20000	16758,00	3242,00		OVH 20000,00	60000	16758,00	43242,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018		3918	60000,00

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance	
	Installed Equipment (PN/SN)				PN Description				Specific Reference											
	WO number	WP	Scheduled check																	
	Inspection				Inspection 2				Overhaul				Life Limit							
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline				
32LMG69 28/10/2008	PISTON HEAD PN : 201590909				SN : 08JL985X150				PISTON HEAD											
LLT													OVH							
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00			60000,00	
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					
32LMG70 28/10/2008	NUT PN : 201590910				SN : 08JL1090X127				RTCN ACT PISTON HEAD NUT											
LLT													OVH							
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00			60000,00	
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					
32LMG71 28/10/2008	GLAND PN : 201590913				SN : 08JL986X50				RETRACTION ACT GLAND											
LLT													OVH							
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00			60000,00	
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					
32LMG72 28/10/2008	PISTON ROD PN : 201590908				SN : 07MDC00557				PISTON ROD											
LLT													OVH							
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00			60000,00	
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918					

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description				AMM				F.I.N				Maint. level		Zone		Phasing		Tolerance			
	Installed Equipment (PN/SN)				PN Description				Specific Reference													
	WO number	WP	Scheduled check																			
		Inspection				Inspection 2				Overhaul				Life Limit								
		BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline					
32LMG73 28/10/2008	PIN				RTCN ACTUATOR PIN (REAR)																	
	PN : 201590901				SN : 07BA58521X46																	
	LLT													OVH								
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00			60000,00			
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918							
32LMG74 28/10/2008	PIN				RTCN ACTUATOR PIN (MAIN)																	
	PN : 201590900				SN : 08BA59459X44																	
	LLT													OVH								
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00	60000	16758,00	43242,00			60000,00			
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918							
32LMG77 28/10/2008	PIN (RETAINING LOWER BEARING)				RETAINIG PIN																TCI	
	PN : 201383608				SN : 08BEV81759009																	
	LLT													OVH								
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00		16758,00							
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918							
32LMG78 28/10/2008	PIN (RETAINING LOWER BEARING)				RETAINIG PIN																TCI	
	PN : 201383608				SN : 08BEV81759010																	
	LLT													OVH								
CY		16758,00				16758,00			20000	16758,00	3242,00		20000,00		16758,00							
Cal Total		3918				3918			3650	3918	-268		26/10/2018		3918							

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance									
	Installed Equipment (PN/SN)		PN Description		Specific Reference												
	WO number	WP	Scheduled check														
		Inspection		Inspection 2		Overhaul		Life Limit									
BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline		
32LMG79 28/10/2008	PIN (RETAINING LOWER BEARING) PN : 201383608		SN : 08BEV81759011	RETAINIG PIN								TCI					
LLT																	
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH	20000,00	16758,00			
Cal Total		3918			3918			3650	3918	-268		26/10/2018	3918				
32LMG83 28/10/2008	BASIC SIDE STAY PN : 201163004-015		SN : AP1269	STAY-SIDE								TCI					
LLT																	
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH	20000,00	16758,00			
Cal Total		3918			3918			3650	3918	-268		26/10/2018	3918				
32LMG84 28/10/2008	LEG AND DRESSINGS-MLG PN : 201582001-040		SN : MDG4845	LEG AND DRESSINGS-MLG								TCI					
LLT																	
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH	20000,00	16758,00			
Cal Total		3918			3918			3650	3918	-268		26/10/2018	3918				
32NG00 28/10/2008	HINGE PIN PN : D65588-1		SN : 07JL113X78	PIN, HINGE													
LLT																	
CY		16758,00			16758,00			20000	16758,00	3242,00		OVH	20000,00	31400	16758,00	14642,00	31400,00
Cal Total		3918			3918			3650	3918	-268		26/10/2018	3918				

Aircraft Remaining Potentials : PR-ONR (PR-ONR) Effectivity : A318

Position Instal. Date	Kardex Description		AMM	F.I.N	Maint. level	Zone	Phasing	Tolerance									
	Installed Equipment (PN/SN)		PN Description		Specific Reference												
	WO number	WP	Scheduled check														
	Inspection				Inspection 2				Overhaul				Life Limit				
	BI	SI	Remain	Deadline	BI2	SI2	Remain	Deadline	BO	SO	Remain	Deadline	LL	SN	Remain	Deadline	
32NG01 28/10/2008	HINGE PIN PN : D65588-1		SN : 07JL113X64	PIN, HINGE													
LLT																	
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			
32NG02 28/10/2008	BARREL PN : D67583		SN : 08B0300X8772	BARREL													
LLT																	
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			
32NG03 28/10/2008	PIN PN : D65589-1		SN : 07JL120X05	PIN													
LLT																	
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			
32NG04 28/10/2008	PIN PN : D65599-1		SN : 07JL121X17	PIN													
LLT																	
CY		16758,00				16758,00			20000	16758,00	3242,00	OVH 20000,00	31400	16758,00	14642,00	31400,00	
Cal Total		3918				3918			3650	3918	-268	26/10/2018		3918			

Pool Request Exchange Order

Number: PX0402122 Date: 02 Feb 2022
 Proforma Invoice Packing Sheet No:



ODPX0402122

Kindly quote above order no. and reference in all correspondence and when sending parts and invoice

PROVIDER

OEMSERVICES
 86 ROUTE DE ROISSY
 93290 - TEMBLAY EN FRANCE
 FRANCE
 E-MAIL: a320.etd@oemservices.aero
 CONTACT: EMEA DEPARTMENT

Priority: REPLENISHMENT

ISSUED BY

ETIHAD AIRWAYS
 NEW AIRPORT ROAD
 KHALIFA CITY A
 35566 - ABU DHABI
 UNITED ARAB EMIRATES
 PHONE: 00971 2 5110000

Responsible: **Abduls. Reynan**
 Email: **RAbduls@etihad.ae**

Item	Qty	UOM	Partnumber	Description	Batch Number
1	1	EA	201590003	RETRACTION ACTUATOR	11B18221
Label No: 1092350		TSN: 34100:08		CSN: 13245	Manufac. Date: 21.Mar.2012
		TSO: 34100:08		CSO: 13245	
		Cost Center: PCOR0608		Cost Type:	
Requested for A/C: EIP			Target Date: 05.Feb.2022		Value for Customs: 0.00 AED
Requested for A/C Serialno, Type: 05095, A			Target Time:		Confirmed Date: --
Condition: UNSERVICEABLE					
*** Requested Partnumber: 201590003 ***					
Item Text					
REMOVED FROM A6-EIP Replace RETRACTION ACTUATOR(201590003/11B18221). SHOP VISIT ON PN: 201590004 ON SN: 14MDC10835					
Requirement					
201590003_SD - REMC	Dimension: Months		TOGO: 47d, 16h		
	Interval: 120	Since new: 21.Mar.2012		Since Req: 21.Mar.2012	
201590003_SD - REMC	Dimension: Flight Cycles		TOGO: 6'755		
	Interval: 20'000	Since new: 13'245		Since Req: 13'245	
201590003_SD - REMC	Dimension: Months		TOGO: 47d, 16h		
	Interval: 120	Since new: 21.Mar.2012		Since Req: 21.Mar.2012	

201590003_SD - REMC	Dimension: Flight Cycles		TOGO: 6'755
	Interval: 20'000	Since new: 13'245	Since Req: 13'245
201590003_SD - REMC	Dimension: Months		TOGO: 47d, 16h
	Interval: 120	Since new: 21.Mar.2012	Since Req: 21.Mar.2012
201590003_SD - REMC	Dimension: Flight Cycles		TOGO: 6'755
	Interval: 20'000	Since new: 13'245	Since Req: 13'245
201590909_LL - DISCA	Dimension: Flight Cycles		TOGO: 46'755
	Interval: 60'000	Since new: 13'245	Since Req: 13'245
201590908_LL - DISCA	Dimension: Flight Cycles		TOGO: 46'755
	Interval: 60'000	Since new: 13'245	Since Req: 13'245
201590901_LL - DISCA	Dimension: Flight Cycles		TOGO: 46'755
	Interval: 60'000	Since new: 13'245	Since Req: 13'245
201590900_LL - DISCA	Dimension: Flight Cycles		TOGO: 46'755
	Interval: 60'000	Since new: 13'245	Since Req: 13'245
201590913_LL - DISCA	Dimension: Flight Cycles		TOGO: 46'755
	Interval: 60'000	Since new: 13'245	Since Req: 13'245
201590910_LL - DISCA	Dimension: Flight Cycles		TOGO: 46'755
	Interval: 60'000	Since new: 13'245	Since Req: 13'245
201697600_LL - DISCA	Dimension: Flight Cycles		TOGO: 46'755
	Interval: 60'000	Since new: 13'245	Since Req: 13'245

SHIP TO

OEMSERVICES
 86 ROUTE DE ROISSY
 93290 - TEMBLAY EN FRANCE
 FRANCE
 E-MAIL: a320.etd@oemservices.aero
 CONTACT: EMEA DEPARTMENT

RETURN TO

ETIHAD AIRWAYS ENGINEERING
 P.O. BOX 46450
 ADJACENT ABU DHABI INT'L AIRPORT
 ABU DHABI
 UNITED ARAB EMIRATES
 CONTACT: +971(0)2-5057332

A PART USED ON THE A/C

04 February 2022
Ref: EY/TR/2022/031

GENERAL NON INCIDENT STATEMENT (NIS) ETD A320 with OEMS

To Whom It May Concern:

We hereby confirm that all aircraft components supplied to OEM SERVICES SAS by ETIHAD AIRWAYS between 01 Jan 2022 to 31 December 2022 were in accordance with Etihad Airways procedures, obtained from industry recognized approved sources and whilst in the possession of Etihad Airways and use on its A320 Family fleet were not removed from an aircraft (or its constituent major components such as engine, APU or landing gear):

1. That was involved in an accident, incident, major failure or fire nor was the engine subjected to extreme stress or heat (e.g., engine failure, fire, accident) as defined by the regulating authority;
2. That was immersed in salt water or otherwise exposed to corrosive agents outside normal operation.
3. That was obtained from, nor operated by any Government, or any military sources.
4. That operated in conjunction with any parts installed that were non type certificate holder repaired parts (no repairs outside of the OEM manuals other than those approved by the OEM) or PMA parts other than those authorized by the OEM.

In the event that a component does not conform to the above, Etihad Airways will issue an individual statement declaring full details.


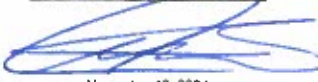
Approved by

Gopinath Paulnadat
Airbus Fleet Manager (Acting)
Tel: +971 2 511 5463
gpaulnadat@etihad.ae

SECTION 3. REPAIR APPROVALS

SECTION 4. MODIFICATION STATUS

MODIFICATION	DESCRIPTION	STATUS
SB 200-32-220	MLG-TORQUE LINK DAMPER UNIT-INTRODUCTION OF A CLAMP.	CW
SB 200-32-230	IDENTIFICATION OF SAFE LIFE OF COMPONENTS AND UNITS.	PCW
SB 200-32-250	INSPECTION OF LOCKING BUSH AND SLIDING TUBE.	CW
SB 200-32-253	MLG - RETRACTION ACTUATOR - NEW SPHERICAL BEARING SUB ASSY.	PCW
SB 200-32-254	MAIN LANDING GEAR - INSTALLATION OF LOCK STAY 201058003 TO 201376 AND 201175 SERIES MAIN LANDING GEARS.	PCW
SB 200-32-261	MLG- INSTALLATION OF TWO STUB BOLTS FOR THE FWD. PINTLE PIN IN PLACE OF THE CROSS BOLT.	PCW
SB 200-32-267	MLG- INTRODUCTION OF A NEW CUFF SUB-ASSEMBLY AND BANANA LINK SUB-ASSEMBLY.	PCW
SB 200-32-268	MLG- TORQUE LINK DAMPER UNIT- INTRODUCTION OF IMPROVED DISK SPRINGS.	PCW
SB 200-32-271	MLG- TO ADD TRACKING NUMBERS TO PARTS LISTED IN AIRBUS DOCUMENT MPD, SECTION 9-1. (RETRACTION ACTUATOR PINS AND FWD. PINTLE PIN CROSS BOLTS).	PCW
SB 200-32-282	MLG- TO ADD TRACKING NUMBERS TO PARTS LISTED IN AIRBUS DOCUMENT MPD, SECTION 9-1 (TORQUE LINK APEX PIN NUT).	PCW
SB 200-32-283	MLG- TO ADD TRACKING NUMBERS TO PARTS LISTED IN AIRBUS DOCUMENT MPD, SECTION 9-1 (SIDE STAY).	PCW
SB 200-32-284	MLG- TO ADD TRACKING NUMBERS TO PARTS LISTED IN AIRBUS DOCUMENT MPD, SECTION 9-1 (RETRACTION ACTUATOR).	PCW
SB 200-32-285	MLG- INTRODUCTION OF A NEW GLAND IN THE LOCK STAY ACTUATOR.	PCW
SB 200-32-286	MLG- INSPECTION OF THE SLIDING TUBE FOR CRACKS.	S/N NOT AFFECTED
SB 200-32-295	LANDING GEAR-MLG-DEFINITION OF RESTORATION TASKS FOR THE MAIN LANDING GEAR.	CW
SB 200-32-297	MLG- INTRODUCTION OF A NEW LOWER BEARING SUB-ASSY.	PCW
SB 200-32-302	MLG- INTRODUCTION OF NEW CHARGING LABELS.	PCW
SB 200-32-312	MLG- INTRODUCTION OF A NEW UPPER CARDAN AND MODIFICATION OF CURRENT P/N. 201163620. MODIFIED TO 201163948.	CW
SB 200-32-313	MLG- BARKHAUSEN NOISE INSPECTION OF MLG SLIDING TUBE AXLES.	S/N NOT AFFECTED
SB 200-32-315	LANDING GEAR - MAIN LANDING GEAR COMPLETE - SIDE STAY COMPLETE - SIDE STAY BASIC - INTRODUCTION OF NE END CAPS AND BOLTS FOR THE UPPER AND LOWER CARDAN JOINTS.	CW
SB 200-32-321	LANDING GEAR- MAIN GEAR INSPECTION SLIDING TUBE / SLAVE LINK LUG CRACK (REFERENCE AIRBUS SB A320-32-1441).	CW
SB 200-32-335	LANDING GEAR - MAIN LANDING GEAR COMPLETE - MAIN LANDING GEAR LEG - INSPECTION OF MAIN LANDING GEAR SLIDING TUBE AXLES.	S/N NOT AFFECTED
SB A320-32A1223	MLG- INTRODUCTION OF A NEW CUFF SUB-ASSEMBLY AND BANANA LINK SUB-ASSEMBLY.	PCW
SB A320-32-1441	"MANDATORY" LANDING GEAR- MAIN GEAR INSPECTION SLIDING TUBE / SLAVE LINK LUG CRACK (REFERENCE SAFRAN SB 200-32-321).	CW
FAA AD 2000-11-09	INSPECTION OF LOCKING BUSH AND SLIDING TUBE. AGREES WITH SB 200-32-250.	CW
FAA AD 2002-06-01	MLG- INSTALLATION OF TWO STUB BOLTS FOR THE FWD. PINTLE PIN IN PLACE OF THE CROSS BOLT. AGREES WITH SB 200-32-261.	PCW

Component Overhaul Services, Corp. FAA Repair Station: #8COR883B EASA 145.6404	CUSTOMER: CAVU AEROSPACE, INC.	WORK ORDER: 12616	
	PART NUMBER: 201582001-040 SERIAL NUMBER: MDG4845		
MODIFICATION	DESCRIPTION		STATUS
FAA AD 2005-17-07	THIS AD RESULTS FROM A REPORT THAT AN AXLE NUT HAD SEPARATED FROM AN AXLE ON A MLG WHEEL DUE TO MISSING LOCKING BOLTS. THE FAA IS ISSUING THIS AD TO DETECT AND CORRECT MISSING LOCKING BOLTS ON THE AXLE NUTS OF THE MLG WHEELS. ABSENCE OF THE LOCKING BOLTS COULD RESULT IN SEPARATION OF A WHEEL (S) FROM THE AXLE AND CONSEQUENT REDUCE THE CONTROLLABILITY OF THE AIRPLANE DURING TAKEOFF AND LANDING, AND POSSIBLE INJURY TO PEOPLE ON THE GROUND (REFERENCE AOT A320-32A1303).		CW
FAA AD 2005-10-04	TO PREVENT FAILURE OF THE MLG SIDE-STAY CUFF LUGS OR DOWN-LOCK SPRING ATTACHMENTS, WHICH COULD RESULT IN IMPROPER DOWN-LOCK OF THE MLG DURING A FREEFALL EXTENSION, AND POSSIBLE COLLAPSE OF THE MLG		PCW
FAA AD 2015-22-08	MLG- INTRODUCTION OF A NEW UPPER CARDAN AND MODIFICATION OF CURRENT P/N: 201163620. MODIFIED TO 201163948. AGREES WITH SB 200-32-312		CW
FAA AD 2017-18-21	MLG - INTRODUCE DOUBLE LOCKING SIDESTAY CARDAN JOINTS. AGREES WITH SB 200-32-315.		CW
FAA AD 2019-14-06	Landing Gear – Main Landing Gear Sliding Tubes – Inspection		S/N NOT AFFECTED
FAA AD 2020-14-08	Landing Gear – Main Landing Gear Torque Link Apex Pin – Replacement / Inspection		CW
FAA AD 2022-12-05	Landing Gear – Main Landing Gear Sliding Tubes – Inspection		CW
FAA AD 2022-22-10	THE FAA IS ISSUING THIS AD TO ADDRESS THE FAILURE OF CERTAIN LIFE-LIMITED PARTS, WHICH COULD RESULT IN REDUCED STRUCTURAL INTEGRITY OF THE AIRPLANE. SEE THE MCAI FOR ADDITIONAL BACKGROUND INFORMATION.		PCW
FAA AD 2023-01-01	Landing Gear – NLG & MLG Landing Gear Sliding Tube – Inspection / Replacement		S/N NOT AFFECTED
FAA AD 2023-23-11	Landing Gear – Landing Gear Sliding Tube – Inspection / Replacement		S/N NOT AFFECTED
EASA AD 2011-0024R1	LANDING GEAR-MAIN LANDING GEAR SIDE STAY CUFF LUG INSPECTION/REPAIR (REFERENCE AIRBUS SB A320-32A1223 AND 32A1224).		PCW
EASA AD 2016-0018R1	LANDING GEAR - MAIN LANDING GEAR COMPLETE - SIDE STAY COMPLETE - SIDE STAY BASIC- INTRODUCTION OF NE END CAPS AND BOLTS FOR THE UPPER AND LOWER CARDAN JOINTS. AGREES WITH SB 200-32-315		CW
EASA AD 2019-0151-E	Landing Gear – Main Landing Gear Sliding Tubes – Inspection		S/N NOT AFFECTED
EASA AD 2020-0130	Landing Gear – Main Landing Gear Torque Link Apex Pin – Replacement / Inspection		CW
EASA AD 2020-0193	Landing Gear – Main Landing Gear Sliding Tubes – Inspection / Replacement		CW
EASA AD 2021-0236	Landing Gear – NLG & MLG Landing Gear Sliding Tube – Inspection / Replacement		S/N NOT AFFECTED
EASA AD 2021-0175	Landing Gear – Main Landing Gear Sliding Tubes – Inspection		S/N NOT AFFECTED
EASA 2022-0204R1	Landing Gear – Main Landing Gear Sliding Tube – Inspection / Replacement		CW
EASA 2024-0066	Time Limits / Maintenance Checks – Safe Life Airworthiness Limitations Items – ALS Part 1		CW
COMPONENT OVERHAUL SERVICES, CORP. CERTIFIES THAT TO THE BEST OF OUR KNOWLEDGE THE COMPONENTS THAT MAKE UP THIS ASSEMBLY WERE NOT OBTAINED FROM ANY U.S. GOVERNMENT OR MILITARY SOURCE AND HAVE NOT BEEN SUBJECTED TO EXTREME HEAT OR STRESS (AS IN A MAJOR ENGINE FAILURE, ACCIDENT, OR FIRE.)			
MODIFICATION STATUS LEGEND CW: COMPLIED WITH PCW: PREVIOUSLY COMPLIED WITH NC: NOT COMPLIED N/A: NOT APPLICABLE		QUALITY ENGINEERING APPROVAL  November 18, 2024	

SECTION 5. INSPECTION STATUS REPORT

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: CUFF

W/ORDER: 12616

P/NUMB.: 201661613

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 07AP0226

SUB W/O: 12616-24

Component Overhaul
Services

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A		2.4840	2.4050	2.4050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2050	201057642	0.0010	2.406
1B			2.4050	2.4050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2060	201057642	0.0010	2.406
2A		1.8120	1.8480	1.8480	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.0450	201057642		
							2.0460			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201057642		0.098
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201057642		0.098
3A	0.6319	NOTE 1	0.6320	0.6320	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-08M	0.0010	0.633
3B	0.6323		0.6320	0.6320	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	4.7237	BCA2-12-08M	0.0010	0.633
3C	0.6319	NOTE 1	0.6320	0.6320	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-08M	0.0010	0.633
3D	0.6323		0.6320	0.6320	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4737	BCA2-12-08M	0.0010	0.633
4A			0.8300	0.8300	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7083	BCA2-12-08M		
	REF.						0.7184			
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-08M		0.061
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-08M		0.061
4B			0.8300	0.8300	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7083	BCA2-12-08M		
	REF.						0.7184			
4B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-08M		0.061
4B2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-08M		0.061
5A	INTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
6A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
7A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: CUFF

W/ORDER: 12616

P/NUMB.: 201661613

OHM CHAP.: 32-11-26

S/NUMB.: 07AP0226

SUB W/O: 12616-24

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT



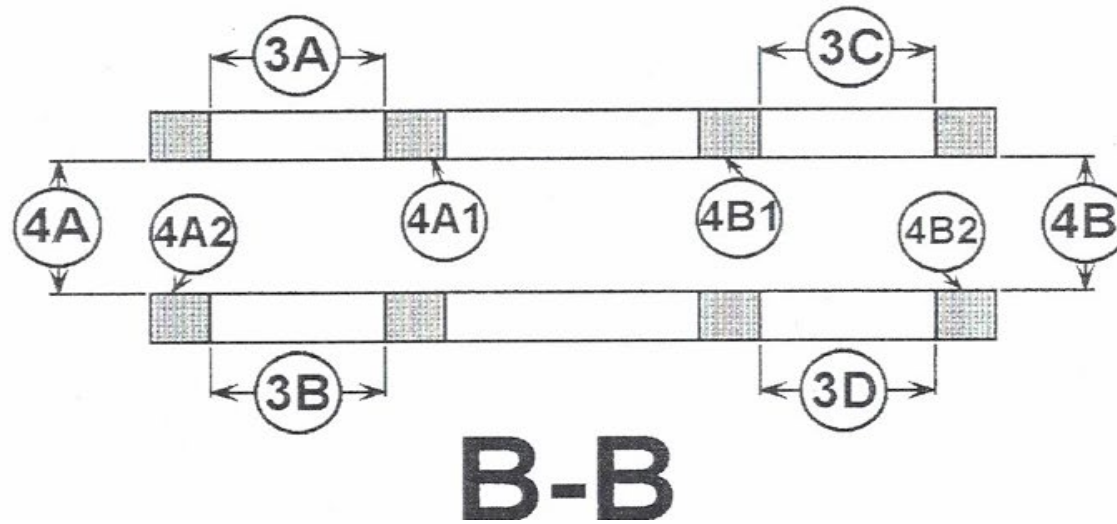
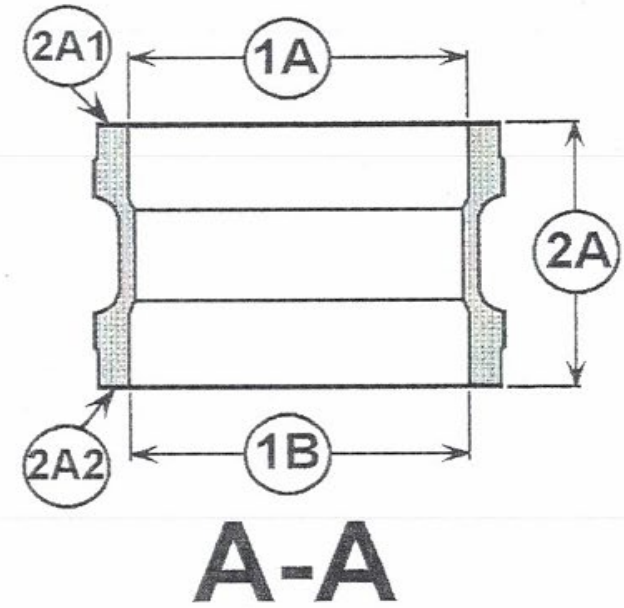
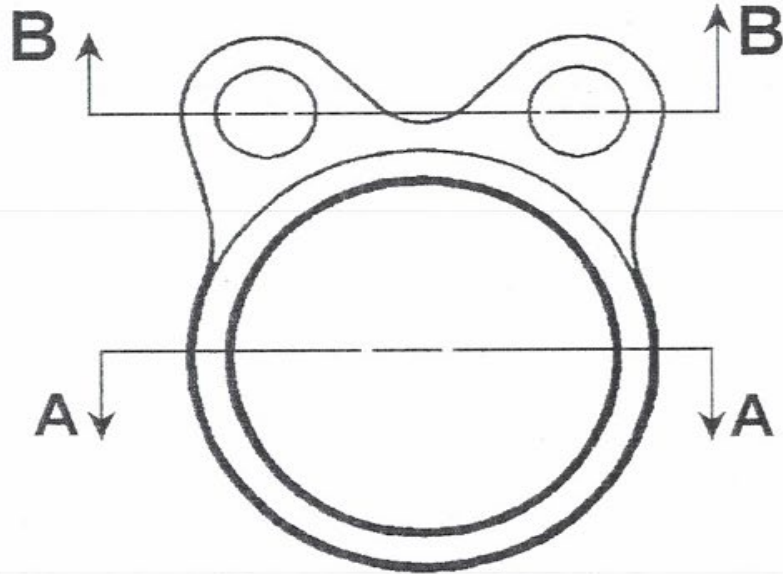
Component Overhaul
Services

NOTE 1: IF BORE IS OVERSIZED OR REQUIURES MACHINING SCRAP THE UNIT.

A/C: A320
P/NAME: CUFF
P/NUMB.: 201661613
S/NUMB.: 07AP0226
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12616
OHM CHAP.: 32-11-26
SUB W/O: 12616-24
C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: TRIANGULAR LINK

W/ORDER: 12616

INSPECTION STATUS REPORT

P/NUMB.: 201661307

OHM CHAP.: 32-11-26



S/NUMB.: 08AP0202

SUB W/O: 12616-102

LIFE LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	0.6306	0.6914	0.6311	0.6311	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-06M	0.0015	0.633
	0.6311						0.4737			
2A		0.4696	0.5080	0.5080	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6192	BCA2-12-06M		
							0.6283			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.056
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.056
3A	0.7506		0.7511	0.7511	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-15-12M	0.0015	0.753
3B	0.7511		0.7511	0.7511	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4737	BCA2-15-12M	0.0015	0.753
4A			2.0850	2.0850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.1940	BCA2-15-12M		
							2.2040			
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-15-12M		0.055
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-15-12M		0.055
5A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
6A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: TRIANGULAR LINK

W/ORDER: 12616

P/NUMB.: 201661307

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT



Component Overhaul
Services

S/NUMB.: 08AP0202

SUB W/O: 12616-102

LIF. LIMIT: 60, 000

C.S.N.: 16,758

NOTE 1: IF BORE OVERSIZED OR REQUIRES MACHINING SCRAP THE UNIT

A/C: A320

P/NAME: TRIANGULAR LINK

P/NUMB.: 201661307

S/NUMB.: 08AP0202

LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.

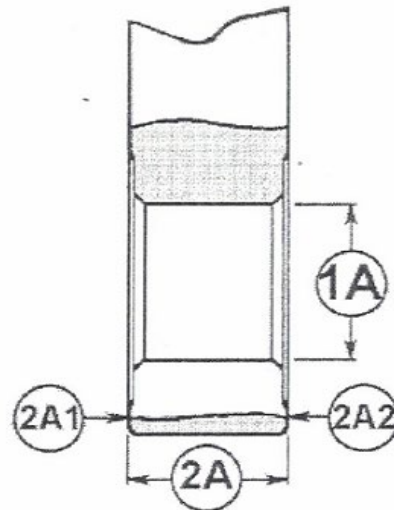
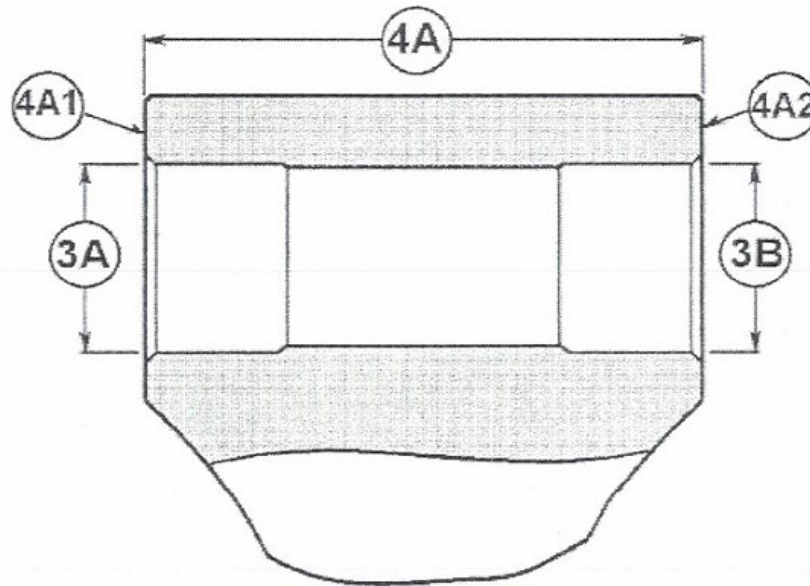
W/ORDER: 12616

OHM CHAP.: 32-11-26

SUB W/O: 12616-102

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: LOWER LOCK LINK

W/ORDER: 12616

P/NUMB.: 201058305

OHM CHAP.: 32-12-27

INSPECTION STATUS REPORT

S/NUMB.: 08AP0041

SUB W/O: 12616-97

Component Overhaul
Services

LIFE LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	0.9843	1.0599	0.9851	0.9851	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058601	0.0010	0.986
1B	0.9851		0.9851	0.9851	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7882	201058601	0.0010	0.986
2A		2.7924	2.8340	2.8340	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.0211	201058601		
							3.0303			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058601		0.094
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058601		0.094
3A	0.6319	0.7240	0.6323	0.6323	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-06M	0.0010	0.633
3B	0.6323		0.6323	0.6323	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4737	BCA2-12-06M	0.0010	0.633
4A		0.4698	0.5090	0.5090	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6196	BCA2-12-06M		
		NOTE 2					0.6283			
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
5A	0.7480	0.8150	0.7489	0.7489	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201058600	0.0010	0.750
5B	0.7489		0.7489	0.7489	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5913	201058600	0.0010	0.750
6A			1.7330	1.7330	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.7717			
							1.7828			
6A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				0.019
6A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				0.019
7A		1.0663	1.0250	1.0250	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.8701	201058600		
							0.8760			
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058600		0.077
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058600		0.077
8A	0.9843	1.0589	0.9851	0.9851	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058612	0.0010	0.986
8B	0.9851		0.9851	0.9851	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7882	201058612	0.0010	0.986
9A			2.0850	2.0850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
9A1	FACES	NOTE 2	COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				-1.043
9A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				-1.043

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: LOWER LOCK LINK

W/ORDER: 12616

P/NUMB.: 201058305

OHM CHAP.: 32-12-27

INSPECTION STATUS REPORT



Component Overhaul Services

S/NUMB.: 08AP0041

SUB W/O: 12616-97

C.S.N.: 16,758

LIFE LIMIT: 60,000

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
10A		1.3425	1.3000	1.3000	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.1417	201058600		
							1.1430			
10A1	FACES	NOTE 2	COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058600		0.079
10A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058600		0.079
11A	EXTERNAL SURFACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
12A	OTHER REJECTIONS		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: LOWER LOCK LINK

W/ORDER: 12616

P/NUMB.: 201058305

OHM CHAP.: 32-12-27

S/NUMB.: 08AP0041

SUB W/O: 12616-97

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT



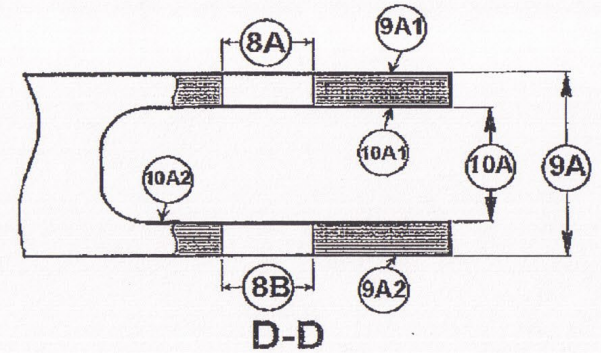
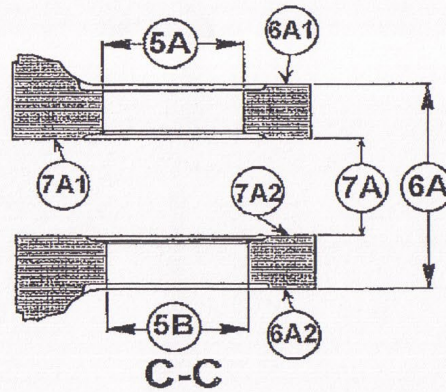
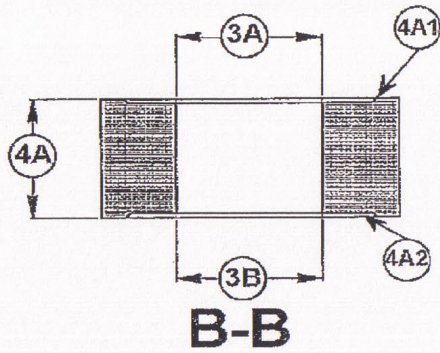
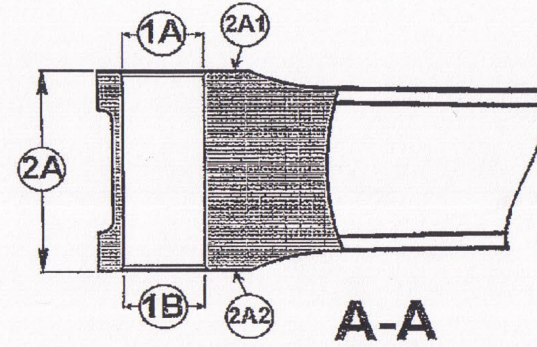
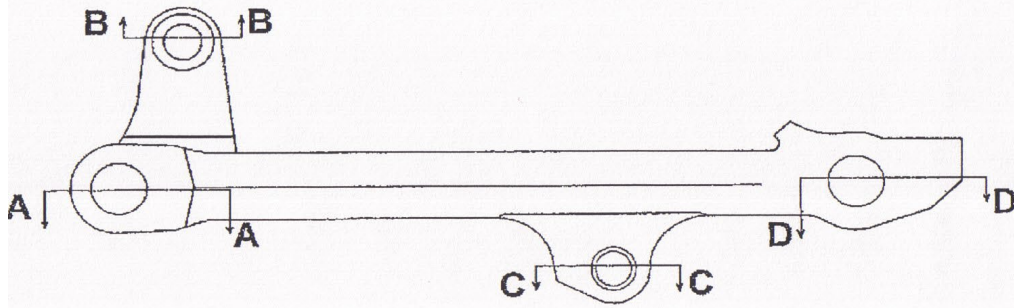
NOTE 1: IF MACHINED IS REQUIRED TO ORDER O/S BUSHING P/N: 899006001 QTY 2

NOTE 2: FACE MACHINING (SPOTFACE DIAMETER OF 1.063)

A/C: A320
P/NAME: LOWER LOCK LINK
P/NUMB.: 201058305
S/NUMB.: 08AP0041
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12616
OHM CHAP.: 32-12-27
SUB W/O: 12616-97
C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: UPPER LOCK LINK

W/ORDER: 12616

P/NUMB.: 201058310

OHM CHAP.: 32-12-27

INSPECTION STATUS REPORT

S/NUMB.: 08AP0005

SUB W/O: 12616-67

Component Overhaul
Services

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	0.9843	1.0600	0.9851	0.9851	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058613	0.0010	0.986
1B	0.9851	NOTE 1	0.9851	0.9851	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7882	201058613	0.0010	0.986
2A	0.9830	0.9428	0.9830	0.9830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.1401	201058613		
	REF.	NOTE 2					1.1409			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058613		0.079
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058613		0.079
3A	0.1929	0.2913	0.1980	0.1980	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1929	201058310-RS3	0.0010	0.199
3B	0.2008		0.1980	0.1980	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.2008	201058310-RS3	0.0010	0.199
3C	0.1929	0.2913	0.1980	0.1980	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1929	201058310-RS3	0.0010	0.199
3D	0.2008		0.1980	0.1980	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.2008	201058310-RS3	0.0010	0.199
4A			COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
4B			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
5A	0.9843	1.0600	0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058601	0.0010	0.986
5B	0.9851	NOTE 1	0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7882	201058601	0.0010	0.986
6A	2.0768		2.0890	2.0890	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.1180			
	2.0965						2.1300			
6A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				0.015
6A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				0.015
7A	1.3010	1.3430	1.3010	1.3010	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.1024	201058601		
	REF.	NOTE 2					1.1037			
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058601		0.099
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058601		0.099
8A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
9A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: UPPER LOCK LINK

W/ORDER: 12616

P/NUMB.: 201058310

OHM CHAP.: 32-12-27

S/NUMB.: 08AP0005

SUB W/O: 12616-67

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT



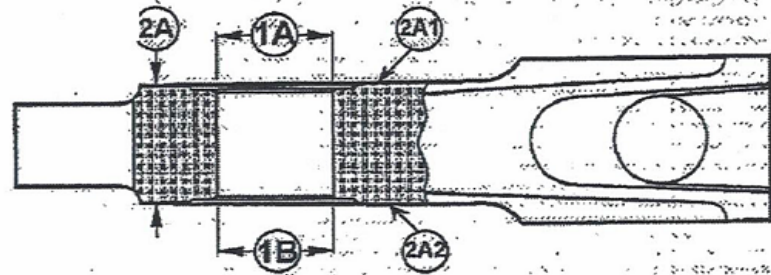
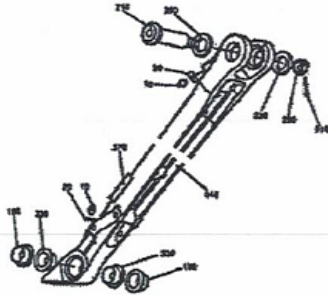
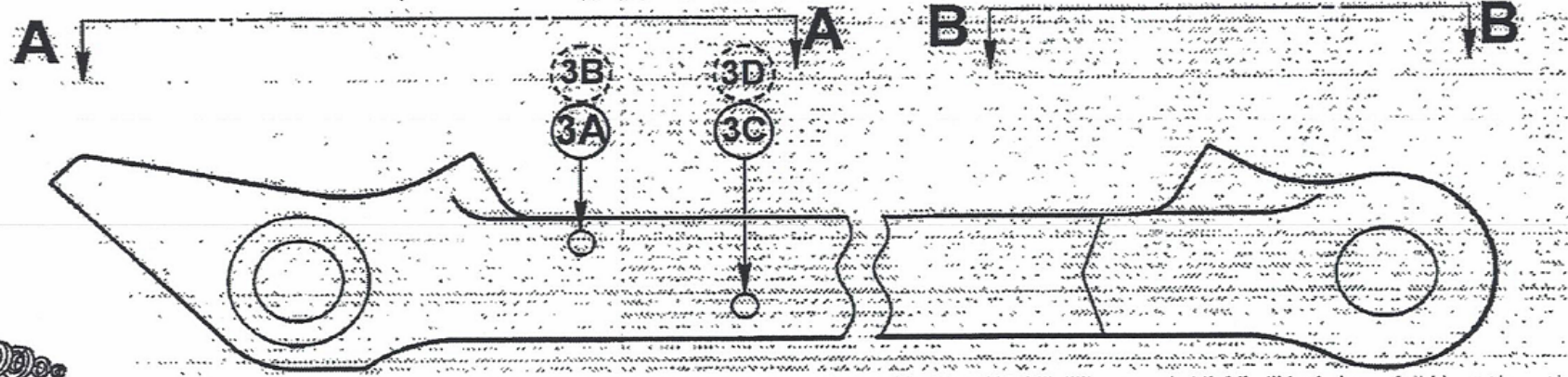
NOTE 1: O/S BUSHINGS

NOTE 2: O/S FLANGE BUSHINGS

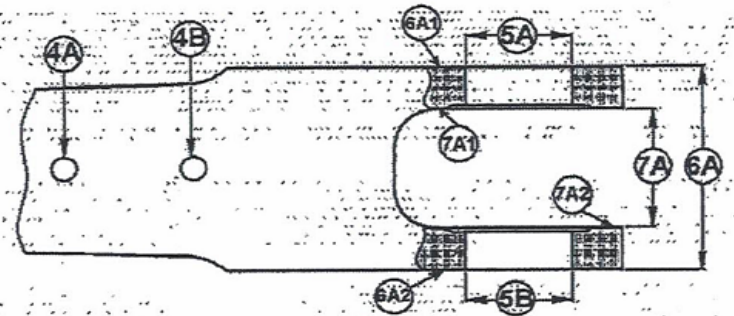
A/C: A320
P/NAME: UPPER LOCK LINK
P/NUMB.: 201058310
S/NUMB.: 08AP0005
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12616
OHM CHAP.: 32-12-27
SUB W/O: 12616-67
C.S.N.: 16,758

INSPECTION STATUS REPORT



A-A



B-B

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: STAY, LOWER SIDE

W/ORDER: 12616

P/NUMB.: 201163301

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 07AP0228

SUB W/O: 12616-122

Component Overhaul
Services

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	2.4016	2.4850	2.4028	2.4028	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2048	201163603	0.0020	2.405
1B	2.4028	NOTE1	2.4028	2.4028	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2059	201163603	0.0020	2.405
2A		7.7660	7.8250	7.8250	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
		NOTE 2								
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
3A		5.6772	5.6350	5.6350	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.4331	201163603		
		NOTE 2					5.4356			
3A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163603		0.101
3A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163603		0.101
4A		1.0490	1.0960	1.0960	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
4B			1.0960	1.0960	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
5A	2.4016	2.4850	2.4028	2.4028	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.0866	201163621	0.0020	2.405
5B	2.4028	NOTE 1	2.4028	2.4028	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.0878	201163621	0.0020	2.405
6A	2.6400	2.6840	2.6400	2.6400	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.4409	201163621		
	REF.						2.4429			
6A1	FACES	NOTE 2	COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163621		0.100
6A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163621		0.100
7A		1.0150	1.0380	1.0380	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7B			1.0380	1.0380	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
8A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
9A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTION		ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: STAY, LOWER SIDE

W/ORDER: 12616

P/NUMB.: 201163301

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 07AP0228

SUB W/O: 12616-122



LIF. LIMIT: 60,000

C.S.N.: 16,758

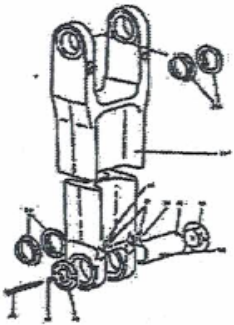
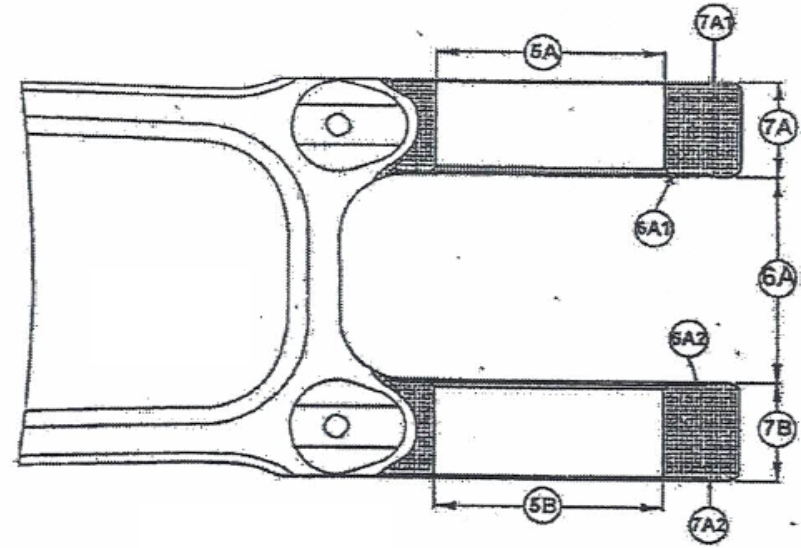
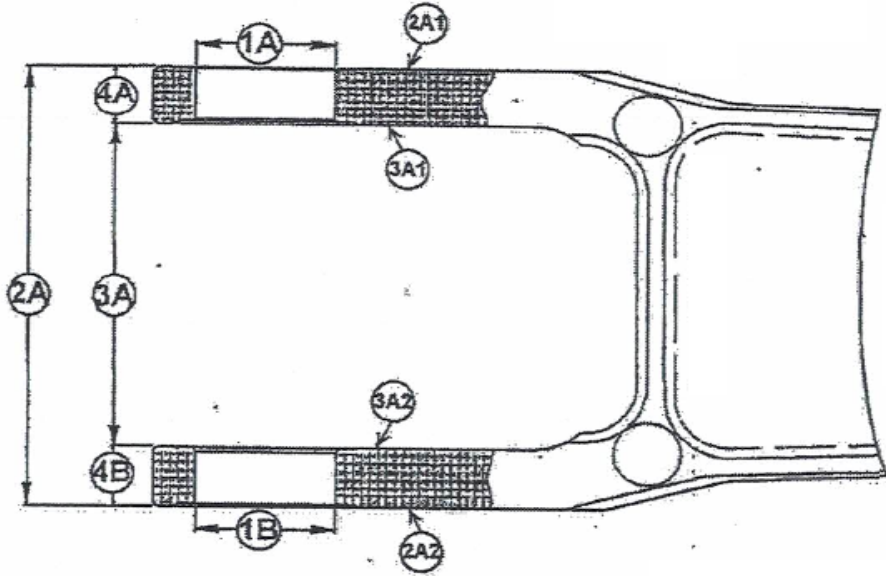
NOTE 1: FOR OVERSIZED

NOTE 2: FOR OVERSIZED FLNGE BUSHINGS

A/C: A320
P/NAME: STAY, LOWER SIDE
P/NUMB.: 201163301
S/NUMB.: 07AP0228
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12616
OHM CHAP.: 32-11-26
SUB W/O: 12616-122
C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: STAY, UPPER SIDE

W/ORDER: 12616

P/NUMB.: 201163300

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: AP042206

SUB W/O: 12616-115

Component Overhaul
Services

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	2.4016	2.4850	2.4020	2.4020	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2048	201163601	0.0015	2.404
1B	2.4028		2.4020	2.4020	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2059	201163601	0.0015	2.404
2A		5.0354	5.0700	5.0700	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.2724	201163601		
		NOTE 5					5.2738			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163601		0.101
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163601		0.101
3A		2.4480	2.4020	2.4020	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2048	201163601		
		NOTE 4					2.2065			
3A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163601		0.099
3A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163601		0.099
4A	0.9843	1.0600	0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201163637	0.0020	0.987
4B	0.9851		0.9850	0.9850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7882	201163637	0.0020	0.987
5A	3.3450		3.3470	3.3470	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	3.1490	201163637		
	REF.						3.1510			
5A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.0950	201163637		0.099
5A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.0950	201163637		0.099
5B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
5B2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
6A	0.7494	0.8236	0.7494	0.7494	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5908	BCA2-15-15M	0.0020	0.751
6B	0.7489		0.7494	0.7494	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5919	BCA2-15-15M	0.0020	0.751
7A			2.4010	2.4010	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2840	BCA2-15-15M		
							2.2980			
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-15-15M		0.059
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-15-15M		0.059
7B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7B2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
8A	0.7874	0.8630	0.7880	0.7880	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5905	201163602	0.0020	0.790
8B	0.7882		0.7880	0.7880	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5912	201163602	0.0020	0.790
9A		2.0910	2.0470	2.0470	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.8543	201163602		
							1.8602			
9A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163602		0.096
9A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163602		0.096

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: STAY, UPPER SIDE

W/ORDER: 12616

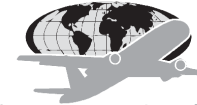
P/NUMB.: 201163300

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: AP042206

SUB W/O: 12616-115

Component Overhaul
Services

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
10A	0.6299	0.6890	0.6306	0.6306	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4726	BCA2-12-12M	0.0020	0.633
10B	0.6306		0.6306	0.6306	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4737	BCA2-12-12M	0.0020	0.633
11A		2.9180	2.6410	2.6410	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.7559	BCA2-12-12M		
							2.7704			
11A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-12M		-0.057
11A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-12M		-0.057
11B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
11B2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
12A	2.4016	2.4850	2.4020	2.4020	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2048	201163600	0.0015	2.404
12B	2.4028		2.4020	2.4020	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2059	201163600	0.0015	2.404
13A			4.7150	4.7150	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	4.7560			
		NOTE 5					4.7680			
13A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				0.021
13A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				0.021
14A		2.6810	2.6400	2.6400	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3623			
							2.3640			
14A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163601		0.139
14A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163601		0.139
15A	EXTERNAL SURFACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
16A	OTHER REJECTIONS		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: STAY, UPPER SIDE

W/ORDER: 12616

P/NUMB.: 201163300

OHM CHAP.: 32-11-26

S/NUMB.: AP042206

SUB W/O: 12616-115

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT



NOTE 1: FOR OVERSIZED USE BUSHING P/N: 450217880

NOTE 2: FOR OVERSIZED USE BUSHING P/N: 450237808

NOTE 3 FOR OVERSIZED USE BUSHING P/N: 450237815

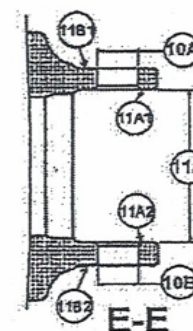
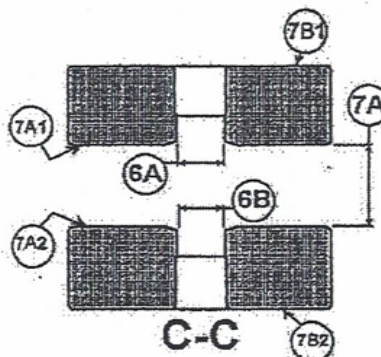
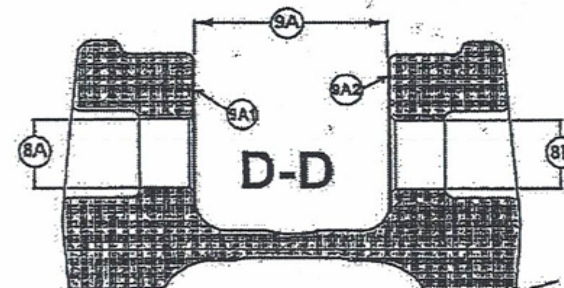
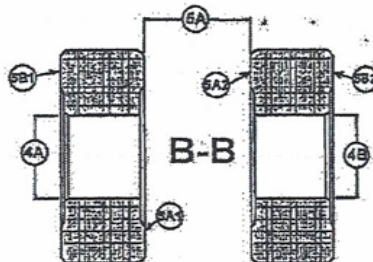
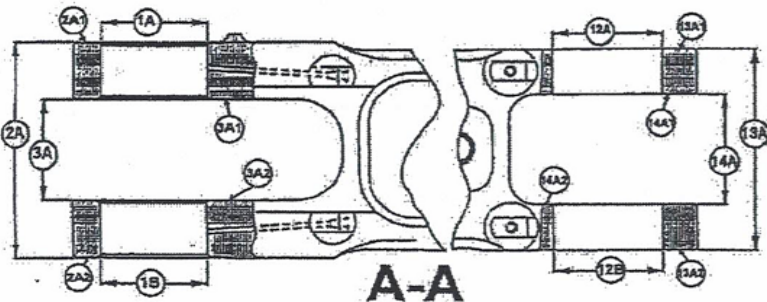
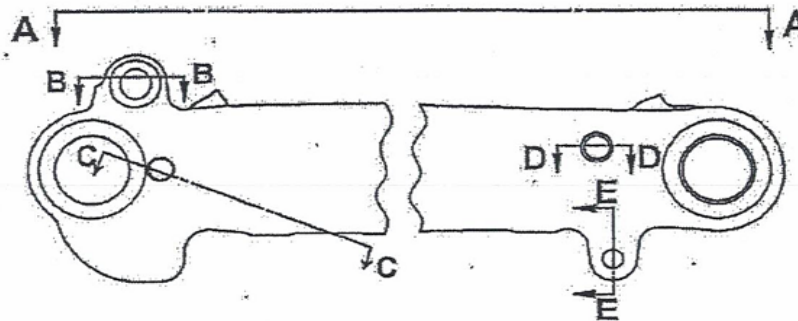
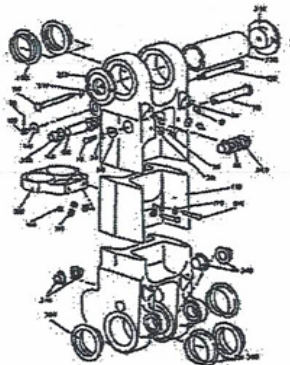
NOTE 4: MAXIMUM DIMENSION OF 1.224 IN FROM THE STAY CENTERLINE TO FACE -3A1- -3A2 PER REPAIR 3-4 MUST BE MAINTAINED.

NOTE 5: MAXIMUM DIMENSION OF 2.5157 IN FROM THE STAY CENTERLINE TO FACE -2A1- -2A2 PER REPAIR 3-4 MUST BE MAINTAINED.

A/C: A320
P/NAME: STAY, UPPER SIDE
P/NUMB.: 201163300
S/NUMB.: AP042206
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12616
OHM CHAP.: 32-11-26
SUB W/O: 12616-115
C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: CARDAN, UPPER

W/ORDER: 12616

INSPECTION STATUS REPORT

P/NUMB.: 201163620

OHM CHAP.: 32-11-26



S/NUMB.: 08B0549X5

SUB W/O: 12616-142

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	3.7370	3.7130	3.7385		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	3.7390	CHROME								
B	2.9500	2.9260	2.9508		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	2.9520	CHROME								
C		2.5200 OHM	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
D	THREADS		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
1A	2.4010	2.4854	2.4020	2.4300	CORROSION	MACHINE	2.2050	201057636	0.0030	2.4330
	REF.	NOTE 2					2.2060			
2A	2.0880	2.0060	2.0840	2.0840	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3580	201057636		
	REF.						2.3600			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201057636		0.137
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201057636		0.137
3A	9.9170		9.9220		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	9.9250									
4A	2.3250	NOTE 1	2.3280		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	2.3280									
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
5A		NOTE 4			INCORPORATE S.B	POLISH TO INCORPORATE				
5B					INCORPORATE S.B	POLISH TO REMOVE DEFECT				
6A	INTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
8A	A/D	SEE NOTE 4	YES (X)		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	2015-22-08		NO ()							
9A	OTHER REJECTIONS	SEE NOTE 3	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: CARDAN, UPPER

W/ORDER: 12616

P/NUMB.: 201163620

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 08B0549X5

SUB W/O: 12616-142



LIF. LIMIT: 60,000

C.S.N.: 16,758

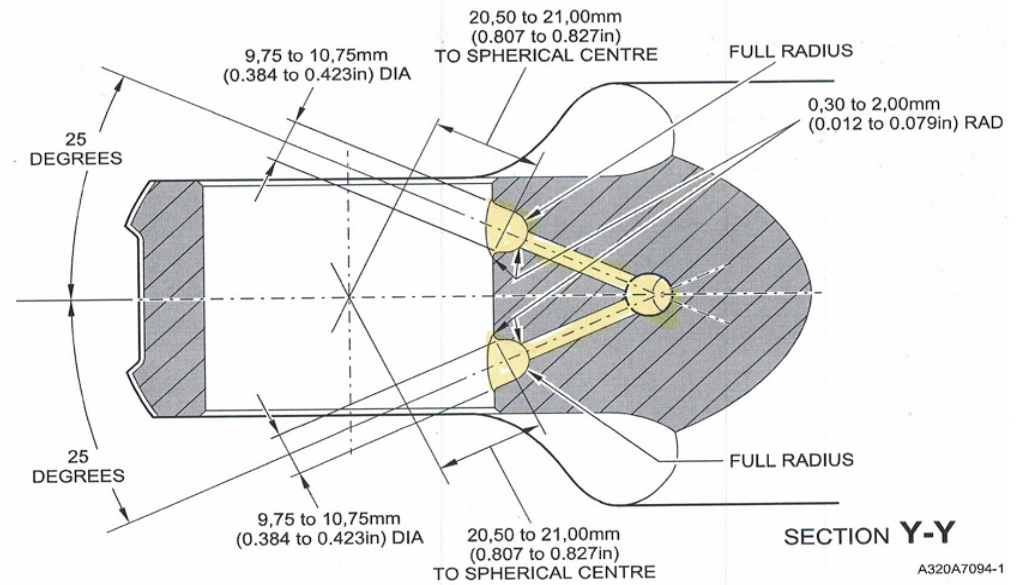
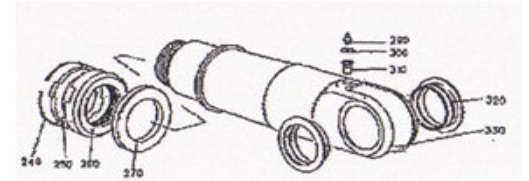
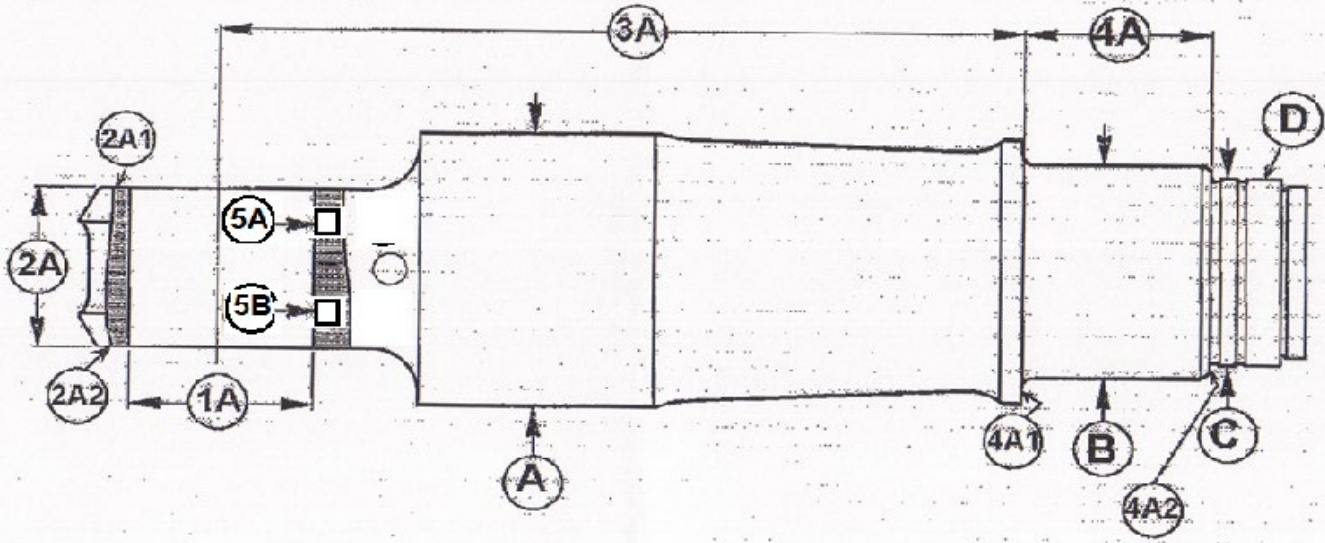
- NOTE 1:** IF LOCATION -4A1- REQUIRES MACHINE, MAINTAIN -4A- DIMENSION TO 2.3318 / 2.336, USE CHROME PLATE PER REPAIR 2-7

IF LOCATION -4A2- REQUIRES MACHINE, MAINTAIN -4A- DIMENSION TO 2.308 MINIMUM, METAL SPRAY (TUNGSTEN CARBIDE, UCAR LW-IN40)
- NOTE 2:** MACHINE DIAMETER -A- JUST SUFFICIENTLY TO REPAIR THE DAMAGE OR WEAR, TO NOT EXCEED A MINIMUM DIAMETER OF 2.323. THE SURFACE FINISH MUST BE 63 MICRO-INCHS.
50 % OF THE SPLINES (94 IN TOTAL, MAY BE REMOVED PROVIDED THAT NOT MORE THAN 33% (61 IN TOTAL) ARE REMOVED FROM ANY ONE 180 DEGREES ARC.
- NOTE 3:** (FOR UAL & TURKISH TECHNIC ONLY) "INCORPORATE S/B 200-32-312 FOR COSTUMERS REQUIREMENTS MANDATOREY"
- NOTE 4:** INCORPORATE A/D 2015-22-08 IT OPENS AND REDUIS THE LUBE HOLES -5A- AND -5B- FROM 0.384 TO 0.423 IT ALSO INCLEASSES THE DEPTH OF THE HOLE FROM 0.807 TO 0.827 (SEE S/B 200-32-312 ,FAA A/D 2015-22-08 OR INPECTION DRAWING SEE DRAWING FOR INFORMATION)
NEW P/N: 201163948

A/C: A320
 P/NAME: CARDAN, UPPER
 P/NUMB.: 201163620
 S/NUMB.: 08B0549X5
 LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.
 W/ORDER: 12616
 OHM CHAP.: 32-11-26
 SUB W/O: 12616-142
 C.S.N.: 16,758

INSPECTION STATUS REPORT



Upper Cardan - 201163948 - After Modification

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: LINK, BANANA

W/ORDER: 12616

P/NUMB.: 201661306

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 08AP0132

SUB W/O: 12616-26



LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	0.6319	NOTE 1	0.6320	0.6320	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-06M	0.0015	0.634
	0.6323						0.4737			
2A		2.0480	0.5090	0.5090	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6192	BCA2-12-06M		
		NOTE 2					0.6283			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
3A	0.6319	NOTE 1	0.6323	0.6323	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	BCA2-12-06M	0.0015	0.634
3B	0.6323						0.4737			
4A			1.6120	1.6120	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.7220	BCA2-12-06M		
							1.7317			
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.055
5A			0.8300	0.8300	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7083	BCA2-12-06M		
							0.7184			
5A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.061
5A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		BCA2-12-06M		0.061
6A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7A	OTHER REJECTIONS		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: LINK, BANANA

W/ORDER: 12616

P/NUMB.: 201661306

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 08AP0132

SUB W/O: 12616-26



LIF. LIMIT: 60, 000

C.S.N.: 16,758

NOTE 1: IF BORE OVERSIZED OR REQUIRES MACHINING SCRAP THE UNIT

A/C: A320

P/NAME: LINK, BANANA

P/NUMB.: 201661306

S/NUMB.: 08AP0132

LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.

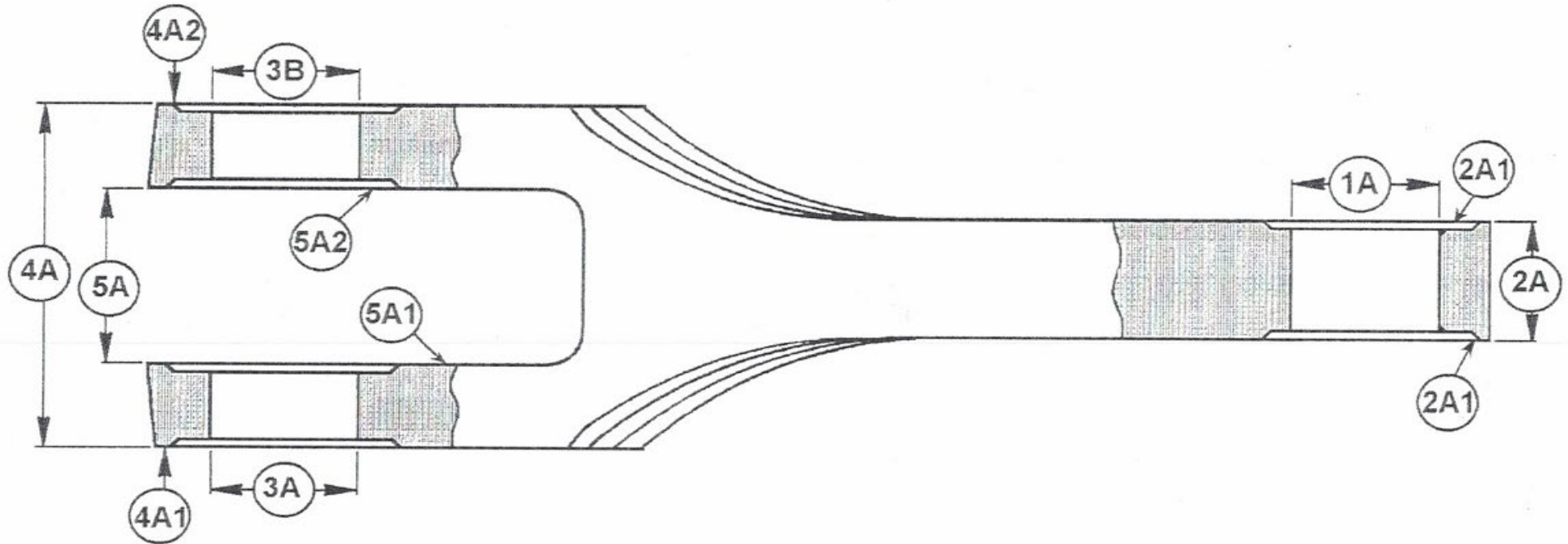
W/ORDER: 12616

OHM CHAP.: 32-11-26

SUB W/O: 12616-26

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320
P/NAME: PIN
P/NUMB.: 201160602
S/NUMB.: 08MDG6745
LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12616
OHM CHAP.: 32-12-25
SUB W/O: 12616-81
C.S.N.: 16,758

INSPECTION STATUS REPORT



NOTE 1: CHROME REPAIR

A/C: A320

P/NAME: PIN

P/NUMB.: 201160602

S/NUMB.: 08MDG6745

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.

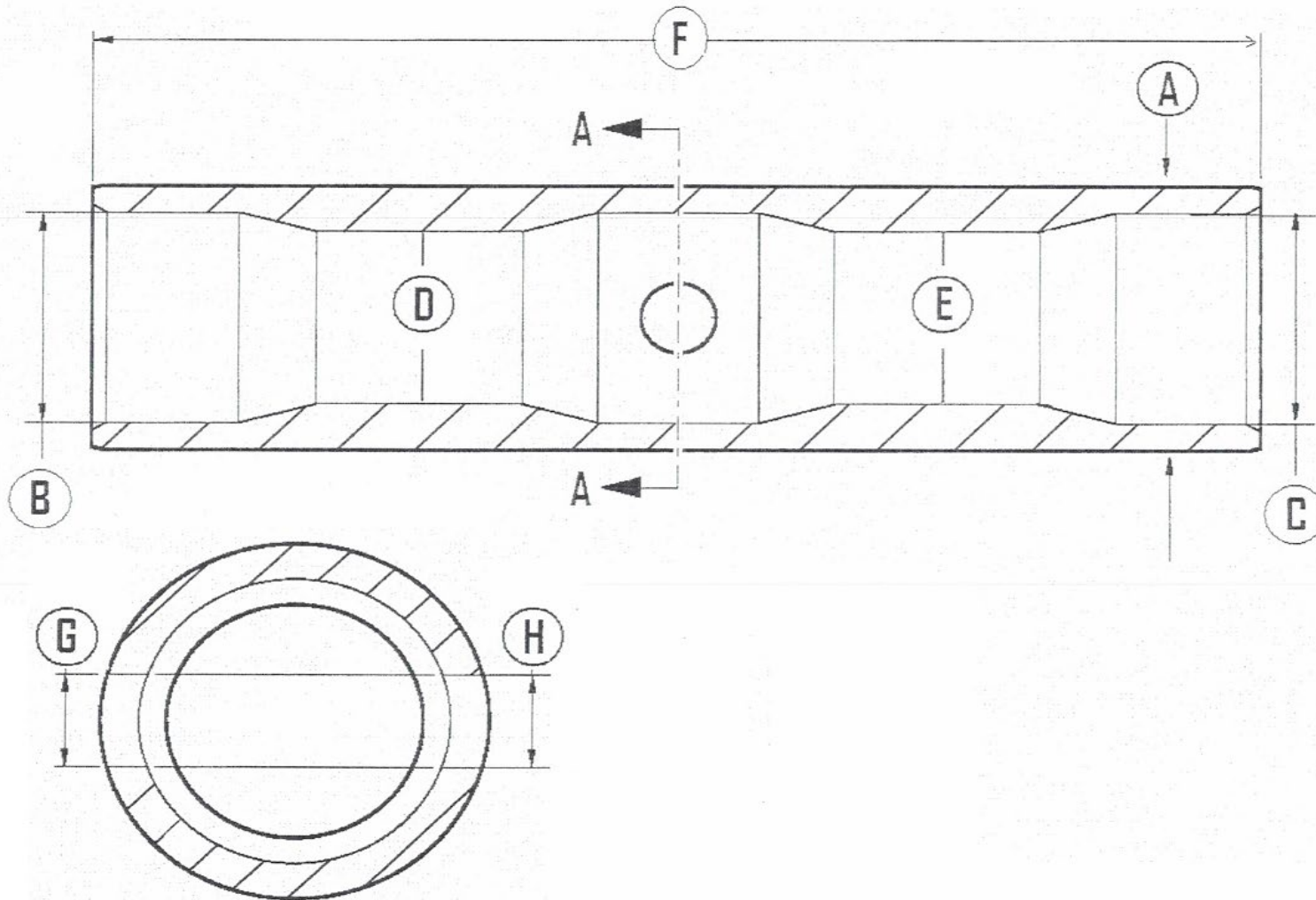
W/ORDER: 12616

OHM CHAP.: 32-12-25

SUB W/O: 12616-81

C.S.N.: 16,758

INSPECTION STATUS REPORT



SECTION A-A
(WITHOUT BUSHES)

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: MLG SLIDING TUBE

W/ORDER: 12616

P/NUMB.: 201371304

OHM CHAP.: 32-11-33 / 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08B5151X6200

SUB W/O: 12616-2



LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	6.9950	6.9730	6.9960	6.9960	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	6.9970		6.9955		6.9955					
A1	6.8471	SUBMIT	6.8480	6.8480	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	6.8487									
B	6.5355	REFER OHM FIG 800'S	6.5370	6.5370	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	6.5379									
C	5.3489	5.3700	5.3500	5.3500	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	5.3512									
D	NON WORKING ID	REFER OHM	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
E	THREADS	REFER OHM	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
F					WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
G	4.6822	4.6596	4.6830	4.6830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
H	4.6836	NOTE 6	4.6830	4.6830	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
I	4.8390	4.8168	4.8400	4.8400	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
J	4.8408	NOTE 6	4.8400	4.8400	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
K	4.9565	4.9349	4.9572	4.9572	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
L	4.9589	NOTE 6	4.9572	4.9470	CHROME BREAK	STRIP				
M	FACES	REFER OHM	COND.		CHROME BREAK	STRIP				
N			ONLY		CHROME BREAK	STRIP				
1A	AXLE I/D	REFER OHM	COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
1B			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
2A	0.2559	NOTE 5	0.2638	0.2638	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
2B	0.2638		0.2638	0.2638	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
2C	0.2559	NOTE 5	0.2638	0.2638	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
2D	0.2638		0.2638	0.2638	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
3A	1.5770	REFERENCE	1.5770	1.5770	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	1.5800	DIMENSION								
3B	1.3820	REFERENCE	1.3820	1.3820	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	1.3830	DIMENSION								
4A	1.8110	1.8947	1.8120	1.8120	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.3976	201383654	0.0010	1.8130
	1.8120						1.4016	NOTE 3		

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: MLG SLIDING TUBE

W/ORDER: 12616

INSPECTION STATUS REPORT

P/NUMB.: 201371304

OHM CHAP.: 32-11-33 / 32-12-25

Component Overhaul
Services

S/NUMB.: 08B5151X6200

SUB W/O: 12616-2

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
5A	5.7030	5.6650	5.7050	5.7050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.9416	201383650		
		NOTE 1					5.9432			
5A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201383650		0.1183
5A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201383650		0.1183
5B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
5B2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
6A	2.5984	2.6822	2.6000	2.6000	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3622	201383650	0.0030	2.6030
6B	2.5996		2.6000	2.6000	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3634	201383650	0.0030	2.6030
7A	5.1150	5.0296	5.1130	5.1130	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.3142	201371617		
	REF.						5.3285			
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1020	201371617		0.1006
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.1020	201371617		0.1006
8A	1.0650	1.1229	1.0650	1.0650	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.8661	201371617	0.0015	1.0665
8B	REF. DIM.		1.0650	1.0650	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.8674	201371617	0.0015	1.0665
9A	0.6693	0.0731	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6714
9B	0.6704	NOTE 4	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6714
9C	0.7559	0.8294	0.7570	0.7570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6590	201056703	0.0010	0.7580
	0.7572	NOTE 2					0.6791			
9D	0.6693	0.0731	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6714
	0.6704	NOTE 4					0.5916			
9E	0.7559	0.8294	0.7570	0.7570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6590	201056703	0.0010	0.7580
	0.7572	NOTE 2					0.6791			
9F	0.6693	0.0731	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6714
9G	0.6704	NOTE 3	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6714
9H	0.6693	0.0731	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6714
9I	0.6704	NOTE 4	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6714
9J	0.7559	0.8294	0.7570	0.7570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6590	201056703	0.0010	0.7580
	0.7572	NOTE 2					0.6791			
9K	0.6693	0.0731	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6714
9L	0.6704	NOTE 3	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6714
10A	0.6693	0.0731	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6714
10B	0.6704	NOTE 4	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6714

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: MLG SLIDING TUBE

W/ORDER: 12616

INSPECTION STATUS REPORT

P/NUMB.: 201371304

OHM CHAP.: 32-11-33 / 32-12-25

Component Overhaul
Services

S/NUMB.: 08B5151X6200

SUB W/O: 12616-2

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
10C	0.7559	0.8294	0.7570	0.7570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6590	201056703	0.0010	0.7580
	0.7572	NOTE 2					0.6791			
10D	0.6693	0.0731	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6714
	0.6704	NOTE 4					0.5916			
10E	0.7559	0.8294	0.7570	0.7570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6590	201056703	0.0010	0.7580
	0.7572	NOTE 2					0.6791			
10F	0.6693	0.0731	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6714
10G	0.6704	NOTE 4	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6714
10H	0.6693	0.0731	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6714
10I	0.6704	NOTE 4	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6714
10J	0.7559	0.8294	0.7570	0.7570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6590	201056703	0.0010	0.7580
	0.7572	NOTE 2					0.6791			
10K	0.6693	0.0731	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5906	201056702	0.0010	0.6714
10L	0.6704	NOTE 4	0.6704	0.6704	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5916	201056702	0.0010	0.6714
11A	VERIFY	S/N NOT LISTED	YES ()		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	NOTE 5	ON S/B	NO ()							
12A	VERIFY	S/N NOT LISTED	YES ()		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	NOTE 6	ON S/B	NO ()							
13A	INTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
14A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
15A	OTHER REJECTIONS		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: MLG SLIDING TUBE

W/ORDER: 12616

P/NUMB.: 201371304

OHM CHAP.: 32-11-33 / 32-12-25

INSPECTION STATUS REPORT



S/NUMB.: 08B5151X6200

SUB W/O: 12616-2

Component Overhaul
Services

LIF. LIMIT: 60,000

C.S.N.: 16,758

NOTE 1: ► DO NOT MACHINE MORE THAN 0.025 FROM EACH FACE. LUG THICKNESS MUST NOT BE LESS THAN 1.415

NOTE 2: ► IF BORE IS IN DESIGN DIMENSION INSTALL BUSHING P/N: 201056703, IF BORE IS O/S OR REQUIRES MACHINING, MACHINE AND INSTALL BUSHINGS IWA CMM 32-12-25 REPAIR 9-9 POAGE 602, AND ORDER THE CORRECT BUSHING

NOTE 3: ► IF NECESSARY, MACHINE FACE TO REMOVE DAMAGE OR WEAR TO A MINIMUM DIMENSION OF: 3.063, SHOWN IN CMM REPAIR 9-10 (CENTER OF AXLE BORE TO FACE) WITH A SURFACE FINISH 63 MICROINCHES THEN RESTORE CHAMFER. INSTALL O/S BUSHING AND MACHINE BUSHINGS TO RESTORE DIMENSION SHOWN IN CMM 3.268 / 3.287 (CENTER OF AXLE BORE TO FACE)

NOTE 4: ► IF BORE IS IN DESIGN DIMENSION INSTALL BUSHING P/N: 201056702, IF BORE IS O/S OR REQUIRES MACHINING, MACHINE AND INSTALL BUSHINGS IWA CMM 32-12-25 REPAIR 9-3 PAGE 601, AND ORDER THE CORRECT BUSHING

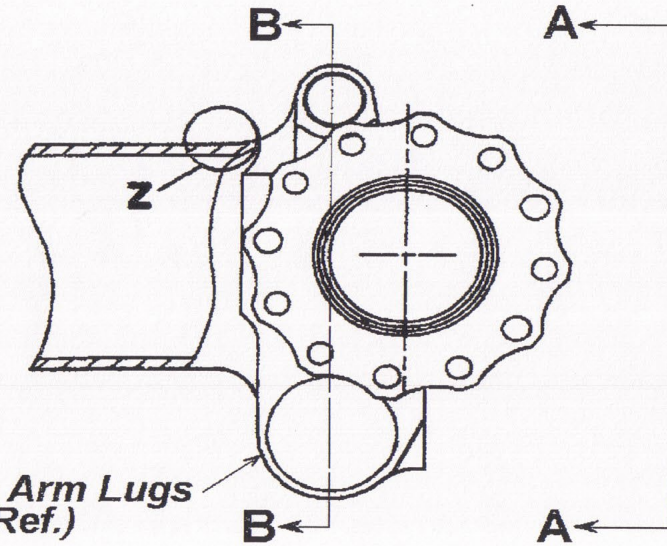
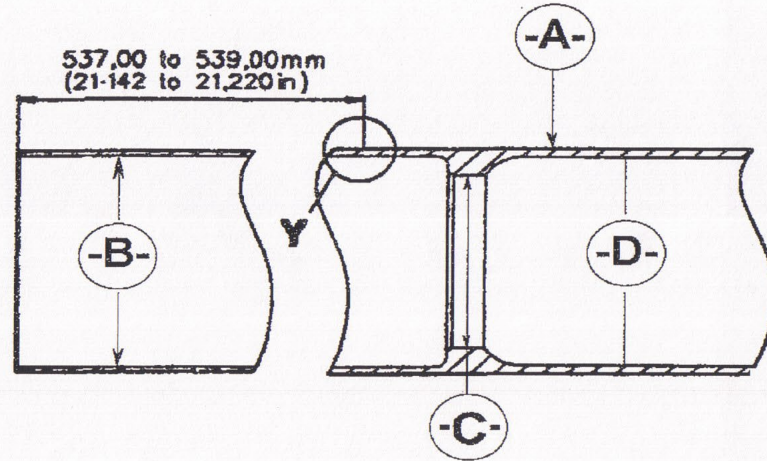
NOTE 5: ► ***** MANDATORY ***** INSPECT COMPONENT AND VERIFY COMPLIANCE WITH (SAFRAN S/B: 200-32-231) AND (AIRBUS S/B A320-32-1441), VERIFY BORES ARE FREE OF DEFECTS (SERVICE LIMITS IS 0.2724)

NOTE 6: ► ***** MANDATORY ***** PER (SAFRAN S/B: 200-32-313) VERIFY PISTON S/N IS NOT LISTED ON THE LIST IN THE S/B 200-32-313
** IF S/N IS LIST ON THE S/B 200-32-313 (CONTACT MESSIER-DOWTY) AND ONLY MESSIER DOWTY QUALIFIED PERSONNEL WILL INSPECT **

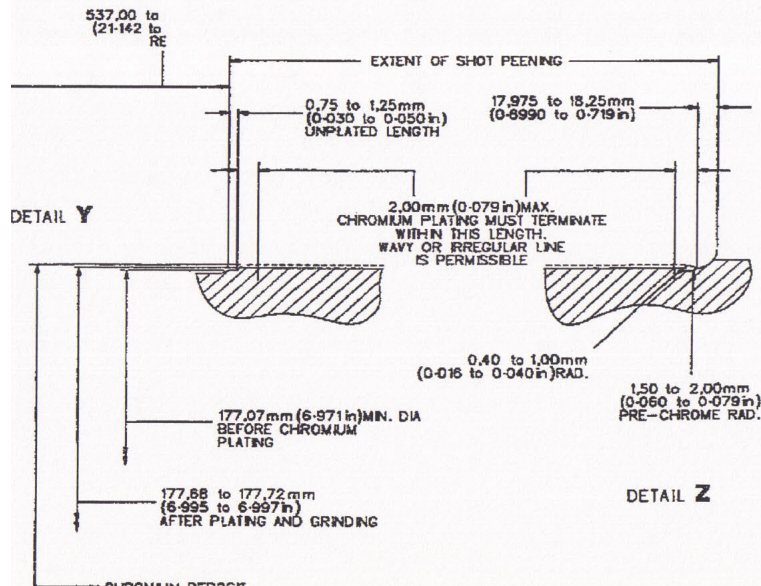
A/C: A320
P/NAME: MLG SLIDING TUBE
P/NUMB.: 201371304
S/NUMB.: 08B5151X6200
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12616
OHM CHAP.: 32-11-33 / 32-12-25
SUB W/O: 12616-2
C.S.N.: 16,758

INSPECTION STATUS REPORT



Torque Arm Lugs
(Ref.)



DRAWING 1 OF 2

A/C: A320

P/NAME: MLG SLIDING TUBE

P/NUMB.: 201371304

S/NUMB.: 08B5151X6200

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.

W/ORDER: 12616

OHM CHAP.: 32-11-33 / 32-12-25

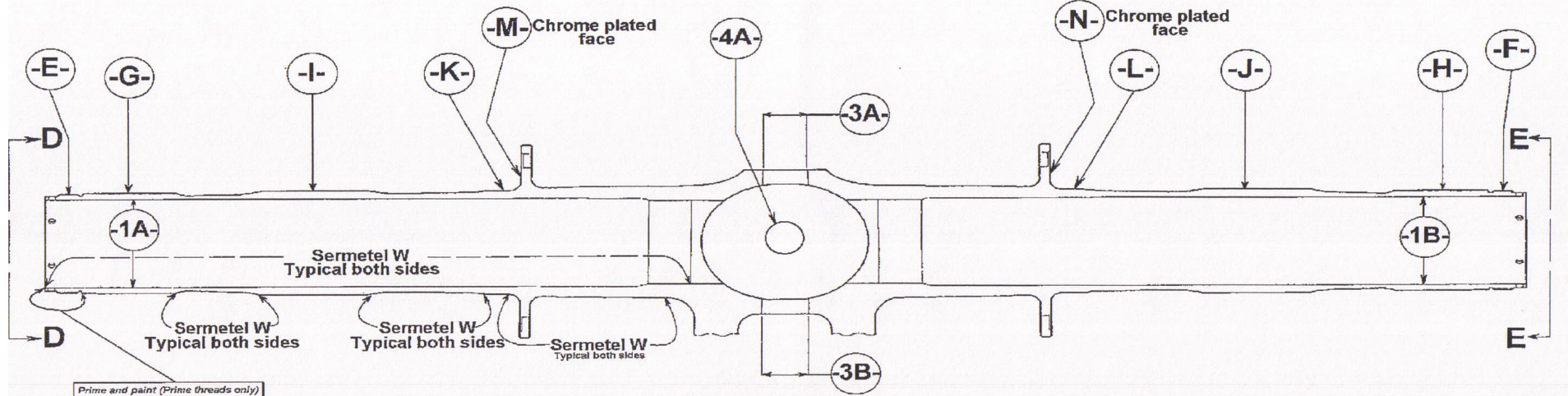
SUB W/O: 12616-2

C.S.N.: 16,758

INSPECTION STATUS REPORT

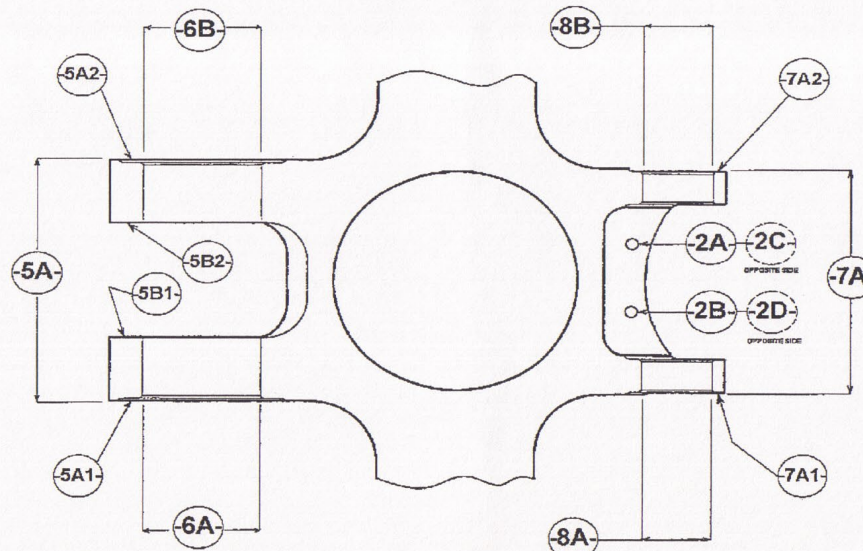


Component Overhaul Services



TORQUE ARM LUGS
DOWN

A - A



R - R

DRAWING 2 OF

A/C: A320

P/NAME: MLG SLIDING TUBE

P/NUMB.: 201371304

S/NUMB.: 08B5151X6200

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.

W/ORDER: 12616

OHM CHAP.: 32-11-33 / 32-12-25

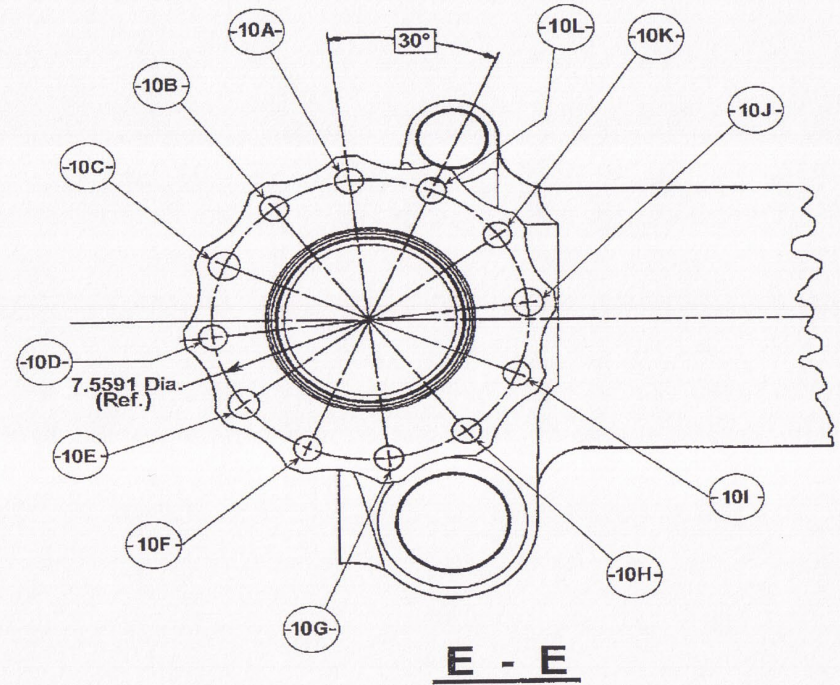
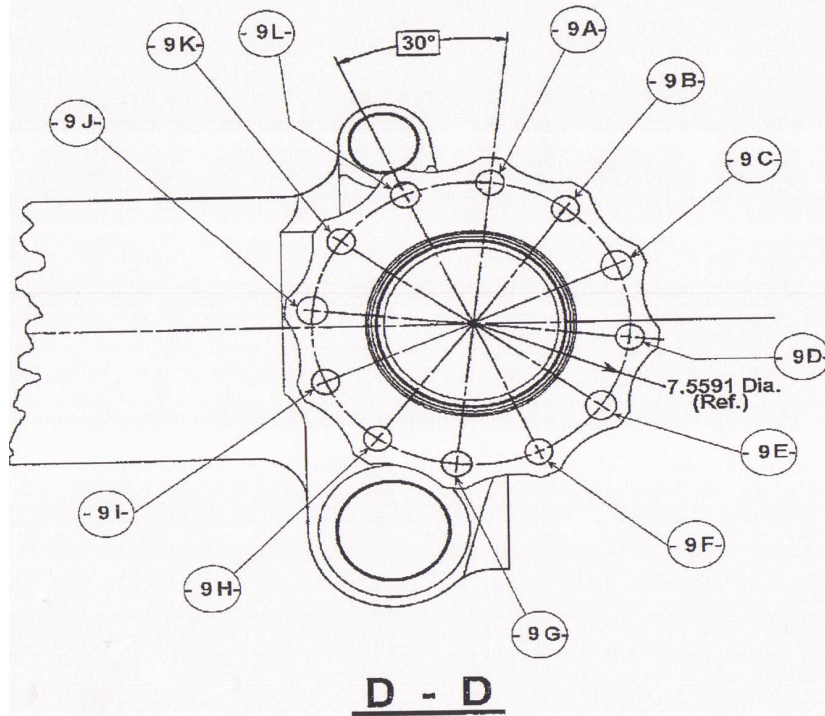
SUB W/O: 12616-2

C.S.N.: 16,758

INSPECTION STATUS REPORT



Component Overhaul
Services



A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: TUBE, UPPER DIAPHRAGM

W/ORDER: 12616

INSPECTION STATUS REPORT

P/NUMB.: 201371615

OHM CHAP.: 32-11-33

Component Overhaul
Services

S/NUMB.: 08B0314X1

SUB W/O: 12616-192

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	S/B 200 32-247		YES (X) NO ()		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	1.4173 1.4183	1.4860	1.4200	1.4200	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	1.2224 1.2234	201522668	0.0010	1.4210
2A	3.3410 REF.	3.3267	3.3410	3.3410	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	3.4803 3.4842	201522668 NOTE 2		
2A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.140	201522668		0.139
2A2					WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
3A	0.7460 REF.	0.6590	0.7460	0.7460	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				0.0000
3A1	FACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	0.060			0.0000
4A	2.0079 2.0091	2.0926	2.0100	2.0100	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS	1.8110 1.8126	201522669	0.0010	2.0110
5A	2.2400 REF.		2.2450	2.2450	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
5A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
5A2					WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS		201522669		-1.1225
6A	2.6340 2.6380	2.6570	2.6370	2.6370	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
6A1	FACE	SEE CMM	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
7A	7.2195 7.2209	7.1610	7.2200	7.2200	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
8A	6.9803 6.9828	6.9230	6.9820	6.9820	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
9A	7.2202 7.2220	7.2140	7.2200	7.2200	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
10A	6.7444 6.7468	6.6870	6.7460	6.7250	LOCAL CORROSION	MACHINE				
11A	7.2202 7.2220	WEAR LIMITS 7.2140	7.2200	7.2200	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
12A	6.9840 REF.	6.9790	6.9840	6.9840	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
		SEE CMM								

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: TUBE, UPPER DIAPHRAGM

W/ORDER: 12616

P/NUMB.: 201371615

OHM CHAP.: 32-11-33

INSPECTION STATUS REPORT

S/NUMB.: 08B0314X1

SUB W/O: 12616-192



LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
13A	0.3974	0.3981	0.3970	0.3970	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	0.3961	WEAR LIMITS								
15A	INTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	SURFACE		ONLY							
16A	EXTERNAL	6.0000	COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	SURFACE		ONLY							
17A	NOTES		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
			ONLY							
18A	0.3620	0.3071	0.3600	0.3600	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.									
19A	4.9016		4.9010	4.9010	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF,									
20A	0.3150	0.2750	0.3160	0.3160	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.									
21A	0.3947		0.3960	0.3960	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	.3961 REF									
22A	0.8071		0.8070	0.8070	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	AS REF.									
23A	0.3800	0.3940	0.3800	0.3800	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	RER.	MINIMUM								
24A	0.9940	1.0240	1.0000	0.9950	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	1.0140	MAX								
24A1	FACE		COND.		LOCAL CORROSION	MACHINE				
			ONLY							
25A	1.1940	1.1810	1.1940	1.1940	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	REF.	MINIMUM								
26A	1.4070	1.4370	1.4250	1.4250	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	1.427	MAX								
27A	0.2126		0.2160	0.2160	WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
	0.2165									
27A1	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN DESIGN LIMITS				
			ONLY							
28A	0..579		0.5850		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	0..589									

A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: TUBE, UPPER DIAPHRAGM

W/ORDER: 12616

P/NUMB.: 201371615

OHM CHAP.: 32-11-33

INSPECTION STATUS REPORT



S/NUMB.: 08B0314X1

SUB W/O: 12616-192

Component Overhaul
Services

LIF. LIMIT: 60, 000

C.S.N.: 16,758

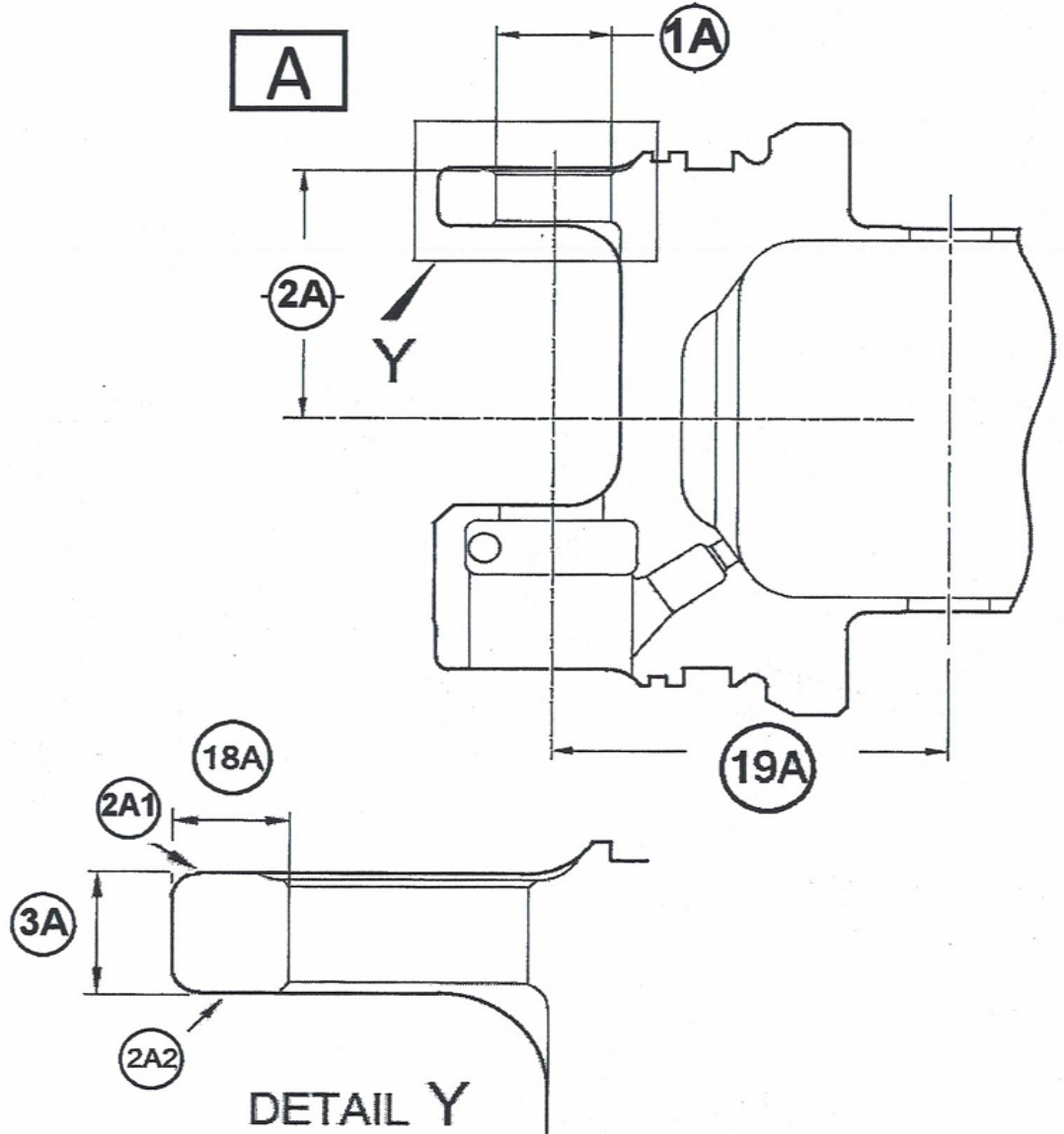
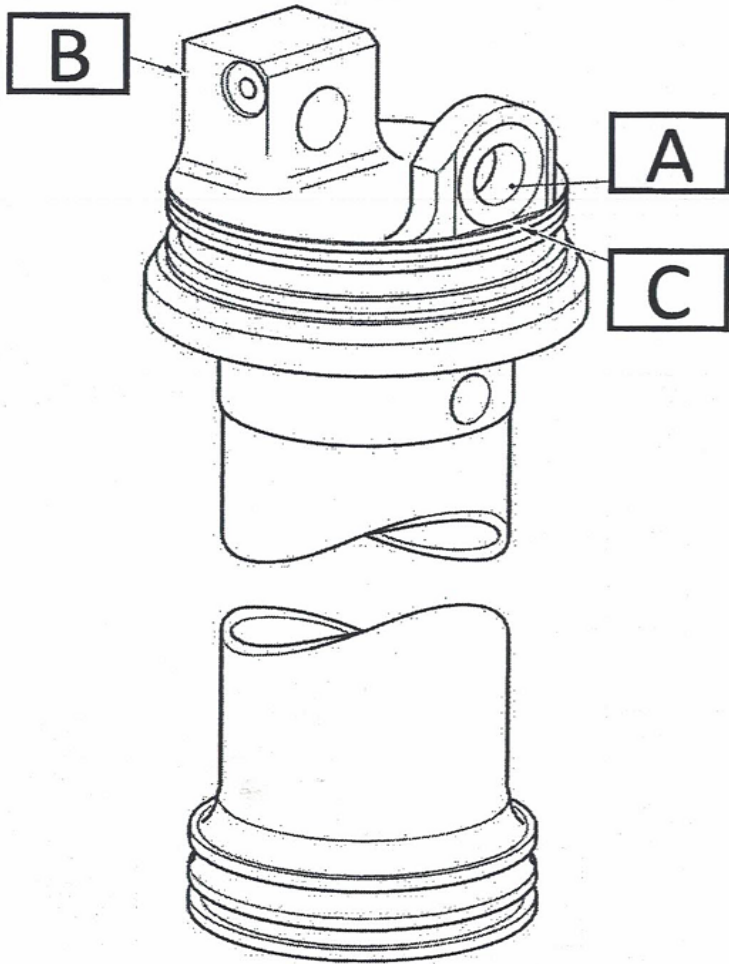
NOTE 1: ▶ PRE S/B 200-32-247 MADE FROM ALUMINUM
▶ POST S/B 200-32-247 MADE FROM STEEL

NOTE 2: ▶ TO OBTAIN DIMENSION -2A- THE PART MUST BE MESURED FROM THE CENTER LINE OF THE TUBE TO THE START OF THE OUTER CHAMFER OF THE BUSHING FLANGE PER CMM 32-11-33 REPAIR NO. 12-1 FIGURE 602 OR CMM 32-12-25 REPAIR NO. 15-1 FIGURE 602

A/C: A320
P/NAME: TUBE, UPPER DIAPHRAGM
P/NUMB.: 201371615
S/NUMB.: 08B0314X1
LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE, INC.
W/ORDER: 12616
OHM CHAP.: 32-11-33
SUB W/O: 12616-192
C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: TUBE, UPPER DIAPHRAGM

W/ORDER: 12616

P/NUMB.: 201371615

OHM CHAP.: 32-11-33

S/NUMB.: 08B0314X1

SUB W/O: 12616-192

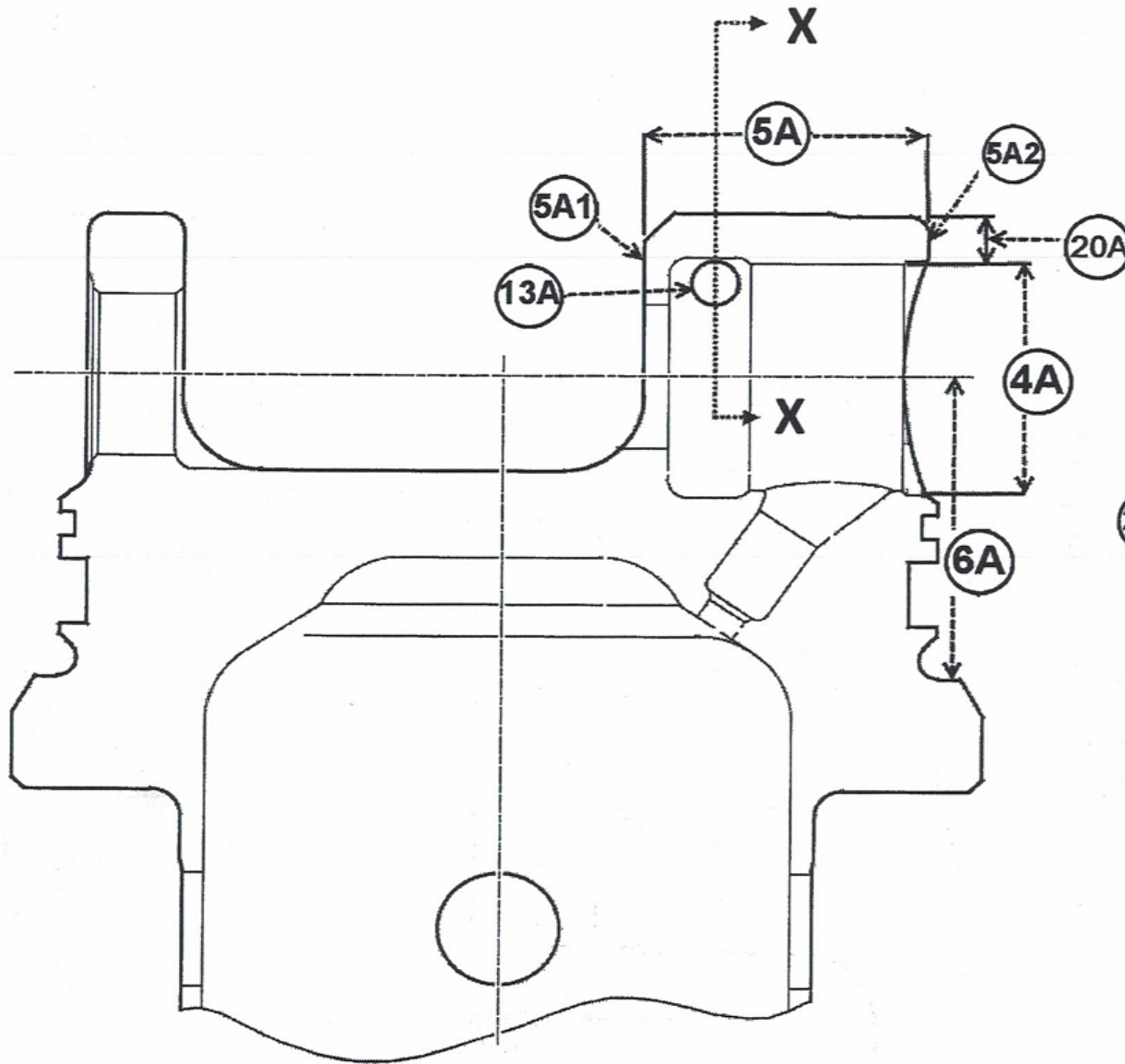
LIF. LIMIT: 60,000

C.S.N.: 16,758

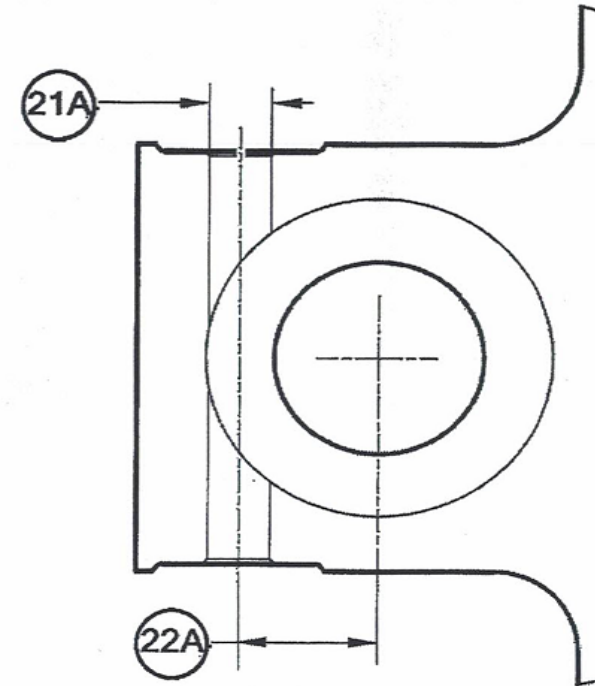
INSPECTION STATUS REPORT



B



SECTION **X - X**



A/C: A320

CUSTOMER: CAVU AEROSPACE, INC.

P/NAME: TUBE, UPPER DIAPHRAGM

W/ORDER: 12616

P/NUMB.: 201371615

OHM CHAP.: 32-11-33

INSPECTION STATUS REPORT



Component Overhaul
Services

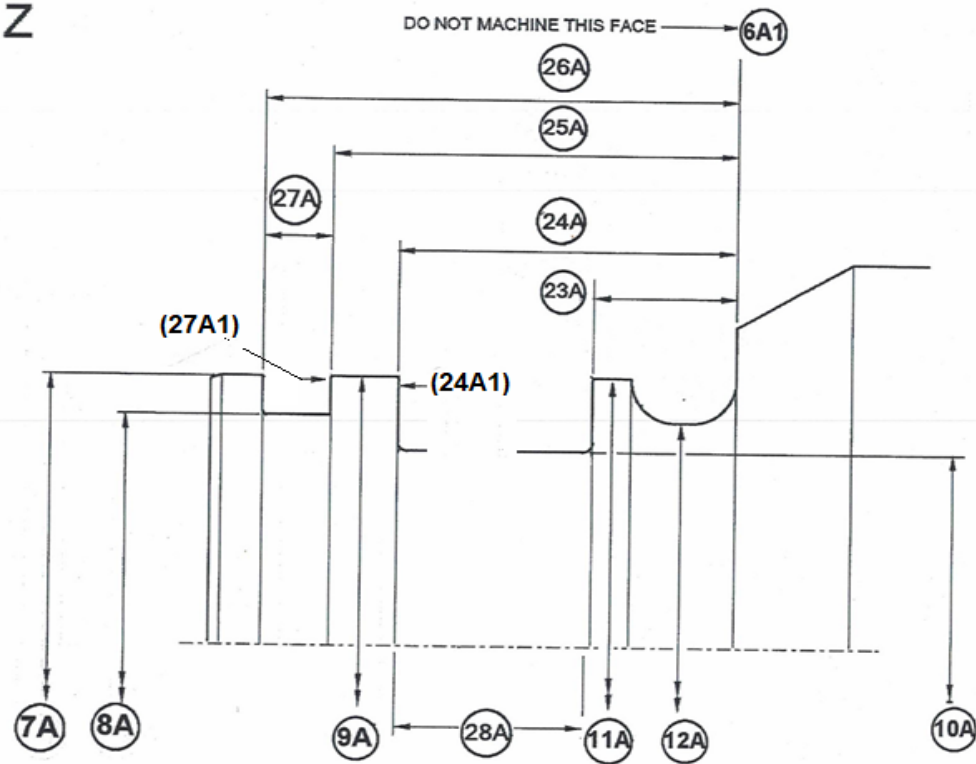
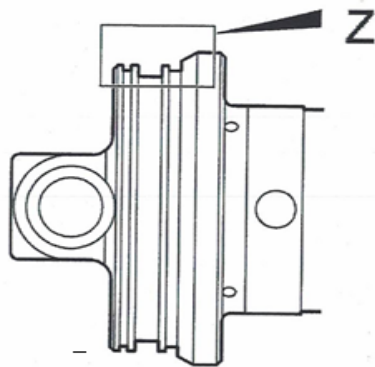
S/NUMB.: 08B0314X1

SUB W/O: 12616-192

LIF. LIMIT: 60,000

C.S.N.: 16,758

C



DETAIL Z

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: CARDAN, LOWER

W/ORDER: 12616

P/NUMB.: 201163307

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT

S/NUMB.: 08B0893X1768

SUB W/O: 12616-123



LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	2.7541	2.7300	2.7547	2.7547	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	2.7547	CHROME								
B	2.7530	2.7300	2.7547	2.7547	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	2.7547	CHROME								
C		2.3230 NOTE 4	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
D	THREADS		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
1A	2.4037	2.4850	2.4040	2.4040	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.0866	201163624	0.0030	2.4070
	2.4044	NOTE 1					2.0878			
2A		2.0480	2.0850	2.0850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2810	201163624		
		NOTE 2					2.2823			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163624		0.0980
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201163624		0.0980
3A	5.4340	NOTE 1	5.4350	5.4350	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	5.4350	NOTE 2								
3A1	FACES	CHROME	COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
3A2		NOTE 3	ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
4A		0.4690	0.4600		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
4B		0.4920	0.4900		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
5A	INTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
6A	EXTERNAL SURFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7A	OTHER REJECTIONS		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: CARDAN, LOWER

W/ORDER: 12616

P/NUMB.: 201163307

OHM CHAP.: 32-11-26

INSPECTION STATUS REPORT



Component Overhaul Services

S/NUMB.: 08B0893X1768

SUB W/O: 12616-123

LIF. LIMIT: 60, 000

C.S.N.: 16,758

NOTE 1: ► METAL SPRAY REPAIR 1-5 (MAKE SURE TO ONLY MACHINE GRIP FACE -3A2- WHEN REPAIRING THIS DIMENSION, DO NOT EXCEED THE REPAIR LIMITS GIVEN OF: 5.4150)

NOTE 2: ► CHROME PLATE REPAIR 1-4 (MAKE SURE TO ONLY MACHINE GRIP FACE -3A1- WHEN REPAIRING THIS DIMENSION, DO NOT EXCEED THE REPAIR LIMIT GIVEN OF 5.445)

***** FOR
UAL REFERENCE DIMENSION IS TIGHTER (REFERENCE DIMENSION IS: 5.434 / 5.435)

NOTE 3: PARTS WHICH ARE REPAIRED PER CMM REPAIR SCHEME 450237620

ARE LIFE LIMITED TO A FURTHER 18,00 LANDINGS AND MUST NOT EXCEED A TOTAL LIFE LIMITS OF 39,500 LANDINGS

IF THE PART WAS INSTALLED IN A TWIN LUG MLG FITTING PRE SB 200-32-183 ASSIGN LIFE LIMITS PER ALS PQRT 1 AIRWORTHINESS LIMITATIONS ITEMS

NOTE 4: MACHINE DIAMETER -A- JUST SUFFICIENTLY TO REPAIR THE DAMAGE OR WEAR, TO NOT EXCEED A MINIMUM DIAMETER OF 2.323. THE SURFACE FINISH MUST BE 63 MICRO-INCHS.

50 % OF THE SPLINES (94 IN TOTAL, MAY BE REMOVED PROVIDED THAT NOT MORE THAN 33% (61 IN TOTAL) ARE REMOVED FROM ANY ONE 180 DEGREES ARC.

A/C: A320

P/NAME: CARDAN, LOWER

P/NUMB.: 201163307

S/NUMB.: 08B0893X1768

LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.

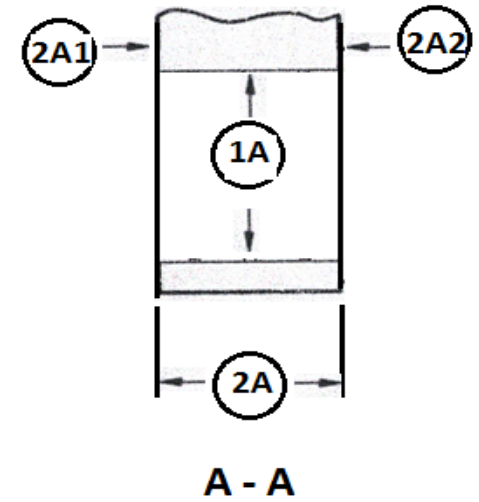
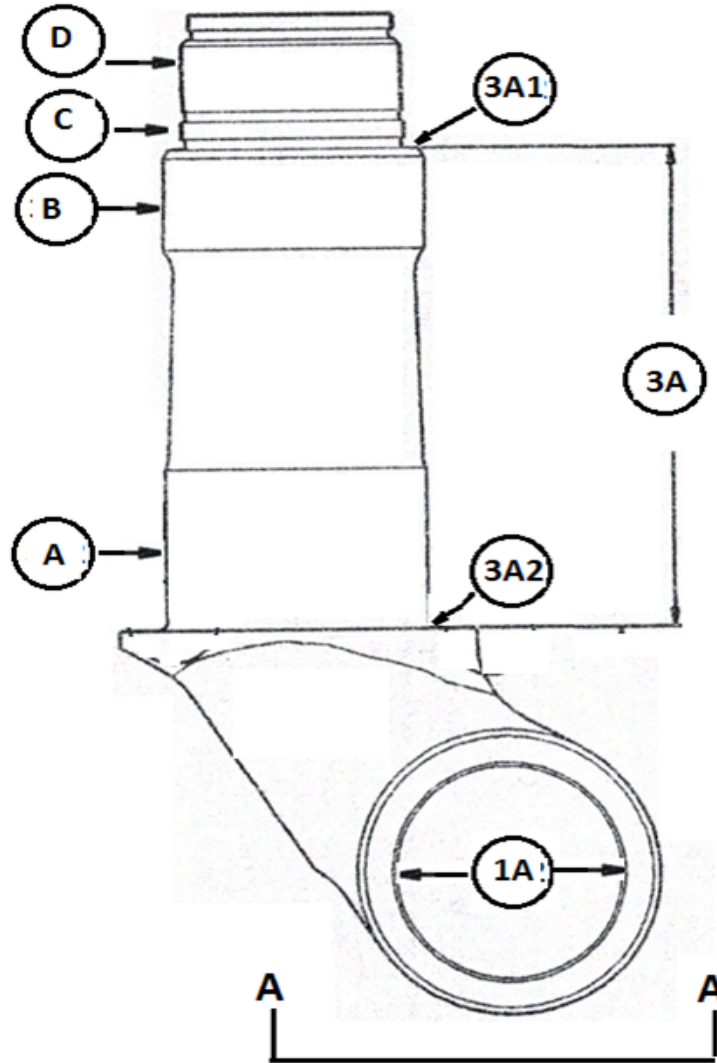
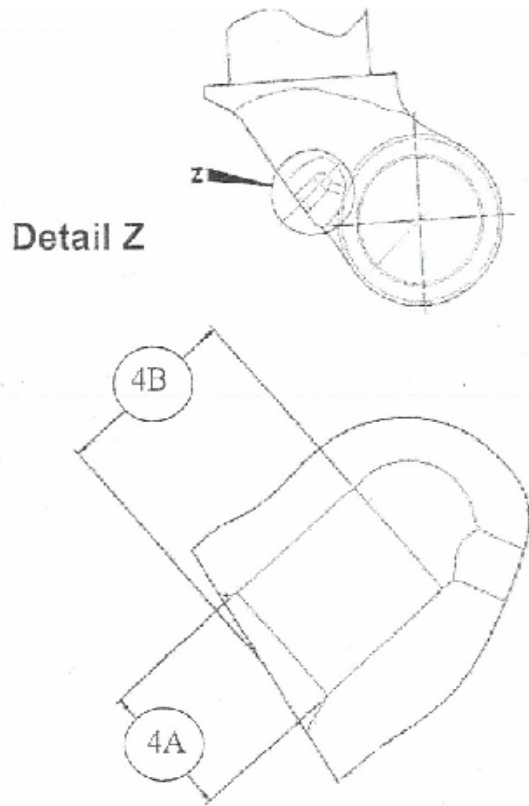
W/ORDER: 12616

OHM CHAP.: 32-11-26

SUB W/O: 12616-123

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: UPPER TORQUE LINK

W/ORDER: 12616

P/NUMB.: 201540303

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08BEL0060

SUB W/O: 12616-77



LIF. LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	1.6535	1.7293	1.6546	1.6546	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.4961	201522603	0.0020	1.6566
	1.6546	MAXIMUM						1.4970		
2A	1.6920	1.6500	1.6900	1.6900	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.8878	201522603		
	REF.	NOTE 1						1.8880		
2A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201522603		0.099
2A2					WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201522603		0.099
3A	5.1380	5.1870	5.1380	5.1380	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.1359			
	REF.							5.1397		
4A	2.5984	2.6820	2.6000	2.6000	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3622	201056605	0.0030	2.6030
								2.3634		
4B	2.5996	2.6820	2.6000	2.6000	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3622	201056605	0.0030	2.6030
								2.3634		
5A	6.3386	6.3843	6.3440	6.3440	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	6.1024	201056605		
	6.3448							6.1048		
5A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056605		0.121
5A2					WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056605		0.121
6A	1.5340	1.4799	1.5340	1.5340	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
6B	1.5340	1.4961	1.5340	1.5340	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
6A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
6B1					WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7A	0.3850	0.4530	0.3850	0.3850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3850	440016734	0.0010	0.3860
7B	AS A REF.		0.3850	0.3850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3850	440016734	0.0010	0.3860
7C	0.3880	0.4530	0.3850	0.3850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3850	440016734	0.0010	0.3860
7D	AS A REF.		0.3850	0.3850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3850	440016734	0.0010	0.3860
8A	LUBE HOLES	NOTE 2	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
9A	EXTERNAL SUFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
10A	NOTES SUFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: UPPER TORQUE LINK

W/ORDER: 12616

P/NUMB.: 201540303

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT



S/NUMB.: 08BEL0060

SUB W/O: 12616-77

LIF. LIMIT: 60,000

C.S.N.: 16,758

NOTE 1: DO NOT REMOVE MORE THAN 0.020 PER FACE

NOTE 2: ► THREE LUBE FITTING HOLES INSTALL LUBRICATION ADAPTER P/N 899005010 (QTY 3)

A/C: A320

P/NAME: UPPER TORQUE LINK

P/NUMB.: 201540303

S/NUMB.: 08BEL0060

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.

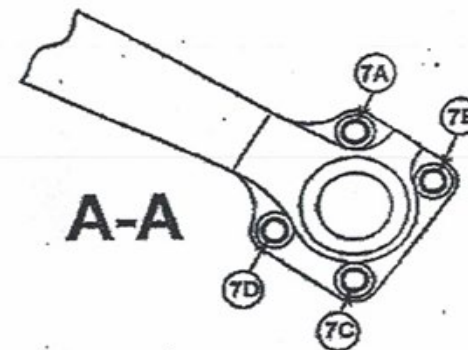
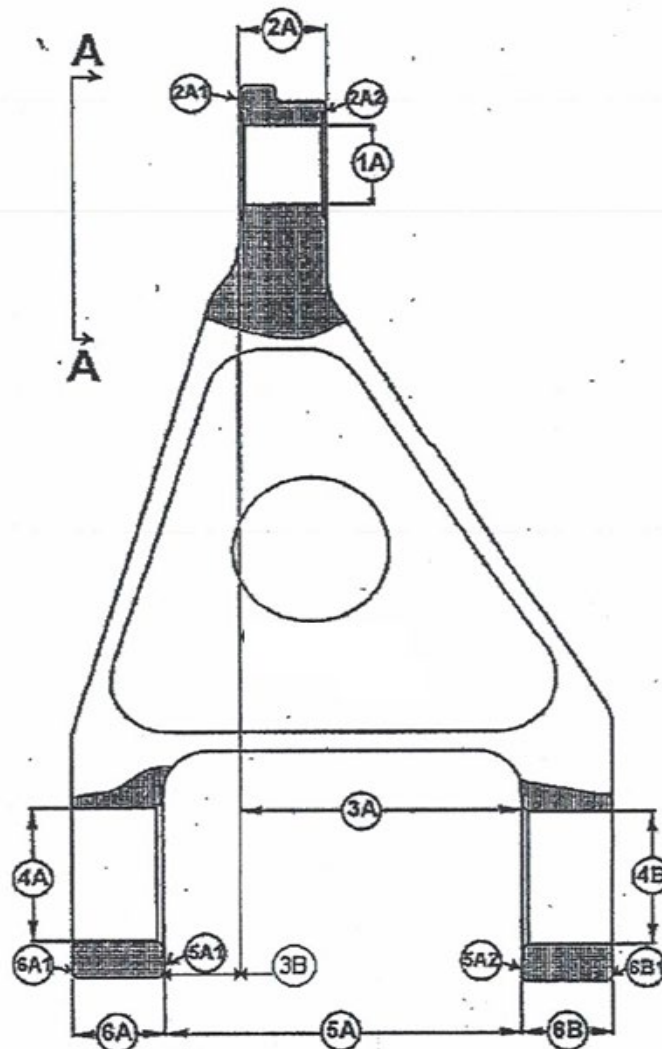
W/ORDER: 12616

OHM CHAP.: 32-12-25

SUB W/O: 12616-77

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: PIN, UPR.

W/ORDER: 12616

INSPECTION STATUS REPORT



Component Overhaul Services

P/NUMB.: 201160603

OHM CHAP.: 32-11-33

S/NUMB.: 08MDG8381

SUB W/O: 12616-74

LIF. LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	2.3603	2.3374	2.3603	2.3603	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	2.3610	NOTE 1								
B	1.8890		1.8890	1.8890	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
C	1.8890		1.8890	1.8890	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
D	1.5570		1.5570	1.5570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
E	1.5570		1.5570	1.5570	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
F	9.6100		9.6100	9.6100	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REF.									
G	0.5632	0.626	0.5632	0.5632	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5632	450237501	0.001	0.5642
H	0.5625	0.649	0.5632	0.5632	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.5625	450237501	0.001	0.5642
I	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320
P/NAME: PIN, UPR.
P/NUMB.: 201160603
S/NUMB.: 08MDG8381
LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12616
OHM CHAP.: 32-11-33
SUB W/O: 12616-74
C.S.N.: 16,758

INSPECTION STATUS REPORT



NOTE 1: CHROME REPAIR

A/C: A320

P/NAME: PIN, UPR.

P/NUMB.: 201160603

S/NUMB.: 08MDG8381

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.

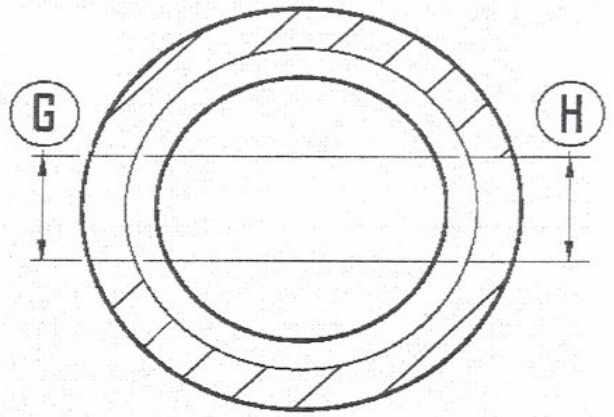
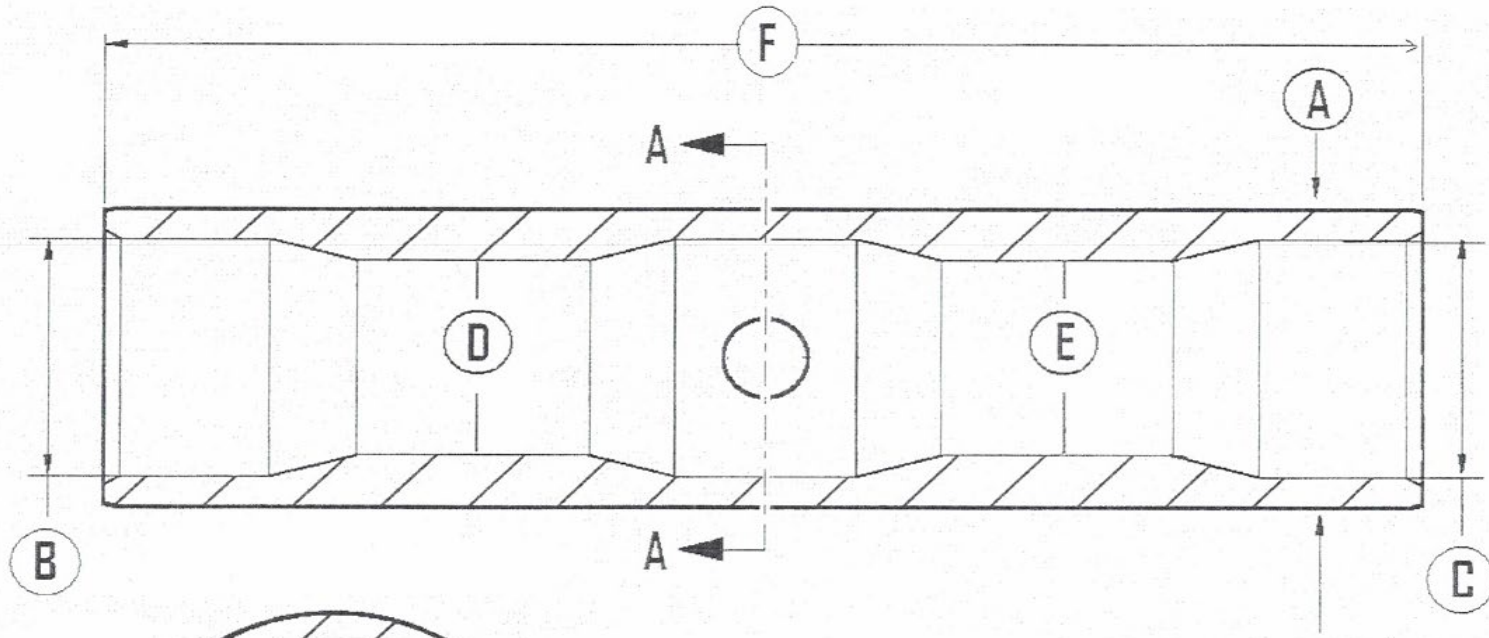
W/ORDER: 12616

OHM CHAP.: 32-11-33

SUB W/O: 12616-74

C.S.N.: 16,758

INSPECTION STATUS REPORT



SECTION A-A
(WITHOUT BUSHES)

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: UPPER TORQUE LINK

W/ORDER: 12616

P/NUMB.: 201540302

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08BEL0113

SUB W/O: 12616-83



LIF. LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	1.6535	1.7293	1.6545	1.6545	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.4567	201383609	0.0020	1.6565
1B	1.6545	MAXIMUM	1.6545	1.6545	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.4577	201383609	0.0020	1.6565
2A	1.6900 REF.	1.6500 MINIMUM	1.6970	1.6970	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.8878			
							1.8888	201383609		
2A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201383609		0.095
2A2					WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201383609		0.095
3A		5.1870 MAXIMUM			WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.1358			
							5.1398			
4A	2.5984	2.6823 MAXIMUM	2.5996	2.5996	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3622	201056605	0.0030	2.6026
							2.3634			
4B	2.5996		2.5996	2.5996	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3622	201056605	0.0030	2.6026
							2.3634			
5A	6.3386	6.3843 MAXIMUM	6.3470	6.3470	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	6.1024	201056605		
	6.3448						6.1048			
5A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056605		0.122
5A2					WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056605		0.122
6A		1.4790 MINIMUM	1.5340	1.5340	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
6B		1.4790 MINIMUM	1.5340	1.5340	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
6A1	FACES		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
6B1					WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
7A	LUBE HOLES	NOTE 1	COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
8A	EXTERNAL SUFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
9A	NOTES SUFACE		COND. ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: UPPER TORQUE LINK

W/ORDER: 12616

P/NUMB.: 201540302

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08BEL0113

SUB W/O: 12616-83



LIF. LIMIT: 60,000

C.S.N.: 16,758

NOTE 1: ► THREE LUBE FITTING HOLES INSTALL LUBRICATION ADAPTER P/NI 899005010 (QTY 3)

A/C: A320

P/NAME: UPPER TORQUE LINK

P/NUMB.: 201540302

S/NUMB.: 08BEL0113

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.

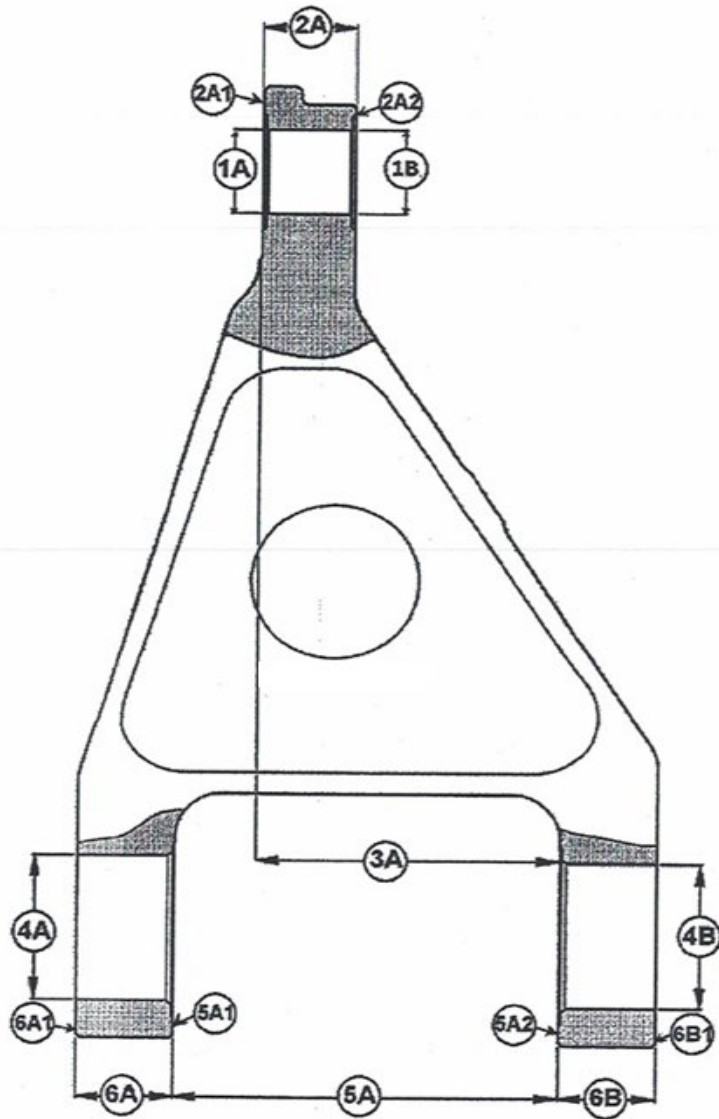
W/ORDER: 12616

OHM CHAP.: 32-12-25

SUB W/O: 12616-83

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: PIN

W/ORDER: 12616

INSPECTION STATUS REPORT

P/NUMB.: 201173600

OHM CHAP.: 32-12-25

Component Overhaul
Services

S/NUMB.: 08MDG6974

SUB W/O: 12616-45

LIF. LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	3.7388	3.7156	3.6390	3.6390	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	3.7397	NOTE 1								
B	4.1325	4.109	4.1330	4.1330	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	4.1334	NOTE 1								
1A	0.630	0.693	0.6300	0.6300	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6300	450239471	0.0015	0.632
1B	0.631	NOTE 2	0.6300	0.6300	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6310	450239471	0.0015	0.632
2A		0.275	0.3180	0.3180	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
		MINIMUM								
3A		0.205	0.2420	0.2420	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
		MINIMUM								
4A	INTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
5A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
6A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320
P/NAME: PIN
P/NUMB.: 201173600
S/NUMB.: 08MDG6974
LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12616
OHM CHAP.: 32-12-25
SUB W/O: 12616-45
C.S.N.: 16,758

INSPECTION STATUS REPORT



NOTE 1: CHROME REPAIR, PER CMM 32-11-33 AND 32-12-25 BOTH JOURNALS -A- AND -B- MUST BE REPAIRED AT THE SAME TIME

NOTE 2: TO REPAIR THIS BORES, MACHINE AND INSTALL BUSHINGS P/N: 450239471

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: PIN

W/ORDER: 12616

P/NUMB.: 201173600

OHM CHAP.: 32-12-25

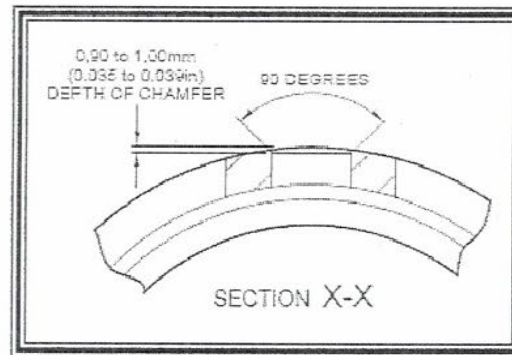
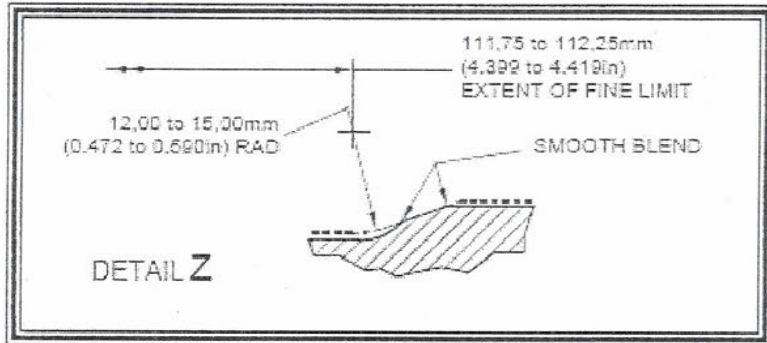
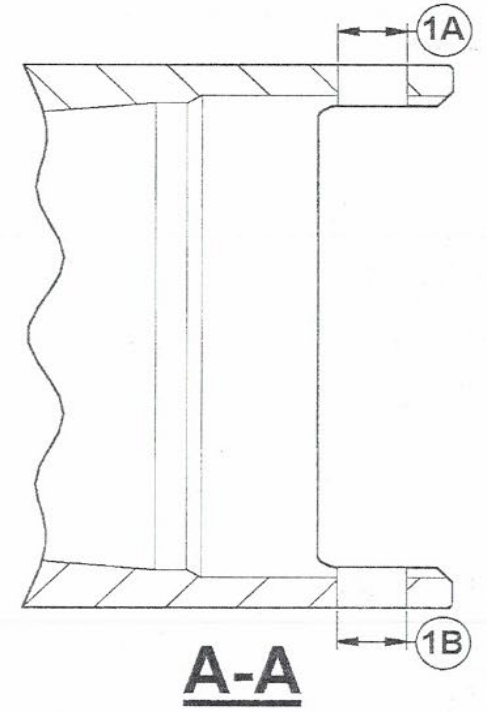
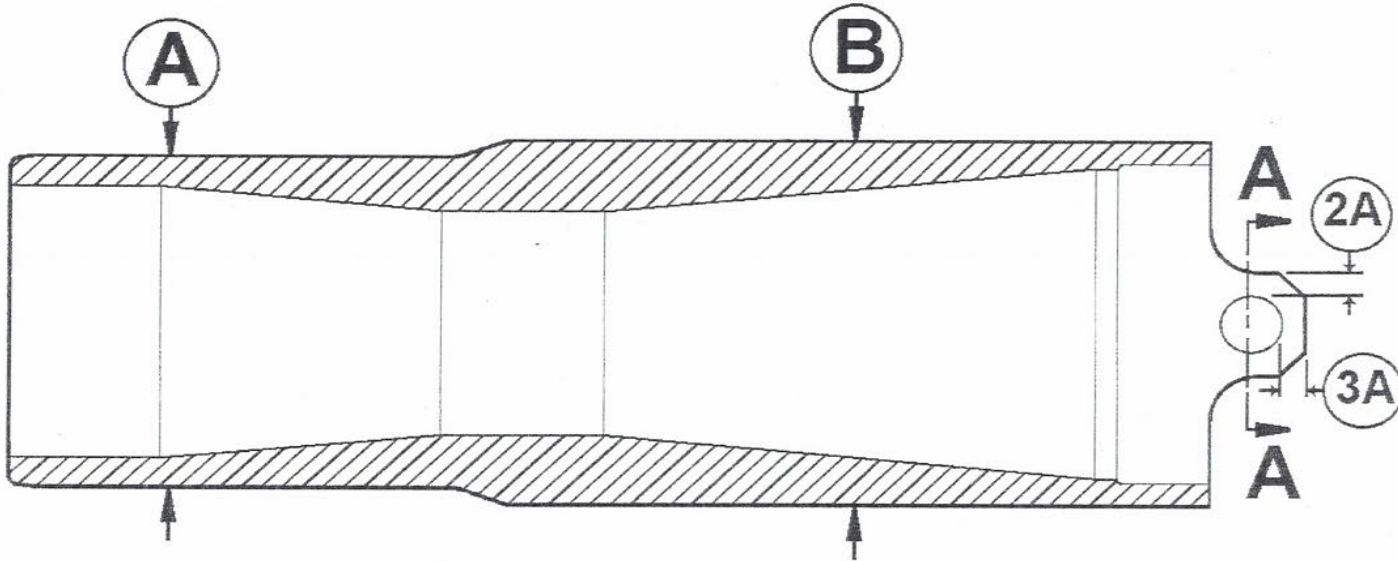
S/NUMB.: 08MDG6974

SUB W/O: 12616-45

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT



BUSHING REPAIR DETAILS

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: CARDAN LOCK STAY

W/ORDER: 12616

P/NUMB.: 201058306

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08COU45295X5723

SUB W/O: 12616-38

Component Overhaul
Services

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	0.9843	1.0599	0.9880	0.9880	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058602	0.0010	0.989
	0.9851						0.7882			
2A			0.7870	0.7870	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.9826	201058602		
							0.9835			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058602		0.098
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058602		0.098
3A			2.0270	2.0270	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.8228	201058620		
							1.8244			
3A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	201058620		0.102
3A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4737	201058620		0.102
4A	0.9873	1.0599	0.9880	0.9880	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058620	0.0020	0.990
	0.9851						0.7882			
4A1	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
4B	0.9873	1.0599	0.9880	0.9880	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058620	0.0020	0.990
	0.9851						0.7882			
4B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
5A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
6A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: CARDAN LOCK STAY

W/ORDER: 12616

P/NUMB.: 201058306

OHM CHAP.: 32-12-25

S/NUMB.: 08COU45295X5723

SUB W/O: 12616-38

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT

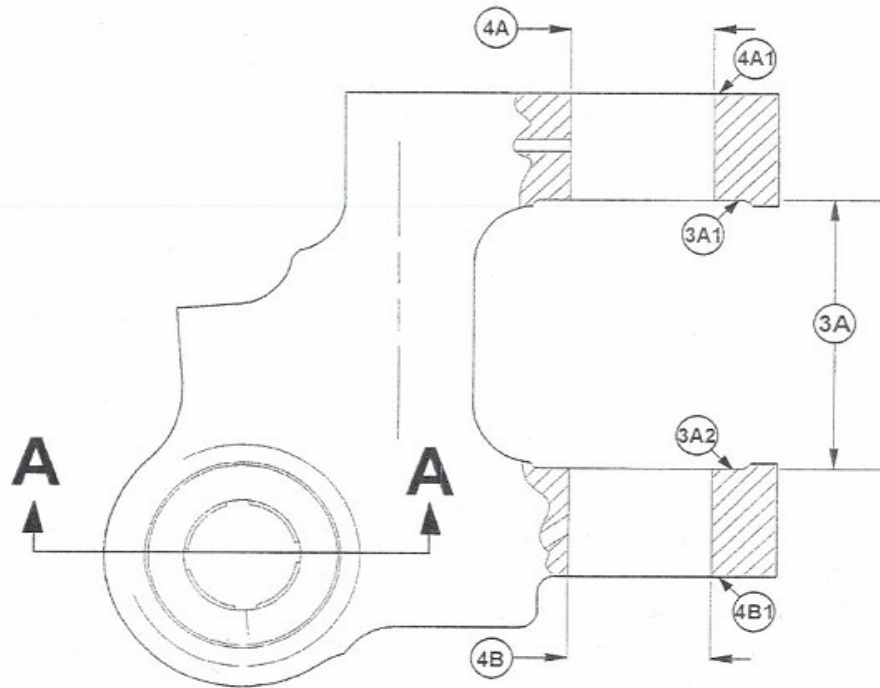
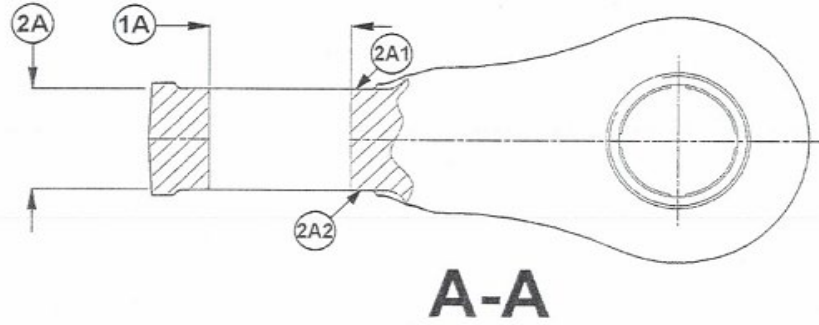


NOTE 1: IF BORE IS OVERSIZED OR REQUIURES MACHINING SCRAP THE UNIT.

A/C: A320
P/NAME: CARDAN LOCK STAY
P/NUMB.: 201058306
S/NUMB.: 08COU45295X5723
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12616
OHM CHAP.: 32-12-25
SUB W/O: 12616-38
C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: MLG LH FITTING

W/ORDER: 12616

P/NUMB.: 201540300

OHM CHAP.: 32-12-25

S/NUMB.: 08MDM2051

SUB W/O: 12616-3

LIFE LIMIT: 60, 000

C.S.N.: 16,758

INSPECTION STATUS REPORT



LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	8.3465	8.408 MAX	8.3510	8.3580	PITTING	STRIP				
	8.3493	NOTE 14	#1							
B	8.2270	8.258 MAX	8.2290	8.2360	PITTING	STRIP				
	8.2241	NOTE 14	#1							
C	8.0020		8.0020	8.0020	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	8.0048	CHROME								
D	7.2270	7.2825	7.2270	7.2270	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	7.2240	NOTE 11								
E			COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
1A	4.3701		4.3750	4.3750	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	4.1339	201160639	0.0060	4.381
	4.3715								4.1352	
2A	7.5886		7.6000	7.6000	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	7.5886			
	7.6082								7.6082	
2A1	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
2A2	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201160639		
			ONLY							
3A	0.7400	0.8158	0.7489	0.7489	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6299	201056715	0.0020	0.751
3B	0.7489	NOTE 1	0.7489	0.7489	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6310	201056715	0.0020	0.751
4A	4.5670				WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	4.5670	201056715		
	4.5767								4.5677	
4A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056715		2.284
4A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056715		2.284
5A		5.3440	5.4000	5.4000	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.5420	201056715		
		NOTE 2							5.5760	
5A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056715		0.071
5A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056715		0.071
6A	2.2047	2.2880	2.2050	2.2050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.0080	201160604	0.0020	2.207
	2.2059								2.0090	
6B	2.2441	2.3270	2.2450	2.2450	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.0019	201160605	0.0020	2.247
	2.2453								2.0091	

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: MLG LH FITTING

W/ORDER: 12616

P/NUMB.: 201540300

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08MDM2051

SUB W/O: 12616-3



LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
7A		2.4400	2.4100	2.4100	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.2050	201160604		
		NOTE 2					2.2080	201160605		
7A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201160604		0.103
7A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201160605		0.103
8A	1.1080	1.0620	1.1040	1.1040	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		NOTE 6		
		MINIMUM								
8B	1.1080	1.0720	1.1040	1.1040	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		NOTE 6		
		MINIMUM								
8A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
8B1			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
9A	6.2992	NOTES	6.3010	6.3010	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201042106	0.0020	6.303
	6.3017	2, 3 AND 4								
10A		0.6100	0.4900	0.5560	CORROSION DEPTH	MACHINE TO REMOVE	0.3750	NOTE 7	0.0020	0.558
							0.3772			
11A		2.6680	2.6790	2.6790	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.6680	NOTE 7		
11A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		NOTE 7		
11A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
12A		0.8465	0.7850	0.7850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6398	201160633	0.0020	0.787
		MAXIMUM					0.6594			
13A	2.7461		2.7850	2.7850	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.7461			
	2.8240						2.8240			
13A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				-0.019
13A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				-0.019
14A	2.0472	2.1310	2.0480	2.0480	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.8436	201056686	0.0020	2.050
							1.8450			
14B	2.0484		2.0480	2.0480	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.4173	201056887	0.0020	2.050
							1.4183			
15A		8.4650	8.5690	8.5690	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	8.7600	201056686		
							8.7690	201056687		
15A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056686		0.095
15A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056687		0.095

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: MLG LH FITTING

W/ORDER: 12616

P/NUMB.: 201540300

OHM CHAP.: 32-12-25

S/NUMB.: 08MDM2051

SUB W/O: 12616-3

LIFE LIMIT: 60, 000

C.S.N.: 16,758

INSPECTION STATUS REPORT



LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
16A	2.9528	3.0370	2.9550	2.9550	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.7559	201056603	0.0020	2.957
16B	2.9539	NOTE 1	2.9550	2.9550	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.7571	201160696	0.0020	2.957
17A		5.0790	5.1250	5.0900	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.4298	201056603		
							5.4313	201160696		
17A1	FACES		COND.		CORROSION DEPTH	MACHINE TO REMOVE		201056603		0.170
17A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201160696		0.170
18A	2.5984	2.6820	2.6000	2.6000	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3623	201056605	0.0020	2.602
18B	2.5996		2.6000	2.6000	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	2.3633	201056605	0.0020	2.602
19A	5.7050	5.6200	5.7050	5.7050	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	5.9416	201056605		
		NOTE 2					5.9430			
19A1	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056605		0.118
19A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201056605		0.118
20A		1.5110	1.5640	1.5640	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
20A1	FACE		COND.	COND.	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY	ONLY						
20B		1.5110	1.5910	1.5910	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
		NOTE 2								
20B1	FACE		COND.	COND.	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY	ONLY						
21A	1.2598	1.3356	1.2600	1.2600	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.0236	201383607	0.0020	1.262
	1.2608	MAXIMUM					1.0244			
21B	1.2598	1.3356	1.2600	1.2600	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.0236	201383607	0.0020	1.262
	1.2608	MAXIMUM					1.0244			
21C	1.2598	1.3356	1.2600	1.2600	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.0236	201383607	0.0020	1.262
	1.2608	MAXIMUM					1.0244			
22A	0.5510	0.6192	0.5520	0.5520	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.3720	201173669	0.0020	0.554
	0.5522						0.3917			

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: MLG RH FITTING

W/ORDER: 3900

P/NUMB.: 201540301

OHM CHAP.: 32-12-25

S/NUMB.: 08MDM2051

SUB W/O: 12616-3

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT



NOTE 1: ► MACHINE THE BORE TO ONE OF THE FOLLOWING REWORKEDS

- 1- 6.3043 / 6.3066 O/S BEARING P/N: 450239401
- 2- 6.3093 / 6.3116 ----- P/N: 450239402
- 3- 6.3143 / 6.3166 ----- P/N: 450239403
- 4- 6.3193 / 6.3216 ----- P/N: 450239404

NOTE 2 5- 6.3243 / 6.3266 ----- P/N: 450239405

- 6- 6.3293 / 6.3316 ----- P/N: 450239406
- 7- 6.3343 / 6.3366 ----- P/N: 450239407
- 8- 6.3393 / 6.3416 ----- P/N: 450239408

NOTE 3: 9- 6.3443 / 6.3466 ----- P/N: 450239409

- 10- 6.3493 / 6.3516 ----- P/N: 450239410
- 11- 6.3543 / 6.3566 ----- P/N: 450239411
- 12- 6.3593 / 6.3616 ----- P/N: 450239412

NOTE 4: ► -14A- BUSHING IS INSTALLED ON THE INBOARD SIDE OF THE MAIN FITTING, SAME SIDE AS THE LOWER CARDAN LUG

NOTE 5: ► REPAIR IAW MESSIER DOWTY CMM 32-12-25 REPAIR 13-28 PG 604
IF DISCREPANCIES REMAIN AND YOU MUST MACHINE OVER RWK LIMITS. THEN CONTACT AN INSPECTOR FOR FURTHER INSTRUCTION

NOTE 6: REFER TO COCESSION NUMBER RC-GL-0027227 FOR REFERENCE DESIGN DIMENSION FOR THE LUG

NOTE 7: MACHINE 0.0197 MAX PER FACE (REF. REPAIR 13-9 FIG. 602 PAGE 604

NOTE 8: ► IF HOLE DIAMETER IS STANDARD, USE BEARING P/N: 201056606 QTY 1 PER CMM 32-12-25 AS REQUIRED
IF PART REQUIRES MACHINING TO REMOVE DEFECTS , THEN MACHINE A MINIMUM OF 0.060 OVER FOR A REPAIR BEARING P/N: 4502378214
QTY. 1 PER CMM 32-12-25 REPAIR NUMBER 13-19

NOTE 9: IF THE LUBRICATION BORE LOCATION 32G REQUIRES REWORKED FOLLOW REPAIR 13-24 FOR THE CORRECT RWK ADAPTER REQUIRED

NOTE 10: IF THE LUBRICATION BORE LOCATION 32H REQUIRES REWORKED FOLLOW REPAIR 13-24 FOR THE CORRECT RWK ADAPTER REQUIRED

NOTE 11: ► REPAIR PER MESSIER DOWTY, CMM 32-11-33 REPAIR NO 1-26 TABLE 601 PG 603
1: DIAMETER 0.4850 / 0.4860 TRANSFER DOWEL 45023951
2- DIAMETER 0.5500 / F.5520 TRANSFER DOWEL 45029352

NOTE 12: DO NOT MACHINE MORE THAN 0.196 IN FROM FACE AND MAKE FINISH 63 MICRO-INCHES

NOTE 13: IF DIAMETER -A- HAS WEARMARKS OR DAMAGE AND REQUIRES REPAIR, DO NOT MACHINE MORE THAN 0.0196 FROM EACH

NOTE 14: IF DIAMETERS -A- OR -B- HAS WEARMARKS OR DAMAGE AND REQUIRES REPAIR, USE THE REPAIRS ON PAGE 604, REPAIR NO 13-25 AS NEEDED TO REPAIR AREAS **(YOU MUST REPAIR BOTH DIAMETERS -A- AND -B- TO THE SAME REPAIR) FLASH CHROME TO 0.0003 / 0.0005**

WITHIN DI WITHIN OHM LIMITS

5.674

A/C: A320

P/NAME: MLG RH FITTING

P/NUMB.: 201540301

S/NUMB.: 08MDM2051

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.

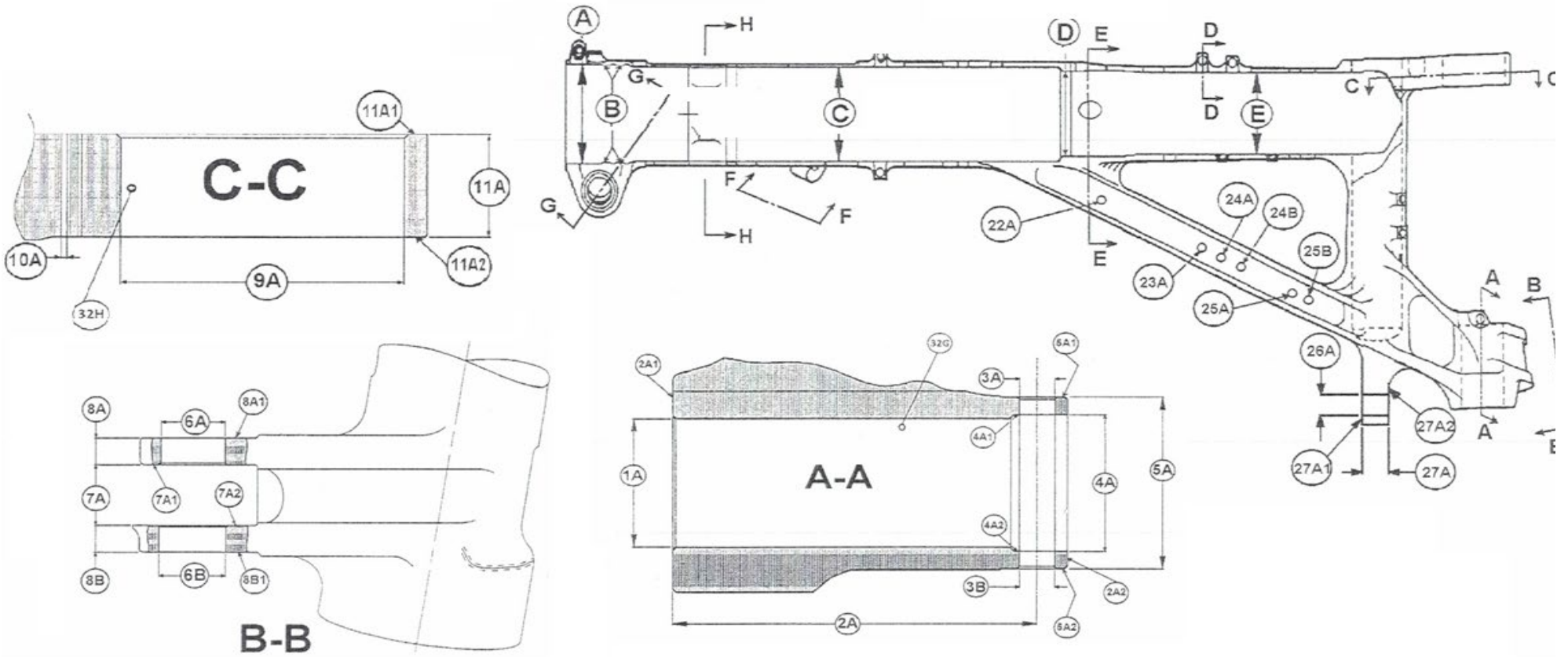
W/ORDER: 3900

OHM CHAP.: 32-12-25

SUB W/O: 12616-3

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

P/NAME: MLG RH FITTING

P/NUMB.: 201540301

S/NUMB.: 08MDM2051

LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.

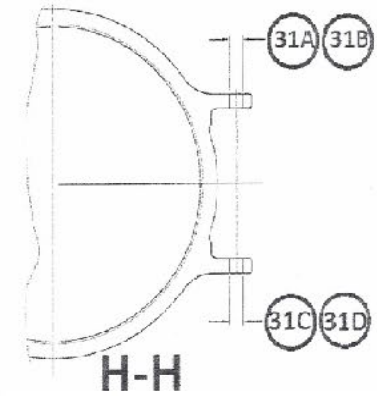
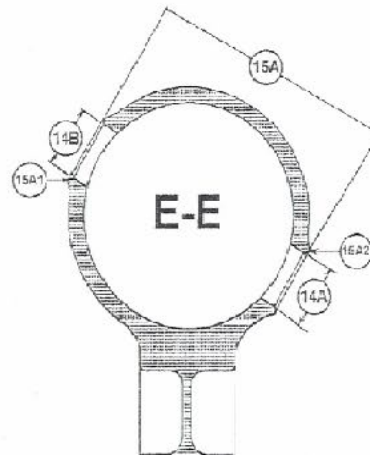
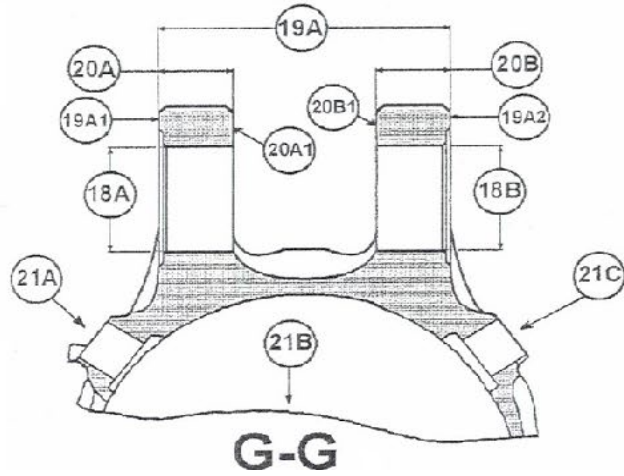
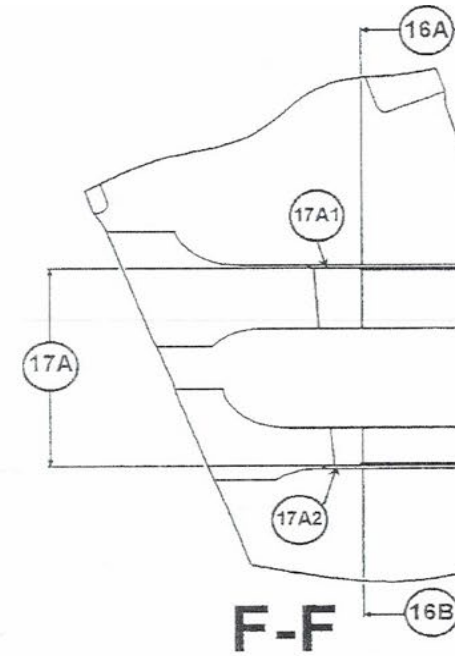
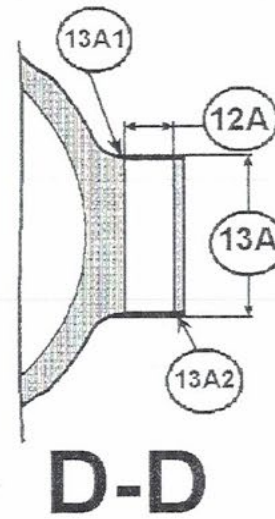
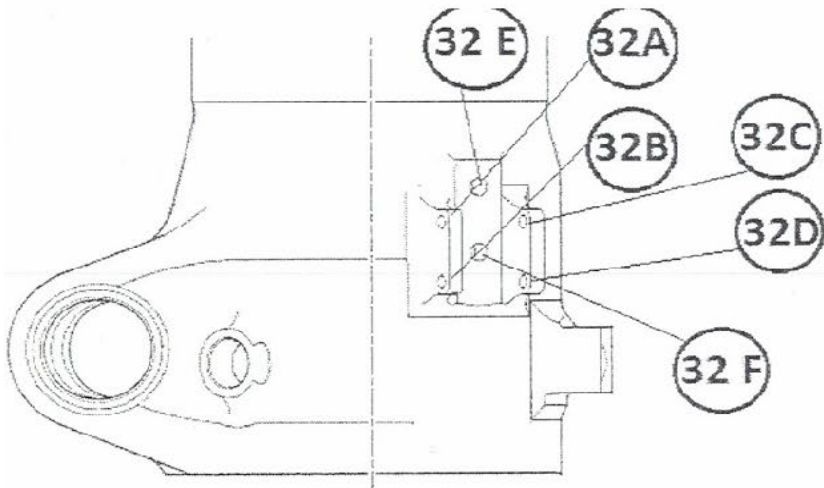
W/ORDER: 3900

OHM CHAP.: 32-12-25

SUB W/O: 12616-3

C.S.N.: 16,758

INSPECTION STATUS REPORT



A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: PIN

W/ORDER: 12616

INSPECTION STATUS REPORT



Component Overhaul Services

P/NUMB.: 201173600

OHM CHAP.: 32-12-25

S/NUMB.: 08MDG6974

SUB W/O: 12616-45

LIF. LIMIT: 60,000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
A	3.7388	3.7156	3.6390	3.6390	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	3.7397	NOTE 1								
B	4.1325	4.109	4.1330	4.1330	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	4.1334	NOTE 1								
1A	0.630	0.693	0.6300	0.6300	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6300	450239471	0.0015	0.632
1B	0.631	NOTE 2	0.6300	0.6300	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.6310	450239471	0.0015	0.632
2A		0.275	0.3180	0.3180	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
		MINIMUM								
3A		0.205	0.2420	0.2420	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
		MINIMUM								
4A	INTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
5A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
6A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320
P/NAME: PIN
P/NUMB.: 201173600
S/NUMB.: 08MDG6974
LIF. LIMIT: 60,000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12616
OHM CHAP.: 32-12-25
SUB W/O: 12616-45
C.S.N.: 16,758

INSPECTION STATUS REPORT



NOTE 1: CHROME REPAIR, PER CMM 32-11-33 AND 32-12-25 BOTH JOURNALS -A- AND -B- MUST BE REPAIRED AT THE SAME TIME

NOTE 2: TO REPAIR THIS BORES, MACHINE AND INSTALL BUSHINGS P/N: 450239471

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: PIN

W/ORDER: 12616

P/NUMB.: 201173600

OHM CHAP.: 32-12-25

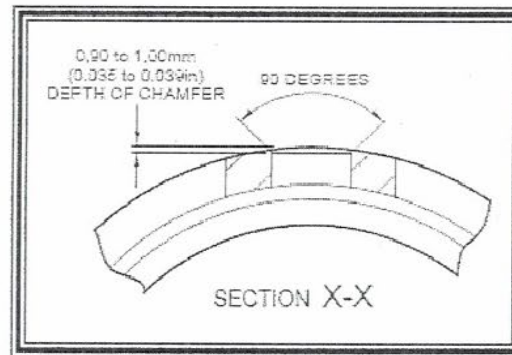
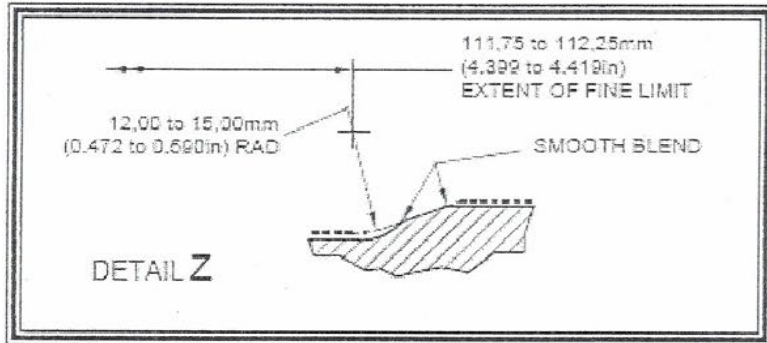
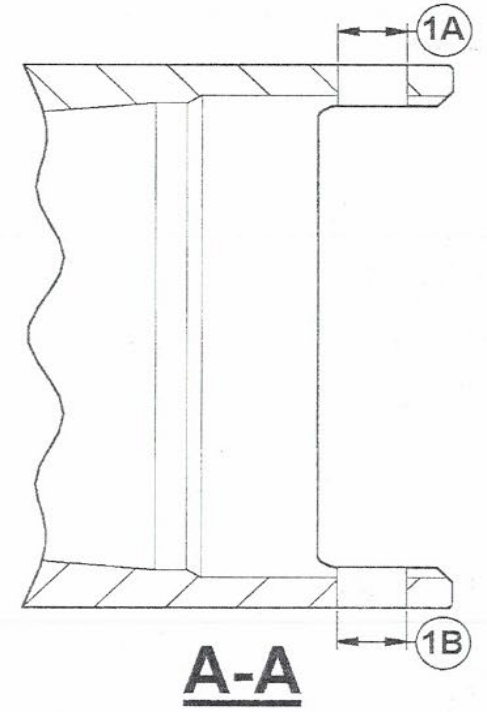
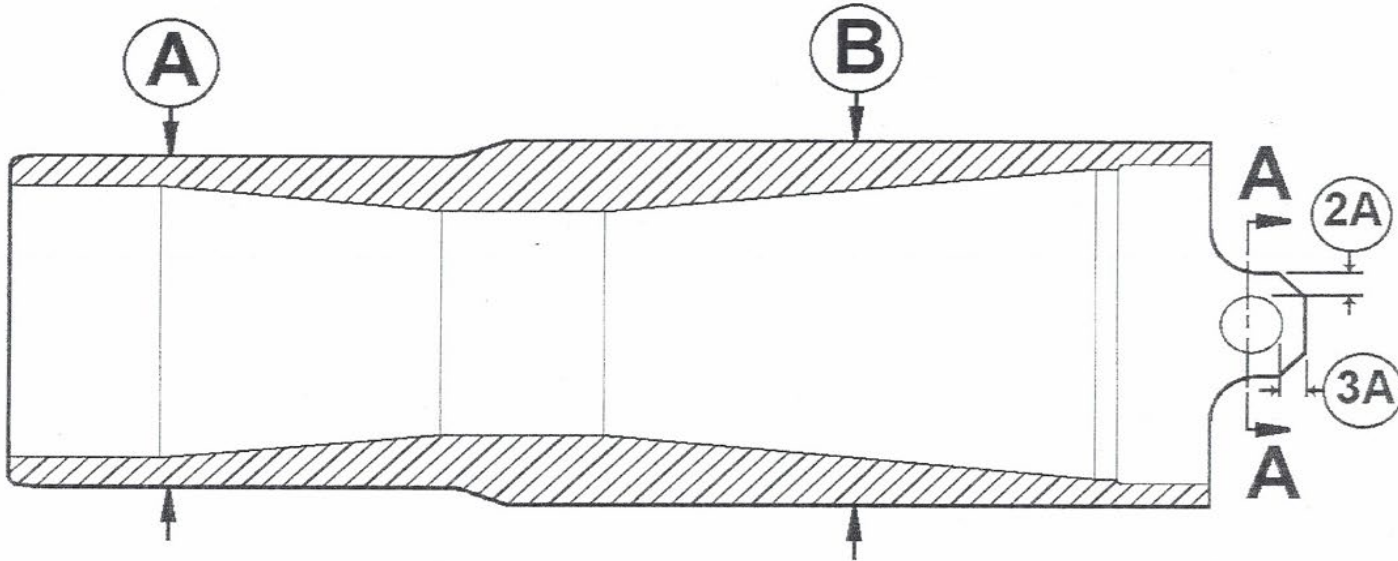
S/NUMB.: 08MDG6974

SUB W/O: 12616-45

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT



BUSHING REPAIR DETAILS

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: CARDAN LOCK STAY

W/ORDER: 12616

P/NUMB.: 201058306

OHM CHAP.: 32-12-25

INSPECTION STATUS REPORT

S/NUMB.: 08COU45295X5723

SUB W/O: 12616-38


Component Overhaul
 Services

LIFE LIMIT: 60, 000

C.S.N.: 16,758

LOCATION	DESIGN DIM.	RWK LIMIT	RECEIVED DIM.	RWK DIM.	CONDITION	DISPOSITION	BUSHING DIM.	PART NUMBER OF BUSHING	PRESS FIT	BUSHING DIM.
1A	0.9843	1.0599	0.9880	0.9880	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058602	0.0010	0.989
	0.9851						0.7882			
2A			0.7870	0.7870	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.9826	201058602		
							0.9835			
2A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058602		0.098
2A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS		201058602		0.098
3A			2.0270	2.0270	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	1.8228	201058620		
							1.8244			
3A1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4727	201058620		0.102
3A2			ONLY		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.4737	201058620		0.102
4A	0.9873	1.0599	0.9880	0.9880	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058620	0.0020	0.990
	0.9851						0.7882			
4A1	FACE		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
4B	0.9873	1.0599	0.9880	0.9880	WITHIN DESIGN LIMITS	WITHIN OHM LIMITS	0.7874	201058620	0.0020	0.990
	0.9851						0.7882			
4B1	FACES		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
			ONLY							
5A	EXTERNAL		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	SURFACE		ONLY							
6A	OTHER		COND.		WITHIN DESIGN LIMITS	WITHIN OHM LIMITS				
	REJECTIONS		ONLY							

A/C: A320

CUSTOMER: CAVU AEROSPACE INC.

P/NAME: CARDAN LOCK STAY

W/ORDER: 12616

P/NUMB.: 201058306

OHM CHAP.: 32-12-25

S/NUMB.: 08COU45295X5723

SUB W/O: 12616-38

LIF. LIMIT: 60,000

C.S.N.: 16,758

INSPECTION STATUS REPORT

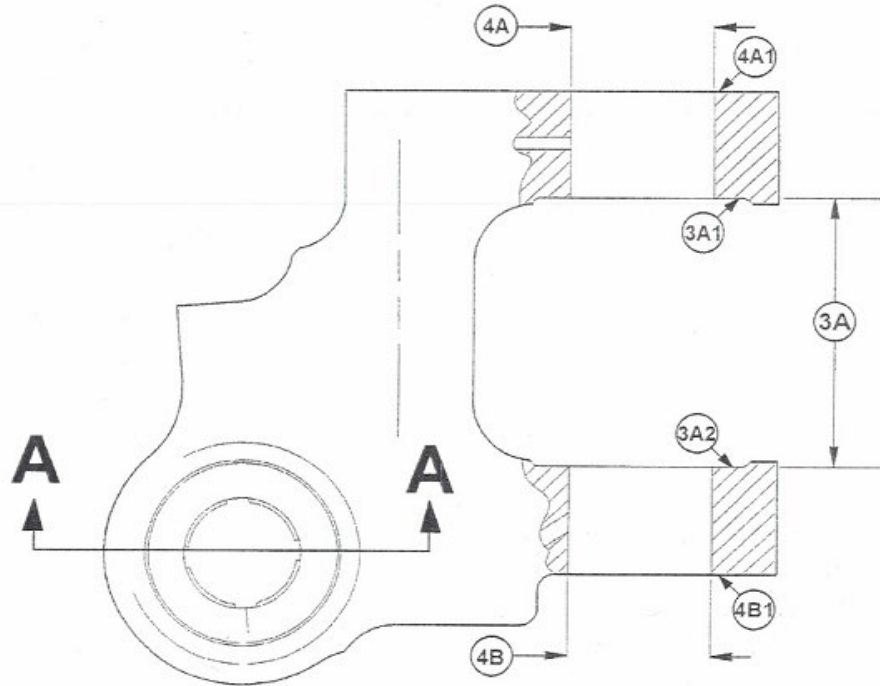
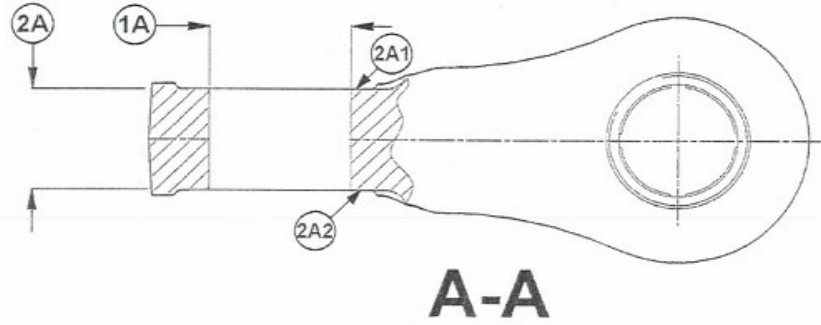


NOTE 1: IF BORE IS OVERSIZED OR REQUIURES MACHINING SCRAP THE UNIT.

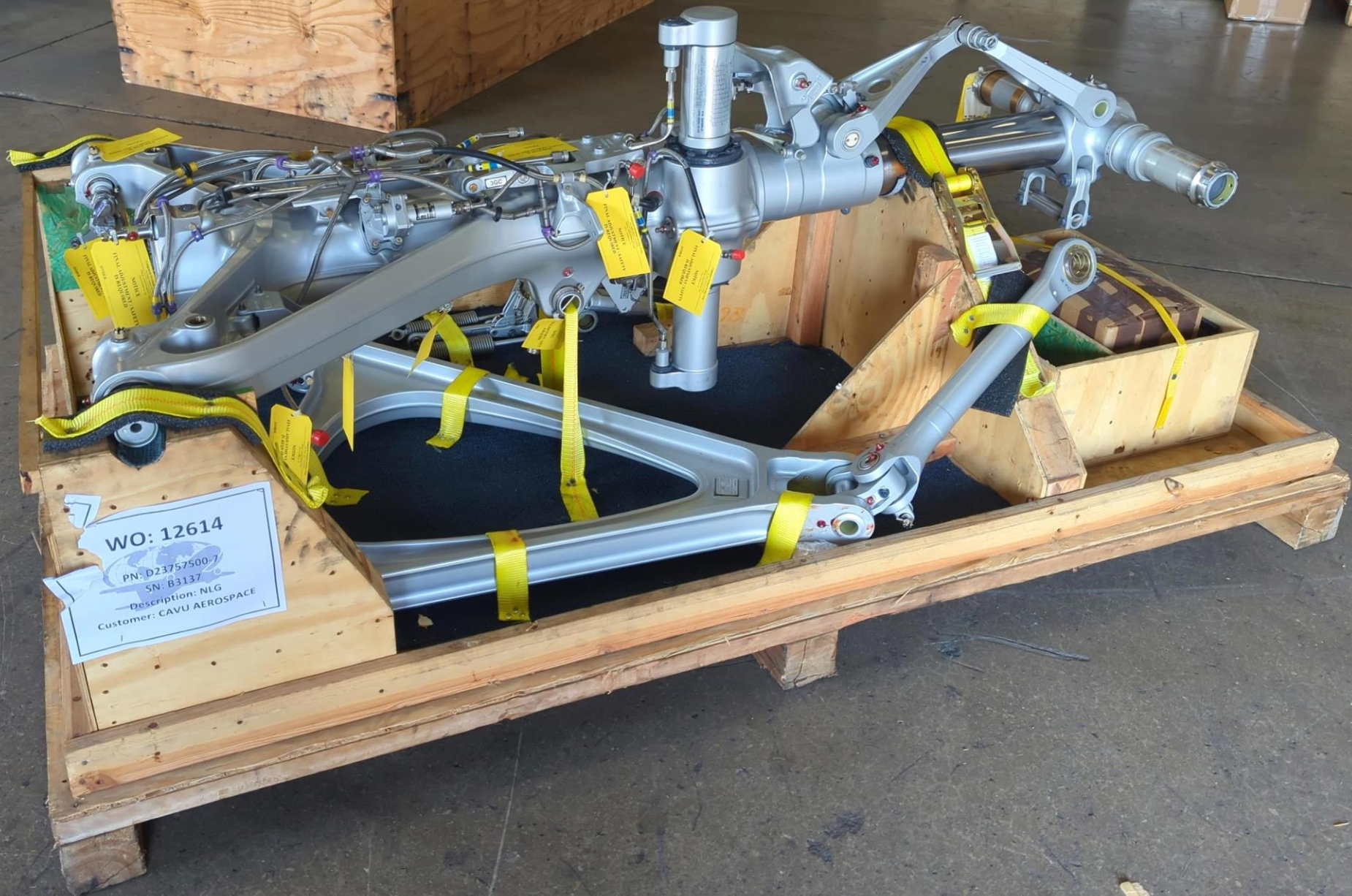
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P/NAME: CARDAN LOCK STAY
P/NUMB.: 201058306
S/NUMB.: 08COU45295X5723
LIF. LIMIT: 60, 000

CUSTOMER: CAVU AEROSPACE INC.
W/ORDER: 12616
OHM CHAP.: 32-12-25
SUB W/O: 12616-38
C.S.N.: 16,758

INSPECTION STATUS REPORT







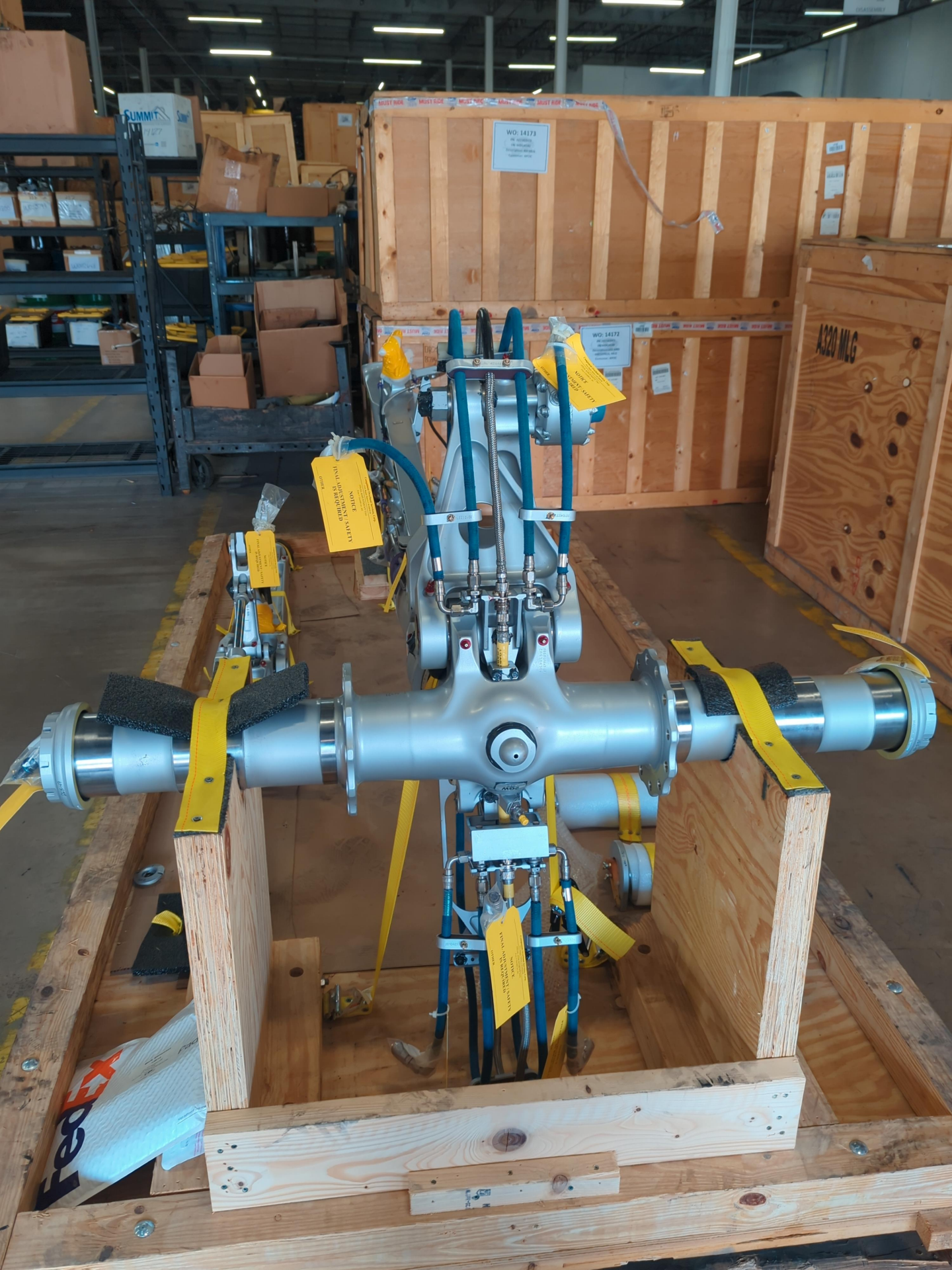
SPARE PARTS
ON SHIPPING
RTC SHELF!!

WO: 12614

WO: 12614
PN: D23757500-7
SN: B3137
Description: NLG
Customer: CAVU AEROSPACE

CAVU AEROSPACE
10000 W. 10th Ave.
Denver, CO 80202
Tel: 303.751.1000
Fax: 303.751.1001
www.cavu.com

CAVU AEROSPACE
10000 W. 10th Ave.
Denver, CO 80202
Tel: 303.751.1000
Fax: 303.751.1001
www.cavu.com



WO: 14173
NO. 14173
NO. 14173
NO. 14173

WO: 14173
NO. 14173
NO. 14173
NO. 14173

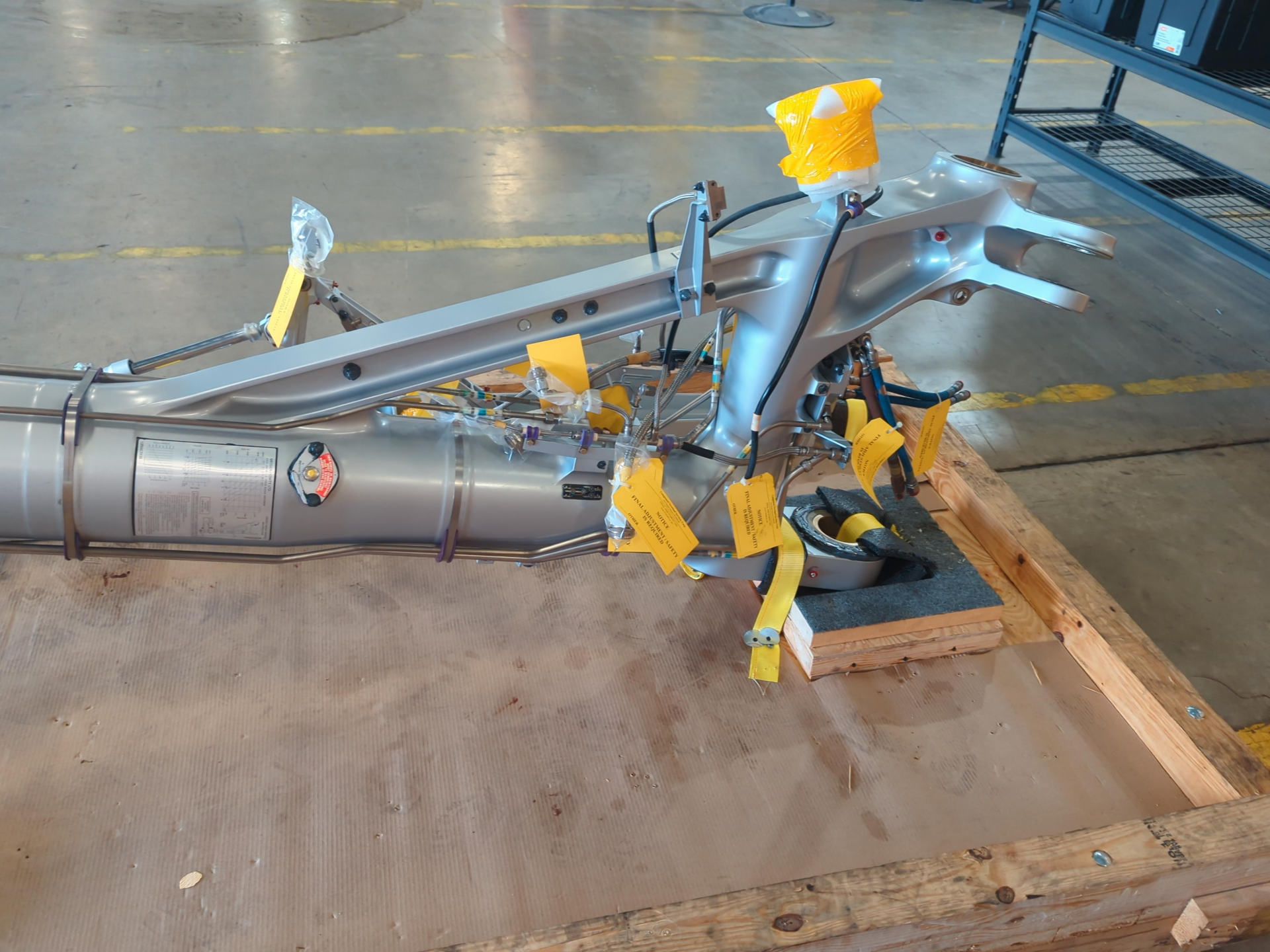
A320 M/G

NOTICE
OF DISPOSITION
AND
REPAIRS

NOTICE
OF DISPOSITION
AND
REPAIRS

FedEx





Technical specifications and diagrams printed on a metal plate on the side of the device.



NOTE: FINAL INSPECTION REQUIRED BEFORE USE

NOTE: FINAL INSPECTION REQUIRED BEFORE USE

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NOTE: FINAL INSPECTION REQUIRED BEFORE USE



Technical specifications and wiring diagrams printed on the side of the component.



CLASSIFICATION
DATE
BY

SUPPLY
INFORMATION
DATE





WO-1675

FINAL INSPECTION
DATE: 10/15/15
BY: [Signature]

FINAL INSPECTION
DATE: 10/15/15
BY: [Signature]

FINAL INSPECTION
DATE: 10/15/15
BY: [Signature]

NOTICE
FINAL INSPECTION
DATE: 10/15/15
BY: [Signature]

Pardec
Part No. 1000000000
Rev. 001



35002
FedEx

1. MAIN HYDRAULIC LINE
2. MAIN HYDRAULIC LINE
3. MAIN HYDRAULIC LINE
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